



708KA

AD COMPLIANCE
LOGBOOK

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: <u>N/A</u>	P/N: <u>N/A</u>
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: <u>N/A</u>	S/N: <u>N/A</u>

AD 83 - 02 - 01

DESCRIPTION: Inspect Tie Bar for corrosion - to prevent corrosive deterioration of the wing strut to fuselage attachment structure

Visually inspect per AD:
 Par a), tie-bar for internal corrosion in accordance with SB paragraph 5 and
 AD Par b), all tie-bar lugs for corrosion in accordance with SB paragraph 9 in the Inspection Section of S/B No. 3/37.

METHOD OF COMPLIANCE: If corrosion is found See AD for additional instructions.

INTERVAL: 12 Mos.

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
26-Jan-23 17,218.5	Per records search, found previously complied with on recurring record from Kenmore Air	<i>Robert B. Gunn</i>	A&P 574601404	January 2024
23-JAN-24 18,440.9	Visually inspected tie-bar for internal corrosion IAW AD A) and tie-bar lugs for corrosion IAW AD B).	<i>Patrick Kuhn</i>	A&P 2922815-	January 2025

DHC-3 Otter

Service Bulletin No. 3/37

Wing Strut to Fuselage Attachment Tie-Bar - Inspection for Corrosion.

This page transmits Revision 'B' of Service Bulletin No. 3/37 dated October 8th, 1982.

Page 1 and 2 of this Service Bulletin reissued to include Canadian Department of Transport Airworthiness Directive.

Remove Page 1 and 2 of existing Service Bulletin No. 3/37 dated May 14th, 1982 and replace with attached.

Previous issues of Service Bulletin No. 3/37:

Initial issue - Page 1 thru 6 dated January 8th, 1982.
Revision 'A' - Page 1 thru 6 dated May 14th, 1982.

Revision 'B' October 8th, 1982

SERVICE BULLETIN

DHC-3 OTTER

SUBJECT: Wing Strut to Fuselage Attachment Tie-Bar - Inspection for Corrosion.

EFFECTIVITY: All DHC-3 Aircraft.

REASON: Several cases of corrosion have been found in the external lug area of the subject tie-bar during routine inspection. Also a case of corrosion due to water ingress and entrapment has been found in the internal channel section under the cabin floor.

DESCRIPTION: The aircraft wings are supported and the wing struts removed from their fuselage attachment. The tie-bar lugs are inspected for corrosion and dye penetrant inspected for cracks. Seats and floor are removed and inspection holes cut in tie-bar cap strip. Internal surface of tie-bar is inspected for corrosion and if tie-bar is serviceable, holes are taped and the floor and seats re-installed. Repetitive inspections for tie-bar inspection is introduced.

COMPLIANCE: Accomplishment of this inspection and rectification if required, should be carried out within six calendar months from effective date of this Service Bulletin. Repeat inspections must be accomplished annually thereafter in conjunction with Special Inspection detailed in Maintenance Manual PSM 1-3-2 Appendix IV Part 4 Item 5.

APPROVAL: The inspection and rectification conveyed by this Service Bulletin has been approved by the Director, Airworthiness, Canadian Department of Transport.

MANHOURS: Approximately six manhours will be required to inspect the tie-bar lugs and approximately three manhours will be required to inspect the tie-bar internally.

KIT COST: C3FS14-5 Tie-bar Assembly.....\$1,196.77 each (Budgetary)

(Price is F.O.B. Toronto and in Canadian funds. Subject to change without notice). A limited number of tie-bars are currently available at de Havilland Aircraft).

January 8th, 1982
Revision 'A', May 14th, 1982
Revision 'B', October 8th, 1982

S/B No. 3/37
Page 1 of 6

WEIGHT AND BALANCE:
None.

PUBLICATIONS AFFECTED:
None.

REFERENCES:
Canadian Department of Transport Airworthiness Directive No.
CF-82-26 dated August 30th, 1982.

ACCOMPLISHMENT INSTRUCTIONS:

Inspection:

To inspect the subject tie-bar assembly C3FS14-5 located under the fuselage floor and extending to the left and right-hand lower wing strut attachment, proceed as follows:

1. Remove the front seven cabin seats, if fitted.
2. Refer to Illustrated Parts Catalogue PSM 1-3-4 Part 2 Figure 25 and remove left and right-hand forward sections of floor extrusion C3FS351-2 and C3FS352-2 following removal of securing screws. Retain all parts.
3. Remove forward three sections of flooring C3FS306-11, C3FS306-12 and C3FS307-7 following removal of securing screws. Retain all parts.

CAUTION

Do not walk on floor diaphragm.
Use scrap wood as walkways.

4. Refer to Figure 1 and cut three 2.50 X .90 inch inspection holes in cap strip C3FS13-4 located over top of tie-bar. Cut one hole each side of cap strip and one at centre of aircraft. Exercise care to prevent damage to tie-bar when cutting holes.
5. Using appropriate light and a long handled mirror, inspect for internal corrosion of the tie-bar. If corrosion is evident or suspected, the tie-bar must be removed from the aircraft as noted in REPLACEMENT paragraph 1 thru 5 and further detailed inspection carried out.
6. If no corrosion is present during initial inspection proceed with tie-bar lugs inspection as detailed below in paragraphs 8 thru 14.
7. Support the aircraft wings either by use of a crane, wing sling and lifting cradles, or by jacking or other approved method of support. Refer to Maintenance Manual PSM 1-3-2 Part 1.
8. Remove left and right-hand strut to fuselage fairings C3W140-11 and -12 respectively and bolt C3W129-3, nut AN320-12, washer AN960D1216L and cotter pin AN381-4-28 securing each strut lower end shackle to the fuselage tie-bar lugs. Retain all hardware except cotter pins. Refer to Illustrated Parts Catalogue PSM 1-3-4 Part 2
WING GROUP.

January 8th, 1982
Revision 'A', May 14th, 1982
Revision 'B' October 8th, 1982

S/B No. 3/37
Page 2 of 6

ACCOMPLISHMENT INSTRUCTIONS: cont'd

10. Inspect the faces of each tie-bar lug for corrosion. If corrosion is evident, the corroded area must be cleaned up. The lug minimum thickness following complete removal of corrosion must not be less than 0.955 inch at any one location. If less than 0.955 inch, the tie-bar must be replaced.
11. If the corrosion is within limits, blend smooth all edges.
12. Using dye penetrant and a strong magnifying glass, inspect each lug for cracking. No cracks are acceptable and if present, tie-bar must be replaced.
13. If no corrosion was found on tie-bar or tie-bar lugs or if corrosion was found and removed with tie-bar lugs still within dimensional limits, the tie-bar may be considered serviceable.
14. The inspection holes in the cap strip over the tie-bar may be covered with adhesive tape and the floor diaphragm, flooring, floor extrusions and seats replaced with retained hardware.
15. Seal all around each tie-bar lug and outer surface of skin using a silicon sealer to ensure a water tight joint. Reconnect wing struts using retained hardware and new cotter pins AN381-4-28. Replace fairings.
16. If no corrosion has been found and tie-bar is declared serviceable, make the following entry in the "Record of Airframe Modifications" or equivalent document: "Service Bulletin No. 3/37 - Inspection of Wing Strut to Fuselage Tie-Bar for Corrosion, accomplished."

REPLACEMENT:

If out of limits corrosion or cracking of the tie-bar lugs is discovered during inspection or any internal corrosion of the tie-bar is evident, the tie-bar must be removed from the aircraft and replaced if necessary. Approximately fourteen manhours will be required to replace tie-bar. Proceed as follows:

1. With the seats, floor diaphragms, floor extrusion and flooring removed as noted in INSPECTION paragraph 7, 8, 9 and 10 above, refer to Maintenance Manual PSM 1-3-2 Part 5 and remove the centre fuel cells.
2. Remove all rivets securing cap strip to tie-bar and remove cap strip.
3. Remove securing bolts, nuts and washers from each end of tie-bar. Identify bolts to ensure correct positioning during reinstallation.

January 8th, 1982

S/B No. 3/37
Page 3 of 6

REPLACEMENT: cont'd

4. Remove rivets securing front and rear frame diaphragm to each side of tie-bar.
5. Remove tie-bar by easing bar through one side of fuselage.
6. With tie-bar removed from aircraft, blend out corrosion. Check depth of corrosion and ensure that the cross-section of the tie-bar has not been reduced more than 0.030 inch on any of the tie-bar channel faces. If corrosion is beyond limit, tie-bar must be replaced. If corrosion is within limits, remove corrosion and blend out. Dye penetrant inspect for cracks. If no cracks are evident, clean, reprotect and reinstall tie-bar in aircraft. Ensure proper sealing for bolts and rivets.
7. If new tie-bar is being installed, position in aircraft and temporarily secure at each end with retained bolts, nuts and washers.
8. Using rivet holes in front and rear frame as a guide, drill holes in tie-bar sides using a No. 30 (0.1285 in. dia.) drill. Secure with spring fasteners as operation proceeds.
9. Remove tie-bar and deburr all holes.
10. Reinstall tie-bar (new or serviceable original) and secure with retained bolts. Place spring fasteners through holes in frame diaphragms and tie-bar web.
11. Secure frame diaphragms to tie-bar web using rivets MS20470AD4-6.
12. Reinstall cap strip using rivets MS20426AD2-3.
13. Cover holes in cap strip with adhesive tape (if necessary) and reinstall floor diaphragm and all securing angles with rivets CR163-4-2.
14. Refer to Maintenance Manual PSM 1-3-2 Part 5 and reinstall center fuel cells.
15. Reinstall floor extrusions, flooring and seats with retained hardware.
16. Seal all around each tie-bar lug and outer surface of skin using a silicon sealer to ensure a water-tight joint. Reconnect wing struts using retained hardware and new cotter pins AN381-4-28. Replace fairings.
17. Following installation of tie-bar, make the following entry in the "Record of Airframe Modifications" or equivalent document: "Service Bulletin No. 3/37 - Inspection of Wing Strut to Fuselage Tie-Bar for Corrosion and Tie-Bar Replacement, accomplished." Note if existing tie-bar reinstalled and amount of corrosion removed.

January 8th, 1982

S/B No. 3/37
Page 4 of 6

KIT OF MATERIALS:

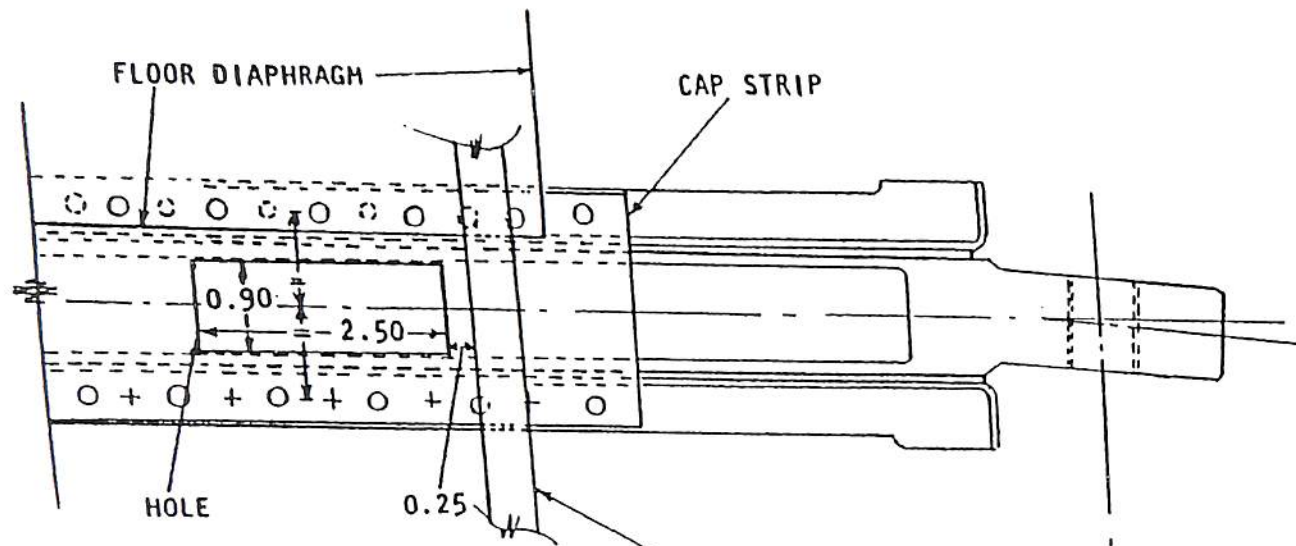
The following parts may be obtained from de Havilland Aircraft on demand:

<u>Part No.</u>	<u>Qty per A/C</u>	<u>Description</u>
C3FS14-5	1	Tie-Bar
MS20470AD4-6	A/R 1	Rivets
MS20426AD2-3	A/R	Rivets
CR163-4-2	A/R	Rivets

January 8th, 1982

S/B No. 3/37
Page 5 of 6

January 8th, 1982



SEAL ALL AROUND
WITH SILICON SEALER

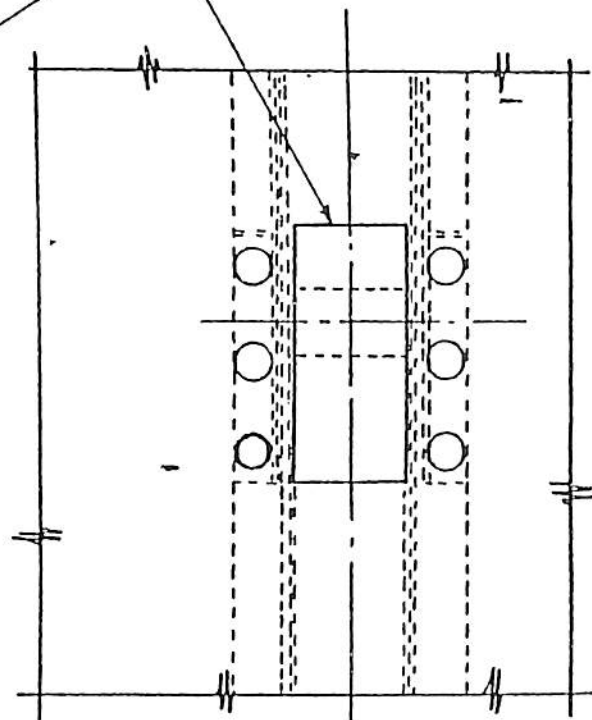


FIGURE 1 TIE BAR INSTALLATION

RECURRING AD COMPLIANCE RECORD

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Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: <u>N/A</u>	P/N: <u>N/A</u>
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: <u>N/A</u>	S/N: <u>N/A</u>

AD 83 - 04 - 05

DESCRIPTION: Inspect Control Column for cracks

Visually inspect the control column lower weld assembly for cracks IAW the Accomplishment Section of de Havilland S/B

METHOD OF COMPLIANCE: No. 3/39. If crack(s) are found See AD for instructions.

INTERVAL: 100 Hrs.

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
5/27/24 19108.1	Inspected IAW AD Par a. Using A Mirror & Flashlight Per SB 3/39, Accomplishment Instruction, Inspection	<i>Robert</i>	AP 574601409	19208.1
6-7-24 19171.6	Inspected per AD, para A no discrepancy noted	<i>Robert</i>	AP 2922915	19271.6
6-27-24 19269.3	Hobbs 5290.1 Inspected per AD paragraph A; no discrepancy noted	<i>Robert</i>	AP 2922915	19369.3
7-6-24 19367.0	Hobbs 5387.8 Inspected per AD paragraph A; no discrepancy noted	<i>Robert</i>	AP 2922915	19467.0
8-6-24 19451.3	Hobbs Visually inspect control column 5472.1 for cracks, para A, none	<i>Robert</i>	AP 2922915	19551.3
9-21-24 1954.6	Inspected per AD para A, No discrepancies noted	<i>Robert</i>	AP 3690633	19642.6
2706724 19609.1	Hobbs 5629.9 Inspected per AD paragraph A; no discrepancy noted	<i>Robert</i>	AP 2922915	19709.1
12/05/24 19708.7	Hobbs 5729.5 Inspected per AD Paragraph A; no discrepancies noted	<i>Robert</i>	AP 3811332	19808.7
12-26-24 19796.2	Inspected per AD para A No Discrepancies noted	<i>Robert</i>	AP 3690633	19896.2
1-14-25 19891.2	Inspected per AD para A. No Discrepancies noted.	<i>Robert</i>	AP 3690633	19991.2

FORM AD01

RECURRING AD COMPLIANCE RECORD

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Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: <u>N/A</u>	P/N: <u>N/A</u>
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: <u>N/A</u>	S/N: <u>N/A</u>

AD 83 - 04 - 05

DESCRIPTION: Inspect Control Column for cracks

Visually inspect the control column lower weld assembly for cracks IAW the Accomplishment Section of de Havilland S/B No. 3/39. If crack(s) are found See AD for instructions.

METHOD OF COMPLIANCE: No. 3/39. If crack(s) are found See AD for instructions.

INTERVAL: 100 Hrs.

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
1/28/25 19949.7	Inspected IAW AD Par a. Using A Mirror & Flashlight Per SB 3/39, Accomplishment Instruction, Inspection	<i>[Signature]</i>	A&P 3690633	20049.7
2-17-25 20147.9	60687. Insp per para A, E SB/3/39. no discrepancies noted.	<i>[Signature]</i>	A&P 2922915	20147.9
3-8-25 20145.6	Inspected IAW AD par a. per SB 3/39 No defects Noted	<i>[Signature]</i>	A&P 3690633	20245.6
3-28-25 20240.5	Inspected IAW AD par a per SB 3/39 no defects noted	<i>[Signature]</i>	A&P 3690633	20340.5
4-10-25 20306.9	Inspected IAW AD Par SB/3/39 No defects Noted	<i>[Signature]</i>	A&P 3690633	20406.9
4-24-25 20325.1	Inspected IAW AD per SB 3/39 no defects noted	<i>[Signature]</i>	A&P 3690633	20425.1
5-1-25 20422.1	Inspected IAW AD per SB 3/39 No Defects noted	<i>[Signature]</i>	A&P 3690633	20522.1
5-23-25 20517.6	Inspected IAW AD Per SB 3/39 no Defects noted	<i>[Signature]</i>	A&P 3811332	20617.6
6-11-25 20615.5	Inspected IAW AD Per SB 3/39 No Defects Noted	<i>[Signature]</i>	A&P 3690633	20715.5
6-17-25 20688.2	Inspected IAW AD per SB 3/39 No defects noted	<i>[Signature]</i>	A&P 3690633	20778.2

FORM AD01

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: <u>N/A</u>	P/N: <u>N/A</u>
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: <u>N/A</u>	S/N: <u>N/A</u>

AD 83 - 04 - 05

DESCRIPTION: Inspect Control Column for cracks

Visually inspect the control column lower weld assembly for cracks IAW the Accomplishment Section of de Havilland S/B

METHOD OF COMPLIANCE: No. 3/39. If crack(s) are found See AD for instructions.

INTERVAL: 100 Hrs.

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
7-9-25 2074.6	Inspected IAW AD Par a. Using A Mirror & Flashlight Per SB 3/39, Accomplishment Instruction, Inspection	<i>Ruth</i>	A+P 3690633	20861.8
7/28/25 20846.6 20846.6	Inspected IAW AD Par a. Using a mirror & Flashlight Per SB 3/39, Accomplishment Instruction, Inspection.	<i>Ruth</i>	A+P 3811332	20846.6
Aug 4, 2025 20882.2	Inspected Visually with a mirror Light Per SB No cracks	<i>Ruth</i>	A+P 574601704	20982.2
8-23-25 20953.9	Inspected IAW AD Par a, Per SB 3/39, No cracks Noted	<i>Ruth</i>	A+P 3690633	21053.9
10-10-25 21024.5	Inspected IAW AD Par A Per SB 3/39 No cracks Noted	<i>Ruth</i>	A+P 3690633	21124.5
10-28-25 21097.1	Inspected IAW AD Par A Per SB 3/39 No cracks Noted	<i>Ruth</i>	A+P 3690633	21197.1
11-23-25 21195.1	Inspected IAW AD Par A Per SB 3/39, No cracks Noted	<i>Ruth</i>	A+P 3690633	21295.4
12-10-25 21275.1	Inspected IAW AD Par A per SB/3/39 No cracks Noted	<i>Ruth</i>	A+P 3690633	21375.1
12-17-25 21309.3	Inspected IAW AD par A per SB 3/39. No cracks Noted	<i>Ruth</i>	A+P 3690633	21409.3
1-3-26 21394.8	Inspected IAW AD Par A Per SB 3/39. No cracks Noted	<i>Ruth</i>	A+P 3848142	21494.8

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S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: <u>N/A</u>	S/N: <u>N/A</u>

AD 83 - 04 - 05

DESCRIPTION: Inspect Control Column for cracks

Visually inspect the control column lower weld assembly for cracks IAW the Accomplishment Section of de Havilland S/B

METHOD OF COMPLIANCE: No. 3/39. If crack(s) are found See AD for instructions.

INTERVAL: 100 Hrs.

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
21453.6 1-22-76	Inspected IAW AD Par a. Using A Mirror & Flashlight Per SB 3/39, Accomplishment Instruction, Inspection	<i>[Signature]</i>	AOP 3690633	21583.6
21524.4 2-5-76	Inspected IAW AD Par A and SB No. 3/39. No crack noted	<i>[Signature]</i>	AOP 3245143	21624.4
21627.7 2-26-76	Inspected IAW AD Par A and SB 3/39 No defects noted	<i>[Signature]</i>	AOP 3690633	21727.7
21707.2 3-16-76	Inspected IAW AD Par A and SB 3/39 No defects noted	<i>[Signature]</i>	AOP 3690633	21807.2
21767.4 3-28-76	Inspected IAW AD Par A and SB 3/39. No defects noted.	<i>[Signature]</i>	AOP 3690633	21867.4

RECURRING AD COMPLIANCE RECORD

AIRCRAFT

APPLIANCE

Reg. #: N/A 708KA MAKE: de Havilland

MAKE: Hartzell P/N: HC-B3TN-3DY/TN+

S/N: N/A MODEL: DHC-3

MODEL: HC-B3TN-3DY S/N: BUA34543

AD 83 . 08 . 01R2

DESCRIPTION: Use proper hardware and torque all serviceable and new propellers at initial installation

METHOD OF COMPLIANCE: Per AD Par. (a) Install all new propellers and serviceable propellers, as follows, before further flight, complete AD Par. (a) items (1) to (6)

INTERVAL: At installation of each new or 0 TSO overhauled propeller

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
25-Jun-19 16031.4	Per records search, found previously complied with on <u>SN: 22120</u> recurring record from Kenmore Air	<i>Robt O</i>	A&P 574601404	At Each Prop Change
<u>Aug 12, 24</u> 19,479.4	Installed Propeller per AD torque sequence and per Hartzell torque sequence. <u>SN: BUA19230</u>	<i>Robt O</i>	A.P 574601404	Prop Change
<u>July 16, 24</u> 20,793.6	Installed Prop per AD Torque Sequence & per Hartzell Torque Sequence <u>SN: BUA31868</u>	<i>Robt O</i>	A.P 574601404	Prop Change

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S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 86 - 16 - 12

DESCRIPTION:	Visually inspect elevator control system - to preclude loss of elevator control and to ensure proper operation
METHOD OF COMPLIANCE:	Visually inspect elevator pushrod, pushrod to rear quadrant area, threaded shank of pushrod rod ends, for cracks and all elevator control system cables and pulleys for damage or distortion and replace unserviceable parts, prior to further flight per "ACCOMPLISHMENT INSTRUCTIONS" in de Havilland S/B No. 3/35, and S/B No. 3/43
INTERVAL:	When a parked airplane has been subjected to wind gusts of 60 mph or greater (including helicopter or jet blasts)

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
29-Mar-23 17,245.2	No Previous record found <i>Found Complied w. 74 ON 700.0</i>	<i>[Signature]</i>	A&P 574601404	Now
April-23 March-47	Visually inspect elevator control system for cracks, damage, or distortion IAW AD Par. (a), (1), (2), & (3).	<i>[Signature]</i>	A&P 574601404	Event Based
<i>13 MAR 23</i> <i>17,219.0</i>	<hr/> <hr/>			

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S/N: <u>59</u>	MODEL: <u>DHC-3</u> MODEL: _____ S/N: _____

AD 2011-18-11

Per AD Par. (f) (1) Inspect Elevator Control Tabs per MM TR 18, 19, and 20. **NOTE:** TRs 18, 19, & 20 are now revised into the PSM1-3-2, MM, Appendix 4, Part 3, 7, & 7A (page 4-11?), Part 2, Par. 2.24.4, (page 44?), Part 2, 2.25.4 (page 46?), & Figure 2-13A.

DESCRIPTION: _____

Visual inspect per AD par (f)(1): end cap and lever for condition and fasteners for security, bearing for condition and security, connecting rod for condition and fasteners for security, the total maximum free play of the elevator tabs is not to exceed 1.0 degrees or 0.070 inches at the trailing edge of the elevator

METHOD OF COMPLIANCE: _____

INTERVAL: 100 Hours

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
6-16-25 20715.5	visually Inspected Per AD Free play less than .070"	<i>[Signature]</i>	AAP 3690633	20715.5
6-17-25 20662.2	visually Inspected Per AD Free play less than .070"	<i>[Signature]</i>	AAP 3690633	20778.2
7-9-25 20761.8	visually Inspected Per AD Free Play less than .070"	<i>[Signature]</i>	AAP 3690633	20861.8
27/28/25 20846.6	visually Inspected Per AD Free play less than 0.070"	<i>[Signature]</i>	AAP 3811332	20946.6
Aug 4, 25 20882.2	visually inspected with a scale Found < .070" Freeplay	<i>[Signature]</i>	AAP 5746014	20982.2
8-23-25 20953.9	visually inspected per AD Free Play less than .070"	<i>[Signature]</i>	AAP 3690633	21053.9
10-10-25 21024.5	visually Inspected Per AD Free Play less than .070"	<i>[Signature]</i>	AAP 3690633	2124.5
10-28-25 21097.1	visually Inspected Per AD Free Play less than .070"	<i>[Signature]</i>	AAP 3690633	21197.1
11-23-25 21195.4	visually Inspected Per AD Free Play less than .070"	<i>[Signature]</i>	AAP 3690633	21295.4

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Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 2011 - 18 - 11

Per AD Par. (f) (1) Inspect Elevator Control Tabs per MM TR 18, 19, and 20. **NOTE:** TRs 18, 19, & 20 are now revised into the PSM1-3-2, MM, Appendix 4, Part 3, 7, & 7A (page 4-11?), Part 2, Par. 2.24.4, (page 44?), Part 2, 2.25.4 (page 46?), & Figure 2-13A.

DESCRIPTION: _____

Visual inspect per AD par (f)(1): end cap and lever for condition and fasteners for security, bearing for condition and security, connecting rod for condition and fasteners for security, the total maximum free play of the elevator tabs is not to exceed 1.0 degrees or 0.070 inches at the trailing edge of the elevator

METHOD OF COMPLIANCE: _____

INTERVAL: 100 Hours

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
1-14-25 19831.2	Visually Inspected per AD Freeplay less than .070"	<i>R. L. P.</i>	A&P 3690633	19991.2
1-28-25 19949.7	Visually Inspected per AD Freeplay less than .070"	<i>R. L. P.</i>	A&P 3690633	20049.7
2/17/25 20087.9	6068-7 Hobbs Visually inspected per AD no discrepancies noted.	<i>Patricia K. M.</i>	AP 292285	20147.9
3-8-25 20145.6	Visually Inspected per AD Freeplay less than .070"	<i>R. L. P.</i>	A&P 3690633	20245.6
3-28-25 20240.5	Visually Inspected per AD Free play less than .070"	<i>R. L. P.</i>	A&P 3690633	20340.5
4-10-25 20306.9	Visually Inspected Per AD Free Play less than .070"	<i>R. L. P.</i>	A&P 3690633	20406.9
4-24-25 20385.1	Visually Inspected Per AD Free Play less than .070"	<i>R. L. P.</i>	A&P 3690633	20485.1
5-1-25 20422.1	Visually Inspected per AD Free Play less than .070"	<i>R. L. P.</i>	A&P 3690633	20522.1
5-23-25 20517.6	Visually Inspected Per AD Free Play less than .070"	<i>[Signature]</i>	A&P 3811332	20617.6

FORM AD01

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 2015 - 24 - 2

DESCRIPTION:	Inspect Wing Corugations & Structure per Par. (f)(3) of this AD. Perform repetitive borescope and visual inspections of both the left-hand and right-hand wing box following Part B of the Accomplishment Instructions of SB V3/0002
METHOD OF COMPLIANCE:	Visually with borescope, mirror & light inspected both the left-hand and right-hand wing box per AD par (f)(3) & Part B of the Accomplishment Instructions of SB V3/0002
INTERVAL:	1,600 hours or 2,100 cycles

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Hours Cycles
26-Jan-23 17,218.5	Per records search, found previously complied with on recurring record from Kenmore Air	<i>Robert D. [Signature]</i>	A&P 574601404	18,818.5 10,292
24 JAN 24 18,440.9	Inspected both the left and right wing box per AD par (f)(3) & SB V3/0002, II., B.	<i>[Signature]</i>	A&P 2922915	20,040.9 12,354.0 22,600
Nov 11 24 19,615.0	Inspected wing structure per AD C:12245 (A)(3) & SB V3/0002, II., B.	<i>Robert D. [Signature]</i>	A&P 574601404	21,212.500 Hours 14,345 Cycles
July 22, 25 20,823.8	Inspected wing structure for cracks & Rivet security per AD 503	<i>Robert D. [Signature]</i>	A&P 574601404	22,423.8 Hours 16,422 Cycles
14,322				



SERVICE BULLETIN TRANSMITTAL SHEET

Service Bulletin No.: V3/0002	Date of Original Issue: 22-September-2010
Revision No.: Revision C	Date of Issue: 30-April-2014

DHC-3 COMPLETE WING BOX INSPECTION – BORESCOPE AND VISUAL

Viking has received SDRs related to cracking of the DHC-3 wing upper skin corrugations as well as cracks in the main spar lower cap. The intent of this Service Bulletin is to detect and repair these cracks and to obtain as much information as possible from operators to evaluate the overall condition of the fleet to assess if further action is required.

NOTE 1: NO RETROACTIVE THRESHOLD INSPECTION IS REQUIRED IF THRESHOLD INSPECTION FROM SB V3/0002 REVISION A OR B HAS BEEN COMPLETED – CONTINUE WITH REPEAT INSPECTIONS IN ACCORDANCE WITH SB V3/0002 REVISION C AT APPROPRIATE REPEAT INSPECTION SCHEDULE.

NOTE 2: ALL OTHER AIRCRAFT MUST INCORPORATE APPLICABLE INSPECTIONS IN ACCORDANCE WITH SB V3/0002 REVISION C.

NOTE 3: This revision supersedes the previous revision of this Service Bulletin.

NOTE 4: Revision bars in the left margin of this Service Bulletin denote changes from the previous revision.

This page transmits Revision 'C' of Service Bulletin V3/0002, pages 1 thru 12, dated 30 April 2014.

Service Bulletin re-issued to include incorporation of MOD 3/1005 (MOD for inspection access holes) and detailed compliance and accomplishment instructions for the inspection of the entire wing box assembly.

Remove Service Bulletin V3/0002 Revision 'B' dated 03 July 2013 and replace with attached.

Previous issues of Service Bulletin V3/0002:

- Initial Issue – (22-Sep-2010) - effectivity – All DHC-3 aircraft
- Revision A – (22-Feb-2013) - effectivity – All DHC-3 aircraft
- Revision B – (03-Jul-2013) - effectivity - All DHC-3 aircraft

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VIKING*

1959 De Havilland Way
Sidney, B.C. Canada
V8L 5V5
Tel: (250) 656 - 7227

DHC-3 OTTER

SERVICE BULLETIN

ATA SYSTEM: 5700

NUMBER: V3/0002

SUBJECT: DHC-3 Complete Wing Box Inspection – Borescope and Visual

I. PLANNING INFORMATION

A. Effectivity

The complete wing box inspection requirement laid out herein, has an effectivity of ALL DHC-3 aircraft.

NOTES: To perform this inspection, improved access is required. Refer to Section II.A. of this Service Bulletin for details.

Before you do this Service Bulletin, read ALL Sections of the bulletin and examine all Supplemental Type Certificate (STC) or equivalent action changes to make sure this Service Bulletin can be completed.

FOR AIRCRAFT EQUIPED WITH STCs or EQUIVALENT CHANGES, THE OPERATOR MUST CONTACT THE STC HOLDER TO EVALUATE CONFLICT POSSIBILITIES.

B. Reason

Viking has received Service Difficulty Reports (SDR's) related to cracking of the DHC-3 wing upper skin corrugations as well as cracks in the main spar lower cap. The intent of this Service Bulletin is to detect and repair these cracks and to obtain as much information as possible from operators to evaluate the overall condition of the fleet and assess if further action is required. All wings require internal inspections in order to maintain long term structural integrity.

The purpose of incorporating additional inspection openings is to provide sufficient access to complete a thorough inspection of the DHC-3 wing.

C. Description

Inspections are required to identify possible cracking, corrosion or any other damage of the wing box assembly.

To accommodate these inspections, access panels are incorporated in the wing lower skin in accordance with MOD 3/1005 (MOD Kit DWG C3MK1005; MOD Kit C3MK1005-1).

30-April-2014
Revision 'C'

S.B. V3/0002
Page 1 of 12

SERVICE BULLETIN V3/0002

The purpose of this Service Bulletin is to:

1. Borescope inspect and visually inspect the wing box for cracks and corrosion.
2. As soon as possible following the first inspection accomplished in accordance with Revision C of the Service Bulletin, complete the Operator Reply Form included with this Service Bulletin even if no cracks or damage is found.
3. Repair damage prior to next flight.

D. Compliance

The compliance time for actions associated with this Service Bulletin is listed in **Table 1** below.

NOTE 1: To accomplish inspection (action) from **Table 1** below, incorporate lower wing skin access openings as defined in Section II.A. of this Service Bulletin.

NOTE 2: If actual flight hours of wing are not known, contact Technical Support at Viking Air Limited (see Sec. I.J for contact information) to assist in the determination.

NOTE 3: If SB V3/0002 Rev A or B has been complied with prior to the issue date of SB V3/0002 Rev C, continue with the repeat inspections, in accordance with **Table 1** below, from the last completed inspection in accordance with SB V3/0002 Rev A or B.

NOTE 4: Damage shall be repaired prior to next flight in accordance with accepted data (such as PSM1-3-3) or other approved data.

TABLE 1 – WING BOX INSPECTION - COMPLIANCE DETAILS

Action	Action Detail	Effectivity	Threshold Inspection	Repeat Inspection
Complete wing box inspection	Do a borescope and visual inspection on ALL areas of the Wing Box in accordance with (Sec. II).	a. <31,200 wing flight hours	Within 800 wing flight hours or within 6 months of issue date of this SB (whichever occurs first) Also see note 3	Every 1600 wing flight hours or 2100 flight cycles (whichever occurs first)
		b. $31,200 \leq$ wing flight hours <31,600	Before accumulating 32,000 wing flight hours or within 6 months of issue date of this SB (whichever occurs first)	
		c. $\geq 31,600$ wing flight hours	Within 400 wing flight hours or 3 months of the issue date of this SB (whichever occurs first)	

E. Approval

The technical content of this Service Bulletin has been approved under the authority of Transport Canada Civil Aviation (TCCA) Design Approval Organization (DAO) 04-V-02.

SERVICE BULLETIN V3/0002

F. Manpower

An estimated 36 man-hours per wing will be required to complete this Service Bulletin (16 man hours to install access panels and 20 hours to inspect the wing components).

All costs to complete the inspection of the wing components are the responsibility of the Operator.

G. Weight and Balance

FOR MOD 3/1005 (Current)		FOR MOD 3/1002 (Original)	
AIRCRAFT STATION	WEIGHT	AIRCRAFT STATION	WEIGHT
STA 139.25	+1.58 lbs	STA 134.15	+1.58 lbs
STA 159.63	+0.60 lbs	STA 159.63	+0.60 lbs

H. Electrical Load Data

Not affected.

I. Publications Affected

None.

J. References

This Service Bulletin contains a Survey to be completed and returned to Viking. The following drawings have been provided with this Service Bulletin to illustrate the technical activities: **C3MK1005**, **C3Z1001**, and **C3Z1002**. For more information please contact:

Viking Air Limited Technical Support
1959 De Havilland Way
Sidney, British Columbia
V8L 5V5
technical.support@vikingair.com
FAX: 250-656-0673
Telephone Numbers:
Regional: 250-656-7227
North America: 1-800-663-8444
International: 1-800-6727-6727

SERVICE BULLETIN V3/0002

II. ACCOMPLISHMENT INSTRUCTIONS

A. INTRODUCTION OF NEW ACCESS HOLES IN THE LOWER WING SKIN

1. To perform this inspection, Viking is providing MOD 3/1005 (MOD KIT DWG C3MK1005) for the incorporation of access holes which is only suitable for incorporation on OEM wings used on OEM aircraft.

NOTE 1: (a) If MOD 3/1002 (MOD Kit DWG C3MK1002) has been incorporated in accordance with SB V3/0002 Revision A or B, the incorporation of MOD 3/1005 (MOD Kit DWG C3MK1005) is not required in order to accomplish the inspection requirements of this Service Bulletin.

(b) If the inspection opening at wing station 53.75 has been incorporated in accordance with SB V3/0002 Revision NC, no change is required at this location, but the remaining inspection openings specified in MOD 3/1005 (MOD Kit DWG C3MK1005) are required unless MOD 3/1002 (MOD Kit DWG C3MK1002) was also incorporated.

NOTE 2: MOD 3/1005 (MOD Kit DWG C3MK1005) is compatible for wings fitted with STC SA94-114 (or US STC SA00287NY) STOL Kit.

NOTE 3: For aircraft fitted with Viking STC SA03-50, refer to Service Bulletin SB 3-STC(03-50)-001 for applicable access instructions.

NOTE 4: For Wings / Aircraft equipped with STCs or equivalent changes, other than mentioned in Section II.A.1. above, the operator must contact the STC holder to evaluate conflict possibilities.

2. Add applicable new access holes in accordance with MOD Kit C3MK1005-1 on both Left Hand (LH) Wing and Right Hand (RH) Wing.
3. Parts can be fabricated by qualified individuals or purchased from Viking Product Support.

B. COMPLETE WING BOX INSPECTION

The Left Hand (LH) Wing:

1. Remove all inspection access panels / plugs between wing root and wing tip, including those for the access holes incorporated in Section II.A. above, and remove wing tip in accordance with PSM1-3-2 Section 2.15.1.
2. Using a flexible borescope, inspect for cracks, corrosion, missing/smoking/working /broken fasteners or other damage on the entire internal structure of wing between wing root and wing tip, upper/lower skins, ribs (webs, upper/lower flanges), fore and aft surfaces of nose spar, main spar and rear spar (including caps and webs). Pay particular attention to the following areas for cracks in fastener holes and open holes:
 - (a.) Skin splice;
 - (b.) Spar Web splice;
 - (c.) Spar cap splice;

SERVICE BULLETIN V3/0002

- (d.) Hinge attachments on rib and rear spar;
 - (e.) Fastener holes on skin (especially those common to the inboard/outboard-end-row of fastener holes on the reinforcement doublers);
 - (f.) Access cutouts and lightening holes on spar/rib webs;
 - (g.) Rib attachment to the webs and caps of the nose spar, main spar, and rear spar, looking for cracks in flanges, bending radii and at fastener heads;
 - (h.) Corrugations at rib attachments (both forward and aft of the main spar). See **Figure 1A** and **1B** for typical corrugation cracks;
 - (i.) All fastener holes on the lower spar caps (main and rear spar), lower stringers and lower skins between wing root and wing tip. Note that fatigue prone areas may be located between Sta 135 and 185. See **Figure 1C** for example of cracks on main spar lower cap.
3. Carry out a detailed inspection of the exterior surface of the wing for cracks, corrosion, missing/smoking/working/broken fasteners or other damage. Pay particular attention to skin splice, and access cutouts.
 4. Do a visual inspection around doubler area of inspection holes at every repeat inspection.
 5. Document all findings and cracks (noting length, direction, wing station position, smoking fasteners, oversized or broken fasteners, etc.); include pictures identifying the location (use station number) and orientation of all findings.
 6. Complete the attached Post Inspection Survey Report Form (even if no damage was found) and forward to Technical Support as soon as possible (see Sec. I.J for contact information).
 7. Repair damage prior to next flight.
NOTE: Repairs shall be in accordance with accepted data such as PSM1-3-3 or other approved data.
 8. Re-install inspection panels and wing tip in accordance with PSM1-3-2 Section 2.15.2.

The Right Hand (RH) Wing:

9. Repeat Step II.B.1 to Step II.B.8 on the RH wing.

D. RECORDING

1. Make an entry in the aircraft flight log book upon completion of access hole incorporation associated with MOD 3/1005: "INCORPORATION OF MOD 3/1005 ACCESS HOLES COMPLETED IN ACCORDANCE WITH SECTION II.A OF SB V3/0002 Revision C AT XX,XXX WING FLIGHT HOURS FOR WING S/N XXX."
2. Complete Operator Reply Form attached following the first inspection incorporated

SERVICE BULLETIN V3/0002

in accordance with Revision C of Service Bulletin V3/0002, and send it to Viking Air Limited Technical Support.

3. Make an entry in the aircraft flight log book upon completion of Section II.B of this Service Bulletin: "INSPECTIONS COMPLETED IN ACCORDANCE WITH SECTION II.B OF SB V3/0002 Revision C AT XX,XXX FLIGHT HOURS FOR WING S/N XXX; INSPECTIONS RESULTS SENT TO VIKING AIR LIMITED."

III. MATERIAL INFORMATION AND TOOLING

A. Material

- Modification Kit C3MK1005-1 (includes MOD Kit DWG C3MK1005 and DWGs C3Z1001 & C3Z1002)

B. Tooling – Price and Availability

- (a) Flexible Borescope Kit with light
- (b) Flashlight and mirror
- (c) Wing Support Stand
- (d) Standard AME hand tools

IV. FIGURES AND TABLES

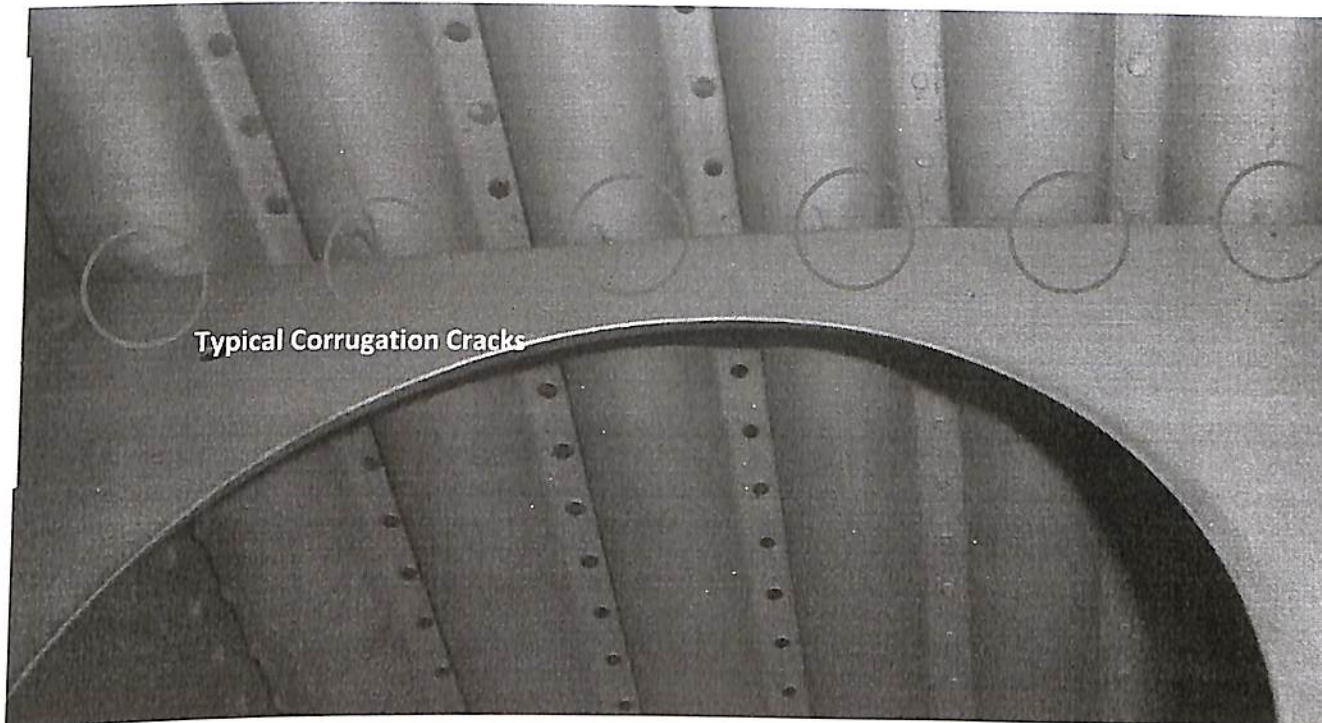


FIGURE 1A – TYPICAL CORRUGATION CRACKS

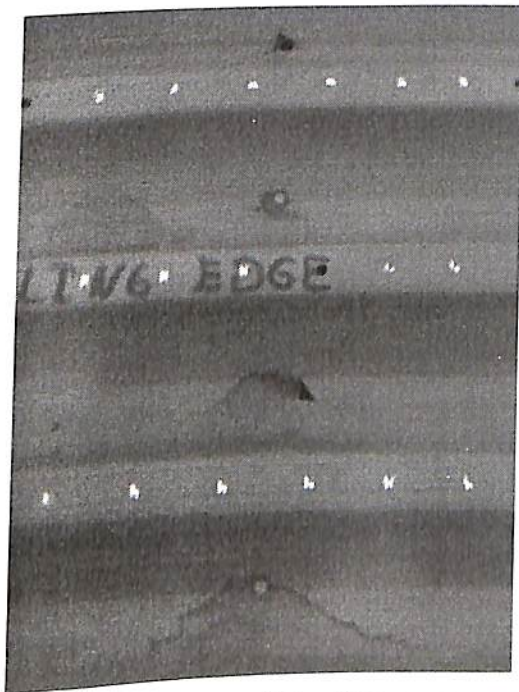


FIGURE 1B – CORRUGATION CRACKS ON ONE OF THE WING SDR

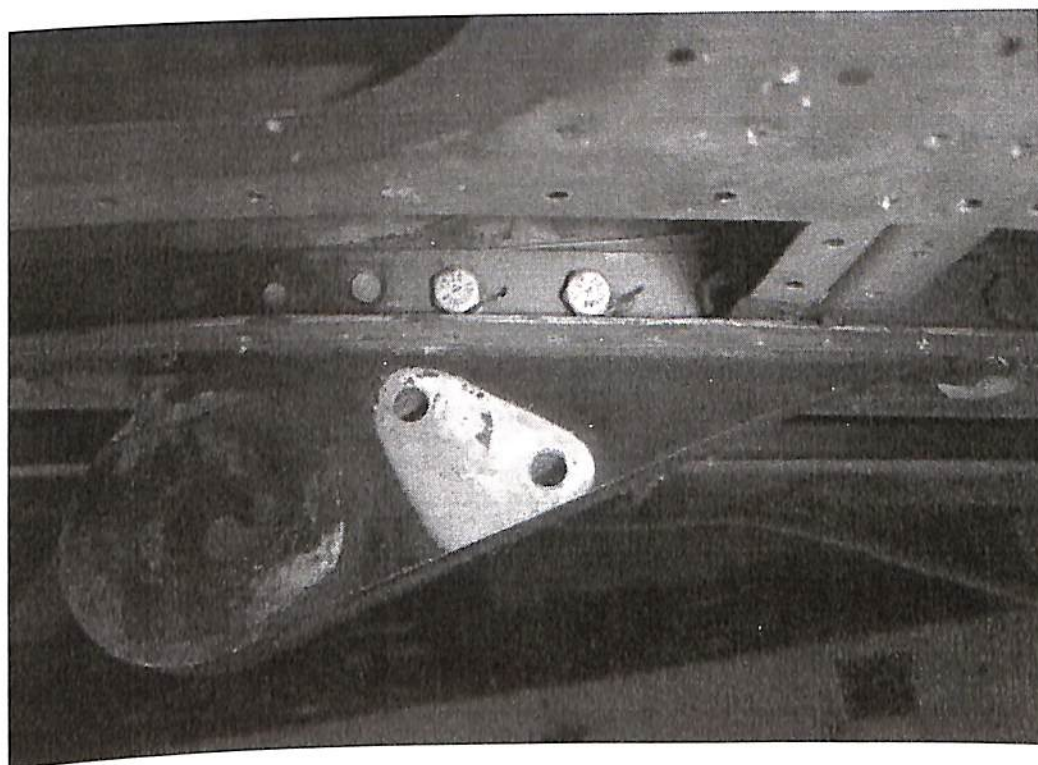
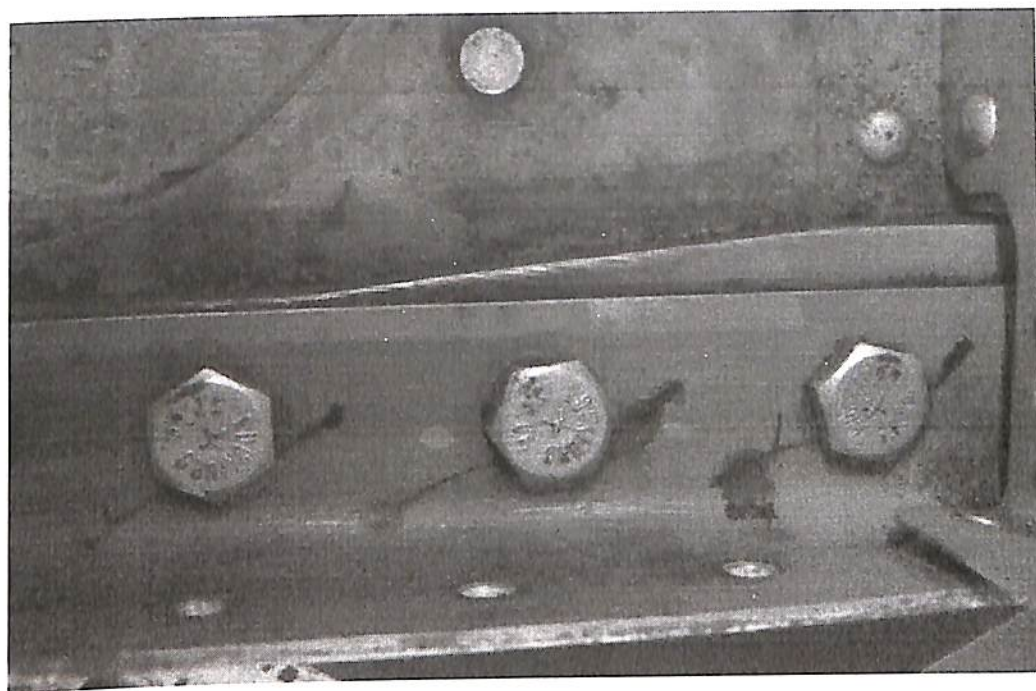
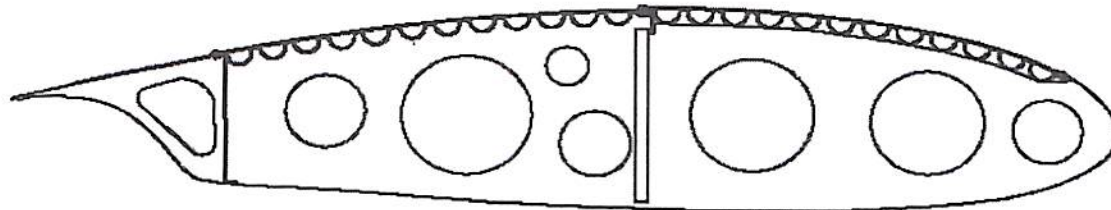


FIGURE 1C – CRACKS ON MAIN SPAR LOWER CAP FROM ONE OF THE WING INSPECTION SDR

SERVICE BULLETIN V3/0002



Rear Spar		Main Spar	Front Spar	Sta. No.
4	3	2	1	37.15
8	7	6	5	47.5
11	10		9	60
14	13		12	72.5
17	16		15	85
20	19		18	97.5
23	22		21	110
26	25		24	122.5
29	28		27	135
32	31		30	147.5
35	34		33	160
38	37		36	172.5
41	40		39	185
44	43		42	197.5
47	46		45	210
50	49		48	222.5
53	52		51	235
56	55		54	247.5
59	58		57	260
62	61		60	272.5
65	64		63	285
68	67		66	297.5
71	70		69	310
74	73		72	322.5
				335

NOTE: NOT TO SCALE - FOR REFERENCE PURPOSES ONLY

FIGURE 2 - FINDING LOCATION DIAGRAM

LEFT HAND WING - LOOKING UP AT TOP SKIN

Document all findings and crack locations on this page
Noting length, WS position, rivet locations, loose, oversized or blind fasteners, etc

30-April-2014
Revision 'C'

S.B. V3/0002
Page 10 of 12

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 2021 - 24 - 18

Due to fatigue damage of the (wing strut lug fitting components or
 DESCRIPTION: the) fuselage to wing strut attachment/tie-bar

For airplanes with a maximum certificated gross weight that has
 ever exceeded 8,000 pounds: Remove from service each left and
 right wing strut fitting and tie-bar per SB V3/0008, and SB 3/37, at
 METHOD OF COMPLIANCE: whichever of the following compliance times that occurs later

INTERVAL: 16,100 hours time in service (32,200 equivalent air time hours)

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due TTAF
30-Dec-15 13,979.2	Per records search, found previously complied with on recurring record & airframe logbook from Kenmore Air	<i>[Signature]</i>	A&P 574601404	30079.2
	Replace Tie-Bar per SB3/37			

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 2021 - 24 - 18

DESCRIPTION: Due to fatigue damage of the Left and Right wing strut lug fitting components (fuselage to wing strut attachment/tie-bar)

For airplanes with a maximum certificated gross weight that has ever exceeded 8,000 pounds: Remove from service each left and right wing strut fitting and tie-bar per SB V3/0008, and SB 3/37, at

METHOD OF COMPLIANCE: whichever of the following compliance times that occurs later

INTERVAL: 16,100 hours time in service (32,200 equivalent air time hours)

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due TTAF
26-Jan-23 17,218.5	Per records search, found previously complied with on recurring record from Kenmore Air	<i>[Signature]</i>	A&P 574601404	33,318.5
	Replaced both wing strut lug fitting components with new			

RECURRING AD COMPLIANCE RECORD

AIRCRAFT	APPLIANCE
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u> MAKE: _____ P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u> MODEL: _____ S/N: _____

AD 2022 - 21 - 51

DESCRIPTION: _____ METHOD OF COMPLIANCE: _____ INTERVAL: <u>110 Hours</u>	Remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs.
Visual inspect per AD par (h)(1): no cracks or corrosion beyond level 1 allowed	

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
11-23-25 21195.4	visually Inspected Per AD par (h)(1) No discrepancies noted	<i>[Signature]</i>	AIP 3690633	21305.1
12-10-25 21275.1	visually Inspected Per AD par (h)(1) No discrepancies noted	<i>[Signature]</i>	AIP 3690633	21385.1
12-18-25 21309.3	visually Inspected Per AD par (h)(1) No discrepancies noted	<i>[Signature]</i>	AIP 3690633	21419.3
21394.8 1-3-26	visually Inspected Per AD par (h)(1) No discrepancies noted	<i>[Signature]</i>	DIP 3245143	21504.8
21483.6 1-22-26	visually Inspected Per AD par (h)(1) No defects noted	<i>[Signature]</i>	AIP 3690633	21593.6
21521.1 2-5-26	visually Inspected Per AD par (h)(1) No defects noted	<i>[Signature]</i>	DIP 3245143	21634.4
21627.7 2-26-26	visually Inspected Per AD par (h)(1) No defects noted	<i>[Signature]</i>	AIP 3690633	21732.7
21707.2 3-16-26	visually Inspected Per AD par (h)(1) No defects noted	<i>[Signature]</i>	AIP 3690633	21807.2
21767.4 3-22-26	visually Inspected Per AD par (h)(1) No defects noted	<i>[Signature]</i>	AIP 3690633	21877.4

FORM AD01

RECURRING AD COMPLIANCE RECORD

AIRCRAFT	APPLIANCE
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u> MAKE: _____ P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u> MODEL: _____ S/N: _____

AD 2022 - 21 - 51

DESCRIPTION: <u>Remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs.</u>	
METHOD OF COMPLIANCE: <u>Visual inspect per AD par (h)(1): no cracks or corrosion beyond level 1 allowed</u>	
INTERVAL: <u>110 Hours</u>	

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
5/27/24 19108.1	Visually Inspected for cracks & corrosion	<i>Pete</i>	AD 574601401	19218.1
6-7-24 19171.6	Visually inspected per AD par (h)(1) no cracks	<i>Pete</i>	AP 2922815	19281.6
6-27-24 19269.3	NOBBS 5290-1: Visually inspected per AD par (h)(1) no cracks	<i>Pete</i>	AP 2922815	19379.3
7-16-24 19367.0	hobbs 5387.8 Inspected per AD - freeplay less than	<i>Pete</i>	AP	19367.0
-	para (h)(1) no cracks detected	<i>K</i>	2922815	-
8-6-24 19451.3	Visually inspected per AD no cracks/corrosion	<i>Pete</i>	AP 2922815	19561.3
9-21-24 19542.6	visually inspected per AD no cracks or corrosion noted	<i>Pete</i>	AD 3690632	19652.6
2706724 19609.1	Visually inspected per AD par (h)(1) no discrepancies noted	<i>Pete</i>	AP 2922815	19619.1
12/05/24 19708.7	visually inspected AD paragraph (h)(1) no discrepancies noted.	<i>Pete</i>	AP 3811332	19818.7

FORM AD01

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____

AD 2022 - 21 - 51

Remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs.

DESCRIPTION: Remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs.

METHOD OF COMPLIANCE: Visual inspect per AD par (h)(1): no cracks or corrosion beyond level 1 allowed

INTERVAL: 110 Hours

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
12-26-21 197962	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	19906.2
1-14-25 198912	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	20,001.2
1-22-25 19949.7	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	20,059.7
2-17-25 20,047.9	6068-7 visually inspected per AD para (h)(1) no discrepancies noted	<i>[Signature]</i>	AP 2422815	20,147.9
3-8-25 201456	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	20255.6
3-22-25 20240.5	visually inspected per AD para (h)(1) no cracks or defects found	<i>[Signature]</i>	AIP 3690633	20350.5
4-10-25 20306.9	visually inspected per AD para (h)(1) no cracks or defects	<i>[Signature]</i>	AIP 3690633	20416.9
4-24-25 20385.1	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	20485.1
5-1-25 20472.1	visually inspected per AD para (h)(1) no cracks noted	<i>[Signature]</i>	AIP 3690633	20532.1

FORM AD01

RECURRING AD COMPLIANCE RECORD

AIRCRAFT		APPLIANCE	
Reg. #: <u>N708KA</u>	MAKE: <u>de Havilland</u>	MAKE: _____	P/N: _____
S/N: <u>59</u>	MODEL: <u>DHC-3</u>	MODEL: _____	S/N: _____



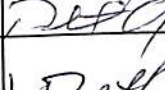
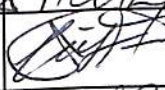
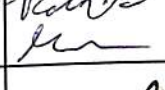
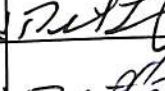
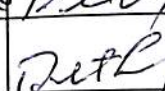


AD 2022 - 21 - 51

Remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs.

DESCRIPTION: repairs.

METHOD OF COMPLIANCE: Visual inspect per AD par (h)(1): no cracks or corrosion beyond level 1 allowed

INTERVAL: 110 Hours

Time Done Date/TTAF	Method Of Compliance	Mechanics Signature	Certificate Type & Number	Next Due Date
05-23-25 20517.6	Visually Inspected Per AD par (H)(I) No cracks noted		A&P 3811332	20627.6
4-11-25 20615.5	Visually Inspected Per AD par (H)(I) No cracks noted		A&P 3690633	20725.5
6-17-25 20662.2	Visually Inspected Per AD par (H)(I) No cracks noted		A&P 3690633	20772.2
7-9-25 20761.2	Visually Inspected Per AD par (H)(I) No cracks or corrosion noted		A&P 3650633	20871.6
07/28/25 20846.6	Visually Inspected Per AD par (H)(I) No cracks or corrosion noted		A&P 3811332	20956.6
Aug 4, 25 20882.2	Visually Inspected w/ light, mirror & Borescope No cracks or corrosion		A&P 57460140	20,992.2
Aug 23, 25 20953.9	Visually Inspected Per AD par (H)(I) No cracks or corrosion noted		A&P 3690633	21063.9
10-10-25 21024.5	Visually Inspected Per AD par (H)(I) No cracks or corrosion noted		A&P 3690633	21134.5
10-22-25 21097.1	Visually Inspected Per AD par (H)(I) No cracks or corrosion noted		A&P 3690633	21207.1

FORM AD01

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 1 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
56-17-02 CF-57-06 CF-56-11	All S/Ns: Flap system must be modified per EB "O" 34, install Mods 3/731, 1 pc. Flap actuator/ratchet valve assy, 3/744 flap interconnect & tab, & 3/745 fiberglass TP leading edge by Nov 1, 1957	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/10/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-54-09	Item #1, due on SN 2-15, 17, 22, 23, 26, 27, & 28 W/O Mod 3/341, within 100 hrs. install jumper IAW TNS "O" 22.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	NA For PT6A STC	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-55-06	Item #2, due on SN 2-15, 17-42, 44, 45, 46, 52-65, & 67-70 W/O Mod 3/558, within 100 hrs. replace breather & box IAW EB "O" 5.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	NA For PT6A STC	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-56-12	Item #3, due on SN 2-96, except 16, 77-79, 91, 93, & 94 W/O Mod 3/680, within 100 hrs. replace exhaust clamps IAW EB "O" 19.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	NA For PT6A STC	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-61-05	Item #4, due on SN 0-390, except 320 W/O Mod 3/865, within 100 hrs. replace prop gov oil line IAW EB "O" 68, issue 3.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	NA For PT6A STC	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-62-05	Item #5, due on SN 0-435, W/O Mod 3/888, within 100 hrs. replace aluminum rivets & add 16 bolts to FW IAW EB "O" 72.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>
76-08-05 CF-54-09	Item #6, due on SN 2-15, & 17, within 50 hrs. insp. former cap strips LH & RH per TNS "O" 24 part A. If additional Rivets needed CW part B	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>R. D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
76-08-05 CF-54-03	Item #7, due on SN 2-15, 17, 22, 23, 23, & 25-28, within 25 hours, install straps C3FF361-* on rear cargo door IAW TNS "O" 17	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-54-07	Item #8, due on SN 2-15, 17, 18, 20-23, & 25-31, W/O Mod 3/349, within 25 hours install handle guard C3FS524 IAW TNS "O" 21.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-53-16	Item #9, due on SN 3-17 except 16, W/O Mos. 3/193. within 100 hours install doublers around tail access holes IAW TNS "O" 9.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-53-16	Item #10, due on SN 3-15, W/O Mod 3/149, within 25 hours replace rudder Quadrant rivets with bolts IAW TNS "O" 3	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-53-16	Item #11, due on SN 2-15, W/O Mod 3/203, within 50 hours install Safety blocks C3FS504in rudder pedal linkage IAW TNS "O" 4.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-57-06	Item #12, due on SN 2-208, except 135, 171, 176, 190, 193, 195, 197, 202-204, & 207 W/O 3/699, within 100 hours replace rudder hinge with C3TR14 IAW EB "O" 28.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-57-06	Item #13, due on SN 2-177 except 16 & 67 W/O Mod 3/670, within 50 hours reinforce Sta. 427 IAW par 6 Mod 3/670, EB "O" 29.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
76-08-05 CF-59-07	Item #14, due on SN 1-300, & 324, within 25 hours inspect rudder BC assy for cracks IAW EB "O" 63 page 2 fig 1.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-69-12	Item #15, for all AC on skis or wheel-skis, due at install and each 250 hours, inspect for cracks and lube bolt IAW SB 3/12 IS 2.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	Recurring	N/A on floats or wheels	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-59-03	Item #16, due on SN 2-410, W/O Mod 3/837 operated on amphibious floats, within 100 hours, reinforce Sta 264 IAW EB "O" 57 I2.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-58-04	Item #17, due on SN 2-253, within 25 hrs, replace horiz stab clinch nuts, IAW par 6 of EB "O" 47, issue 2.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-68-13	Item #18, W/O Mod 3/909, to prevent flutter, within 50 hrs placard VNE until SB "O" 3/4 is C/W, within 3 mos., change VNE C/W SB "O" 3/6.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-60-03	Item #19, Due on SN 1-356, within 50 hrs, insp. Struts C3W101-9 IAW EB "O" 66 for cracks, repair per par 3 or replace.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A As Equipped	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-57-06	Item #20, Due on SN 2-155 W/O Mod 3/735, within 100 hrs, inst. No push placards per par 5, EB "O" 32.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 4 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type 3. 574601404 4. Robert D. Grace	3. Cert. No. 4. Author. By
76-08-05 CF-57-06	Item #21, due on SN 2-26, W/O Mod 3/721, within 50 hrs reinforce horiz stab attach & replace rivets, IAW par 6, of EB "O" 33,	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	
76-08-05 CF-57-06	Item #22, due on S/N 2-192, except 176, 178, 181, & 190, W/O Mod 3/729, within 50 hrs install bolt C3-WF-54-3, IAW EB "O" 30.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-57-06	Item #23, due on S/N 2-126, except 67, 106, 114, & 120, W/O Mod 3/711, within 50 hrs reinforce IB nose flap, IAW par 6, EB "O" 24.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-58-15	Item #24, due on SN 2-270, 272, 275, 276, & 278, W Mod 3/288 & W/O 3/807, within 25 hrs, replace flap control rod, IAW par 6, of EB "O" 49.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-56-03	Item #25, due on SN 2-73, except 16,25, 27, 34, 43, 47-51, & 67, within 100 hrs insp rivets/replace IAW par 5, EB "O" 17.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05	Item #26, Due on SN 2-71, except 16,25, 27, 34, 43, 47-51, & 54, W/O Mod 3/458, within 25 hrs & each 100 hrs insp. rudder syst IAW par 6, of EB "O" 1	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-57-06	Item #27, Due on SN 2-54 within 50 hrs, replace parts IAW EB "O" 3, or EB "O" 22.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 5 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type 3. 574601404 4. Robert D. Grace	3. Cert. No. 4. Author. By
76-08-05 CF-57-08	Item #28, due on SN 1-201, with Mod 3/731, W/O 3/772, within 50 hrs, remove sintered filter IAW par 6 of EB "O" 38,	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-54-04	Item #29, due on S/N 1-40, within 50 hrs, modify MLG strut, IAW TNS "O" 23, amendment No. 1	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A As Equipped	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-53-16	Item #30, due on S/N 2-15, except 7, W/O tail wheel steering but with Mod 3/132, within 25 hrs, install parts IAW TNS "O" 6.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A As Equipped	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-61-04	Item #31, due on wheel-skis IAW Mod 3/874, at install or within 05 hrs, replace axle pin, IAW EB "O" 70, issue 2 & amendments 1 & 2.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A As Equipped	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
76-08-05 CF-54-09	Item #32, due on SN 2-37, except 16, & 19, W/O 3/374, within 50 hrs install eye bolt IAW TNS "O" 25.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Printed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
83-02-01 CF-82-26R1	Par a), inspect wing strut tie-bar each 12 months, internally for corrosion IAW SB 3/37, rev A, par 5.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1-26-23.	Recurring	See Recurring Log	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
83-02-01 CF-82-26R1	Par (b), inspect wing strut tie-bar each 12 months, external lugs for corrosion IAW SB 3/37, rev A, par 9.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1-26-23.	Recurring	See Recurring Log	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland Model: DHC-3 Turbine Otter		N 708KA S/N: 59		Page: 6 Of 14			
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
83-04-05 CF-82-34	Par a. & b., inspect control column P/N C3CF 39-15/-13 lower welds for cracks, each 100 hrs. Par c. d. & e., if cracks are found repair or replace IAW SB 3/39.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1-26-23.	Recurring	See Recurring Log	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	
83-04-05 CF-82-34	Par f., repeat inspection in Par a. may be discontinued when control column lower assy C3CF 39-21, mod 3/930 is installed.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1-26-23.	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
86-16-12 CF-85-04	Par (b) note: when a parked AC has been subject to wind gusts of 60 MPH or greater, helo or jet blats, inspect IAW SB 3/35, & par a. & b. of AD.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02".	Recurring	See Recurring Log	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
86-16-12 CF-85-04	Par (d) within 60 days replace elevator pushrod, rudder & elevator pushrod to quadrant bolt, nut, washer, & cotter pin IAW SB 3/43.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02".	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
88-15-08 CF-85-03R1	To prevent folding seat forward leg from disengaging, inspect or modify forward seat leg per SB 3/42, Mod 3/932.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
92-08-08 CF-91-16	For VAZAR 135/135A equipped AC. To prevent inadvertent engine shutoff, inspect/install fuel condition Lever lock IAW Serv-Aero SB001	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/14/2015.	N/A	N/A As Equipped	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
96-09-04 CF-89-20R1	Par (a) & (b), for all -3 AC W/O Mod 3/935, SB 3/50, rev A. To prevent tailplane failure inspect tailplane for cracks and repair any cracks IAW SB 3/46 rev B.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/16/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland Model: DHC-3 Turbine Otter		N 708KA S/N: 59		Page: 7 Of 14			
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
96-09-04 CF-89-20R1	Par (c), Install angle & plates Mod 3/935, IAW SB 3/50 Rev A, terminates repeat insp.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/16/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
98-18-08 CF-99-05	To prevent parcel or complete electrical system failure and possible fire, replace Voltage Reg., Voltage-ammeter and Mod Aux. Buss IAW A.M. Luton Drawing 20075 R G & E	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/16/2015.	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2004-05-01 R1 CF-2006-02R	By July 19, 2004 install redundant control linkage per STC No. SA01059SE.	6/5/2023 17,513.9	AD Complied With By Kenmore Air, See Annual Logbook Entry, Item 28), Dated 1/27/2023, For Installation Of STC SA01059SE (Previously Had Viking Air STC SA01243NY Installed With AMOC)	One Time	N/A Post Mod	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2004-21-06	Fire wall cannon plug lock wire	6/5/2023 17,513.9	Superseded By AD2008-12-12	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2004-24-01 CF-2002-48	Only applies to AC with STC ST01243NY or SA01243NY 524085 BC, LTD. (Viking Air))	6/5/2023 17,513.9	This AD Is Not Applicable Due To STC SA01243NY Removed, See Annual Logbook Entry, Item 28), Dated 1/27/2023, For Installation Of STC SA01059SE	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2004-24-10 CF-2002-38	Part (1), inspect P3 heat system wiring & re-wire if needed per A. M. Luton SIL-00-10-10.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2004-24-10 CF-2002-38	Part (2), After compliance with Par (1), replace AFMS per A. M. Luton SIL-00-10-10. Rev G, March 28, 2001	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland Model: DHC-3 Turbine Otter		N 708KA S/N: 59		Page: 8 Of 14			
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
2004-24-10 CF-2002-38	Part (3), inspect Circuit breaker switch & replace if needed per A. M. Luton SIL-00-10-10.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	
2005-01-19	Only Applies to Garmin transponders with software versions: 3.00, 3.01, 3.02, 3.04, or 3.05, CW Garmin SB 0409.	6/5/2023 17,513.9	I Found No Record Specifically About AD2005-01-19, -However- AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". -Also- I Physically Checked And Found Garmin Transponder Has Software Version 8.04 Installed	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2005-25-08	Remove from service & do not install any ADC 2000s P/N: 962830A-1-S-8, -2-S-8, or -3-S-8. configurations B, C, or D, unless upgraded per Shadin SB28-05-002, Rev C.	6/5/2023 17,513.9	Not Applicable As Equipped	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2008-12-12 CF-2001-37R	Par (h), for all aircraft with piston engines, install ignition harness firewall cannon plugs Mod Kit C3VMK0001-1, per SB V3/0002	6/5/2023 17,513.9	Not Applicable As Equipped	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2011-12-02	Only applies to Texas Turbans STC SA09866SC	6/5/2023 17,513.9	Not Applicable As Equipped	N/A	N/A	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2011-18-11	Par (f), (1), & (2), inspect elevator tabs for free-play each 100 hrs & if needed repair before flight per MM TR 18, 19, & 20.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1-26-23.	Recurring	See Recurring Log	1. PM Air. 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2014-17-11 (f) (1)	Within 100 hours or 90 days, inspect TP (trim jack) for movement or defects following per SB V3/0005	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02".	One Time	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland Model: DHC-3 Turbine Otter		N 708KA S/N: 59		Page: 9 Of 14			
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2015-24-02 (f) (1)	Corrugation/spar cap cracking Within 30 days after January 4, 2016 determine the accumulated wing flight cycles or wing flight hours by	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2015-24-02 (f) (2) (i)	Corrugation/spar cap cracking Within 30 days after January 4, 2016 determine all installed STCs or modifications affecting the wing. Based on accumulated air time (from (f) (1)) install access panels: (i) - if there are no STCs or Mods install inspection panels IAW SB V3/0002, Rev C	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2015-24-02 (f) (2) (ii)	Corrugation/spar cap cracking Within 30 days after January 4, 2016 determine all installed STCs or modifications affecting the wing. Based on accumulated air time (from (f) (1)) install access panels: (ii) - if AC has STC SA2009NY install inspection panels IAW SB 3-STC (03-50)-001, Rev NC	6/5/2023 17,513.9	Paragraph N/A do to STC SA2009NY not installed.	N/A	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 10 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2015-24-02 (f) (2) (iii)	Corrugation/spar cap cracking Within 30 days after January 4, 2016 determine all installed STCs or modifications affecting the wing. Based on accumulated air approved AMC using time (from (f) (iii) - if there are STCs or Mods that affect the wing the installer must have an FAA- approved AMOC using guidance developed by the STC holder or Mod developer	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Compliance, See Kenmore AD Document Signed 12/21/2015.	One Time	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2015-24-02 (f) (3)	Corrugation/spar cap cracking Based on accumulated air from (f) (1), perform initial & repetitive borescope & visual inspection per Part B Accomplishment Instructions of SB V3/0002, Rev. C.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1/26/23.	Recurring	See Recurring Log	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2015-24-02 (f) (4)	Corrugation/spar cap cracking If any cracks are found, contact Technical Support at Viking Air Limited for an FAA- approved repair and incorporate the repair before further flight.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Date 1/26/23.	Repairs Needed Determined By Inspection	N/A	PM Air LLC 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2016-01-03	Leveling & Weighing (Barron STOL Kit) Remove Any FMS other Installed FMS #4 Rev. 3 Per STC & AD.	6/5/2023 17,513.9	Barron STOL Kit STC Installed At/By Kenmore Air See 337 Dated 1/20/23. AD's Complied With By Kenmore Air, See Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". - Also- I Physically Checked And Found No FMS. I Installed FMS #4 Rev. 3 Per STC & AD.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2018-21-02	Wing Strut Bolts	6/5/2023 17,513.9	NA for Harbour Air Wing Struts	N/A	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 11 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
2021-24-18 (f) (1) Left And Right Wing Strut Lugs	Within 3 months after the effective date of this AD, determine and record the number of equivalent air time hours on each wing by doubling the total hours time-in-service (TIS) accumulated on each part	6/5/2023 17,513.9	Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02".	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2021-24-18 (f) (1) Tie-Bar	Within 3 months after the effective date of this AD, determine and record the number of equivalent air time hours on each tie-bar by doubling the total hours time-in-service (TIS) accumulated on each part	6/5/2023 17,513.9	Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02".	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2021-24-18 (f) (3) Left And Right Wing Strut Lug	For airplanes with a maximum certificated gross weight that has ever exceeded 8,000 pounds: Remove from service each wing strut fitting following the Accomplishment Instructions in SB V3/0008, Revision NC: (i) Before the part accumulates 32,200 equivalent air time hours, or (ii) Within 12 months after the effective date (Jan 24, 2022) of this AD whichever occurs later:	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Dated 1/26/23.	Recurring	See Recurring Log	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 12 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
2021-24-18 (f) (3) Tie-Bar	For airplanes with a maximum certificated gross weight that has ever exceeded 8,000 pounds: Remove from service each tie-bar by following the Replacement section of the Accomplishment instructions in SB 3/37, Revision B,(i) Before the part accumulates 32,200 equivalent air time hours, or(ii) Within 12 months after the effective date (Jan 24, 2022) of this AD whichever occurs later.	6/5/2023 17,513.9	AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2023-02". For Previous Recurring Compliance, See Kenmore "ADCOMP" Form Entry Dated 1/26/23.	Recurring	See Recurring Log	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2022-05-11	This AD applies to DHC-3 airplanes, all serial numbers, certificated in any category with a wing strut assembly part number (P/N) C3W100 (all dash numbers) installed.	6/5/2023 17,513.9	This AD is NA due to heavy duty wing struts, fittings, & bolts Previously installed per Harbour Air STC SA2711NY	N/A	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2022-21-51 (Effective Date: Oct 4, 2022) (h) (1). Initial Inspection	Within 10 Hrs or 3 days, unless already done within the last 90 days, and thereafter at intervals not to exceed 110 hours TIS, remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs. For this AD, structural reinforcements are not considered previous repairs.	6/5/2023 17,513.9	Found On "ADCOMP" Record Complied With On 1/26/23 (Recorder As AD2022-51-22) AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". This Indicates To Me That Kenmore Has Complied With This AD Per Their Repair Station Maintenance Manual Policies And Procedures.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 13 Of 14

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2022-21-51 (Effective Date: Oct 4, 2022) (h) (1). Repeat Inspection	At intervals not to exceed 110 hours TIS, remove the left-hand elevator tab from the elevator and perform a detailed visual inspection of the entire left-hand elevator auxiliary spar for cracks, corrosion, and previous repairs. For the this AD, structural reinforcements are not considered previous repairs.	6/5/2023 17,513.9	Found On "ADCOMP" Record Complied With On 1/26/23 (Recorder As AD2022-51-22) AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". This Indicates To Me That Kenmore Has Complied With This AD Per Their Repair Station Maintenance Manual Policies And Procedures.	Recurring	See Recurring Log	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2022-21-51 (Effective Date: Oct 4, 2022) (h) (1). Repeat Inspection	Within 10 days, report the results of the inspection. Include AC serial number, hours TIS, auxiliary spar hours TIS (if known), and any crack, corrosion beyond Level 1, or previous repair that is found.	6/5/2023 17,513.9	Found On "ADCOMP" Record Complied With On 1/26/23 (Recorder As AD2022-51-22) AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". This Indicates To Me That Kenmore Has Complied With This AD Per Their Repair Station Maintenance Manual Policies And Procedures.	Recurring	When any crack, corrosion beyond Level 1, or previous repair that is found.	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
AD 2022-23-08 (g)	Within 10 Hrs of this AD, perform a visual inspection of the stabilizer actuator to confirm that the stabilizer actuator lock ring is present, correctly seated in the groove in the upper housing, and engaged in the clamp nut. If the stabilizer actuator lock ring is missing or not correctly installed, before further flight, repair....	6/5/2023 17,513.9	I Found No Record Specifically About AD2022-23-08 -However- AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". This Indicates To Me That Kenmore Has Complied With This AD Per Their Repair Station Maintenance Manual Policies And Procedures.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record							
Make: de Havilland		N 708KA		Page: 14 Of 14			
Model: DHC-3 Turbine Otter		S/N: 59					
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2022-23-08 (h)	Before further flight after the inspection required by paragraph (g) of this AD, apply a torque seal to the clamp nut and lock ring.	6/5/2023 17,513.9	Found On "ADCOMP" Record Complied With On 1/26/23 Visual Inspection Found Torque Seal Applied	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
AD 2022-23-08 (i)	Within 10 days after paragraph (g) or within 10 days after the effective date, whichever occurs later, report the results of the inspection to the FAA at 9-avs-nyaco-cos@faa.gov. Include the AC serial number, hours TIS, hours TIS since last actuator overhaul (if known), and whether the lock ring was present, missing, or incorrectly installed.	6/5/2023 17,513.9	I Found No Record Specifically About AD2022-23-08 -However- AD's Complied With By Kenmore Air, See Kenmore Airs, Annual Logbook Entry Dated 1/27/2023 "Searched AD's Through Biweekly 2023-02". So I Would Assume That Kenmore Has Complied With This AD Per Their Repair Station Maintenance Manual Policies And Procedures.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
AD 2023-10-02	Airplanes equipped with a radio (also known as radar) altimeter	6/5/2023 17,513.9	This AD Is Not Applicable For This Aircraft As Equipped With No Radio (Radar) Altimeter	N/A	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record							
Make: de Havilland		N 708KA		Page: 15 Of 19			
Model: DHC-3 Turbine Otter		S/N: 59					
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-18-03, Par (g), (1)	Within 90 days after the effective date of this AD, incorporate into the existing maintenance records for your airplane, the actions and associated thresholds and intervals, including life limits, specified in Parts 2 and 3 of Viking DHC-3 Otter PSM 1-3-5. Do each initial task within 6 months after the effective date of this AD or at the threshold for each applicable task, whichever occurs later	Date: Jan 8, 2024 ACTT: 18,380.9	Updated AC Due Lists for individual Inspections and intervals in PSM1-3-5 Part 2 and 3	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-22-17, Par (g), (1) Jan 19, 2024	Within 6 months after the effective date of this AD, do a detailed visual inspection of the lower engine mount pickup fittings part numbers (P/Ns) C3FS46-7 and C3FS46-8 and the upper engine mount pickup fittings P/Ns C3FS42-5 and C3FS42-6 for cracking, deformation (altered form or shape), corrosion, fretting or wear, paint or surface coating damage (loose, delaminating, flaking, peeling, chipping of the coating or paint, exposed bare metal, or corroded), and loose, missing, or broken fasteners, in accordance with Part A, steps 1 through 8, of the Accomplishment Instructions in SB V3/0012.	Date: Jan 8, 2024 ACTT: 18,380.9	Inspected with light, Mirror & Borescope Per SB Part A, steps 1 through 8	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
AD 2023-22-17, Par (g), (2)	If any crack or deformation (altered form or shape) of any engine mount pickup fitting is found during the DVI above, before further flight, replace the fitting with a new or serviceable part, IAW Part A, step 10, of the Accomplishment Instructions in SB V3/0012. For purposes of this AD, "new" means zero hours time-in-service.	Date: Jan 8, 2024 ACTT: 18,380.9	No crack or deformation (altered form or shape) of any engine mount pickup fitting found	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	

Signature: *[Handwritten Signature]*

Signature: *[Handwritten Signature]*

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-22-17, Par (g), (3)	If any paint or surface coating of the engine mount pickup fitting is found damaged (loose, delaminating, flaking, peeling, chipping of the coating or paint, exposed bare metal, or corroded) during the DVI above, before further flight, repair the fitting in accordance with Part 1 of PSM 1-3-5, and Part A, step 12, of the Accomplishment Instructions in SB V3/0012. Where Part 1 of PSM 1-3-5, specifies contacting Viking if the alloy and condition of an affected engine mount pickup fitting cannot be identified, this AD requires contacting the Manager, International Validation Branch, FAA; Transport Canada; or Viking's Transport Canada Design Approval Organization (DAO) for instructions.	Date: Jan 8, 2024 ACTT: 18,380.9	Painted fitting Per Part 1 of Viking PSM 1-3-5, DHC-3, and Part A, step 12, of the Accomplishment Instructions in SB V3/0012.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	

Signature: *[Handwritten Signature]*

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 18 Of 19

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-22-17, Par (g), (4)	If any loose, missing, or broken fastener is found during the DVI above, before further flight, replace the fastener with a new, do a DVI of the fastener hole to detect cracking, corrosion, an elongated bore hole, bore surface roughness, or other defects, and repair any damage found or replace the engine mount pickup fitting with a new or serviceable part, IAW Part 1 of PSM 1-3-3 and Part A, step 9, of the Accomplishment Instructions in SB V3/0012.	Date: Jan 8, 2024 ACTT: 18,380.9	All Fasteners looked Good	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
Signature: <i>Robert D. Grace</i>							

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 19 Of 19

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-22-17, Par (g), (5)	If any corrosion, wear, or fretting to any engine mount pickup fitting is found during the DVI above, before further flight, contact the Manager, International Validation Branch, FAA; Transport Canada; or Viking's Transport Canada DAO to obtain instructions for an approved repair and, within the compliance timeframe specified therein, do the repair. If approved by the DAO, the approval must include the DAO-authorized signature. Alternatively, before further flight, replace the engine mount pickup fitting with a new or serviceable part in accordance with Part A, step 10, of the Accomplishment Instructions in SB V3/0012.	Date: Jan 8, 2024 ACTT: 18,380.9	No corrosion, wear, or fretting on engine mount pickup fitting found.	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
Signature: <i>Robert D. Grace</i>							

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 20 Of 20

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
AD 2023-22-17, Par (g). (5)	Report the inspection results from the DVI re-quired by paragraph (g)(1) at the applicable time specified in paragraph (h)(1) or (2) of this AD in accordance with Part A, step 14, of the Accomplishment Instructions in SB V3/0012. (1) For inspections done on or after the effective date of this AD: Submit the report within 30 days after the inspection. (2) For inspections done before the effective date of this AD: Submit the report within 30 days after the effective date of this AD.	Date: Feb 12, 2024, TTAF: 18,537.9	Report for Inspection Submitted	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
						Signature: <i>Robert D. Grace</i>	

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59



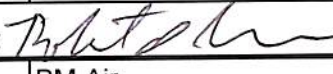
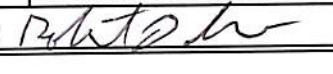
AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type 3. 574601404 4. Robert D. Grace	3. Cert. No. 4. Author. By
AD 2025-17-08, Par (g)	Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, Transport Canada AD CF-2024-46.	Oct 6, 2025 21,022.0	See individual requirements for AD CF-2024-46 listed below	See below	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
CF AD- 2024-46, Part I - Initial Inspection, A.	Within 25 hours or 90 days, from Sep 29, 2025, perform a visual inspection to confirm that the stabilizer actuator lock ring is present, correctly seated in the groove in the upper housing, and fully engaged in the clamp nut IAW the inspection instructions of DHC3-SL-27-001 Iss1.	Oct 6, 2025 21,022.0	Inspected/instaled actuator with properly seated lock ring per DHC3-SL-27-001 Iss1 <i>13.2 10-7 HRS From EFFECTIVE DATE</i>	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
CF AD- 2024-46, Part I - Initial Inspection, B.	If the lock ring is missing or incorrectly installed, before further flight, rectify in accordance with DHC3-SL-27-001 Iss1, or replace the stabilizer actuator with a serviceable part.	Oct 6, 2025 21,022.0	Inspected/instaled actuator with properly seated lock ring per DHC3-SL-27-001 Iss1	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	
CF AD- 2024-46, Part II - Addition of Witness Mark	Within 25 hours TIS or 90 days, from Sep 29, 2025, accomplished, apply a witness mark perpendicular across clamp nut, lock ring & upper housing barrel with a semi-permanent method. This witness mark serves as an indicator of the relative position of the lock ring and the clamp nut when the lock ring is fully engaged.	Oct 6, 2025 21,022.0	Inspected/instaled actuator with witness mark perpendicular across clamp nut, lock ring & upper housing barrel per DHC3-SL-27-001 Iss1	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace	

Airworthiness Directive Compliance Record

Make: de Havilland
Model: DHC-3 Turbine Otter

N 708KA
S/N: 59

Page: 22 Of 22

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One Time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type 3. Cert. No. 4. Author. By
CF AD-2024-46, Part III - Repetative Inspection, A.	Repear inspection not exceeding 110 hours AD, visual inspection to confirm that the stabilizer actuator lock ring is present, correctly seated in the groove in the upper housing and fully engaged in the clamp nut and verify that the witness mark is intact. IAW DHC3-SL 27-001	Oct 6, 2025 21,022.0	Per FAA AD Par (h) and TC AD Part V this paragragg is Not Applicable for aircraft with STC SA02761SE) or PDA23-10	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace Signature: 
CF AD-2024-46, Part III - Repetative Inspection, B.	If the lock ring is missing, incorrectly installed or the clamp nut shows signs of rotation based on the witness mark, before further flight, rectify IAW DHC3-SL-27-001 or replace the stabilizer actuator with a serviceable part.	Oct 6, 2025 21,022.0	Inspected/instaled actuator confirmed properly seated lock ring per DHC3-SL-27-001 Iss1	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace Signature: 
CF AD-2024-46, Part III - Repetative Inspection, C.	If the witness mark is missing or not intact, reapply a witness mark in accordance with Part II of this AD. Any residue from the previous witness mark shall be removed prior to the reapplication.	Oct 6, 2025 21,022.0	Inspected/instaled actuator with witness mark perpendicular across clamp nut, lock ring & upper housing barrel per DHC3-SL-27-001 Iss1	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace Signature: 
CF AD-2024-46, Part IV - Parts Installation Prohibition	As of the effective date of this AD, it is prohibited to install any stabilizer actuator unless it is a serviceable part.	Oct 6, 2025 21,022.0	Noted	One Time	N/A	PM Air 2. A&P 3. 574601404 4. Robert D. Grace Signature: 

Airworthiness Directive Compliance Record

Make: Hartzell
Model: HC-B3TN-3DY

N 708KA
S/N: BUA19230

Page 1 of 3

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
70-05-06	Applies to T10176() & T10176H() blades on HC-B3TN-5C propellers on AiResearch TPE331-() engines.	Aug 12, 2024 19,479.4	AD Not aplicable per blade instaled & STC	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
72-08-04	Applies to T10173() & T10176() blades installed on HC-B3TN-2, -3, -5, HC-B3TF-7, & HC-B4TN-3 propellers on PT6A-, TPE331- & 250-B engines.	Aug 12, 2024 19,479.4	AD Not aplicable per blade instaled	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
74-14-01	Applies to Hartzell T10173H(+), T10176H(+), T10178H(-), & T10282H(+) hard alloy blades on HC-B3TN-2(), -3(), 5(), -7(), HC-B3TF-7(), HC-B4TN-3() & HC-B4TN-5() propellers on PT-6A-() & TPE 331()-() engines on Swearingen SA-226, Short SC-7 Series 3, Aero Commander 690, & DHC-6 300 aircraft.	Aug 12, 2024 19,479.4	AD Not aplicable per STC	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
83-08-01 R2	Aapplicable to HCB3TN-2, HC-B3TN-3, -5, HC-B4TN-3, -5, HC-B4MN-5, & HC-B5MP-3 propellers. installed on PT6A-(), TPE-331-() engines	Aug 12, 2024 19,479.4	AD Complied with at install	Recurring	See Recurring Log	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
86-05-12	Applies to HC-B3TN-5() /T10282() propellers on Fairchild Model SA226-TC aircraft modified by Garrett STC SA344GL or Par Air STC SA4872SW with TPE331-10UA engines	Aug 12, 2024 19,479.4	AD Not aplicable per STC	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: Hartzell
Model: HC-B3TN-3DY

N 708KA
S/N: BUA19230

Page 2 of 3

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type 3. 574601404 4. Robert D. Grace	3. Cert. No. 4. Author. By
96-15-04	Applies to HC-B3TN, HC-B5MP, HC-E4A, & HC-D4N propellers, equipped with blades manufactured between March 1992 and June 1996	Aug 12, 2024 19,479.4	AD Is Previously Complied With Per Prop Overhaul	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
96-18-14	Applies to Various props including HC-B3TN-3() installed on any AC	Aug 12, 2024 19,479.4	AD Is Previously Complied With Per Prop Overhaul	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
2003-04-23	Applies to HC-B3TN-5() propellers, with T10176H(B)-5, T10176H(K)-5, T10176H-5, T10178H-11, T10178H-11R, T10178H(B)-11, & T10178H(B)-11R blades installed on Mitsubishi Ltd, MU-2 airplanes	Aug 12, 2024 19,479.4	AD Not applicable per Model	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
2003-13-17	Applies to Hartzell, McCauley, Sensenich, & Raytheon propellers returned to service by T and W Propellers, Inc. of Chino, CA,	Aug 12, 2024 19,479.4	AD Not applicable per overhil facility	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
2005-04-08	Applies to model HC-B3TN-5()/T10282() propellers	Aug 12, 2024 19,479.4	AD Not applicable per Model	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
2005-14-11	Applies to Hartzell, McCauley, & Sensenich propeller models returned to service by Southern California Propeller Service	Aug 12, 2024 19,479.4	AD Not applicable per overhil facility	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

Airworthiness Directive Compliance Record

Make: Hartzell
Model: HC-B3TN-3DY

N 708KA
S/N: BUA19230

Page 3 of 3

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
2005-14-12	Applies to Various models including HC-B3TN-3, propellers installed with propeller mounting bolts, part number (P/N) B-3339.	Aug 12, 2024 19,479.4	AD NA due to mounting bolts used	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2005-18-20	Applies to Goodrich De-icing and Specialty Systems "FASTprop" propeller de-icers	Aug 12, 2024 19,479.4	AD Not aplicable as equipped	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2006-24-07	AD applies to Hartzell & McCauley propellers that have been serviced by Oxford Aviation Limited, doing business as CSE Aviation.	Aug 12, 2024 19,479.4	AD Not aplicable per overhil facility	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>
2009-10-14	AD applies to Hartzell steel hub propellers with counterweight slug attachment bolts P/N B-3386-14H, LFC manufacturing lot 224, installed	Aug 12, 2024 19,479.4	AD Is Previously Complied With Per Prop Overhaul	N/A	N/A	1. PM Air 2. A & P 3. 574601404 4. Robert D. Grace	Signature: <i>[Signature]</i>

Airworthiness Directive Compliance Record

Make: de Havilland
Model: PT6A-34

N 708KA
S/N: PCE-RB0658

Page: 2 of 2

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
2002-23-13 CF-2002-47	To prevent failure of the Exh. Duct due to cracking of welds repaired by welding, inspect Exh. ducts on engines that have been overhauled or repaired IAW SB PT6A-72-1610	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
AD 2011-20-51	(-34) For engines with PMA Timken 1st stage sun gears replaced since February 3, 2010.	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
AD 2011-25-12	(-34) For engines with PMA Timken 1st stage sun gears replaced since December 22, 2008.	6/5/2023 17,513.10	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
AD 2014-11-05 CF-2013-33R1	PT Containment ring replace per P&WC A1427	6/5/2023 17,513.11	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>
2022-08-13	low-time fractures of compressor turbine (CT) blades	6/5/2023 17,513.12	AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	Signature: <i>Robert D. Grace</i>

REPRODUCTION OF THIS DOCUMENT IS PROHIBITED

Airworthiness Directive Compliance Record

Make: de Havilland
Model: PT6A-34

N 708KA
S/N: PCE-RB0658

Page: 1 of 2

AD Number	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility		3. Cert. No.	
						2. Cert. Type	4. Author. By		
AD 70-16-01	Woodward Governor safety wire on PT6A engines	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
AD 75-06-01	(-34 engines) To detect RGB deterioration inspect per AD par (a) until par 2 of SBs 1138, 1216, & 1220 are complied with.	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
AD 75-11-04	(-34 engines) With PT shaft housing 3010548; par 1. inst. chip detector, par 2. insp each 25 hrs until OH, 3500 hrs until modified.	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A Post	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
AD 80-08-12	(-34 & -135 engines) With fuel pump coupling 02-14910-01 to prevent failure replace fuel pump drive coupling Ref SB 1316	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
AD 90-12-05	(-34) Engines repaired by Gregory Flying Service/ Airforce Turbine Services. PCE 38093 & PCE 56090	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
AD 92-15-11	Due on various PT6A-*** engines on Beech Aircraft	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace
97-04-12 CF-1996-24R1	(-34 & 135A) To prevent power loss due to BOV malfunction, due on BOV with supplier code # 8070 per SB 1538 R3,	6/5/2023 17,513.9	All AD's Complied With At Engine Overhaul, See Standard Aero Logbook Entry Dated Mar 21/2019. AD's Complied With By Kenmore Air, See Kenmore Air Logbook Entry Dated 8/30/2022 "Searched AD's Through Biweekly 2022-18".	One Time	N/A	1. PROMECH INC.	2. A & P /IA	3. 574601404	4. Robert D. Grace

Airworthiness Directive Compliance Record

Make: Appliances
Model: Various

N 708KA
S/N: Various

AD No.	Description	Complied Date/time	Status Method of Compliance/Applicability	One time/ Recurring	Next Due Date/Time	1. Facility 2. Cert. Type	3. Cert. No. 4. Author. By
70-16-01	To prevent uncontrolled feather within 50 hours inspect OSG & CSU for socket head screws & proper safety wire security of solenoid valve.	5-Jun-23 17513.9	NA by overhaul date	one time	N/A Per date	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
75-05-04	To prevent possible fire, if P/N A427 Whelen Strobe Flash Tube is installed Comply with AD	5-Jun-23 17513.9	NA per PN	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
93-12-04	To prevent smoke in the Cockpit Remove or Replace Precise Flight, Inc Pulselite model 1210-2405-2 with 1210-2405-2A	5-Jun-23 17513.9	No per MN	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
2004-13-20	Garmin GTX50/55/60/65 with Softwear Version 3.0 to 3.4	5-Jun-23 17513.9	AD N/A Per GPS PN	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
2025-01-19	GTX330 Softwear Version	5-Jun-23 17513.9	AD N/A Per softwear version installed	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
2005-25-08	Remove from service & do not install any ADC 2000s P/N: 962830A-1-S-8, -2-S-8, or -3-S-8, configurations B, C, or D, unless upgraded per Shadin SB28-05-002, Rev C.	5-Jun-23 17513.9	AD N/A Per ADC Part No,	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	
2019-01-02	Aspen Avionics EFD1000 W/ Softwear Version 2.9	5-Jun-23 17513.9	AD N/A Softwear Version B3,2,5 installed	N/A	N/A	1. PROMECH INC. 2. A & P /IA 3. 574601404 4. Robert D. Grace	
						Signature:	

PM Air - NON-ROUTINE

N708KA HOBBS: 6,947.4 ETSO: 4,968.9 PTSO: 2,757.5
 DATE: 18-Aug-25 TTAF: 20,926.6 ETT: 8,778.6 PTT: 16,749.4 PAGE: 1 of 1
 AREA: STARTS: 14,497

Discrepancy		CORRECTIVE ACTION		
ITEM	1	Annual AD Audit	Completed AD search on FAA website. No new ADs for Airframe, Engine, or Propeller since AD2023-22-17 (Effective date: January 19, 2024).	
INIT			SIGN: <i>Robert D. [Signature]</i>	DATE: <i>Aug 18 25</i> INSP
RG		R II Y: N: X	CERT TYPE: A&P # <i>Robert D. [Signature]</i>	
ITEM				
INIT			SIGN: DATE: INSP	
		R II Y: N:	CERT TYPE: A&P #	
ITEM				
INIT			SIGN: DATE: INSP	
		R II Y: N:	CERT TYPE: A&P #	
ITEM				
INIT			SIGN: DATE: INSP	
		R II Y: N:	CERT TYPE: A&P #	
ITEM				
INIT			SIGN: DATE: INSP	
		R II Y: N:	CERT TYPE: A&P #	

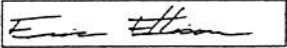
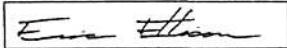
N-Number: 708KA

MAKE: deHavilland

Model: DHC-3

SERIAL NUMBER: 59

Print Date: 12/21/2015

A.D. NUMBER	A.D. DESCRIPTION	M.O.C.	C.W. DATE	SIGNATURE
5B17-02				
1. 57-17-02	Flap Control Restriction	Modification kits previously installed per AD log	12/10/2015	
Reference Docs	Description			
EB34	Describes Install			
MOD3/731	New Actuator & Filter Assy	Verified 2/22/1999		
MOD3/744	Elevator Interconnect	Verified 2/22/1999		
MOD3/745	Fiberglass Tailplane Leading Edge	Verified 2/22/1999		
2. 76-08-05	Fuel Boost Pump Jumper Wire	Complied with each of the 32 items listed below.		
Reference Docs	Description			
1 MOD3/341	Heavier Gauge Wire	N/A by installation of Vazar STC SA3777NM		
2 MOD3/558	Breather Pipe Assy	N/A by installation of Vazar STC SA3777NM		
3 MOD3/680	Exhaust Pipe	N/A by installation of Vazar STC SA3777NM		
4 MOD3/865	Improved CSU Oil Line	N/A by installation of Vazar STC SA3777NM		
5 MOD3/388 - EB72	Monel Rivets + add sixteen bolts on Firewall	Verified Previous Compliance	12/10/2015	
6 TNS24	Cap Strips at 153.77 and 174.22	Verified Previous Compliance	12/10/2015	
7 TNS17	Rear Cargo Door Straps C3-FF351, C3-FF-361-5	Complied with TNS17 by installation of door straps.	12/21/2015	
8 TNS21	Rear Cargo Door Handle Guards C3-FS-524	Complied with TNS21 by installation of handle guards.	12/21/2015	
9 TNS09	Skin Stiffeners at Rear Access Holes	Verified Previous Compliance	12/10/2015	
10 MOD3/149, TNS03	Replace Rivets with bolts in Rudder Support C3-FS-169	Verified Previous Compliance	12/10/2015	
11 MOD3/203, TNS04	Install Rudder Safety Block C3-FS-504	N/A by serial number. Blocks installed at factory.	12/10/2015	
12 MOD3/699, EB28	Install Improved Rudder Hinge Assy C3-TR-14	Verified Previous Compliance	12/14/2015	
13 MOD3/670, EB29	Reinforcement of the Fuselage at the Horizontal Attach Bracket	Verified Previous Compliance	12/14/2015	
14 EB63	Rudder Bellcrank C3-CF-15, CS-CF15-7	Verified Previous Compliance	12/14/2015	
15 SB3/12	Aircraft on Skis / Wheel Skis - Gear to Fuselage Bolt Cracks	NA - Skis not installed	12/14/2015	
16 MOD3/837, EB57	Amphibians - Reinforce Station 264	NA - Straight Floats Installed. Straps previously installed.	12/14/2015	
17 EB47	Replace Clinch Nuts with PN 22-NC4-048	Verified Previous Compliance	12/14/2015	
18 MOD3/909, SB3/4, SB3/6	Elevator Tab Horn	Flutter kit installed / no blind fasteners installed	12/14/2015	

19	EB66	Inspection of Wing Strut Spars	New wing struts installed	12/14/2015
20	MOD3/735, EB32	"No Push" Signs on Tailplane	No Push signs installed on tailplane and elevators per EB32.	12/21/2015
21	MOD3/721, EB33	Tailplane Doubler Plates, Replace Rivets With Bolts	N/A by serial number	12/14/2015
22	MOD3/729, EB30	Aileron Center Hinge Bearing Bolt	Verified Previous Compliance	12/14/2015
23	MOD3/711, EB24	Nose Flap Center Hinge Rib Reinforcing Angles	Verified Previous Compliance	12/14/2015
24	MOD3/288, EB49	Replacement of the Flap Control Rod Assy C3CF170-4 with -11	Verified Previous Compliance	12/14/2015
25	EB17	Top of Wing Rivet Spacing - Corrugation Support	Verified Previous Compliance	12/14/2015
26	MOD3/458, EB1	Rudder Control System Damage Due to Wind Gusts	NA by serial number	12/14/2015
27	EB22	Elevator Quadrant C3-CF-127: add six rivets	Verified Previous Compliance	12/14/2015
28	MOD3/772, EB38	Remove Filter From Flap Jack	NA by P/N installed / OH date.	12/14/2015
29	TNS23	Main Gear Compression Strut Piston Tube	N/A wheel gear not installed	12/14/2015
30	MOD3/132, TNS6	Prevent Loss of Tailwheel in Flight	N/A wheel gear not installed	12/14/2015
31	MOD3/874, EB70	Aircraft on Skis - Lubrication of Axle Pin	N/A skis not installed	12/14/2015
32	MOD3/374, TNS25	Rear Cabin Seat Eyebolt Attachment	Seats modified per TNS 25	12/14/2015

3.	83-02-01	Tie Bar Inspection	New tie bar P/N C3FS14-2 installed at this time.	12/14/2015	
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Reference Docs	Description
S83/37	Visually inspect wing strut tiebar

4.	83-04-05	Lower Control Column Inspection	Complied with by means of visual inspection.	12/14/2015	
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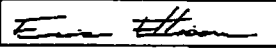
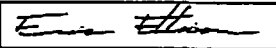
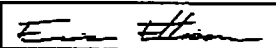
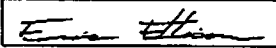
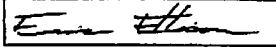
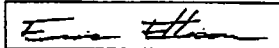
Reference Docs	Description
S83/39, MOD3/390	Visually inspect for cracks longer than 1". Repair or Replace

5.	86-16-12	Elevator Control System Inspection	Verified Previous Compliance / New Hardware Installed	12/21/2015	
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Reference Docs	Description
S83/35, 3/43	Pushrod Ends, Cables, Pulleys, Replace Parts

6.	88-15-08	Prevent Disengagement of The Folding Seat Leg	All seats modified with Mod 3/932.	12/21/2015	
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Reference Docs	Description
3/932, S83/42	Test Lock for Slippage - Replace

7.	90-12-08	Superseded			
	Reference Docs	Description			
8.	92-08-08	Condition Lever Lock to Prevent Inadvertent Fuel Cutoff	Installed condition lever lock IAW Serv Aero SB 0001.	12/14/2015	
	Reference Docs	Description			
	Serv Aero SB001	Installation of Condition Lever Lock			
9.	96-09-04C	Tailplane Main Rib Inspection and Repair	Verified Previous Compliance	12/16/2015	
	Reference Docs	Description			
	MOD3/935, SB3/46, 3/50	Additional Reinforcement of Center Section of Tailplane			
10.	98-18-08	Modification to Vazar Otter Electrical System	Verified Rev levels G (page 1) and E (page 2&3)	12/16/2015	
	Reference Docs	Description			
	AM Luton SA-SIL-98-11-03	Replacement of the Voltage Regulator and Ammeter Gauge IAW Vazar Drawings Rev. G and E			
11.	2004-05-01 R1	Vazar Otter Servo Tab Flutter	Viking Anti-Flutter Kit STC SA01243NY Installed IAW AMOC Letter dated July 13, 2004.	12/14/2015	
	Reference Docs	Description			
	SA010595E	Installation of redundant control linkage			
12.	2004-21-06	Superseded by 2008-12-12			
	Reference Docs	Description			
13.	2004-24-01	Replacement of Elevator Servo Tab	Installed servo tab introduced by SBV3/01 and kit V3MK1148.	12/16/2015	
	Reference Docs	Description			
	SBV3/01	Install Redesigned Servo Tab and Balance			
14.	2004-24-10	Vazar Otter P3 and Py Wiring	Verified Py and P3 are wired in parallel not series, and CB size is increased.	12/21/2015	

Reference Docs**Description**

Vazar SIL-00-10-10

Wire Indicator Lights in Parallel not Series

15. 2008-12-12

Magneto Harness Cannon Plug

N/A turbine engine installed

12/14/2015

Reference Docs**Description**

SBA3/53, SBV3/0001

Cannon Plug Safety Wire Inspection

SBV3/0001, C3vMK0001-1

Modification Kit

16. 2011-05-02

Superseded by 2011-18-11

Reference Docs**Description**

17. 2011-12-02 C

Texas Turbine Otters - Airspeed Limitation

N/A Vazar turbine conversion installed

12/14/2015

Reference Docs**Description**

18. 2011-18-11

Repetitive Inspection of Elevator Control Tabs

Trim tab freeplay measured and found to be within limits.

12/14/2015

Reference Docs**Description**

AD2011-05-02

Update to previous AD adds Texas Turbine Aircraft

TR 18, 19, 20

Specifications for freeplay measurement.

19. 2014-17-01

Trim Jack Actuator Bracket

Complied with trim jack actuator inspection IAW Viking SB V3/0005.

12/21/2015

Reference Docs**Description**

V3/0005

Inspection of trim jack lower mounting bracket.

20. 2015-24-02

To detect and correct cracking found at various wing stations and on the main spar lower cap

Viking SB V3/0002, Revision C previously installed on wings. Performed visual inspection with no defects noted.

12/21/2015

Reference Docs**Description**

V3/0002

Wing modification and inspection.

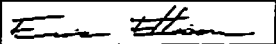
Component: Engine

Make: Pratt & Whitney Canada

Model: PT6A-34

S/N: PCE-RB0910

Print Date: 12/21/2015

A.D. NUMBER	A.D. DESCRIPTION	M.O.C.	C.W. DATE	SIGNATURE
1.	Airworthiness Directives researched through BI-weekly 2015-25. No new ADs apply at this time. Date of engine manufacture: June 29, 2015. See Recurrent AD Log for recurring AD Note compliance and next due.		12/21/2015	

Component: Propelle Make: Hartzell

Model: HC-B3TN-3DY


Serial Number: BUA33422

Print Date: 12/21/2015

A.D. NUMBER	A.D. DESCRIPTION	M.O.C.	C.W. DATE	SIGNATURE
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1. Airworthiness Directives researched through Bi-weekly 2015-25. No new ADs apply at this time. Date of propeller manufacture: June 1, 2015. See Recurrent AD Log for recurring AD Note compliance and next due.

12/21/2015



N708KA
 Registration Number
 59
 Serial Number
 DHC-3
 Aircraft Type

Control Column

83-04-05
 AD #
 3/1/1983
 Effective Date
 (Rev. Date)

DATE	ACTT	METHOD OF COMPLIANCE	NEXT DUE		SIGNATURE
			ACTT	DATE	
4/3/18	15125	Visual Inspection	15225	NA	<i>[Signature]</i>
5/15/18	15215	VISUAL INSPECTION	15315	N/A	<i>[Signature]</i>
6/7/2018	15245.6	VISUAL INSPECTION	15345.4	N/A	<i>[Signature]</i>
7/5/2018	15349.1	VISUAL INSPECTION	15491	N/A	<i>[Signature]</i>
7/24/18	15489.0	Visual Insp. 100hr	15589.0	NA	<i>[Signature]</i>
8/15/18	15589.5	Visual Insp. 100hr	15689.5		<i>[Signature]</i>
8/21/18	15689.5	Visual Insp. 100hr	15745.0		<i>[Signature]</i>
11/12/18	15738.1	Visual Insp. 100hr	15838.1		<i>[Signature]</i>
3/24/19	15830	VISUAL 100HR	15930		<i>[Signature]</i>
6/21/19	16018	Visual Insp	16118	NA	<i>[Signature]</i>
7/11/19	16106	Visual Inspection - 100hr	16208	N/A	<i>[Signature]</i>
7/29/19	16193.5	Visual Insp. 100hr	16293.5		<i>[Signature]</i>
8/27/19	16223.8	Visual Insp. 100hr	16423.6		<i>[Signature]</i>
9/1/19	16355.8	Visual Insp. 100hr	16455.8		<i>[Signature]</i>
12/30/19	16483.5	Visual Insp	16533.5		<i>[Signature]</i>
10/15/20	16483.7	visual inspection 100h	16583.7		<i>[Signature]</i>
7/22/21	16675.4	vis. insp. 100hr	16675.4		<i>[Signature]</i>
8/16/21	16673.7	vis. insp. 100hr	16773.7		<i>[Signature]</i>
9/8/21	16761.1	vis. insp. 100hr	16861.1		<i>[Signature]</i>
4/4/22	16772.3	vis. insp. 100hr	16872.3		<i>[Signature]</i>
06/14/22	16865.3	Vis. Insp 100hr	16965.3	N/A	<i>[Signature]</i>
07/11/22	16952.5	Vis Inspection 100 hr	17052.5		<i>[Signature]</i>
08/08/22	17048.4	Vis Inspection 100hr	17148.4		<i>[Signature]</i>
08/30/22	17119.6	Vis Insp. 100hr	17219.6		<i>[Signature]</i>
11/26/23	17214.5	vis insp. 100hr	17314.5		<i>[Signature]</i>

N708KA
 Registration Number
 59
 Serial Number
 DHC-3
 Aircraft Type

Trim Tab Freeplay

2011-18-11
 AD #
 10/3/2011
 Effective Date
 (Rev. Date)

DATE	ACTT	METHOD OF COMPLIANCE	NEXT DUE		SIGNATURE
			ACTT	DATE	
4/3/18	15125	Visual Inspection	15225	NA	<i>[Signature]</i>
5/15/18	15215	VISUAL INSPECTION	15315	N/A	<i>[Signature]</i>
6/3/2018	15245	VISUAL INSPECTION	15395	N/A	<i>[Signature]</i>
7/5/2018	15391	VISUAL INSPECTION	15491	N/A	<i>[Signature]</i>
7/24/18	15489	visual insp 100hr	15589	NA	<i>[Signature]</i>
8/8/18	15568.5	visual insp. 100hr	15668.5		<i>[Signature]</i>
8/27/18	15645.0	visual insp. 100hr	15745.0		<i>[Signature]</i>
11/2/18	15738.1	visual insp. 100hr	15838.1		<i>[Signature]</i>
3/29/19	15836	VIS INSP 100HR	15936		<i>[Signature]</i>
6/21/19	16018	Visual Insp	16118	NA	<i>[Signature]</i>
7/11/19	16106	Visual Inspection - 100hr	16206	N/A	<i>[Signature]</i>
7/24/19	16193.5	visual insp. 100hr	16293.5		<i>[Signature]</i>
8/27/19	16223.8	visual insp. 100hr	16423.8		<i>[Signature]</i>
10/1/19	16353.4	visual insp. 100hr	16453.4		<i>[Signature]</i>
12/30/19	16433.5	Visual Insp	16533.5		<i>[Signature]</i>
10/15/20	16483.7	visual inspection 100h	16583.7		<i>[Signature]</i>
7/22/21	16575.4	V.S. Insp. 100hr	16675.4		<i>[Signature]</i>
8/16/21	16673.7	V.S. Insp. 100hr	16773.7		<i>[Signature]</i>
4/5/21	16744.0	V.S. Insp. 100hr	16844.0		<i>[Signature]</i>
4/4/22	16761.1	V.S. Insp. 100hr	16861.1		<i>[Signature]</i>
	16769.6	V.S. Insp. 100hr	16869.6		<i>[Signature]</i>
	16772.3	V.S. Insp. 100hr	16872.3		<i>[Signature]</i>
06/14/22	16853	Vis. Insp 100 hr	16965.3		<i>[Signature]</i>
07/11/22	16865.3	Vis. Insp 100hr	17052.5		<i>[Signature]</i>
08/08/22	17048.4	Vis. Insp 100hr	17148.4		<i>[Signature]</i>
08/30/22	17119.6	Vis Insp 100hr	17219.6		<i>[Signature]</i>
1/26/23	17216.5	V.S. Insp. 100hr	17318.5		<i>[Signature]</i>

AIRCRAFT, ENGINE, PROPELLER, APPLIANCE			MAKE	MODEL	SERIAL NUMBER		
			de Havilland	DHC-3	59		
AD NO. & REV. DATE	SUBJECT	DATE AND HOURS AT COMPLIANCE	METHOD OF COMPLIANCE	ONE TIME	RECURRING	NEXT COMP DUE DATE/HRS, CYCLES	AUTHORIZED SIGNATURE AND NUMBER
83-02-01	Tie Bar	12/30/15 13979.2	New tie bar installed.	↓	X	Dec 2016	<i>[Signature]</i>
83-04-05	Control Column	12/30/15 13979.2	Visual Insp / 100hrs	↓	X	14079.2	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	12/30/15 13979.2	Vis Insp / 100hrs	↓	X	14079.2	<i>[Signature]</i>
2015-24-02	Wing Box Inspection	12/1/15 13979.2	Vis Insp / 1600hrs - 2100 cycles	↓	X	15579.2 2100 cycles	<i>[Signature]</i>
83-04-05	Control Column	7/12/16 14067.4	Vis Insp / 100 Hrs	↓	X	14167.4	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	7/12/16 14067.4	Vis Insp / 100hrs	↓	X	14167.4	<i>[Signature]</i>
83-04-05	Control Column	8/1/16 14165.0	Vis Insp / 100 hrs	↓	X	14265.0	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	8/1/16 14165.0	Vis Insp / 100hrs	↓	X	14265.0	<i>[Signature]</i>
83-04-05	CONTROL COLUMN	8-22-16 14258.5	vis Insp / 100 hrs	↓	X	14358.5	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	8-22-16 14258.5	Vis Insp / 100 hrs	↓	X	14358.5	<i>[Signature]</i>
83-04-05	CONTROL COLUMN	9-8-16 14328.0	vis Insp / 100 hrs	↓	X	14428.0	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	9-8-16 14328.0	vis Insp / 100 hrs	↓	X	14428.0	<i>[Signature]</i>
2015-24-02	Wing Box Inspection	10/27/16 14363.6	borescope Insp. 1600hrs/2100cycles	↓	X	15703.6 2100	<i>[Signature]</i>
83-02-01	tie bar	10/27/16 14363.6	Visual Insp. 1yr.	↓	X	10/17	<i>[Signature]</i>
83-04-05	control column	10/27/16 14363.6	visual Insp. 100hrs	↓	X	14463.6	<i>[Signature]</i>
2011-18-11	trim Tab Freeplay	10/27/16 14363.6	Visual Insp. 100hrs	↓	X	14463.6	<i>[Signature]</i>
83-04-05	Control column	05/03/17 14458	Visual Insp. 100hrs	↓	X	14558	<i>[Signature]</i>
2011-18-11	Trim tab freeplay	05/03/17 14458	Visual Insp. 100hrs	↓	X	14558	<i>[Signature]</i>
83-04-05	CONTROL COLUMN	6/8/17 14551	VISUAL INSP/100HRS	↓	X	14651	<i>[Signature]</i>
2011-18-11	TRIM TAB FREEPLAY	6/8/17 14551	VISUAL INSP/100HRS	↓	X	14651	<i>[Signature]</i>
83-04-05	Control Column	7/6/17 14648	Visual Insp/100hrs	↓	X	14748	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	7/6/17 14648	Visual Insp/100hrs	↓	X	14748	<i>[Signature]</i>
83-04-05	Control Column	7/24/17 14731	Visual Insp / 100hrs	↓	X	14831	<i>[Signature]</i>
2011-18-11	Trim Tab freeplay	7/24/17 14731	Visual Insp / 100hrs	↓	X	14831	<i>[Signature]</i>
83-04-05	Control Column	8/8/17 14808.2	Visual Insp / 100hrs	↓	X	14908.2	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	8/8/17 14808.2	Measure / 100hrs	↓	X	14908.2	<i>[Signature]</i>
83-04-05	Control Column	8/23/17 14884.8	Visual / 100 hrs	↓	X	14984.8	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	8/23/17 14884.8	Measure / 100 hrs	↓	X	14984.8	<i>[Signature]</i>
83-04-05	Control Column	9/7/17 14942.4	Visual / 100 HR	↓	X	15042.4	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	9/7/17 14942.4	Measure / 100 HR	↓	X	15042.4	<i>[Signature]</i>
2015-24-02	Wing box	10/20/17 15035.6	borescope 1600hrs/2100cycles	↓	X	16635.6/2100	<i>[Signature]</i>
83-02-01	Tie bar	10/20/17 15035.6	Visual Insp. 1hr	↓	X	10/18	<i>[Signature]</i>
83-04-05	control column	10/20/17 15035.6	visual / 100hr	↓	X	15135.6	<i>[Signature]</i>
2011-18-11	Trim Tab Freeplay	10/20/17 15035.6	Measure / 100 hr	↓	X	15135.6	<i>[Signature]</i>
2018-24-02	wing strut bolts	11/12/18	inspected bolts	X	-	-	<i>[Signature]</i>
2022-51-22	elevator axle spar	11/21/23	insp. 10hr	X	X	17325.5	<i>[Signature]</i>
2022-08-08	trim jack	1/26/23	visual insp.	X	-	-	<i>[Signature]</i>
2021-05-11	wing struts	1/26/23	Previously complied with	X	-	-	<i>[Signature]</i>
2021-24-18	wing lug	1/26/23	replaced wing lugs	X	-	-	<i>[Signature]</i>