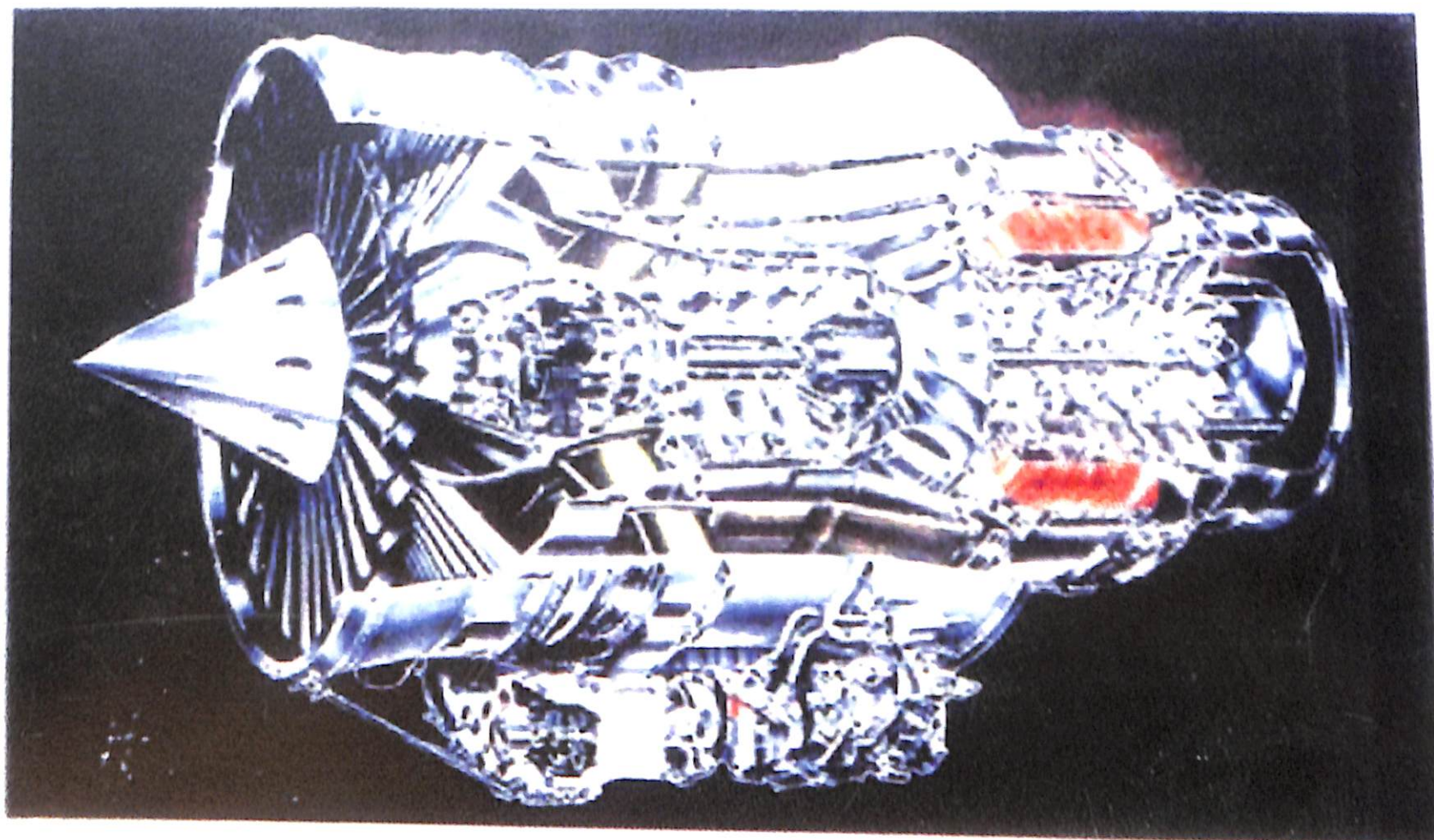
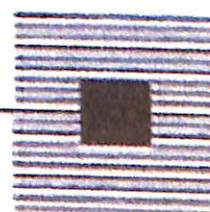




# TFE731



## ENGINE LOG BOOK

**Honeywell**

HONEYWELL International Inc.  
Engines & Systems  
111 S. 34th Street  
Phoenix, Az. 85034-2892

S/N P137180

P137180-000

DSC-A1629 , REV.

Honeywell Engine TFE731-40BR-1B  
Honeywell Part Number 3062075-1  
ENGINE SERIAL NUMBER P137180

TEST DATE: 02/11/2014

| SEA-LEVEL<br>U.S. STANDARD DAY<br>STATIC CONDITIONS                                                     |          |            |      | SEA-LEVEL<br>103.0 F<br>STATIC CONDITIONS |            |      |      |
|---------------------------------------------------------------------------------------------------------|----------|------------|------|-------------------------------------------|------------|------|------|
| TAKEOFF                                                                                                 |          |            |      | TAKEOFF                                   |            |      |      |
| PARAMETER                                                                                               | REQUIRED | TEST VALUE | TEST | REQUIRED                                  | TEST VALUE | TEST | TEST |
| NET THRUST LBS                                                                                          | 3850.    | 3850.      | I    | 3850.                                     | 3850.      | I    | I    |
|                                                                                                         | MINIMUM  | I          | I    | MINIMUM                                   | I          | I    | I    |
| THRUST SPECIFIC FUEL CONSUMPTION LB/HR-LB                                                               | 0.446    | 0.431      | I    | 0.469                                     | 0.455      | I    | I    |
|                                                                                                         | MAXIMUM  | I          | I    | MAXIMUM                                   | I          | I    | I    |
| HIGH PRESSURE TURBINE DISCHARGE TEMPERATURE, F                                                          | 1505.    | I          | I    | 1668.                                     | I          | I    | I    |
|                                                                                                         | I        | I          | I    | I                                         | I          | I    | I    |
| ENGINE SPEED, N1, RPM                                                                                   | 18458.   | I          | I    | 19219.                                    | I          | I    | I    |
|                                                                                                         | I        | I          | I    | I                                         | I          | I    | I    |
| ACCEPTANCE TEST CONDUCTED WITH AIR-OIL COOLERS INSTALLED.<br>NO PERFORMANCE ADJUSTMENTS HAVE BEEN MADE. |          |            |      | SIGNATURE DATE                            |            |      |      |
| N1 COMPENSATOR PART NO. 3060735-4                                                                       |          |            |      | TEST LABORATORY                           |            |      |      |
|                                                                                                         |          |            |      | 2/11/14                                   |            |      |      |
|                                                                                                         |          |            |      | FT 223                                    |            |      |      |
|                                                                                                         |          |            |      | QUALITY ASSURANCE                         |            |      |      |
|                                                                                                         |          |            |      | 2/12/14                                   |            |      |      |
|                                                                                                         |          |            |      | I                                         |            |      |      |

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DSC-A1630 , REV.

Honeywell Engine TFE731-40BR-1B  
Honeywell Part Number 3062075-1  
ENGINE SERIAL NUMBER P137180

TEST DATE: 02/11/2014

| AUTOMATIC PERFORMANCE RESERVE<br>PERFORMANCE AT TAKEOFF<br>SEA LEVEL STATIC CONDITIONS                   |                              |                        |            |                                 |                                 |  |  |
|----------------------------------------------------------------------------------------------------------|------------------------------|------------------------|------------|---------------------------------|---------------------------------|--|--|
| TEST VALUES                                                                                              |                              |                        |            |                                 |                                 |  |  |
| POWER<br>SETTING                                                                                         | AMBIENT<br>TEMP.<br>DEG. C/F | REQUIRED<br>THRUST LBS | THRUST LBS | N2 RPM<br>(MAX VALUE<br>31,200) | N1 RPM<br>(MAX VALUE<br>19,611) |  |  |
| APR                                                                                                      | 44.4/112.0                   | 3850                   | 3850.      | 30686.                          | 19371.                          |  |  |
| ACCEPTANCE TEST CONDUCTED WITH AIR-OIL<br>COOLERS INSTALLED: NO PERFORMANCE<br>ADJUSTMENT HAS BEEN MADE. |                              |                        |            | SIGNATURE                       | DATE                            |  |  |
|                                                                                                          |                              |                        |            | TEST<br>LABORATORY              | 2-11-14                         |  |  |
|                                                                                                          |                              |                        |            | QUALITY<br>ASSURANCE            | 2/12/14                         |  |  |

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**ENGINE DATA**



## ENGINE DATA

Model No. TFE731-40BR-1B

Part Number 3062075-1

Serial No. P137180

Net Thrust - Lbs. Takeoff 3850

Maximum Continuous 3500

Specific Fuel Consumption - Lb/Hr. 0.446

High Pressure Turbine Discharge Temperature

Deg F -----

Deg C -----

Fuel - See Aircraft Flight Manual

Oil - See Aircraft Flight Manual

**TOTAL ENGINE WEIGHT** 871 **LBS.**

Honeywell Engine TFE731-40BR-1B  
Honeywell Part Number 3062075-1  
ENGINE SERIAL NUMBER P137180

TEST DATE: 02/11/2014

| SEA-LEVEL<br>U.S. STANDARD DAY<br>STATIC CONDITIONS                                                     |          |               |          |               | SEA-LEVEL<br>103.0 F<br>STATIC CONDITIONS |               |          |               |  |
|---------------------------------------------------------------------------------------------------------|----------|---------------|----------|---------------|-------------------------------------------|---------------|----------|---------------|--|
| TAKEOFF                                                                                                 |          |               |          |               | TAKEOFF                                   |               |          |               |  |
| PARAMETER                                                                                               | REQUIRED | TEST<br>VALUE | REQUIRED | TEST<br>VALUE | REQUIRED                                  | TEST<br>VALUE | REQUIRED | TEST<br>VALUE |  |
| NET THRUST LBS                                                                                          | 3850.    | 3850.         |          |               | 3850.                                     | 3850.         |          |               |  |
|                                                                                                         | MINIMUM  |               |          |               | MINIMUM                                   |               |          |               |  |
| THRUST SPECIFIC FUEL<br>CONSUMPTION LB/HR-LB                                                            | 0.446    | 0.431         |          |               | 0.469                                     | 0.455         |          |               |  |
|                                                                                                         | MAXIMUM  |               |          |               | MAXIMUM                                   |               |          |               |  |
| HIGH PRESSURE TURBINE<br>DISCHARGE TEMPERATURE, F                                                       |          | 1505.         |          |               |                                           | 1668.         |          |               |  |
|                                                                                                         |          |               |          |               |                                           |               |          |               |  |
| ENGINE SPEED, N1, RPM                                                                                   |          | 18458.        |          |               |                                           | 19219.        |          |               |  |
|                                                                                                         |          |               |          |               |                                           |               |          |               |  |
| ACCEPTANCE TEST CONDUCTED WITH AIR-OIL COOLERS INSTALLED.<br>NO PERFORMANCE ADJUSTMENTS HAVE BEEN MADE. |          |               |          |               | SIGNATURE DATE                            |               |          |               |  |
| N1 COMPENSATOR PART NO. 3060735-4                                                                       |          |               |          |               | TEST LABORATORY                           |               |          |               |  |
|                                                                                                         |          |               |          |               | QUALITY ASSURANCE                         |               |          |               |  |

DSC-A1650 , REV.

Honeywell Engine TFE731-40BR-1B  
Honeywell Part Number 3062075-1  
ENGINE SERIAL NUMBER P127180

TEST DATE: 02/11/2014

| AUTOMATIC PERFORMANCE RESERVE<br>PERFORMANCE AT TAKEOFF<br>SEA LEVEL STATIC CONDITIONS                   |                              |                        |                                                                      |
|----------------------------------------------------------------------------------------------------------|------------------------------|------------------------|----------------------------------------------------------------------|
| TEST VALUES                                                                                              |                              |                        |                                                                      |
| POWER<br>SETTING                                                                                         | AMBIENT<br>TEMP.<br>DEG. C/F | REQUIRED<br>THRUST LBS | TEST VALUES                                                          |
|                                                                                                          |                              |                        | THRUST LBS    N2 RPM (MAX VALUE 31,200)    N1 RPM (MAX VALUE 19,611) |
| APR                                                                                                      | 44.4/112.0                   | 3850                   | 3850.    30686.    19371.                                            |
| SIGNATURE    DATE                                                                                        |                              |                        |                                                                      |
| ACCEPTANCE TEST CONDUCTED WITH AIR-OIL<br>COOLERS INSTALLED: NO PERFORMANCE<br>ADJUSTMENT HAS BEEN MADE. |                              |                        |                                                                      |
| TEST<br>LABORATORY                                                                                       |                              |                        |                                                                      |
| QUALITY<br>ASSURANCE                                                                                     |                              |                        |                                                                      |

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## CONFIGURATION REPORT

Unison Industries  
7575 Baymeadows Way, Jacksonville, FL 32256 USA  
Phone: 904-739-4201 Facsimile: 904-739-4006

**CERTIFICATE FOR SEALED RADIOACTIVE SOURCE**

Model number: 9060280-1 (Honeywell P/N 3061186-1)  
Serial number: J13460387  
Radionuclide: <sup>85</sup>Kr  
Description: Beta radiation source  
Gas mixture containing <sup>85</sup>Kr in sealed glass/metal envelope  
ISO Classification: ISO/99/C11111  
Estimated content activity: 215.4 kBq (107.7 kBq/spark gap)  
Radiation output:  
Quantity measured: Air kerma rate at 0.1 m from the source  
Results: < 0.002 µGy/hr (minimum level detectable by dose rate meter used)  
Test for freedom from surface contamination: N/A, inert gas  
Test for freedom from leakage:  
Method: ISO 9978 Hot-liquid bubble test (See 6.2.2)  
Results: Passed Date: 12/10/2013 (mm/dd/yyyy)

This certificate and the information contained herein complies with the requirements of ISO 2919

Bruce D. Markert Bruce D. Markert Radiation Safety Officer



# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description           | Rev | Serial Number   | Lot Number | Batch         | Ref.Des | Quantity |
|-----|-------------|--------------------------------|-----|-----------------|------------|---------------|---------|----------|
| 1   | 3070685-11  | VALVE, PRESSURE LIM            | U   | 18894           | 1          |               |         | 1.0      |
| 2   | 3070580-4   | BODY, VALVE - PCD LIMITER      | C   |                 |            |               |         | 1.0      |
| 2   | 3070581-1   | BELLOWS REGULATING,PCD LIMITER | R   | 131070005893    |            |               |         | 1.0      |
| 2   | 3070583-1   | RETAINER,SPRING                | B   |                 |            |               |         | 1.0      |
| 2   | 3070586-1   | SLEEVE SPACER                  | B   |                 |            |               |         | 1.0      |
| 2   | 3070587-1   | STRAP RETAINER                 | -   |                 |            |               |         | 1.0      |
| 2   | 3070596-1   | SEAT VALVE,BALL                |     |                 |            |               |         | 1.0      |
| 2   | 3070872-1   | SPRING, HELICAL, COMPRESSION   | -   |                 |            |               |         | 1.0      |
| 2   | 3070878-1   | SPRING,HELICAL COMP            | B   |                 |            |               |         | 1.0      |
| 2   | S21025-3    | PLATE,IDENTIFICATION           | A   |                 |            |               |         | 1.0      |
| 2   | 3070584-1   | LUG RETAINER,SPRING            | B   |                 |            |               |         | 1.0      |
| 1   | 394476-2    | VALVE OUTL,FLOW DIVIDER        |     | 3725            | 1          |               |         | 1.0      |
| 1   | 3060150-3   | SEAL, SEGMENTED                |     | 13-169020-00186 |            |               |         | 1.0      |
| 1   | 3070816-14  | SENSOR, P2T2                   |     | 1316901724313   |            |               |         | 1.0      |
| 1   | 3060116-1   | COUPLING SFT RIGID             |     |                 |            | P4575791      |         | 1.0      |
| 1   | 3060155-6   | FAN ROTOR ASSEMBLY             | F   | 13-162246-00720 |            | P4812555      |         | 1.0      |
| 2   | 3060154-3   | FAN BLADE                      |     |                 |            | 13-181879-27C |         | 12.0     |
| 2   | 3060154-3   | FAN BLADE                      |     |                 |            | 13-181879-47C |         | 3.0      |
| 2   | 3060154-3   | FAN BLADE                      |     |                 |            | 13-181879-48C |         | 13.0     |
| 2   | 3060154-3   | FAN BLADE                      |     |                 |            | 13-181879-34C |         | 2.0      |
| 2   | 3060287-2   | DISK, FAN                      | F   | 13-162246-00720 |            | 131034289     |         | 1.0      |
| 3   | 3900358-111 | DISK, FAN, ULTRASONIC SHAPE    | A   | 13-162246-00720 |            |               |         | 1.0      |
| 1   | 3060156-2   | SPINNER ENG INL                |     | 13-5060-00131   |            | 13-5060-267   |         | 1.0      |
| 1   | 3060270-3   | HOUSING, SEAL NO. 3            |     |                 |            | 13-162545-000 |         | 1.0      |
| 1   | 3060272-3   | GEAR HLCL SUN                  | E   | 13-1034-38478   |            | P4571238      |         | 1.0      |

Subject to U.S. Export Regulations

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description              | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|-------------|-----------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 2   | 3060272-751 | GEAR HELICAL SUN                  | C   | 13-1034-38478   | 5018143171    |       |         | 1.0      |
| 3   | 9-8925      | 6.00 CBS600 AMS6255BAR STOCK      |     |                 |               |       |         | 5.8      |
| 1   | 3060709-4   | COOLER FUEL OIL                   | D   | 3748            | 2             |       |         | 1.0      |
| 2   | 160506-4    | COOLER FUEL OIL                   | J   | 3748            |               |       |         | 1.0      |
| 1   | 3060720-5   | TANK, OIL, ENGINE                 |     | 14-5040-80003   |               |       |         | 1.0      |
| 1   | 3060735-4   | COMPENSATOR                       |     | 1316909400636   |               |       |         | 1.0      |
| 1   | 3060757-3   | HARNESS, WIRING                   |     | 13-150262-34905 |               |       |         | 1.0      |
| 1   | 3061186-1   | IGNITION UNIT                     |     | J13460387       |               |       |         | 1.0      |
| 1   | 3071870-3   | SHAFT SHOULDERED                  |     |                 | 13-169019-673 |       |         | 1.0      |
| 1   | 3072653-1   | HOUSING, TOWER SHAFT              |     |                 | 13-100159-323 |       |         | 1.0      |
| 1   | 3074672-2   | SEAL AIR OIL                      |     |                 | 13-162313-062 |       |         | 1.0      |
| 1   | 70060003-1  | PUMP, FUEL - MOTIVE FLOW          |     | 13-30003-00097  |               |       |         | 1.0      |
| 1   | 3060022-930 | PLANET GEAR ASSEMBLY              |     | LA049428        |               |       |         | 1.0      |
| 2   | 3060121-8   | CARRIER, PLANETARY REDUCTION GEAR |     | 1216224301178   | 13-3300-313   |       |         | 1.0      |
| 2   | 3060129-7   | PLANETARY GEARSHAFT               | A   | 14P64280        |               |       |         | 1.0      |
| 3   | 3060255-1   | BEARING, ROLLER PLANET-FWD        |     | 13-162255-25350 |               |       |         | 1.0      |
| 3   | 3060256-1   | BEARING, ROLLER PLANET-AFT        |     | 13-162255-29655 |               |       |         | 1.0      |
| 3   | 3060273-2   | GEAR HELICAL PLANETARY            | E   | 13-1034-39027   | P1592370      |       |         | 1.0      |
| 4   | 3060273-751 | GEAR HELICAL PLANETARY            | D   | 13-1034-39027   | LN131034193   |       |         | 1.0      |
| 5   | 9-8915      | 3.00 CBS600 AMS-6255 BAR STOCK    |     |                 |               |       |         | 5.0      |
| 2   | 3060129-7   | PLANETARY GEARSHAFT               | A   | 14P64281        |               |       |         | 1.0      |
| 3   | 3060255-1   | BEARING, ROLLER PLANET-FWD        |     | 13-162255-25352 |               |       |         | 1.0      |
| 3   | 3060256-1   | BEARING, ROLLER PLANET-AFT        |     | 13-162255-29663 |               |       |         | 1.0      |
| 3   | 3060273-2   | GEAR HELICAL PLANETARY            | E   | 13-1034-39025   | P1592370      |       |         | 1.0      |
| 4   | 3060273-751 | GEAR HELICAL PLANETARY            | D   | 13-1034-39025   | LN131034193   |       |         | 1.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description           | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|-------------|--------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 5   | 9-8915      | 3.00 CBS600 AMS-6255 BAR STOCK |     |                 |               |       |         | 5.0      |
| 2   | 3060129-7   | PLANETARY GEARSHAFT            | A   | 14P64282        |               |       |         | 1.0      |
| 3   | 3060255-1   | BEARING, ROLLER PLANET-FWD     |     | 13-162255-25353 |               |       |         | 1.0      |
| 3   | 3060256-1   | BEARING, ROLLER PLANET-AFT     |     | 13-162255-29668 |               |       |         | 1.0      |
| 3   | 3060273-2   | GEAR HELICAL PLANETARY         | E   | 13-1034-39041   | P1592370      |       |         | 1.0      |
| 4   | 3060273-751 | GEAR HELICAL PLANETARY         | D   | 13-1034-39041   | LN131034193   |       |         | 1.0      |
| 5   | 9-8915      | 3.00 CBS600 AMS-6255 BAR STOCK |     |                 |               |       |         | 5.0      |
| 2   | 3060129-7   | PLANETARY GEARSHAFT            | A   | 14P64283        |               |       |         | 1.0      |
| 3   | 3060255-1   | BEARING, ROLLER PLANET-FWD     |     | 13-162255-25356 |               |       |         | 1.0      |
| 3   | 3060256-1   | BEARING, ROLLER PLANET-AFT     |     | 13-162255-29674 |               |       |         | 1.0      |
| 3   | 3060273-2   | GEAR HELICAL PLANETARY         | E   | 13-1034-39047   | P1592370      |       |         | 1.0      |
| 4   | 3060273-751 | GEAR HELICAL PLANETARY         | D   | 13-1034-39047   | LN131034193   |       |         | 1.0      |
| 5   | 9-8915      | 3.00 CBS600 AMS-6255 BAR STOCK |     |                 |               |       |         | 5.0      |
| 2   | 3060129-7   | PLANETARY GEARSHAFT            | A   | 14P64284        |               |       |         | 1.0      |
| 3   | 3060255-1   | BEARING, ROLLER PLANET-FWD     |     | 13-162255-25357 |               |       |         | 1.0      |
| 3   | 3060256-1   | BEARING, ROLLER PLANET-AFT     |     | 13-162255-29749 |               |       |         | 1.0      |
| 3   | 3060273-2   | GEAR HELICAL PLANETARY         | E   | 13-1034-39040   | P1592370      |       |         | 1.0      |
| 4   | 3060273-751 | GEAR HELICAL PLANETARY         | D   | 13-1034-39040   | LN131034193   |       |         | 1.0      |
| 5   | 9-8915      | 3.00 CBS600 AMS-6255 BAR STOCK |     |                 |               |       |         | 5.0      |
| 2   | 3060212-1   | BEARING, BALL-LP-THRUST        |     | 12-162255-27358 |               |       |         | 1.0      |
| 2   | 3000005-64  | WASHER SPRING TNSN             |     |                 | 13-151587-101 |       |         | 1.0      |
| 1   | 3060022-980 | BYPASS DUCT ASSEMBLY           |     | LA049424        |               |       |         | 1.0      |
| 2   | 109696-6-1  | VALVE,SOLENOID-CONTROLLER      | K   | 14190           |               |       |         | 1.0      |
| 2   | 3060769-1   | VALVE, OIL REDUCTION           |     | 13-182215-12199 |               |       |         | 1.0      |
| 2   | 160554-5    | COOLER, OIL, SURGACE - OUTLINE | J   | 3692            | 1             |       |         | 1.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description                | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|-------------|-------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 3   | MFG3-6706   | CORE AY                             |     | 13110262        |               |       |         | 1.0      |
| 4   | R2722-150   | .003XMASTER 4047F AMS4185           |     |                 | 4912790056    |       |         | 1.0      |
| 2   | 160554-6    | COOLER, OIL, SURFACE - OUTLINE      | K   | 3743            | 2             |       |         | 1.0      |
| 3   | MFG3-6706   | CORE AY                             |     | 13100254        |               |       |         | 1.0      |
| 4   | R2722-150   | .003XMASTER 4047F AMS4185           |     |                 | 4910703486    |       |         | 1.0      |
| 2   | 3061300-8   | DUCT, BYPASS - LOWER                |     | 13-181898-01058 | 13-181898-024 |       |         | 1.0      |
| 2   | 3061300-7   | DUCT, BYPASS - UPPER                |     | 13-181898-01017 | 13-181898-024 |       |         | 1.0      |
| 2   | 160552-4    | COOLER, OIL, SURFACE - OUTLINE      | M   | 2279            |               |       |         | 1.0      |
| 3   | 157024-7    | CORE ASSEMBLY, OIL COOLER           | R   | 13100005        |               |       |         | 1.0      |
| 1   | 3060188-7   | SHAFT AND HOUSING ASSY, FAN SUPPORT | N   | LA049425        |               |       |         | 1.0      |
| 2   | 3061510-1   | BEARING, ROLLER                     |     | 13-162111-19837 |               |       |         | 1.0      |
| 2   | 3075657-1   | RUNNER ASSY SEAL NO1                |     | 13-162397-03934 |               |       |         | 1.0      |
| 2   | 3060130-1   | WASHER SPRING TNSN                  |     |                 | 12-162298-734 |       |         | 1.0      |
| 2   | 3060162-4   | HOUSING ASSEMBLY, FAN SUPPORT       |     |                 | 13-3300-256   |       |         | 1.0      |
| 2   | 3060180-2   | SHAFT FAN                           | B   | 1216205002353   | P4425693      |       |         | 1.0      |
| 3   | 3060180-901 | MACHINED FORGING SHA                |     | 1216205002353   | 5017178480    |       |         | 1.0      |
| 2   | 3060257-1   | BEARING BALL NO2                    |     | 13-162111-08661 |               |       |         | 1.0      |
| 2   | 3060277-4   | GEAR HELICAL RING                   | A   | 1215611408534   | P4365751      |       |         | 1.0      |
| 3   | 3060277-751 | GEAR HELICAL RING                   | B   | 1215611408534   | LN131034081   |       |         | 1.0      |
| 4   | 3900214-115 | FORGING GEAR RING                   | D   | 1215611408534   |               |       |         | 1.0      |
| 2   | 3060457-1   | SEAL, HYDRODYNAMIC CARBON, NO. 1    |     | 13-152826-46959 |               |       |         | 1.0      |
| 1   | 3060022-925 | POWER SECTION ASSEMB                |     | LA049423        |               |       |         | 1.0      |
| 2   | 3060601-2   | GEAR, EXCITER                       |     |                 | 13-163248-001 |       |         | 1.0      |
| 2   | 3060714-5   | SENSOR, SPEED (N1 MONOPOLE)         |     | 14910300013     |               |       |         | 1.0      |
| 2   | 3060718-4   | LEAD, N1 MONOPOLE                   |     | 14-150262-37555 |               |       |         | 1.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Serial Number P137180  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material     | Material Description                 | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|--------------|--------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 2   | 3060722-11   | LEAD IGNITER                         |     | S14020008       |               |       |         | 1.0      |
| 2   | 3060722-12   | LEAD IGNITER                         |     | S14010045       |               |       |         | 1.0      |
| 2   | 158465-6     | VALVE OIL TEMP CONT                  | L   | 14059           | 1             |       |         | 1.0      |
| 2   | 3071740-1    | NUT RND SLOTTED                      |     |                 | 13-162313-106 |       |         | 1.0      |
| 2   | 3060022-907  | PLENUM ASSEMBLY                      |     | LA049431        |               |       |         | 1.0      |
| 3   | 3060584-1    | RETAINER VANE HPT                    |     |                 | 12-108737-007 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 13-163498-01150 | 13-163498-011 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 13-163498-01171 | 13-163498-012 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 13-163498-01174 | 13-163498-012 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 13-163498-01177 | 13-163498-012 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00018 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00019 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00020 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00021 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00024 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00026 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00028 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060540-2    | NOZZLE FUEL INJ                      |     | 14-163498-00063 | 14-163498-001 |       |         | 1.0      |
| 3   | 3060318-8    | DIFFUSER, CENTRIFUGAL COMPRESSOR, HP |     |                 | 14-3300-008   |       |         | 1.0      |
| 3   | 3060534-4    | CASE, COMBUSTION CHAMBER             |     | 13-3300-02954   | 14-3300-018   |       |         | 1.0      |
| 4   | MS20605R3W2  | MS20605R3W2 Per NASM20605 RIVET      |     |                 | 010008347233  |       |         | 80.0     |
| 4   | MS20470AD6-6 | RIVET, SOLID, UNIVERSAL HEAD         |     |                 | 010004909875  |       |         | 1.0      |
| 4   | MS21062-3    | NUT PLATE                            |     |                 | 010007113195  |       |         | 2.0      |
| 4   | MS21076-3    | NUT PLATE                            |     |                 | 010007751050  |       |         | 38.0     |
| 4   | 3060545-2    | BOLT, D HEAD                         |     |                 | 010008088619  |       |         | 40.0     |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material       | Material Description           | Rev | Serial Number   | Lot Number    | Batch      | Ref.Des | Quantity |
|-----|----------------|--------------------------------|-----|-----------------|---------------|------------|---------|----------|
| 4   | 3072443-935    | SUPPORT                        |     |                 |               |            |         | 1.0      |
| 4   | VZOREK 28      | SAMPLE TH3.18 INCO 718         |     |                 | 6004248082    |            |         | 1.0      |
| 4   | 3060534-937    | BOSS                           |     |                 | 010008322389  |            |         | 3.0      |
| 4   | 3060534-977    | CLIP                           |     |                 | 010008010470  |            |         | 1.0      |
| 4   | 3060534-979    | GUSSET, Find Number 10/MBP     |     |                 | 010006542385  |            |         | 2.0      |
| 4   | 3072443-927    | MACHINED BOSS                  |     |                 | 010008432019  |            |         | 1.0      |
| 4   | 3060534-907.51 | WELDED SUBASSY                 |     |                 | 6004443780    |            |         | 1.0      |
| 4   | 3060534-991.51 | WELDED FLANGE                  |     |                 | 6004443791    |            |         | 1.0      |
| 3   | 3060573-2      | LINER, OUTER TRANSITION        |     | 13-3300-02988   | 13-3300-326   |            |         | 1.0      |
| 4   | 4662098        | PLASMA POWDER METCO 81VF-NS    |     |                 |               | 0001720065 |         | 150.0    |
| 4   | 3060573-01     | SUB ASSY                       |     |                 | 6004490967    |            |         | 1.0      |
| 4   | 3060573-902    | FLANGE                         |     |                 | 6004377006    |            |         | 1.0      |
| 4   | 3060573-903    | LINER                          |     |                 | 6004413050    |            |         | 1.0      |
| 3   | 3060579-1      | SWIRL RING, COOLING AIR, BLADE |     |                 | 13-3300-316   |            |         | 1.0      |
| 3   | 3060766-1      | CHAMBER, COMBUSTION            |     | 13-3300-03522   | 13-3300-334   |            |         | 1.0      |
| 4   | 3060766-914    | OUTER PANEL                    | W   |                 |               |            |         | 1.0      |
| 4   | 3060766-916    | INNER SUB ASSY                 | P   |                 |               |            |         | 1.0      |
| 4   | 3073026-1      | IGNITER GROMMET                |     |                 | 010008426677  |            |         | 2.0      |
| 4   | 3075542-2      | NOZZLE GROMMET                 |     |                 | 010008309673  |            |         | 12.0     |
| 4   | 4662081        | PLASMA POWDER AMDRY 964        |     |                 |               | 0001908289 |         | 0.4      |
| 4   | 4665696        | PLASMA POWDER AMDRY 204 NS-1   |     |                 |               | 0001929369 |         | 1.0      |
| 2   | 3060022-912    | TURBINE/REAR MOUNT ASSY        |     | LA049426        |               |            |         | 1.0      |
| 3   | 3070344-2      | ROTOR SEAL                     |     |                 | 13-162313-147 |            |         | 1.0      |
| 3   | 3071383-2      | SEAL AIR OIL                   |     | 13-150211-00287 | 13-150211-015 |            |         | 1.0      |
| 3   | 358750-2       | BEARING, ROLLER                |     | 13-162111-13306 |               |            |         | 1.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description                 | Rev | Serial Number   | Lot Number    | Batch      | Ref.Des | Quantity |
|-----|-------------|--------------------------------------|-----|-----------------|---------------|------------|---------|----------|
| 3   | 3060626-8   | TURBINE ROTOR ASSY - LP, THIRD STAGE | H   | 12-156103-00929 | P4806391      |            |         | 1.0      |
| 4   | 3060671-2   | DISK, ROTOR - LP TURBINE, STAGE 3    |     | 12-156103-00929 | 13-169025-499 |            |         | 1.0      |
| 4   | 3075307-1   | SEAL ROTOR LPT 3STG                  |     | 12-169025-00441 | 13-169025-832 |            |         | 1.0      |
| 4   | 3061823-2   | BLADE, TURBINE, ROTOR, LPT, STAGE 3  |     |                 | HG-4024       |            |         | 1.0      |
| 4   | 3061823-2   | BLADE, TURBINE, ROTOR, LPT, STAGE 3  |     |                 | HG-4025       |            |         | 55.0     |
| 3   | 3060653-7   | NOZZLE TURB LP 2STG                  |     |                 | 13-3300-318   |            |         | 1.0      |
| 3   | 3060655-6   | NOZZLE ASSY LPT 3STG                 |     |                 | 13-3300-340   |            |         | 1.0      |
| 3   | 3060662-1   | PIN, RETAINER RING                   |     |                 | 13-5060-333   |            |         | 1.0      |
| 3   | 3060674-3   | SHAFT AND SEAL AY LPT                |     |                 | P4708621      |            |         | 1.0      |
| 3   | 3060694-8   | DUCT, TRANSITION, TURBINE INTERSTAGE |     | 13-3300-02416   | 13-3300-303   |            |         | 1.0      |
| 4   | 3060694-921 | FLANGE                               |     |                 | 010008183544  |            |         | 1.0      |
| 4   | 3060694-964 | PIN                                  |     |                 | 010006358376  |            |         | 6.0      |
| 4   | 3075293-1   | BOLT ENG MOUNT AFT                   |     |                 | 010007801304  |            |         | 19.0     |
| 4   | 4662081     | PLASMA POWDER AMDRY 964              |     |                 |               | 0001781837 |         | 0.2      |
| 4   | 4665696     | PLASMA POWDER AMDRY 204 NS-1         |     |                 |               | 0001781836 |         | 0.4      |
| 4   | 3060694-927 | FLANGE TX T.OWEN                     |     |                 | 010007815370  |            |         | 1.0      |
| 4   | 3060694-901 | SUB ASSY                             |     | 13-3300-02416   |               |            |         | 1.0      |
| 5   | 3060694-923 | FLANGE                               |     |                 | 010008328243  |            |         | 1.0      |
| 5   | 3060694-925 | FLANGE                               |     |                 | 010008094156  |            |         | 1.0      |
| 5   | 3060694-933 | CONE                                 |     |                 | 010008222779  |            |         | 1.0      |
| 4   | 3060694-920 | FWD FLANGE                           |     |                 |               |            |         | 1.0      |
| 4   | 3060694-951 | PAD                                  |     |                 | 010006376613  |            |         | 1.0      |
| 4   | 3060694-961 | MACHINED PAD                         |     |                 | 010008021092  |            |         | 4.0      |
| 3   | 3060763-3   | HARNESS ITT                          |     | 1316901740684   |               |            |         | 1.0      |
| 3   | 3061497-3   | RING, RETAINING, HP SHROUD           |     |                 | 12-182405-013 |            |         | 1.0      |

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# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material     | Material Description                     | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|--------------|------------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 3   | 3074108-2    | SEAL AIR ROTTUR2STG                      |     | 12-169025-00239 | 13-169025-692 |       |         | 1.0      |
| 3   | 3074743-1    | SHAFT TIE                                |     |                 | 13-156138-A4W |       |         | 1.0      |
| 3   | 3074767-1    | SEAL ROTATING                            |     |                 | 12-162313-032 |       |         | 1.0      |
| 3   | 3000005-5    | WASHER SPRING                            |     |                 | 11-162298-725 |       |         | 1.0      |
| 3   | 3075443-4    | BEARING HOUSING, REAR - NO. 6            |     | 13-3300-03337   | 13-3300-354   |       |         | 1.0      |
| 4   | 423-603-9011 | INSERT, SCREW THREAD,-KEY LOCKED (THIN W |     |                 | 010008212033  |       |         | 11.0     |
| 4   | 3072889-5    | PIN ALIGNMENT                            |     |                 | 010006950805  |       |         | 1.0      |
| 4   | 3074809-955  | PLUG                                     |     |                 | 010006497551  |       |         | 1.0      |
| 4   | 3075443-111  | HOUSING BEARING                          |     |                 | 010008659527  |       |         | 1.0      |
| 4   | 3075443-902  | SUB ASSY                                 |     |                 |               |       |         | 1.0      |
| 4   | 3075443-931  | STIFFENER                                |     |                 | 010008048877  |       |         | 1.0      |
| 4   | 4665708      | PLASMA POWDER METCO 307 NS               |     |                 | 010007771570  |       |         | 0.2      |
| 3   | 3060610-5    | TURBINE ROTOR ASSY - LP, FIRST STAGE     | -   | 12-156103-01094 | P4702468      |       |         | 1.0      |
| 4   | 3060612-1    | BLADE, TURBINE ROTOR, LPT - STAGE 1      |     |                 | HM-3317       |       |         | 17.0     |
| 4   | 3060612-1    | BLADE, TURBINE ROTOR, LPT - STAGE 1      |     |                 | HM-3316       |       |         | 25.0     |
| 4   | 3060612-1    | BLADE, TURBINE ROTOR, LPT - STAGE 1      |     |                 | HM-3312       |       |         | 8.0      |
| 4   | 3060612-1    | BLADE, TURBINE ROTOR, LPT - STAGE 1      |     |                 | HM-3315       |       |         | 14.0     |
| 4   | 3060625-1    | DISK, ROTOR - LP TURBINE, 1ST STAGE      |     | 12-156103-01094 | 13-169025-536 |       |         | 1.0      |
| 3   | 3060608-5    | TURBINE ROTOR ASSY - LP, SECOND STAGE    | -   | 12-156103-00810 | P5298284      |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4154(ECWJ) |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECCW) |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECUH) |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECTY) |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECWM) |       |         | 1.0      |
| 4   | 3075424-2    | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECWL) |       |         | 2.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material    | Material Description                     | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|-------------|------------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECWC) |       |         | 9.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4150(ECWA) |       |         | 5.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4149(ECWM) |       |         | 2.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4149(ECWE) |       |         | 1.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4149(ECWJ) |       |         | 2.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4149(ECWK) |       |         | 12.0     |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4149(ECWL) |       |         | 10.0     |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4113(ECWD) |       |         | 2.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4113(ECWC) |       |         | 1.0      |
| 4   | 3075424-2   | BLADE, TURNINE ROTOR 2DN STAGE LP        |     |                 | HM-4113(ECCW) |       |         | 1.0      |
| 4   | 3060611-1   | DISK, TURBINE ROTOR - LOW PRESSURE, STG  |     | 12-156103-00810 | 13-169025-961 |       |         | 1.0      |
| 3   | 3060666-11  | NOZZLE ASSY, LPT, STAGE 1                |     | 13-3300-02512   | 13-3300-340   |       |         | 1.0      |
| 4   | NAS1398C5-4 | "RIVET, BLIND, PROTRUDING HEAD, LOCKED S |     |                 | 010008390849  |       |         | 1.0      |
| 4   | NAS1398C5-4 | "RIVET, BLIND, PROTRUDING HEAD, LOCKED S |     |                 | 010008390849  |       |         | 9.0      |
| 4   | 3060679-1   | DEFLECTOR FORWARD                        |     |                 | 010008377847  |       |         | 1.0      |
| 4   | 3060680-1   | DEFLECTOR-AFT                            |     |                 | 010008431760  |       |         | 1.0      |
| 4   | 3060693-1   | SUPPORT, STATIONARY SEAL, AIR            |     | 13-3300-03364   |               |       |         | 1.0      |
| 5   | 3060693-111 | SUPPORT-INVEST CSTG                      |     |                 | 010008578133  |       |         | 1.0      |
| 4   | 3073765-12  | SEAL, AIR, STATIONARY                    |     | 13-3300-03420   |               |       |         | 1.0      |
| 5   | 3073765     | SEAL                                     |     |                 | 6004508584    |       |         | 1.0      |
| 4   | 3001797-13  | BUSHING TURB NOZ                         |     |                 | 010008509620  |       |         | 9.0      |
| 4   | 3001797-6   | BUSHING TURB NOZ                         |     |                 | 010008378403  |       |         | 9.0      |
| 4   | NAS1198-5-7 | RIVET                                    |     |                 | 010008432008  |       |         | 9.0      |
| 4   | NAS1398C5-2 | "RIVET, BLIND, PROTRUDING HEAD, LOCKED S |     |                 | 010008371282  |       |         | 2.0      |
| 4   | NAS1398C5-2 | "RIVET, BLIND, PROTRUDING HEAD, LOCKED S |     |                 | 010008371282  |       |         | 15.0     |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material     | Material Description                     | Rev | Serial Number   | Lot Number    | Batch | Ref.Des | Quantity |
|-----|--------------|------------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 4   | 3060651-10   | NOZZLE TURBINE LP, STAGE 1               | -   | 13-3300-02512   |               |       |         | 1.0      |
| 5   | 3060651-115  | NOZZLE                                   |     |                 | 010007938570  |       |         | 1.0      |
| 3   | 3073718-3    | SUPPORT, BEARING HOUSING, REAR - NO. 6   |     | 13-3300-01989   | 13-3300-257   |       |         | 1.0      |
| 4   | MS20605R3W2  | MS20605R3W2 Per NASM20605 RIVET          |     |                 | 010006515301  |       |         | 24.0     |
| 4   | MS20501-1032 | NUT                                      |     |                 | 010007581898  |       |         | 12.0     |
| 4   | 3075384-902  | SUB ASSEMBLY FIND #3 AND #4              |     |                 |               |       |         | 1.0      |
| 4   | 3073718-113  | SUPPORT, BEARING HOUSING, REAR - NO. 6 - |     |                 | 010007957713  |       |         | 1.0      |
| 2   | 3060022-913  | LPT1 SHROUD ASSEMBLY                     |     | LA049427        |               |       |         | 1.0      |
| 3   | 3061494-1    | HPT SHROUD SEGMENT                       |     |                 | 13-150285-018 |       |         | 9.0      |
| 3   | 3061494-1    | HPT SHROUD SEGMENT                       |     |                 | 13-150285-017 |       |         | 3.0      |
| 3   | 3061495-2    | CASE HP TURBINE                          | B   | 13-156113-18883 |               |       |         | 1.0      |
| 4   | 3061495-903  | CASE                                     |     | 13-156113-18883 | 010008881118  |       |         | 1.0      |
| 2   | 3060022-950  | FRONT FRAME ASSEMBLY                     | CA  | LA049430        |               |       |         | 1.0      |
| 3   | 3060163-5    | VANE ASSEMBLY, FAN STATOR                | -   | 14P63992        | P4787479      |       |         | 1.0      |
| 3   | 3060435-1    | HOUSING, FAN INLET                       |     |                 | 13-3300-277   |       |         | 1.0      |
| 3   | 3060421-3    | FRAME ASSY, FRONT - 9 STRUT MACHINING    |     | 13-101376-00047 | 13-3300-330   |       |         | 1.0      |
| 4   | 423-603-9031 | INSERT, SCREW THREAD, -KEY LOCKED (THIN  |     |                 | 010008426750  |       |         | 58.0     |
| 4   | 423-603-9031 | INSERT, SCREW THREAD, -KEY LOCKED (THIN  |     |                 | 010008426750  |       |         | 30.0     |
| 4   | 423-603-9031 | INSERT, SCREW THREAD, -KEY LOCKED (THIN  |     |                 | 010008426750  |       |         | 28.0     |
| 4   | 682-557-2109 | STUD KEY LKD                             |     |                 | 010008524751  |       |         | 88.0     |
| 4   | 682-557-2111 | STUD KEY LKD                             |     |                 | 010008425669  |       |         | 14.0     |
| 4   | 682-557-2111 | STUD KEY LKD                             |     |                 | 010008619935  |       |         | 9.0      |
| 4   | 682-557-2111 | STUD KEY LKD                             |     |                 | 010008619935  |       |         | 68.0     |
| 4   | 3060189-1    | PIN                                      |     |                 | 010007959301  |       |         | 5.0      |
| 4   | 3060207-12   | BUSHING                                  |     |                 | 010006669415  |       |         | 8.0      |

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material      | Material Description                     | Rev | Serial Number   | Lot Number    | Batch      | Ref.Des | Quantity |
|-----|---------------|------------------------------------------|-----|-----------------|---------------|------------|---------|----------|
| 4   | 3060207-13    | BUSHING                                  |     |                 | 010007209145  |            |         | 2.0      |
| 4   | 3060207-14    | BUSHING                                  |     |                 | 010008050079  |            |         | 2.0      |
| 4   | 3060207-15    | BUSHING                                  |     |                 | 010007630065  |            |         | 8.0      |
| 4   | 3060207-2     | BUSHING                                  |     |                 | 010005066364  |            |         | 2.0      |
| 4   | 3060207-4     | BUSHING                                  |     |                 | 010007102395  |            |         | 2.0      |
| 4   | 3060416-1     | PLUG FRONT FRAME                         |     |                 | 01003824413   |            |         | 2.0      |
| 4   | 682-557-2211  | STUD KEY LKD                             |     |                 | 010008535082  |            |         | 18.0     |
| 4   | 423-603-9022  | INSERT, SCREW THREAD, -KEY LOCKED (THIN  |     |                 | 010007692336  |            |         | 4.0      |
| 4   | 682-5504-2401 | STUD, KEY LOCKED - WEIGHT - REDUCED, INC |     |                 | 010008619396  |            |         | 18.0     |
| 4   | 3060420-111   | FRAME, FRONT - 9 STRUT CASTING           |     | 13-101376-00047 | 010008454945  |            |         | 1.0      |
| 2   | 3060300-17    | COMPRESSOR ASSEMBLY                      | AC  | 14P07071        |               |            |         | 1.0      |
| 3   | 3071824-5     | RETAINER BRG HP RTR                      |     |                 | 13-169006-216 |            |         | 1.0      |
| 3   | 3075016-1     | BEARING BALL THR RTR                     | J   | 13-151725-06988 |               | 0001972012 |         | 1.0      |
| 3   | 3075633-2     | SEAL RUNNER ASSY NO5                     |     | 13-162397-07434 |               |            |         | 1.0      |
| 3   | 3075639-2     | SEAL RUNNER ASSY NO4                     |     | 13-162397-08717 |               |            |         | 1.0      |
| 3   | 358751-1      | BEARING, ROLLER                          |     | 13-162111-20334 |               |            |         | 1.0      |
| 3   | 358752-3      | BEARING, BALL                            |     | 13-151725-06554 |               |            |         | 1.0      |
| 3   | 358759-1      | BEARING, ROLLER CYLINDRICAL-HP ROTOR     |     | 13-162255-34891 |               |            |         | 1.0      |
| 3   | 3060316-4     | CASE HP IMPELLER                         |     |                 | 13-3300-339   |            |         | 1.0      |
| 3   | 3060366-3     | SHAFT SHOULDERED                         | -   | 14P64042        | P4574637      |            |         | 1.0      |
| 4   | 3060366-751   | SHAFT SHOULDER                           |     |                 | 5018673608    |            |         | 1.0      |
| 3   | 3060371-1     | COMPRESSOR, ROTOR, CENTRIFUGAL           | J   | 13-182449-09493 | 141034008     |            |         | 1.0      |
| 4   | 3900232-111   | FORGING, IMPELLER                        | B   | 13-182449-09493 |               |            |         | 1.0      |
| 3   | 3060382-1     | SPACER SLEEVE                            |     |                 | 13-5060-336   |            |         | 1.0      |
| 3   | 3060436-1     | SEAL, HYDRODYNAMIC CARBON, NO. 5         |     | 14-152826-49353 |               |            |         | 1.0      |

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 Honeywell International Inc  
 111 South 34th Street  
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|-----|-------------|------------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 3   | 3060437-1   | SEAL, HYDRODYNAMIC CARBON, NO. 4         |     | 13-152826-47021 |               |       |         | 1.0      |
| 3   | 3060502-2   | COUPLING, SHAFT HP TURBINE               | E   | 13-1034-40068   | 131034289     |       |         | 1.0      |
| 4   | 3060502-751 | COUPLING SFT HP TURB                     | D   |                 |               |       |         | 1.0      |
| 4   | 3073609-5   | RING SEAL METAL                          | K   |                 |               |       |         | 1.0      |
| 4   | 3060502-915 | PIN                                      | -   |                 |               |       |         | 2.0      |
| 3   | 3060514-1   | SEAL PLATE, HPT                          | J   | 13-182449-08997 | 131034347     |       |         | 1.0      |
| 4   | 3060514-751 | SEAL PLATE HPT                           | E   | 13-182449-08997 |               |       |         | 1.0      |
| 5   | 3900219-111 | FORGING, ROTATING                        | B   | 13-182449-08997 |               |       |         | 1.0      |
| 3   | 3060555-5   | HPT ROTOR ASSEMBLY                       | H   | 12-156111-00365 | P4787495      |       |         | 1.0      |
| 4   | 3060556-1   | BLADE, TURBINE ROTOR, HP                 |     |                 | HM-3360       |       |         | 7.0      |
| 4   | 3060556-1   | BLADE, TURBINE ROTOR, HP                 |     |                 | HM-3361       |       |         | 49.0     |
| 4   | 3060841-1   | DISK, TURBINE ROTOR - HP                 | J   | 12-156111-00365 | 131034296     |       |         | 1.0      |
| 5   | 3900246-111 | FORGING, DISK, HPT                       |     | 12-156111-00365 |               |       |         | 1.0      |
| 3   | 3060698-1   | SEAL, AIR, ROTATING                      | F   | 12-162053-47086 | 14-1034-017   |       |         | 1.0      |
| 4   | 3060698-761 | SEAL AIR ROTATING                        |     | 12-162053-47086 |               |       |         | 1.0      |
| 5   | 3900222-111 | FORGING ROTATING                         | A   | 12-162053-47086 |               |       |         | 1.0      |
| 3   | 3060804-1   | GEARSHAFT, BEVEL                         | B   | 14P63798        | P4606762      |       |         | 1.0      |
| 4   | 3060804-751 | MACHINED BLANK                           |     |                 | LN131034219   |       |         | 1.0      |
| 3   | 3060805-1   | GEAR, BEVEL, ACCESSORY DRIVE             |     |                 | P4674811      |       |         | 1.0      |
| 3   | 3070024-1   | SCOOP, OIL                               |     |                 | 13-107281-006 |       |         | 1.0      |
| 3   | 3070027-1   | TUBE TRANSFER                            |     |                 | 09-162806-001 |       |         | 1.0      |
| 3   | 3070151-4   | TIE SHAFT NUT                            |     |                 | 13-5040-420   |       |         | 1.0      |
| 3   | 3070369-2   | SUPPORT ASSY                             |     |                 | 13-156138-DRG |       |         | 1.0      |
| 3   | 3070490-1   | SCOOP OIL                                |     |                 | 14-5060-015   |       |         | 1.0      |
| 3   | 3071463-1   | CARRIER ASSEMBLY, ROLLER BEARING - HP CO |     |                 | 13-3300-303   |       |         | 1.0      |

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|-----|-------------|--------------------------------------------|-----|---------------|---------------|-------|---------|----------|
| 3   | 3071863-1   | RETAINER BRG                               |     |               | 12-169006-666 |       |         | 1.0      |
| 3   | 3071866-3   | CARRIER BEARING                            |     |               | 13-162313-122 |       |         | 1.0      |
| 3   | 3060599-3   | COUPLING, REAR SEAL - HP SHAFT             | H   | 13-1034-40494 | 131034336     |       |         | 1.0      |
| 2   | 3061704-2   | NOZZLE, ASSEMBLY, H.P. TURBINE             | A   | 14P07070      |               |       |         | 1.0      |
| 3   | 3060581-1   | SEAL CARR HP TURB                          |     |               | 13-156162-010 |       |         | 1.0      |
| 3   | 3061496-2   | HP NOZZLE RETAINING RING                   |     |               | 12-182382-762 |       |         | 1.0      |
| 2   | 3061704-902 | A-4 NOZZLE ASSEMBLY                        |     | LA049432      |               |       |         | 1.0      |
| 3   | 3060840-4   | NOZZLE SEGMENT, HP, COOLED                 |     |               | 13-1082-325   |       |         | 4.0      |
| 3   | 3060840-5   | NOZZLE SEGMENT, HP, COOLED                 |     |               | 13-1082-213   |       |         | 1.0      |
| 3   | 3060840-6   | NOZZLE SEGMENT, HP, COOLED                 |     |               | 14-1082-027   |       |         | 4.0      |
| 3   | 3060840-6   | NOZZLE SEGMENT, HP, COOLED                 |     |               | 14-1082-024   |       |         | 2.0      |
| 3   | 3060840-6   | NOZZLE SEGMENT, HP, COOLED                 |     |               | 14-1082-016   |       |         | 4.0      |
| 2   | 3060022-971 | LPC STACK                                  |     | LA049429      |               |       |         | 1.0      |
| 3   | 978812-7-1  | VALVE OUTL MODULATING & SHUTOFF PNEUMATIG  |     | 14833         | 1             |       |         | 1.0      |
| 3   | 3060334-1   | SEAL                                       |     |               | 13-3300-330   |       |         | 1.0      |
| 3   | 3070279-17  | COMPRESSOR STATOR ASSEMBLY, THIRD STAGE AF |     | 13P63456      | P4708714      |       |         | 1.0      |
| 4   | 3060348-4   | VANE, COMPRESSOR STATOR - STAGE 3          |     |               | 13-3500-106   |       |         | 21.0     |
| 4   | 3060348-4   | VANE, COMPRESSOR STATOR - STAGE 3          |     |               | 13-3500-071   |       |         | 8.0      |
| 4   | 3060348-4   | VANE, COMPRESSOR STATOR - STAGE 3          |     |               | 12-3500-045   |       |         | 11.0     |
| 4   | 3060348-5   | VANE, COMPRESSOR STATOR - STAGE 3          |     |               | 13-3500-049   |       |         | 1.0      |
| 4   | 3060348-6   | VANE STATOR LPC-3                          |     |               | 13-3500-145   |       |         | 18.0     |
| 4   | 3072464-6   | RING RTNR VANE 3STG                        |     |               | 13-108737-003 |       |         | 1.0      |
| 3   | 3070280-9   | COMPRESSOR STATOR ASSY, 2ND STAGE - LOW M  |     | 13P63454      | P4708928      |       |         | 1.0      |
| 4   | 3060347-4   | VANE STATOR LPC-2                          |     |               | 13-3500-103   |       |         | 2.0      |
| 4   | 3060347-4   | VANE STATOR LPC-2                          |     |               | 13-3500-052   |       |         | 8.0      |

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|-----|-------------|--------------------------------------------|----------|-----------------|---------------|-------|---------|----------|
| 4   | 3060347-4   | VANE STATOR LPC-2                          |          |                 | 13-3500-068   |       |         | 6.0      |
| 4   | 3060347-4   | VANE STATOR LPC-2                          |          |                 | 13-3500-051   |       |         | 2.0      |
| 4   | 3060347-4   | VANE STATOR LPC-2                          |          |                 | 12-3500-309   |       |         | 40.0     |
| 4   | 3060347-5   | VANE STATOR LPC-2                          |          |                 | 12-3500-085   |       |         | 1.0      |
| 4   | 3060347-6   | VANE STATOR LPC-2                          |          |                 | 08-07641-670  |       |         | 2.0      |
| 4   | 3072482-2   | RING RETAINING                             |          |                 | 12-156138-A4F |       |         | 1.0      |
| 4   | 3072485-1   | STOP, VANE, STATOR - COMPRESSOR, SECOND    |          |                 | 09-163248-001 |       |         | 1.0      |
| 3   | 3070281-12  | COMPRESSURE STATOR ASSY, 1ST STAGE LOW P V | 13P63452 |                 | P4708712      |       |         | 1.0      |
| 4   | 3060346-4   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 13-3500-122   |       |         | 1.0      |
| 4   | 3060346-4   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 13-3500-110   |       |         | 14.0     |
| 4   | 3060346-4   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 13-3500-073   |       |         | 5.0      |
| 4   | 3060346-4   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 13-3500-143   |       |         | 25.0     |
| 4   | 3060346-5   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 12-3500-287   |       |         | 1.0      |
| 4   | 3060346-6   | VANE, COMPRESSOR STATOR - STAGE 1          |          |                 | 0707641184    |       |         | 1.0      |
| 4   | 3072481-2   | RING VANE RETAINER, FIRST STAGE CMPSR      |          |                 | 13-108737-002 |       |         | 1.0      |
| 4   | 3072484-1   | STOP STATOR VANE                           |          |                 | 13162301068   |       |         | 1.0      |
| 3   | 3070289-9   | SHROUD LPC-2                               |          |                 | 13-182127-003 |       |         | 1.0      |
| 3   | 3070292-9   | SHROUD, COMPRESSOR WHEEL, THIRD STAGE      |          |                 | 13-182127-001 |       |         | 1.0      |
| 3   | 3071740-1   | NUT RND SLOTTED                            |          |                 | 14-162313-010 |       |         | 1.0      |
| 3   | 3072417-12  | COMPRESSOR ROTOR ASSY, AXIAL, TURBINE EN   | AK       | 13-162045-00656 | P4775847      |       |         | 1.0      |
| 4   | 3060343-1   | BLADE CPRSR RTR 3STG                       |          |                 | 13-3500-187   |       |         | 39.0     |
| 4   | 3060374-1   | DISK, AXIAL COMPRESSOR, TURBINE ENGINE -   | H        | 13-162045-00656 | 131034242     |       |         | 1.0      |
| 5   | 3060374-111 | DISK, AXIAL COMPRESSOR, TURBINE ENGINE -   | H        | 13-162045-00656 |               |       |         | 1.0      |
| 4   | 3072404-4   | SHROUD, AXIAL COMPRESSOR, TURBINE ENGINE   |          |                 | 131034341     |       |         | 1.0      |
| 3   | 3074744-1   | TIE ROD, THREADED ENDS                     | P        | 13P62792        |               |       |         | 1.0      |

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|-----|-------------|-------------------------------------------|-----|-----------------|---------------|-------|---------|----------|
| 4   | 9-8934      | 1.375 718 EMS55476 BAR STOCK              |     |                 | 5018814423    |       |         | 36.0     |
| 3   | 3075351-12  | COMPRESSOR ROTOR ASSY, AXIAL - STAGE 4    | AA  | 13-162053-54870 | P4783796      |       |         | 1.0      |
| 4   | 3060344-1   | BLADE LPC-4                               |     |                 | 13-3500-183   |       |         | 10.0     |
| 4   | 3060344-1   | BLADE LPC-4                               |     |                 | 13-3500-206   |       |         | 32.0     |
| 4   | 3072405-4   | SHROUD, AXIAL COMPRESSOR, TURBINE ENGINE  |     |                 | 141034015     |       |         | 1.0      |
| 4   | 3074803-5   | SHAFT, SHOULDERED, LOW PRESSURE COMPRESSW |     | 14P63884        | P4668145      |       |         | 1.0      |
| 5   | 3074803-753 | SHAFT SHLDR LP CPRSR                      |     |                 | 5018756532    |       |         | 1.0      |
| 4   | 3075326-5   | DISK, AXIAL COMPRESSOR, - STAGE 4         | AB  | 13-162053-54870 | 131034263     |       |         | 1.0      |
| 5   | 3075324-111 | FORGING                                   | B   | 13-162053-54870 |               |       |         | 1.0      |
| 3   | 3060361-5   | CASE AND VANE ASSY STAGE 4                |     | 13-3300-03751   | 13-3300-352   |       |         | 1.0      |
| 4   | MS20427F3-4 | MS20427F3-4 PER NASM20427 RIVT            |     |                 | 0100064700633 |       |         | 6.0      |
| 4   | MS21070-3   | NUT, SELFLOCKING                          |     |                 | 010008268465  |       |         | 3.0      |
| 4   | 3075466-2   | PIN STATOR LPC 4STG                       |     |                 | 0100085364    |       |         | 1.0      |
| 4   | 3060361-921 | MACHINED HUB                              |     |                 |               |       |         | 1.0      |
| 4   | 3060361-927 | SHROUD                                    |     |                 |               |       |         | 1.0      |
| 4   | 3060361-933 | RING                                      |     |                 | 0100086524    |       |         | 1.0      |
| 4   | 3060361-999 | SEALING WRAP/MBP                          |     |                 |               |       |         | 1.0      |
| 4   | 3073797-2   | VANE, LP COMPRESSOR, STAGE 4              |     |                 | 6004682946    |       |         | 50.0     |
| 4   | 3073797-2   | VANE, LP COMPRESSOR, STAGE 4              |     |                 | 6004670374    |       |         | 34.0     |
| 4   | VZOREK 134  | SAMPLE TH1,6 CRES 17-4PH                  |     |                 | 6004608866    |       |         | 1.0      |
| 3   | 3071873-6   | SHROUD LPC 1STG                           |     |                 | 13-182127-003 |       |         | 1.0      |
| 3   | 3070440-24  | COMPRESSOR ROTOR ASSY 1ST STAGE           | AV  | 13-162045-00698 | P4550010      |       |         | 1.0      |
| 4   | 3060314-3   | SHAFT, SHOULDERED, LP COMPRESSOR - FWD    | G   | 13P61652        | P4387023      |       |         | 1.0      |
| 5   | 3060314-751 | SHAFT SHLDR LP CPRSR                      |     |                 | LN131034080   |       |         | 1.0      |
| 4   | 3060341-1   | BLADE LPC-1                               |     |                 | 13-3500-104   |       |         | 31.0     |

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|-----|-------------|------------------------------------------|-----|-----------------|---------------|------------|---------|----------|
| 4   | 3060345-1   | LPC RETAINER RING                        |     |                 | 13-169015-001 |            |         | 1.0      |
| 4   | 3060372-1   | DISK, AXIAL COMPRESSOR, TURBINE ENGINE - | H   | 13-162045-00698 | 131034248     |            |         | 1.0      |
| 5   | 3060372-111 | DISK, AXIAL COMPRESSOR, TURBINE ENGINE - | G   | 13-162045-00698 |               |            |         | 1.0      |
| 3   | 3075841-3   | COMPRESSOR ROTOR ASSY - STG 2            | -   | 13-162045-00818 | P4770490      |            |         | 1.0      |
| 4   | 3075839-1   | DISK, LPC ROTOR-STG 2                    | E   | 13-162045-00818 | 131034294     |            |         | 1.0      |
| 5   | 3060373-111 | DISK, AXIAL COMPRESSOR, TURBINE ENGINE - | F   | 13-162045-00818 |               |            |         | 1.0      |
| 4   | 3075840-3   | BLADE, COMPRESSOR ROTOR - STAGE 2        |     |                 | 13-3500-096   |            |         | 1.0      |
| 4   | 3075840-3   | BLADE, COMPRESSOR ROTOR - STAGE 2        |     |                 | 13-3500-095   |            |         | 2.0      |
| 4   | 3075840-3   | BLADE, COMPRESSOR ROTOR - STAGE 2        |     |                 | 13-3500-211   |            |         | 5.0      |
| 4   | 3075840-3   | BLADE, COMPRESSOR ROTOR - STAGE 2        |     |                 | 13-3500-213   |            |         | 3.0      |
| 4   | 3075840-3   | BLADE, COMPRESSOR ROTOR - STAGE 2        |     |                 | 13-3500-024   |            |         | 28.0     |
| 4   | 3072403-6   | SHROUD, AXIAL COMPRESSOR                 |     |                 | 141034010     |            |         | 1.0      |
| 1   | 3070800-47  | FUEL CONTROL ASSY MAIN TURBINE ENGINE    | CL  | 20398           | 6             |            |         | 1.0      |
| 2   | 2683881     | VALVE ASSEMBLY METERING (MTCHD)          | C   | 163194-3276     |               | 0001852489 |         | 1.0      |
| 2   | 3070643-4   | VALVE ASSEMBLY BODY & HSG                | U   | 163194-3118     |               |            |         | 1.0      |
| 2   | 3070690-8   | BODY FUEL CONTROL                        | AW  | 162260-CF4178   |               |            |         | 1.0      |
| 2   | 3070973-2   | SET VALVE,MTCHD-RELIEF                   | K   | 10128-13-0094   |               |            |         | 1.0      |
| 2   | 3074127-2   | SERVO PRESS REG VALVE SET,MATCHED        | K   | 150529-1540     |               |            |         | 1.0      |
| 2   | 3074131-2   | VALVE ASSEMBLY ROTARY-MATCHED SET        | R   | 126102500223    |               |            |         | 1.0      |
| 2   | 3074253-3   | FLYWEIGHT ASSY GOVERNOR                  | G   | GOV6004654705   |               |            |         | 1.0      |
| 3   | 3073852-4   | SET FLYWEIGHT,GOV,MTCHD                  | H   | 12107006298     |               |            |         | 1.0      |
| 2   | 2690663     | VALVE ASSEMBLY,METERING                  | A   | MS6004673318    |               |            |         | 1.0      |
| 3   | 3074139-3   | BELLOWS SENSOR,COMPRESSOR DISCHAR        | T   | 1315076800098   |               |            |         | 1.0      |
| 3   | 3074276-3   | FILTER PLATE VALVE                       | D   | 5511            |               |            |         | 1.0      |
| 2   | 2690665     | CONTROL ASSEMBLY,SERVO VALVE             | A   | HS6004715141    |               |            |         | 1.0      |

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| BOM | Material     | Material Description                     | Rev | Serial Number   | Lot Number    | Batch      | Ref.Des | Quantity |
|-----|--------------|------------------------------------------|-----|-----------------|---------------|------------|---------|----------|
| 3   | 2687608      | BELLOWS REGULATING,SERVO VALVE           | A   | 13107010774     |               | 0001990144 |         | 1.0      |
| 3   | 3070521-5    | TORQUE MOTOR                             | AG  | D9002S          |               |            |         | 1.0      |
| 2   | 2691329      | LEVER ASSEMBLY,POWER                     | B   | PLA6004708663   |               |            |         | 1.0      |
| 3   | 3074110-1    | RESISTOR VAR,LIN,PRECISION               | K   | 1316280100206   |               |            |         | 1.0      |
| 3   | 442580-2     | SOLENOID AND VALVE,INSTALLATION OF       |     | SVEBA10571      |               |            |         | 1.0      |
| 3   | 442581-2     | SOLENOID AND VALVE                       |     | SVEBB10567      |               |            |         | 1.0      |
| 2   | 441886-ISS6  | FUEL CONTROL,MAIN,TURBINE ENGINE,INSTL O | 06  | 20398           |               |            |         | 1.0      |
| 1   | 2118882-8001 | TFE-731 DIGITAL ELECTRONIC ENGINE CONTRO | BD  | 93-BP0058       |               |            |         | 1.0      |
| 2   | 2118883-11   | TFE-731 DIGITAL ELECTRONIC ENGINE CONTRO | AL  | 101-22/401      |               |            |         | 1.0      |
| 3   | 4241318-13   | MODULE ASSY                              | AA  | 1351-00977      |               |            |         | 1.0      |
| 4   | 4241319-13   | SUB, FADEC SUBASSEMBLY                   | P   | 1350-00976      |               |            |         | 1.0      |
| 5   | 4242010-905  | SUB, DEEC :                              | M   | 1343-00967      |               |            |         | 1.0      |
| 5   | 4241325-909  | SUB FADEC                                |     | 1346-00954      |               |            |         | 1.0      |
| 3   | 4242225-803  | Q-XDCR ASSY                              | E   | 1343-00946      |               |            |         | 1.0      |
| 4   | 4242710-2    | CIRCUIT CARD ASSEMBLY EMI/FLEX           | N   | 1343-00946      |               |            |         | 1.0      |
| 3   | 4241309-11   | CIRCUIT CARD ASSEMBLY, POWER SUPPLY      | BH  | 1342-00939      |               |            |         | 1.0      |
| 1   | 3062075-904  | GEARBOX ASSEMBLY                         |     | LA049433        |               |            |         | 1.0      |
| 2   | 3070639-1    | SHAFT, SHOULDER                          |     |                 | 11-169019-807 |            |         | 1.0      |
| 2   | 895349-1     | COUPLING RGD SP                          |     |                 | 13-169006-137 |            |         | 1.0      |
| 2   | 3060760-4    | FUEL PUMP                                |     | 3547            |               |            |         | 1.0      |
| 2   | 3060800-1    | SUPPORT ASSEMBLY                         |     |                 | 13-169019-939 |            |         | 1.0      |
| 2   | 3060801-1    | MOUNT ASSY ACCESSORY                     |     |                 | 13-169019-794 |            |         | 1.0      |
| 2   | 3070093-16   | GEARBOX, TRANSFER                        | BP  | 14P64285        |               |            |         | 1.0      |
| 3   | 358753-2     | BEARING, BALL, THRUST                    |     | 12-162255-10441 |               |            |         | 1.0      |
| 3   | 358754-2     | BEARING, ROLLER                          |     | 13-162111-17843 |               |            |         | 1.0      |

Subject to U.S. Export Regulations

07/21/2014

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# Configuration Report

# Honeywell

Material Number 3062075-1 Serial Number P137180  
 Material Desc ENGINE ASSEMBLY TURBOFAN; TFE731-40BR-1B  
 Mfg. Revision A

ESS - Engines - Phoenix  
 Honeywell International Inc  
 111 South 34th Street  
 PHOENIX AZ 85072  
 USA

| BOM | Material  | Material Description               | Rev | Serial Number   | Lot Number  | Batch | Ref.Des | Quantity |
|-----|-----------|------------------------------------|-----|-----------------|-------------|-------|---------|----------|
| 3   | 358755-1  | BEARING, ROLLER                    |     | 13-162111-14548 |             |       |         | 1.0      |
| 3   | 358756-1  | BEARING, ROLLER                    |     | 13-162111-17165 |             |       |         | 1.0      |
| 3   | 3060828-2 | HOUSING GRBX XFR                   |     |                 | 13-5060-305 |       |         | 1.0      |
| 3   | 3070212-3 | GEARSHAFT, BEVEL TRANSFER GEARBOX  |     |                 | P4680517    |       |         | 1.0      |
| 3   | 3070219-2 | GEARSHAFT, BEVEL, TRANSFER GEARBOX |     |                 | P4606761    |       |         | 1.0      |
| 3   | 3070722-8 | TRANSDUCER                         |     | 13910300192     |             |       |         | 1.0      |
| 3   | 358757-1  | BEARING BALL                       |     | 13-151725-03565 |             |       |         | 1.0      |
| 2   | 3070201-2 | SHAFT STRAIGHT                     |     |                 | P4346176    |       |         | 1.0      |
| 2   | 3060785-3 | PUMP - LUBE WITH REGULATOR & SCAV  |     | 1316225400153   |             |       |         | 1.0      |
| 2   | 3060827-4 | GEARBOX ASSEMBLY ACCESSORY DRIVE   |     | S-4238          | 1           |       |         | 1.0      |

Subject to U.S. Export Regulations

07/21/2014

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**SERVICE RECORD**

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

Page 1 of 2

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 09-Mar-2022 | 0       | 0       |           |           | 3206.2    | 3206      |         |            |  |  |

| No. | Task No./Position | Description                                         |
|-----|-------------------|-----------------------------------------------------|
| 1   | 761330A           | NO. 2 ENGINE - DOWNLOAD ENGINE (DEEC) COMPUTER ECTM |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
CURT TAUNTON

Certificate #:

A&P 3788607

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

Page 1 of 1

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 18-Jan-2022 | 0       | 0       |           |           | 3109      | 3115      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
CURT TAUNTON

Certificate #: A&P 3788607

Work Inspected By:

\_\_\_\_\_

Certificate #:

\_\_\_\_\_

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

Page 1 of 1

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 05-Feb-2022 | 0       | 0       |           |           | 3150.3    | 3157      |         |            |  |  |

| No. | Task No./Position | Description                                         |
|-----|-------------------|-----------------------------------------------------|
| 1   | 761330A           | NO. 2 ENGINE - DOWNLOAD ENGINE (DEEC) COMPUTER ECTM |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
CURT TAUNTON

Certificate #: A&P 3788607

Work Inspected By:

\_\_\_\_\_

Certificate #:

\_\_\_\_\_

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

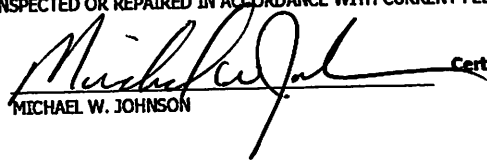
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 21-Sep-2021 | 0       | 0       |           |           | 2949.9    | 2940      |         |            |  |  |

| No. | Task No./Position                      | Description                                                                                                                                                                                                                                     |
|-----|----------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 1   | TFE731-72-5256 R01 (D201506000058 R01) | NO. 2 ENGINE - BYPASS SECTION - IDENTIFY AND REPLACE SPECIFIC FAN DISKS, PN 3060287-2, WITH SUSPECT SHOT PEENED ROLLED EDGES ON TFE731-20 AND -40 ENGINE MODELS, SERVICE BULLETIN NOT APPLICABLE - NA BY SERIAL NUMBER<br>Method Of Compliance: |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

MECHANIC AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

Page 1 of 2

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

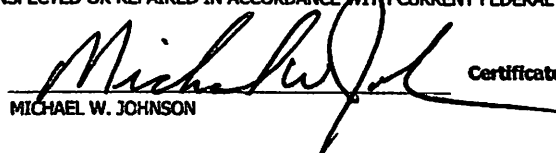
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 30-Sep-2021 | 0       | 0       |           |           | 2963.3    | 2953      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

MECHANIC AP3481311

Work Inspected By:

Certificate #:

## SERVICE RECORD ENTRY

StandardAero

Bush Field Airport  
1550 Hangar Road  
Augusta GA 30906-9684

FAA Repair Station No: GW4R221M  
Telephone (706)771-5600  
Fax (706)793-0469

|                                |                           |                     |
|--------------------------------|---------------------------|---------------------|
| CUSTOMER NAME: GEORGIA CROWN   | Work Order Number: 362786 | Date: 19/APR/2021   |
| ENGINE serial Number: P-137180 | ENGINE Time: 2745.7       | ENGINE CYCLES: 2730 |

Performed logbook research and found that at this time, this engine qualifies for the first use of a 3500/7000 hour MPI/CZI interval in accordance with Honeywell maintenance manual 72-03-06 Rev. 13, TR 5-18. For subsequent use of the 3500 Hr MPI/7000 Hr CZI interval, the service bulletins in the final paragraph of temporary revision 5-18 will have to be complied with at the first event.

Date: 19/APR/2021 Signed: Jay Shealy AG 27 For  
StandardAero, CRS# GW4R221M, Augusta, Ga 30906

## SERVICE RECORD ENTRY

|                    |                      |                     |                      |                    |  |
|--------------------|----------------------|---------------------|----------------------|--------------------|--|
| Georgia Crown      |                      | Work Order: 363356  |                      | Date: 04/June/2021 |  |
| Serial Number: 496 | Tail Number: N877W   | Model: Learjet 45   | ACTT: 2,848.4        | LNDS: 2,843        |  |
| Engine Position: 2 | Engine S/N: P-137180 | Engine TSN: 2,799.0 | Engine Cycles: 2,799 |                    |  |

- Complied with **150-250** Hour Inspection in accordance with Honeywell TFE731-60 LMM 72-03-06.
  - Analyze engine oil and filter.
  - Inspect chip detector for contamination.
- Complied with **500-650** Hour Inspection in accordance with Honeywell TFE731-60 LMM 72-03-06.
  - Chip detector.
  - Replace fuel filter element.
  - Inspect motive flow fuel pump drive spline.
- Complied with **500-850** Hour Inspection in accordance with Honeywell TFE731-60 LMM 72-03-06.
  - Inspect electrical wiring and connections.
  - Inspect plumbing lines and connections.
  - Inspect inlet/fan area.
  - Inspect tail-pipe and LPT3 rotor assembly.
- Complied with **1500-1650** Hour Inspection in accordance with Honeywell TFE731-60 LMM 72-03-06.
  - Carbon seal health check.
  - Inspect ignitor plug and leads.
- Accomplished engine run, performed vibration survey with all parameters within allowable limits, leak and ops check are satisfactory. All work done in accordance Honeywell TFE731-40 Light Maintenance Manual 72-06-12 Rev 13.
- Installed **new** starter cannon plug back-shell p/n 450DA001N08G IAW L75 AWD 20-20-01-201.

All maintenance performed on this aircraft/engine was performed in accordance with current regulations of the Federal Aviation Administration and is approved for return to service with respect to the work performed. Details are on file under W/O# **363356**.

Date: 04/June/2021 Signed: Jonah Richie AG 79 for

StandardAero CRS No., GW4R221M Augusta, GA 30906



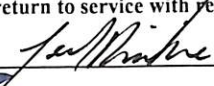
# SERVICE RECORD ENTRY

|                    |                      |                     |                      |                       |  |
|--------------------|----------------------|---------------------|----------------------|-----------------------|--|
| Georgia Crown      |                      | Work Order: 360266  |                      | Date: 27/October/2020 |  |
| Serial Number: 496 | Tail Number: N877W   | Model: Learjet 45   | ACIT: 2,598.3        | LNDS: 2,845           |  |
| Engine Position: 2 | Engine S/N: P-137180 | Engine TSN: 2,548.9 | Engine Cycles: 2,510 |                       |  |

- Accomplished engine run, performed vibration survey with all parameters within allowable limits, leak and ops check are satisfactory. All work done in accordance Honeywell TFE731-40 Light Maintenance Manual 72-06-12 Rev 13.

All maintenance performed on this aircraft/engine was performed in accordance with current regulations of the Federal Aviation Administration and is approved for return to service with respect to the work performed.

Details are on file under W/O# 360266.

Date: 27/October/2020 Signed: Jonah Richie  for

StandardAero CRS No., GW4R221M Augusta, GA 30906

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P-137180

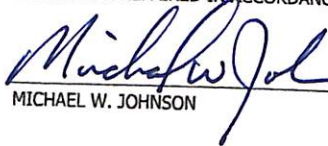
On Aircraft Reg: N877W Serial: 45-496

| Work Order #: |         |        |           |           |           |           |         |            |  |
|---------------|---------|--------|-----------|-----------|-----------|-----------|---------|------------|--|
| Date          | A/C HRS | A/CAFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |
| 09-Mar-2021   | 0       | 0      |           |           | 2668.6    | 2654      |         |            |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

MECHANIC AP3481311

Work Inspected By:

Certificate #:



GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

Page 1 of 2

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 31-Aug-2020 | 0       | 0       |           |           | 2494.1    | 2449      |         |            |  |  |

Work Order #:

No. Task No./Position

Description

NO. 2 ENGINE - DOWNLOAD ENGINE (DEEC) COMPUTER ECTM

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 14-Sep-2020 | 0       | 0       |           |           | 2506.4    | 2470      |         |            |  |  |

Work Order #:

No. Task No./Position

Description

NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

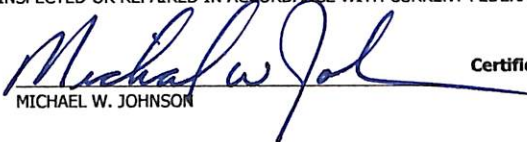
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 22-Feb-2020 | 0       | 0       |           |           | 2313.2    | 2242      |         |            |  |  |

| No. | Task No./Position | Description                                                                                                                                                                                                                                                   |
|-----|-------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 1   | 793310            | NO. 2 ENGINE - OIL FILTER BYPASS VALVE<br>Removed 3060782-2 S/N# K18005 From Aircraft Reg: N877W Serial: 45-496 Reason Removed B-FAILURE<br>Installed 3060782-2 S/N# 213859 On Aircraft Reg: N877W Serial: 45-496 Installed Part Times TSN: 0 MOS 0 HRS 0 ENC |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 05-Mar-2020 | 0       | 0       |           |           | 2335.9    | 2259      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 23-Aug-2019 | 2185.7  | 2101    |           |           | 2136.3    | 2057      |         |            |  |  |

No. Task No./Position

Description

1 057110

NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

### SERVICE RECORD ENTRY

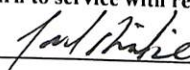
|                    |                    |                      |  |                     |                        |                      |
|--------------------|--------------------|----------------------|--|---------------------|------------------------|----------------------|
| Georgia Crown      |                    | Work Order: 353460   |  |                     | Date: 13/February/2020 |                      |
| Serial Number: 496 | Tail Number: N877W | Model: Learjet 75    |  | ACTT: 2,342.1       |                        | LNDS: 2,277          |
| Engine Position: 2 |                    | Engine S/N: P-137180 |  | Engine TSN: 2,292.7 |                        | Engine Cycles: 2,233 |

- Complied with 500 to 650 Hour Inspection in accordance with Honeywell TFE731-40 Light Maintenance Manual 72-00-00.
  - 730120 - Replace fuel filter element.
  - 732120 - Inspect motive flow fuel pump drive spline.
  - 790150 - Inspect chip detector for contamination and perform resistance check.
- Complied with 1,000 to 1,300 Hour Inspection in accordance with Honeywell TFE731-40 Light Maintenance Manual 72-00-00.
  - 727010 - Inspect bypass stator vane assembly.
  - 790140 - Perform oil change.
  - 793315 - Check oil filter bypass valve circuit/thermal lockout switch.
- Accomplished required operational and leak checks in accordance with Honeywell TFE731-40 Light Maintenance Manual.

All maintenance performed on this aircraft/engine was performed in accordance with current regulations of the Federal Aviation Administration and is approved for return to service with respect to the work performed. Details are on file under W/O# 353460.

Date: 13/February/2020

Signed: Jonah Richie



AG  
79

for

StandardAero CRS No., GW4R221M Augusta, GA 30906



GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 03-Dec-2018 | 1850.3  | 1764    |           |           | 1800.9    | 1720      |         |            |  |  |

| No. | Task No./Position | Description                                                                                                                                                                                                                                                     |
|-----|-------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 1   | 793310            | NO. 2 ENGINE - OIL FILTER BYPASS VALVE<br>Removed PN3060782-2 S/N# F15932 From Aircraft Reg: N877W Serial: 45-496 Reason Removed B-FAILURE<br>Installed 3060782-2 S/N# K18005 On Aircraft Reg: N877W Serial: 45-496 Installed Part Times TSN: 0 MOS 0 HRS 0 ENC |

I CERTIFY THAT THIS ENGINE HAS BEEN REPAIRED IN ACCORDANCE WITH HONEYWELL TFE731-40L MAINTENANCE MANUAL AND IS APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

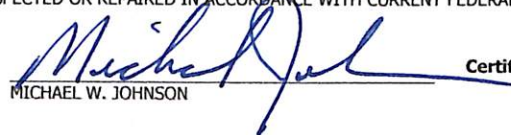
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 01-Apr-2019 | 0       | 0       |           |           | 1940.9    | 1850      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

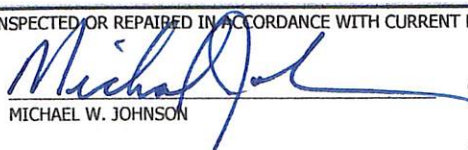
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 03-May-2019 | 0       | 0       |           |           | 1991.8    | 1900      |         |            |  |  |

| No. | Task No./Position | Description                                                                         |
|-----|-------------------|-------------------------------------------------------------------------------------|
| 1   | 057140            | NO. 2 ENGINE - 500 TO 850 HOUR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

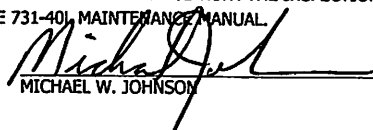
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 03-Dec-2018 | 1850.3  | 1764    |           |           | 1800.9    | 1720      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |
| 2   | 057130            | NO. 2 ENGINE - 500 TO 650 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS ENGINE HAS BEEN INSPECTED IN ACCORDANCE WITH THE INSPECTION AND PROCEDURES FOR 150/250-500/650 HOUR ENGINE  
INSPECTION REFERENCING HONEYWELL TFE 731-40J MAINTENANCE MANUAL.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

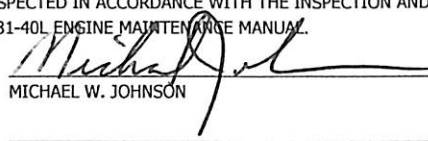
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 25-Jul-2018 | 1684.8  | 1599    |           |           | 1635.4    | 1555      |         |            |  |  |

| No. | Task No./Position | Description                                                                       |
|-----|-------------------|-----------------------------------------------------------------------------------|
| 1   | 057110            | NO. 2 ENGINE - 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS |

I CERTIFY THAT THIS ENGINE HAS BEEN INSPECTED IN ACCORDANCE WITH THE INSPECTION AND PROCEDURES FOR 150/250 HOUR INSPECTION IN ACCORDANCE WITH THE HONEYWELL TFE 731-40L ENGINE MAINTENANCE MANUAL.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 04-May-2018 | 1560.6  | 1478    |           |           | 1511.2    | 1434      |         |            |  |  |

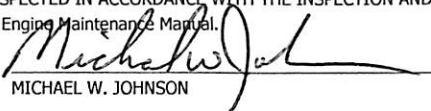
No. Task No./Position Description

1 057110

NO. 2 ENGINE - PERFORM 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS  
INSTRUCTIONS

I CERTIFY THAT THIS ENGINE HAS BEEN INSPECTED IN ACCORDANCE WITH THE INSPECTION AND PROCEDURES FOR 150/250 HOUR INSPECTION IN  
ACCORDANCE WITH THE TFE 731-40L Light Engine Maintenance Manual.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

GEORGIA CROWN DISTRIBUTING CO.  
5290 ARMOUR ROAD  
COLUMBUS, GA, USA, 31909

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

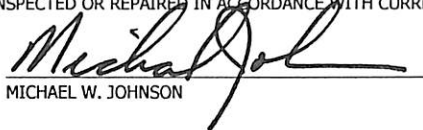
| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 30-Nov-2017 | 1402.6  | 1341    |           |           | 1353.2    | 1297      |         |            |  |  |

No. Task No./Position Description

1 057110 NO. 2 ENGINE - PERFORM 150 TO 250 HR ROUTINE PERIODIC INSP PER MANUFACTURERS INSTRUCTIONS

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F) (3) AND APPROVED FOR RETURN TO SERVICE

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #: AP3481311

Work Inspected By:

Certificate #:



GEORGIA CROWN DISTRIBUTING CO.  
5290 ARMOUR ROAD  
COLUMBUS, GA, USA, 31909

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #: \_\_\_\_\_

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 04-Aug-2017 | 1274.4  | 1219    |           |           | 1225      | 1175      |         |            |  |  |

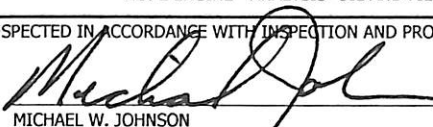
No. Task No./Position Description

1 72-AS-720010

NO. 2 ENGINE - ANALYSIS OIL AND FILTER (BASELINE/SPECIAL REQUIREMENT)

I CERTIFY THAT THIS ENGINE HAS BEEN INSPECTED IN ACCORDANCE WITH INSPECTION AND PROCEDURES FOR \_\_\_\_\_ INSPECTION UNDER \_\_\_\_\_

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

ENGINE ☐  
MODULE ☐  
SERIAL NUMBER

## SERVICE RECORD

PAGE

## DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER.**

## SERVICE RECORD ENTRY

[illegible]

|                                                                    |                           |                                                  |
|--------------------------------------------------------------------|---------------------------|--------------------------------------------------|
| Georgia Crown Distributing Co.<br>Learjet 45 (75), S/N: 496, N877W | Work Order Number: 336874 | Date: July 26, 2017<br>ACTT: 1,242.7 LDGS: 1,201 |
| Engine Serial Number: P-137180<br>Position: R/H                    | Engine Time: 1,193.3      | Engine Cycles: 1,157                             |

1. Installed Engine in **R/H** Position of Learjet 45 (75), N877W, S/N: 496 with total time of 1,242.7 Hours and 1,201 Landings. Accomplished all necessary ground checks and adjustments in accordance with the Honeywell TFE 731-20/-40/-50/-60 Light Maintenance Manual and Learjet 45 Maintenance Manual. Serviced Engine with Mobil 254 oil.

**Note: Oil and Filter analysis due after initial 25 Hours of Engine Operation.**

All maintenance performed on this aircraft / engine was performed in accordance with current regulations of the Federal Aviation Administration and is approved for return to service with respect to the work performed. Details are on file under W/O# 336874.

**Date:** July 26, 2017

Signed: Stephen E. Cline

Steph E Cline

for

StandardAero CRS No., GW4R221M Augusta, GA 30906

**TOTALS TO DATE**

## MEDRA 6.0.83 - Data Reduction - Performance Margins

## TEST IDENTIFICATION

FE731-40BR-1B S/N: 137180 Work order number: 336035  
 Test date: 07/10/2017 Test time: 16:03:05  
 RED Version: 6.40 PGM Version: 6.77  
 Reduced date: 07/10/2017 Reduced time: 16:24:15  
 Reason for test: Unscheduled Core Entry

Time since new: 1193.3 Cycles since new: 1157  
 Build/Run: 1/2 EEC/DEEC SG:

Customer: GEORGIA CROWN Aircraft S/N: Unk  
 Aircraft model: Lear 75 Engine location: 1  
 Test Cell Code: GASAGS

Data recorded by: 004117 Data entered by: AGS  
 Comments:

## Test Cell Acquired Data

## Takeoff Rating Margins

|              |          | at<br>T/O Thrust | at<br>T/O N1 |
|--------------|----------|------------------|--------------|
| TAMB -       | °F       | 103.0            | 103.0        |
| Thrust -     | LB force | 3850             |              |
| N1 -         | RPM      |                  | 19308        |
| Fn Margin -  | LB force | 0 @              | 21 @ 3878    |
| T45 Margin - | °F       | 100 @ 1779       | 94 @ 1779    |
| N1 Margin -  | RPM      | 243 @ 19458      | 150 @ 19458  |
| N2 Margin -  | RPM      | 0 @              | 1039 @ 31485 |

## APR Rating Margins

|              |          | at<br>APR T/O Fn | at<br>APR T/O N1 |
|--------------|----------|------------------|------------------|
| TAMB -       | °F       | 112.0            | 112.0            |
| Thrust -     | LB force | 3850             |                  |
| N1 -         | RPM      |                  | 19461            |
| Fn Margin -  | LB force | 0 @ 3850         | 21 @ 3878        |
| T45 Margin - | °F       | 107 @ 1819       | 100 @ 1819       |
| N1 Margin -  | RPM      | 245 @ 19611      | 150 @ 19611      |
| N2 Margin -  | RPM      | 0 @              | 1121 @ 31800     |

## Atmospheric Conditions

Average Ambient T1 = 85.7 °F  
 Average Ambient Pressure = 14.67 psia  
 Average Relative Humidity = 70.29 %  
 N1 Compensation Value = 4.0 ( 150 RPM)

**CERT. SHEETS**

**P-137180**

**W/O # 336035**

**July 3, 2017**



**StandardAero**  
**CRS# GW4R221M**

ENG/  
MOD  
SERI



Bush Field Airport  
1550 Hangar Road  
Augusta, GA 30906-9684

FAA Repair Station # GW4R221M  
Telephone: (706) 771-5600  
Fax: (706) 793-0469

NUMBER.

DATE

|                                |                                  |                            |
|--------------------------------|----------------------------------|----------------------------|
| Customer: <b>GEORGIA CROWN</b> | Work Order Number: <b>336035</b> | Date: <b>July 3, 2017</b>  |
| Engine S/N: <b>P-137180</b>    | Engine Time: <b>1193.3</b>       | Engine Cycles: <b>1157</b> |

- Disassembled Engine as Necessary for Repair for Bad SOAP. Installed the Following Parts:

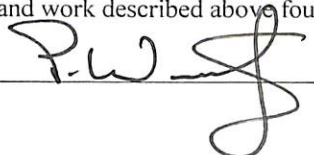
| Part Nomenclature              | Part Number | Part Serial Number | Quantity | Condition |
|--------------------------------|-------------|--------------------|----------|-----------|
| FAN STATOR VANE ASSEMBLY       | 70060130-1  | 17P92287           | 1        | NEW       |
| PLANETARY CARRIER              | 3060121-8   |                    | 1        | NEW       |
| CARBON SEAL #3                 | 3060150-3   | 15-169020-02100    | 1        | NEW       |
| CARBON SEAL #1                 | 3060457-1   | 17-152826-47223    | 1        | NEW       |
| CARBON SEAL #4                 | 3060437-1   | 17-152826-49175    | 1        | NEW       |
| #5 BEARING                     | 3075016-1   | 17-151725-00436    | 1        | NEW       |
| CARBON SEAL #5                 | 3060436-1   | 17-152826-48948    | 1        | NEW       |
| HPT SEAL RING                  | 3073609-5   |                    | 1        | NEW       |
| IGNITER PLUG                   | 3074541-4   |                    | 2        | NEW       |
| IGNITER LEAD                   | 3060722-12  | S16320065          | 1        | NEW       |
| A/G/B BEARING - ALTERNATOR FWD | 3072883-1   | 14-162111-06754    | 1        | NEW       |
| OIL TEMP CONTROL VALVE         | 158465-6    | 14317              | 1        | NEW       |
| ALTERNATOR CARBON SEAL         | 3075818-1   | 17-152826-44949    | 1        | NEW       |

- HPT nozzle AE: 12.442
- Accomplished **150-250, 300-440, 500-650, 500-850, 1000-1300, 1500-1650 HR** Periodic Routine Inspections.
- See Appropriate Section of Logbook for Service Bulletin addressed this event.
- Reassembled Engine Using New Consumables. Accomplished Engine Certification Test Cell Run, Vibration Survey, Leak Check, and Ops Check. All work Accomplished I/A/W TFE731-40 LMM 72-03-06 Rev. 12 and HMM 72-03-07 Rev. 11.

**NOTE: S.O.A.P. SAMPLE DUE IN 25 HOURS.**

This aircraft or component has been inspected and work described above found airworthy for return to service.

DATE: 03/JUL/2017, SIGNED: Peter Weitz  
Standard Aero; Augusta, GA  
Repair Station GW4R221M

  
AG 14

TIONS



**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER.**

**TOTALS TO DATE**



N877W  
RB-107

***Georgia Crown Aviation Department***

5290 Amour Road  
Columbus Georgia 31909

TFE731-40BR-1B

Engine 2 Part Number: 3062075-1

Serial Number: 137180

TTSN: 497.80

TCSN: 489

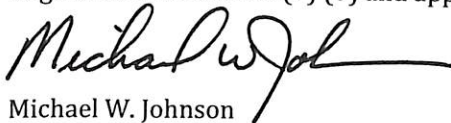
Date: December 14, 2015

Completed 150/250 Hour Engine Routine Periodic Inspection per Honeywell

TFE731-40 LMM.

- Completed Oil and Filter (SOAP) Inspection
- Inspected Chip Detector for Contamination

I certify this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and approved for return to service.



Michael W. Johnson

Georgia Crown Aviation Department

Aviation Maintenance Technician

Airframe and Powerplant License Number: **3481311**



**Georgia Crown Aviation Department**

Lear 75  
N877W  
45-496

201 Airport Parkway  
LaGrange Georgia 30240

Part Number 3062075-1  
Model TFE731-40BR-1B  
Engine TTSN: 614.0

Serial number P137180  
Engine TCSN: 600

Date: March 5, 2016

Completed the following tasks:

1. **057110**      Number 2 Engine 150-250 Routine Periodic Inspection
  - **790130**      Oil Analysis and Filter
  - **790160**      Inspect Chip Detector for Contamination
  
2. **057130**      Number 2 Engine 500-650 Routine Periodic Inspection
  - **730120**      Replace Fuel Filter
  - **732120**      Inspect Motive Flow Fuel Pump Drive Spline
  - **790150**      Inspect Chip Detector for Contamination

I certify that this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3). All test operational tests were performed in accordance with the manufacturer's engine maintenance manual or component maintenance manual. This aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: **3481311**





**Georgia Crown Aviation Department**

Lear 75  
N877W  
45-496

201 Airport Parkway  
LaGrange, Georgia 31909

Aircraft TTSN: 801.5      Aircraft TCSN: 803  
Engine 2 Part Number: 3062075-1  
Engine 2 Model: TFE731-40BR-1B  
Engine 2 Serial Number: P137180  
Engine TTSN: 801.5  
Engine TCSN: 803  
Date: July 7, 2016

Completed the following tasks:

1. 057110      150-250 Hour Routine Periodic Inspection
2. 057140      500-850 Hour Routine Periodic Inspection
3. 710130      Inspect Tail Pipe/LPT3 Rotor
4. 710140      Inspect Inlet/ Fan for FOD Damage
5. 714010      Inspect Electrical Wires and Connections
6. 761330A      DEEC Download

I certify that this engine has been inspected in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and current aircraft and engine manufacturer's maintenance manuals. All functional and operational checks were satisfactory and this aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: **3481311**



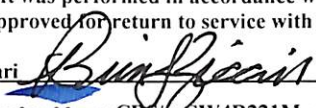
Bush Field Airport  
1550 Hangar Road  
Augusta, GA 30906-9684

FAA Repair Station # GW4R221M  
Telephone: (706) 771-5600  
Fax: (706) 793-0469

|                                |                      |                        |                   |                     |                    |
|--------------------------------|----------------------|------------------------|-------------------|---------------------|--------------------|
| Georgia Crown Distributing Co. |                      | Work Order: 327791     |                   | Date: July 16, 2016 |                    |
| Serial Number: 45-496          | Tail Number: N877W   | Model: Learjet 45 / 75 | ACTT: 809.9       |                     | LNDS: 812          |
| Engine Position: 2             | Engine S/N: P-137180 |                        | Engine TSN: 809.9 |                     | Engine Cycles: 812 |

- Dynamically Balanced Fan Assembly to .07 IPS. All work done IAW TFE 731-20/- 40/-50/-60 LMM 72-03-06 Rev. 12
- Painted Fan Hub and Fan Blades in accordance with Honeywell TFE 731-20/- 40/-50/-60 Light Maintenance Manual 72-03-06 Rev. 12.
- Accomplished Vibration Survey in accordance with Honeywell TFE 731-20/- 40/-50/-60 Light Maintenance Manual 72-03-06 Rev. 12. All Parameters found to be within limits. Results on file.

All maintenance performed on this aircraft was performed in accordance with current regulations of the Federal Aviation Administration and is approved for return to service with respect to the work performed. Details are on file under W/O# 327791.

Date: July 16, 2016, Signed: Brian J. Zicari   for

StandardAero CRS# GW4R221M, Augusta, GA.30906



Lear 75  
N877W  
45-496

**Georgia Crown Aviation Department**

201 Airport Parkway  
LaGrange Georgia 30240

Part Number 3062075-1

Model TFE731-40BR-1B

Serial number P137180

Engine TTSN: 907.7

Engine TCSN: 906

Date: October 12, 2016

**0157110**

Engine  $\frac{1}{2}$  150/250 Routine Inspection

**790130**

Oil and Filter SOAP

**790160**

Inspect Chip Detector

I certify that this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3). All test operational tests were performed in accordance with the manufacturer's engine maintenance manual or component maintenance manual. This aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: **3481311**



**Georgia Crown Aviation Department**

Lear 75  
N877W  
45-496

201 Airport Parkway  
LaGrange Georgia 30240

Part Number 3062075-1  
Model TFE731-40BR-1B  
Engine TTSN: 1074.3

Serial number P137180  
Engine TCSN: 1046

Date: 2.21.2017

2. 057110                      NO. 2 ENGINE - PERFORM 150 TO 250 HR ROUTINE PERIODIC INSP  
PER MANUFACTURERS INSTRUCTIONS

I certify that this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3). All test operational tests were performed in accordance with the manufacturer's engine maintenance manual or component maintenance manual. This aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: 3481311



Lear 75  
N877W  
45-496

**Georgia Crown Aviation Department**

201 Airport Parkway  
LaGrange, Georgia 31909

Aircraft TTSN: 1186.9      Aircraft TCSN: 1148  
Engine 2 Part Number: 3062075-1  
Engine 2 Model: TFE731-40BR-1B  
Engine 2 Serial Number: P137180  
Engine TTSN: 1186.9  
Engine TCSN: 1148  
Date: June 1, 2017

**Completed the following tasks:**

1. 731001      NO. 2 ENGINE - MOTIVE FLOW FUEL PUMP ASSEMBLY  
Removed 70060003-1 S/N# 13-30003-0097  
Installed 70060003-1 S/N# 12-30003-00010
  
2.              Engine 2 Alternator Air/Oil Seal  
Removed part number 3075815-1  
sn 12-152826-19908  
Installed new part number 3075815-1  
Sn 17-152826-46017

I certify that this engine has been inspected in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and current aircraft and engine manufacturer's maintenance manuals. All functional and operational checks were satisfactory and this aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: 3481311

**SERIAL NUMBER**

## SERVICE RECORD

### DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER**

PAGE \_\_\_\_\_

[illegible]

N877W  
RB-107

**Georgia Crown Aviation Department**

5290 Amour Road  
Columbus Georgia 31909

TFE731-40BR-1B

Engine 2 Part Number: 3062075-1

Serial Number: 137180

TTSN: 293.9

TCSN: 300

Date: July 20, 2015

Completed 150/250 Hour Engine Routine Periodic Inspection per Honeywell

TFE731-40 LMM.

- Completed Oil and Filter (SOAP) Inspection
- Inspected Chip Detector for Contamination

I certify this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and approved for return to service.

Michael W. Johnson

Georgia Crown Aviation Department

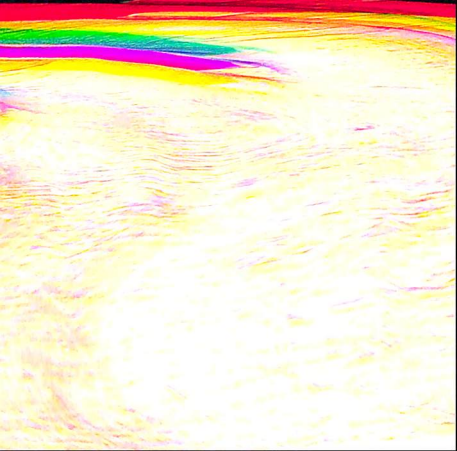
## Aviation Maintenance Technician

Airframe and Powerplant License Number: 3481311

3MED

**D**





ENGINE ☒ A  
MODULE ☐

SERIAL NUMBER

## SERVICE RECORD

DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER

PAGE \_\_\_\_\_

### ACCUMULATIVE TOTALS

DATE

HOURS  
THIS  
FLIGHT

CYCLES  
THIS  
FLIGHT



N877W  
RB-107

*Georgia Crown Aviation Department*

5290 Amour Road  
Columbus Georgia 31909

TFE731-40BR-1B

Engine 2 Part Number: 3062075-1

Serial Number: 137180

TTSN: 162.5

TCSN: 158

Date: March 26, 2015

Completed 150/250 Hour Engine Routine Periodic Inspection per Honeywell

TFE731-40 LMM.

- Completed Oil and Filter (SOAP) Inspection
- Inspected Chip Detector for Contamination

I certify this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and approved for return to service.

  
Michael W. Johnson

Georgia Crown Aviation Department

Aviation Maintenance Technician

Airframe and Powerplant License Number: **3481311**



**SERIAL NUMBER**

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER**

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ENGINE ☒  
MODULE ☐  
SERIAL NUMBER

## SERVICE RECORD

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DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS  
ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER

DATE HOURS  
THIS  
FLIGHT



N877W  
RB-107

Georgia Crown Aviation Department  
5290 Amour Road  
Columbus Georgia 31909

TFE731-40BR-1B

Engine 2 Part Number: 3062075-1

Serial Number: 137180

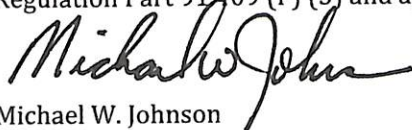
TTSN: 40.8

TCSN: 47

Date: December 9, 2014

1. Performed recommended 25 Hour SOAP sample per Honeywell LMM. No leaks noted on engine run. Service engine with Mobil 254.

I certify this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and approved for return to service.

  
Michael W. Johnson

Georgia Crown Aviation Department

Aviation Maintenance Technician

Airframe and Powerplant License Number: 3481311

TOTALS TO DATE

ENGINE ☒  
MODULE ☐  
SERIAL NUMBER

# SERVICE RECORD

PAGE \_\_\_\_\_

### DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER**

[illegible]



the evolution of mobility

FAA/CRS # LK2R981K

## Service Order

Page of

THIS FORM IS USED TO DOCUMENT MAINTENANCE PERFORMED IN ACCORDANCE WITH PROCEDURE 3.5 – CRT/AOG

|                                        |                |                                                                                |                |                                                                                 |                |
|----------------------------------------|----------------|--------------------------------------------------------------------------------|----------------|---------------------------------------------------------------------------------|----------------|
| 1. Work Order Number:<br><i>115733</i> |                | 2. Registration:<br><i>N877W</i>                                               |                | 3. Aircraft Model and S/N:<br><i>45-496</i>                                     |                |
| 4a. Aircraft                           |                | 4b. Engine 1                                                                   |                | 4c. Engine 2                                                                    |                |
| 4d. APU                                |                |                                                                                |                |                                                                                 |                |
| Hrs: <i>29.7</i>                       | Cyc: <i>37</i> | Hrs: <i>29.7</i>                                                               | Cyc: <i>37</i> | Hrs: <i>29.7</i>                                                                | Cyc: <i>37</i> |
|                                        |                | 5. Preliminary inspection performed (stamp or signature)<br><i>[Signature]</i> |                | 6. INDICATE RII OR OTHER SPECIAL 135/121 OPERATOR REQUIREMENT<br><i>12-3-11</i> |                |

|                                                                                         |          |                |            |          |         |                    |             |
|-----------------------------------------------------------------------------------------|----------|----------------|------------|----------|---------|--------------------|-------------|
| 7. SvO #                                                                                | 11351027 | 8. WRITTEN BY: | L. Larkins | 9. DATE: | 12-3-14 | 10. AUTHORIZED BY: | [Signature] |
| 11. DISCREPANCY/TASK: 0003- L/H Engine Roll Back and went to M.H, with Amber L/H Fault. |          |                |            |          |         |                    |             |

12. CORRECTIVE ACTION: Found Fault Code 112 on L/H DEEC. Swapped DEEC's L/H to R/H and R/H to L/H All work Done DPA/WHITNEYWELL-40 L.H.H. Rev. 12 Chapter 72-00

|                    |                                                                                           |                                                         |                                            |                                                                    |  |
|--------------------|-------------------------------------------------------------------------------------------|---------------------------------------------------------|--------------------------------------------|--------------------------------------------------------------------|--|
|                    |                                                                                           |                                                         |                                            | 6. RII: YES <input type="checkbox"/> / NO <input type="checkbox"/> |  |
| 13. ESTIMATE HOURS | 14. WARRANTY CONSIDERATION:<br>YES <input type="checkbox"/> / NO <input type="checkbox"/> | 15. PERFORMED BY & DATE:<br>49085A<br>CT-JOG<br>12-3-14 | 16. CHECKED BY & DATE:<br>47378<br>12-3-14 |                                                                    |  |

|                         |          |             |  |       |  |                |  |
|-------------------------|----------|-------------|--|-------|--|----------------|--|
| SvO #                   | 11351028 | WRITTEN BY: |  | DATE: |  | AUTHORIZED BY: |  |
| DISCREPANCY/TASK: 0004- |          |             |  |       |  |                |  |

**CORRECTIVE ACTION:**

|                 |  |  |                                                                                      |  |                      |  |                    |  |                                                                 |  |
|-----------------|--|--|--------------------------------------------------------------------------------------|--|----------------------|--|--------------------|--|-----------------------------------------------------------------|--|
| ESTIMATE HOURS: |  |  | WARRANTY CONSIDERATION<br>YES <input type="checkbox"/> / NO <input type="checkbox"/> |  | PERFORMED BY & DATE: |  | CHECKED BY & DATE: |  | RII: YES <input type="checkbox"/> / NO <input type="checkbox"/> |  |
|-----------------|--|--|--------------------------------------------------------------------------------------|--|----------------------|--|--------------------|--|-----------------------------------------------------------------|--|

| SERIALIZED or ROTABLE COMPONENT INFORMATION |           |                 |              |                |             |                         |
|---------------------------------------------|-----------|-----------------|--------------|----------------|-------------|-------------------------|
| OPERATION #                                 | NOMENCLAT | PART NUMBER OFF | SERIAL # OFF | PART NUMBER ON | SERIAL # ON | TSN/TSO (IF APPLICABLE) |
| 17                                          | 18        | 19              | 20           | 21             | 22          | 23                      |
| 0003                                        | LIN DEEC  | 211882-8001     | 98-BP0058    | 211882-8001    | 103-BP0061  |                         |
| 0003                                        | RIN DEEC  | 211882-8001     | 103-BP0061   | 211882-8001    | 98-BP0058   |                         |
|                                             |           |                 |              |                |             |                         |
|                                             |           |                 |              |                |             |                         |

## FINAL INSPECTION AND AIRWORTHINESS RELEASE

The aircraft identified above was repaired and inspected in accordance with current Federal Aviation Regulations and found airworthy with respect to the work performed/completed. (Airworthiness Release may exclude discrepancies not worked/completed)

|                                               |                                  |
|-----------------------------------------------|----------------------------------|
| 24. INSPECTOR'S SIGNATURE<br><i>Carl Case</i> | 25. DATE OF RELEASE<br>12/3/2014 |
|-----------------------------------------------|----------------------------------|

ENGINE ☒  
MODULE ☐  
SERIAL NUMBER

## SERVICE RECORD

PAGE \_\_\_\_\_

### DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER**

[illegible]

Engine Serial : 137180

VERNON DREHER  
LEAR JET  
8220 W HARRY  
WICHITA,KS  
United States-67209

**S.O.A.P.**  
**SPECTROMETRIC OIL ANALYSIS PROGRAM**  
For Honeywell  
REPORT DATE : 12/04/2014  
CONTACT PHONE : 316-946-3094  
CONTACT EMAIL : vernon.dreher@aero.bombardier.com  
FAX NUMBER : 316-946-2246

Honeywell Engines & Systems  
111 S. ... Street  
Phoenix, AZ  
United States-85034  
**Honeywell**  
ENGINE MODEL : TFE731-40BR-1B  
AIRCRAFT TYPE : 45  
AIRCRAFT SERIAL : 496  
OIL TYPE : MOBIL JET OIL 254  
SERVICE CONTRACT# :

**CURRENT SEND IN ANOTHER SAMPLE IN 25 HOURS (+/- 20%) AND CONTINUE FLIGHTS; RESAMPLE RESULTS MAY DICTATE MORE ACTION.**

Reviewed by: Clark Rodgers, Honeywell Engr

LAB DATE 12/04/2014  
SAMPLE DATE 11/22/2014  
RESP EMP Rick Fenton  
TSN 6  
TSO 0  
CYCLES 0  
OIL HOURS 6  
FILTER HOURS 6  
FILTER WEIGHT 19 MGS

| ***FILTER RESULTS*** ANALYSIS PER WORK INSTRUCTIONS MLTC WI 1440. |    |    |    |    |    |    |     |      |      |      |      |      |      |
|-------------------------------------------------------------------|----|----|----|----|----|----|-----|------|------|------|------|------|------|
|                                                                   | ST | ST | CB | ST | AL | ST | M50 | COPR | SILV | MAGN | ALUM | GRIT | MISC |
| AMT                                                               |    |    | TR |    |    |    |     |      | TR   |      | TR   |      | MA   |
| TYPE                                                              |    |    |    |    |    |    |     |      |      |      |      |      |      |
| FORM                                                              |    |    |    |    |    |    |     |      |      |      |      |      |      |

LAB CODE : K SAMPLE NUMBER : 18 CONTACT CODE : LEARV

-----OIL AND FILTER ANALYSIS FINDINGS-----

FILTER MG OVER MAXIMUM

| ***OIL RESULTS*** ANALYSIS PER WORK INSTRUCTIONS MLTC WI 1437. |    |       |    |           |    |       |    |            |   |  |
|----------------------------------------------------------------|----|-------|----|-----------|----|-------|----|------------|---|--|
| FE                                                             | CU | NI    | CR | AG        | MG | AL    | PB | BE         | S |  |
| TAN                                                            |    | WATER |    | VISCOSITY |    | RULER |    | FLASHPOINT |   |  |

**RECOMMEND Re-Sample Continue**

MAINT PERFORMED :  
LAB COMMENTS :



SERIAL NUMBER P137180

## SERVICE RECORD

PAGE 3

### DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

**ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER**

[illegible]

# TFE731-40BR

# Honeywell

|              |      |                              |                 |
|--------------|------|------------------------------|-----------------|
| TRACKING NO: | 002B | FSE WORK INSTRUCTION NUMBER: | 731-40-P14-002B |
|--------------|------|------------------------------|-----------------|

**DESCRIPTION OF WORK:** Reason for revision – correct blade retainer part number, change recording of blade machining lot numbers to optional and add requirement for LPT2 rotor life limit card entry.

Disassemble engine to remove the LPT2 rotor assembly for inspection.

Please record each of the 52 blade casting lot numbers on the sheet provided and forward the information to engineering: [anthony.dangelo@honeywell.com](mailto:anthony.dangelo@honeywell.com).

Recording the machining lot number of the blades is desirable but not required.

## REFERENCED DOCUMENTS

| DOCUMENT NUMBER | REV. | TITLE               |
|-----------------|------|---------------------|
| 3060608-5       | --   | LPT2 ROTOR ASSEMBLY |

## REQUIRED PARTS / MATERIAL

| PART NUMBER          | QTY REM. | QTY INST. | F/N | SERIAL NUMBER            | NOMENCLATURE                    |
|----------------------|----------|-----------|-----|--------------------------|---------------------------------|
| 3061829-1            |          | AR        |     |                          | Blade Retainer                  |
| PCS5720              |          | AR        |     |                          | Compound                        |
| 73563-21             |          | 1         |     |                          | T-Lock, Turbine                 |
| PCS5723              |          | AR        |     |                          | Compound                        |
| 73563-10             |          | 1         |     |                          | T-Lock, Turbine monopole sensor |
| 3060642-1            |          | 1         |     |                          | Gasket, Aft Bearing Cover       |
|                      |          | AR        |     |                          | Scotch tape                     |
| M83248/1-012         |          | 1         |     |                          | Packing                         |
| 651-534-9612         |          | 1         |     |                          | E-Seal, Turbine                 |
| M83248/1-013         |          | 2         |     |                          | Packing                         |
| M83248/1-011         |          | 2         |     |                          | Packing                         |
| Engine Serial Number |          |           |     | LPT2 Rotor Serial Number |                                 |
| P137180              | 1        |           |     | 12-156103-00730          |                                 |
| P137180              |          | 1         |     | 12-156103-00810          |                                 |

## AFFECTED ENGINE SERIAL NUMBERS:

P137149, P137150, P137159, P137160, P137171, P137172, P137173, P137174, P137175, P137176, P137177, P137178, P137179, P137180, P137181, P137182, P137183, P137184, P137185, P137186, P137187, P137188, P137189, P137190, P137191, P137192, P137193, P137194



# TFE731-40BR

# Honeywell

|              |      |                              |                 |
|--------------|------|------------------------------|-----------------|
| TRACKING NO: | 002B | FSE WORK INSTRUCTION NUMBER: | 731-40-P14-002B |
|--------------|------|------------------------------|-----------------|

**COMMENTS:** Activity is to access the LPT2 rotor assembly and inspect the LPT2 blades. Also, record the casting lot numbers (machining lot numbers optional) on pages 4 and 5 of this instruction. The location of the lot numbers is shown in the detailed instructions attached.

Rotor assemblies shall have each of the 52 blades inspected by eddy current. As an option, the rotor assembly may be returned to Honeywell for inspection. Replacement rotor assemblies will have eddy current inspection stamps on each blade (see detailed instructions).

Any blades that fail eddy current inspection must be replaced with a blade with an eddy current inspection stamp, or, perform the eddy current inspection on any blade that does not contain the stamp.

Please return blades that fail eddy current inspection to Honeywell at the following address:

**Honeywell Aerospace**  
**3131 East Airline**  
**Phoenix, AZ 85034**  
**Attn: NDE Engineering MS 403-NDT**

All LPT2 rotor assemblies shall be returned to:

**Honeywell Engines**  
**RHAC TEAM PLANT 1015**  
**111 South 34<sup>TH</sup> Street**  
**Phoenix, AZ. 85072**

|                                         | Signature                                                                                                                                                          | Date          |
|-----------------------------------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------------|
| FSE Work Instruction Approved by:       | Tony D'Angelo<br><small>Digitally signed by Tony<br/>DN: cn=Tony, o, ou,<br/>email=anthony.dangelo@honeywell.com, c=US<br/>Date: 2014.06.26 06:50:40 -0700</small> |               |
| Work Completed by (HON Representative): | Jim Beasley HON-FSE                                                                                                                                                | 11-13-14 (ST) |
| Work Accepted by (Honeywell FSE):       | Jim Beasley HON-FSE                                                                                                                                                | 11-13-14 (ST) |
| Field Service Report (FSR) Number:      |                                                                                                                                                                    |               |

|              |      |                              |                 |
|--------------|------|------------------------------|-----------------|
| TRACKING NO: | 002B | FSE WORK INSTRUCTION NUMBER: | 731-40-P14-002B |
|--------------|------|------------------------------|-----------------|

**GENERAL:**

This instruction is required to remove the LPT2 rotor assembly for inspection. Detailed instructions are attached for engine disassembly, inspection and reassembly.

Engine log book and LPT2 life limit card entries will be required.

**DETAILED INSTRUCTIONS:**

1. See attached for detailed instructions.
2. Check and ensure proper engine operation – perform leak check and vibration survey.
3. Make an entry in the engine log book.
  - a. Date, T/T, inspection performed per FSE WI 731-40-P14-002B. If required, record the LPT2 Rotor Assy PN, SN removed and LPT2 Rotor Assy PN, SN installed. Work completed by (FSE name) (quality stamp). Honeywell Production Certificate PC413NM.
4. Provide a copy of the engine log book page after sign-off by scanning it and email to engineering: [anthony.dangelo@honeywell.com](mailto:anthony.dangelo@honeywell.com).
5. LPT2 Rotor assemblies eddy current inspected and installed into engines must have an entry made on the back of the Life Limit Card as follows:
  - a. DATE – date on which the inspection was performed
  - b. Signature and Stamp – shall be that of the eddy current inspector

| LIFE LIMITED PART REPAIR RECORD |                         |                      |
|---------------------------------|-------------------------|----------------------|
| DATE                            | MAINTENANCE PERFORMED   | AUTHORIZED SIGNATURE |
| 6/24/2014                       | Inspection per 21-15477 | Name and UQR stamp   |

# TFE731-40BR

# Honeywell

|                     |             |                                     |                        |
|---------------------|-------------|-------------------------------------|------------------------|
| <b>TRACKING NO:</b> | <b>002B</b> | <b>FSE WORK INSTRUCTION NUMBER:</b> | <b>731-40-P14-002B</b> |
|---------------------|-------------|-------------------------------------|------------------------|

TFE731-40BR ENGINE S/N

[illegible]

See Attached

# Honeywell

## Outgoing LPT2 Blade PN 3075424 Data Form

**NOTE:** Document the blade dash number, blade casting, and machining lot numbers for each blade.

If the blade dash number or machining lot number is not visible on the blade, it is not required to record the dash number or machining lot number for that blade, but casting lot number is required.

After form has been completed and is ready for submission, click "Submit Form".

Enter the engine serial number as the new file name, and click "Save" to save to your computer hard drive.  
An email window will open with the file attached, add any email address for copy, if desired, and press send.

Date of Inspection: 7/12/14

Inspector: Mark Stroot

Inspection Facility: Bombardier Wichita

Aircraft SN: 45-485 446 QB

Engine SN: P-137180

Engine TSN: 0

Engine CSN: 0

LPT2 Rotor Assy PN: 3060608-5

LPT2 Rotor Assy SN: 12-156103-00810

LPT2 Rotor Assy TSN: 0

LPT2 Rotor Assy CSN: 0

| Record Data for LPT2 Blades, PN 3075424 Being Installed |                   |                          |                            |                |                   |                          |                            |
|---------------------------------------------------------|-------------------|--------------------------|----------------------------|----------------|-------------------|--------------------------|----------------------------|
| Blade Position                                          | Blade Dash Number | Blade Casting Lot Number | Blade Machining Lot Number | Blade Position | Blade Dash Number | Blade Casting Lot Number | Blade Machining Lot Number |
| 1                                                       | -2                | ECWK                     | HM-4149                    | 27             | -2                | ECWM                     | HM-4150                    |
| 2                                                       | -2                | ECWK                     | HM-4149                    | 28             | -2                | ECWC                     | HM-4150                    |
| 3                                                       | -2                | ECWK                     | HM-4149                    | 29             | -2                | ECWK                     | HM-4149                    |
| 4                                                       | -2                | ECWL                     | HM-4149                    | 30             | -2                | ECTY                     | HM-4150                    |
| 5                                                       | -2                | ECWL                     | HM-4149                    | 31             | -2                | ECWL                     | HM-4149                    |
| 6                                                       | -2                | ECWJ                     | HM-4154                    | 32             | -2                | ECWD                     | HM-4113                    |
| 7                                                       | -2                | ECWC                     | HM-4150                    | 33             | -2                | ECWA                     | HM-4150                    |
| 8                                                       | -2                | ECWM                     | HM-4149                    | 34             | -2                | ECCW                     | HM-4150                    |
| 9                                                       | -2                | ECWA                     | HM-4150                    | 35             | -2                | ECWK                     | HM-4149                    |
| 10                                                      | -2                | ECWC                     | HM-4150                    | 36             | -2                | ECWK                     | HM-4149                    |
| 11                                                      | -2                | ECWL                     | HM-4150                    | 37             | -2                | ECWD                     | HM-4113                    |
| 12                                                      | -2                | ECWA                     | HM-4150                    | 38             | -2                | ECWL                     | HM-4149                    |
| 13                                                      | -2                | ECWL                     | HM-4149                    | 39             | -2                | ECWL                     | HM-4149                    |
| 14                                                      | -2                | ECWA                     | HM-4150                    | 40             | -2                | ECWL                     | HM-4149                    |
| 15                                                      | -2                | ECUH                     | HM-4150                    | 41             | -2                | ECWC                     | HM-4150                    |
| 16                                                      | -2                | ECWC                     | HM-4150                    | 42             | -2                | ECWC                     | HM-4150                    |
| 17                                                      | -2                | ECWC                     | HM-4150                    | 43             | -2                | ECWJ                     | HM-4149                    |
| 18                                                      | -2                | ECWC                     | HM-4113                    | 44             | -2                | ECWK                     | HM-4149                    |
| 19                                                      | -2                | ECWL                     | HM-4149                    | 45             | -2                | ECWA                     | HM-4150                    |
| 20                                                      | -2                | ECCW                     | HM-4113                    | 46             | -2                | ECWC                     | HM-4150                    |
| 21                                                      | -2                | ECWE                     | HM-4149                    | 47             | -2                | ECWM                     | HM-4149                    |
| 22                                                      | -2                | ECWK                     | HM-4149                    | 48             | -2                | ECWL                     | HM-4149                    |
| 23                                                      | -2                | ECWJ                     | HM-4149                    | 49             | -2                | ECWK                     | HM-4149                    |
| 24                                                      | -2                | ECWK                     | HM-4149                    | 50             | -2                | ECWL                     | HM-4150                    |
| 25                                                      | -2                | ECWL                     | HM-4149                    | 51             | -2                | ECWL                     | HM-4150                    |
| 26                                                      | -2                | ECWK                     | HM-4149                    | 52             | -2                | ECWK                     | HM-4149                    |

**NOTE:** All blades must be identified with an EC inspection acceptance ink stamp.  
Form will be emailed to [tfedata@honeywell.com](mailto:tfedata@honeywell.com)



Submit Form

**TFE731-40BR**

# Honeywell

|                     |             |                                     |                        |
|---------------------|-------------|-------------------------------------|------------------------|
| <b>TRACKING NO:</b> | <b>002B</b> | <b>FSE WORK INSTRUCTION NUMBER:</b> | <b>731-40-P14-002B</b> |
|---------------------|-------------|-------------------------------------|------------------------|

[illegible]

# TFE731-40BR

## DIS-ASSEMBLY INSTRUCTIONS TO REMOVE AND REPLACE P/N 3060608-5 LPT2 ROTOR ASSEMBLY

See supplemental instructions (following) for performing eddy current inspection of the LPT2 rotor assembly and for LPT2 blade replacement.

REMOVE TAILPIPE COVER

REMOVE #6 OIL SUPPLY AND SCAVENGE TUBES AND CLAMPS (NEXT PAGE) AS SHOWN.

|      |              |          |                                      |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|--------------|----------|--------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 70060013-906 | P/N Desc | 70060013 LMAASA -40 P&E INSTALLATION | Alternate MOT# | 000* | Rev        | G | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 500          | OP Desc  | number 6 BRG TUBES - INSTALL         | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

**Instructions**
Reference Documents [5]

**CAUTION:**

Use back-up wrenches  
when securing tubing.

**CAUTION:**

Maintain proper clearance  
between tubing and surrounding  
hardware when installing.

1. Install FN112 and D129  
oil tubes with hardware  
indicated.
2. Install FN E071 & FN E072 to N1 lead  
and FN D112 as shown using hardware  
indicated.

**Attachments**

**SECTION X-X**

**Parts [14]**

| SEQ | Part         | FindNo | Name             | Qty | AltQty |
|-----|--------------|--------|------------------|-----|--------|
| 1   | 3060909-1    | D112   | TUBE ASSY, METAL | 1   |        |
| 2   | 3060964-1    | D129   | TUBE ASSY        | 1   |        |
| 3   | 3060646-1    | D250   | RETAINER PLATE   | 1   |        |
| 4   | 3060647-1    | D251   | RETAINER PLATE   | 1   |        |
| 5   | M83248/1-013 | D305   | PACKING          | 2   |        |
| 6   | M83248/1-011 | D360   | Packing, O-Ring  | 2   |        |
| 7   | MS9556-06    | D425   | SCREW, MACHINE   | 1   |        |

Parts [14]
Tools [0]



|      |              |          |                                      |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|--------------|----------|--------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 70060013-906 | P/N Desc | 70060013 LMAASA -40 P&E INSTALLATION | Alternate MOT# | 000* | Rev        | G | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 500          | OP Desc  | number 6 BRG TUBES - INSTALL         | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

|              |                         |             |
|--------------|-------------------------|-------------|
| Instructions | Reference Documents [5] | Attachments |
|--------------|-------------------------|-------------|

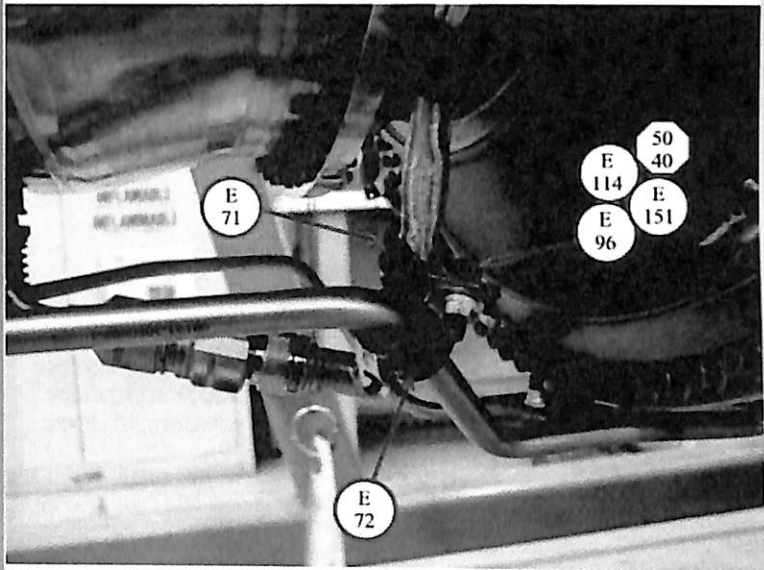
CAUTION:

Use back-up wrenches  
when securing tubing.

CAUTION:

Maintain proper clearance  
between tubing and surrounding  
hardware when installing.

1. Install FN112 and 129  
oil tubes with hardware  
indicated.
2. Install FN E071 & FN E072 to N1 lead  
and FN D112 as shown using hardware  
indicated.



| Parts [14] |                 |        |                           |     |        |  |
|------------|-----------------|--------|---------------------------|-----|--------|--|
| SEQ        | Part            | FindNo | Name                      | Qty | AltQty |  |
|            | MS9556-06       | D427   | SCREW, MACHINE            | 4   |        |  |
|            | PCS5720(QS3166) | D490   | COMPOUND (NPC60301-1 REF) |     | A/R    |  |
|            | 211-536-9104    | E071   | CLAMP                     | 1   |        |  |
|            | 211-536-9106    | E072   | CLAMP LOOP                | 1   |        |  |
|            | MS9556-06       | E096   | SCREW, MACHINE            | 1   |        |  |
|            | MS21043-3       | E114   | NUT, SELF-LOCKING         | 1   |        |  |
|            | PCS5723         | E151   | ANTISEIZE, MOLYDISULFIDE  |     | A/R    |  |

Parts [14]
Tools [0]

## REMOVE N1 MONOPOLE AND COVER

|      |            |          |                     |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|---------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY  | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0230       | OP Desc  | N1 SENSOR - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

**NOTE:**  
Ensure cover is seated flush and maintain downward force when securing.

1. Install FN87 gasket and FN88 exciter/cover assy with bolts.

**INITIAL TORQUE OF COVER ON THIS OPERATION.  
FINAL TORQUE TO BE DONE AT LATER OPERATION**

2. Install FN138 sensor as shown.

Parts [6]

| SEQ | Part            | FindNo | Name                      | Qty | AltQty |
|-----|-----------------|--------|---------------------------|-----|--------|
|     | 3060642-1       | 0087   | GASKET COVER              | 1   |        |
|     | 3060658-5       | 0088   | COVER - EXCITER           | 1   |        |
|     | 3060714-4       | 0138   | SENSOR SPEED NI MONO      |     | REF    |
|     | MS9556-07       | 0210   | BOLT                      | 9   |        |
|     | MS9556-31       | 0211   | MACHINE BOLT              | 2   |        |
|     | PCS5720(QS3166) | 0301   | COMPOUND (NPC60301-1 REF) |     | A/R    |

Tools [2]

| Rev | Part    | Name          |
|-----|---------|---------------|
| -   | MAT2386 | BREAKER BAR   |
| -   | T216734 | TANGED WRENCH |

Attachments

REMOVE EXCITER GEAR NUT. (SEE NEXT PAGE FOR TOOLING)

|      |            |          |                            |                |      |            |   |           |  |         |  |             |    |             |    |         |      |
|------|------------|----------|----------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|------|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY         | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline | Cell |
| Step | 0210       | OP Desc  | EXCITER GEAR NUT - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |      |

**Instructions**    Reference Documents [5]

**- TRACEABLE ITEM THIS OPERATION -**

1. Install FN84 nut with FN304 lubricant and FN85 lock-key as shown.
2. Torque nut per procedure below and record on build sheet FORMULA 8.

**TORQUING PROCEDURE:**

- Initial torque nut to 1200 in/lbs
- Locate lock-key slot that best aligns
- Back-off nut and install lock-key
- Retorque nut to 1200 in.lbs
- Apply additional torque for alignment

- DO NOT EXCEED 1400 IN/LBS -

- Bend lock-key into slot

**Attachments**

**Parts [3]**

| SEQ | Part            | FindNo | Name               | Qty | AltQty |
|-----|-----------------|--------|--------------------|-----|--------|
| 1   | 3070163-1       | 0084   | SLOTTED NUT        | 1   |        |
| 2   | 73563-10        | 0085   | KEY NUT LKG        | 1   |        |
| 3   | PCS5723(QS3043) | 0304   | MOLY LUBE (QS3043) |     | A/R    |

Parts [3]    Tools [2]

# TOOLING FOR REMOVING EXCITER GEAR NUT OR ALTERNATE EQUIVALENT TOOLING

|      |            |          |                            |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|----------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY         | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0210       | OP Desc  | EXCITER GEAR NUT - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

**- TRACEABLE ITEM THIS OPERATION -**

1. Install FN84 nut with FN304 lubricant and FN85 lock-key as shown.
2. Torque nut per procedure below and record on build sheet FORMULA 8.

**TORQUING PROCEDURE:**

- Initial torque nut to 1200 in/lbs
- Locate lock-key slot that best aligns
- Back-off nut and install lock-key
- Retorque nut to 1200 in.lbs
- Apply additional torque for alignment

**- DO NOT EXCEED 1400 IN/LBS -**

- Bend lock-key into slot

Attachments

A) INITIAL TQ 1200 IN/LBS

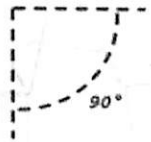
B) LOCATE LOCK-TAB SLOT THAT BEST ALIGNS

C) BACK-OFF NUT AND INSTALL LOCK-KEY IN SLOT

D) RE-TORQUE TO 1200 IN/LBS

E) APPLY ADDITIONAL TQ TO ALIGN LOCK-KEY WITH SLOT

**- DO NOT EXCEED 1400 IN/LBS -**



**- MAINTAIN 90° ANGLE -**



**STANDARD TQ WRENCH**



**T216734**



**MAT2386 (OR EQUIVALENT)**

Tools [2]

| Rev | Part    | Name          |
|-----|---------|---------------|
| -   | MAT2386 | BREAKER BAR   |
| -   | T216734 | TANGED WRENCH |

Parts [3]

Tools [2]

# REMOVE EXCITER GEAR (SEE NEXT PAGE FOR TOOLING)

|      |            |          |                        |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY     | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0200       | OP Desc  | EXCITER GEAR - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

- TRACEABLE ITEM THIS OPERATION -

1. Install FN83 gear using T216732 as shown.

Attachments

Parts [1]

| SEQ | Part      | FindNo | Name                 | Qty | AltQty |
|-----|-----------|--------|----------------------|-----|--------|
|     | 3060601-2 | 0083   | EXCITER, SPEED SENSO | 1   |        |

Parts [1]

Tools [1]

## TOOLING FOR EXCITER GEAR REMOVAL (OR EQUIVALENT)

|      |            |          |                        |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY     | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0200       | OP Desc  | EXCITER GEAR - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

Attachments

**- TRACEABLE ITEM THIS OPERATION -**

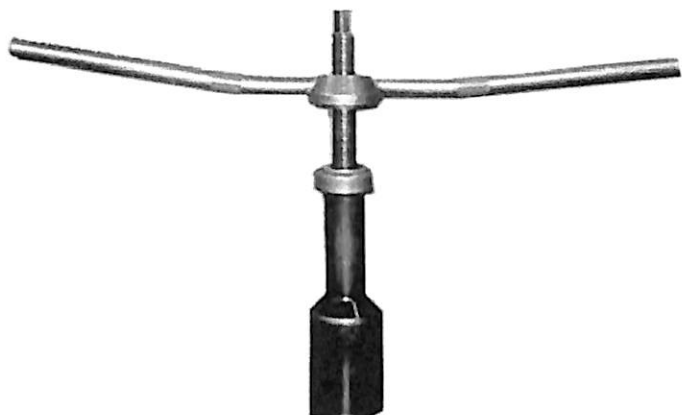
1. Install FN83 gear using T216732 as shown.

Tools [1]

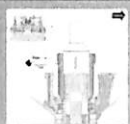
| Rev | Part    | Name               |
|-----|---------|--------------------|
| -   | T216732 | INSTALL #6 BEARING |

Parts [1]

Tools [1]



**T216732**




SECURE TOP REAR MOUNT TO LIFTING DEVICE, REMOVE BRACKETS AND LPT FLANGE BOLTS AS SHOWN.

|      |            |          |                     |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|---------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY  | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0170       | OP Desc  | LPT FLANGE - SECURE | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

Attachments

1. Remove LPT module slave bolts at this time.

2. Install washer on 'BOLT HEAD' side in positions identified by blue circles when securing flange.

3. Install fn's D201, E033, and G002 in positions identified by red circles using hardware shown.

4. Install fn 207 bolts (93) places as shown.

5. Tighten all bolts and nuts to torque shown.

Parts [8]

| SEQ | Part            | FindNo | Name                                    | Qty | AltQty |
|-----|-----------------|--------|-----------------------------------------|-----|--------|
|     | AS3237-10       | 0207   | BOLT, MACHINE - DOUBLE HEXAGON EXTENDED | 93  |        |
|     | MS21043-4       | 0236   | NUT                                     | 120 |        |
|     | NAS620C416      | 0272   | WSHR.FLT 1/4-RDC OD.063                 | 2   |        |
|     | AS3237-11       | 0293   | BOLT, MACHINE - DOUBLE HEXAGON EXTENDED | 27  |        |
|     | PCSS720(QS3166) | 0301   | COMPOUND (NPC60301-1 REF)               |     | A/R    |
|     | 3073465-1       | D201   | BRACKET OIL TUBES                       | 1   |        |
|     | MS9593-037      | E033   | BRACKET                                 | 1   |        |
|     | 3072586-1       | G002   | BRACKET                                 | 11  |        |

VIEW X-X



UNSTRETCH GROUP PER INSTRUCTIONS BELOW USING TOOLING SHOWN OR EQUIVALENT  
REMOVE FN 102 NUT (3071740-1)

|      |            |          |                    |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|--------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0150       | OP Desc  | LPT - STRETCH      | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

Instructions

Reference Documents [5]

1. Install stretch shaft tool (approx. 9 full turns) then back-off shaft 1/4 turn.
2. Install remaining tooling.
3. Apply approx. 4500 psig. and secure nut.
4. Remove tooling.

---

**CAUTION:** Ensure pick-up gear is removed prior to unstretching of tie-shaft (if installed).

**UNSTRETCH:**

- Install stretch shaft (approx. 9 turns)
- Back off shaft 2 & 1/2 turns
- Bottom out shaft nut
- Back off shaft nut 3 full turns
- Apply approx. 4500 psig.
- Back off LPT nut
- Release psig. and remove tooling

Attachments

**4500 PSIG.**

**UNSTRETCH**

**T216776**

**MAT2212-3**

**T216775-10 (NUT)**

**T216777-2 (SHAFT)**

**30 TON PUMP & CYLINDER**

**T216777-130 (ADAPTER)**

**T216777 (NUT TOOL)**

Parts [0]

CAREFULLY REMOVE TURBINE AND REAR MOUNT FROM ENGINE. NOTE: PHOTOS DEPICT VERTICAL REMOVAL. HORIZONTAL REMOVAL POSSIBLE SUPPORTING WEIGHT OF LPT MODULE FROM CENTER MOUNT CLEVIS USING OVERHEAD CRANE OR LIFT. REMOVE ALL FASTENERS, WITH LPT ASSY HOOKED TO VERTICAL LIFT. GENTLY ROCK LPT ASSY AFT FOR REMOVAL . PLACE LPT MODULE ON BENCH FOR FURTHER DISASSEMBLY.

|      |            |          |                          |                |      |            |   |           |  |         |  |             |    |             |    |         |  |      |  |
|------|------------|----------|--------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|----|-------------|----|---------|--|------|--|
| Part | 3060022-26 | P/N Desc | POWER SECTION ASSY       | Alternate MOT# | 000* | Rev        | A | Rel. Date |  | Prog_ID |  | Eng Chg Ltr | AK | P/L Chg Ltr | CJ | Outline |  | Cell |  |
| Step | 0120       | OP Desc  | TAIL PIPE ASSY - INSTALL | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |    |             |    |         |  |      |  |

**Instructions**
Reference Documents [5]

**Attachments**

**CAUTION:**

Exercise care not to damage curvics when lowering tail pipe assembly onto tie-shaft.

- Carefully lower the LP turbine module down over the LP tie shaft. Exercise care not to damage the curvics when lowering the turbine assembly onto the tie-shaft. Align the off-set bolt pattern of the transition duct and plenum case at the 12 o'clock position. Initially align the dykem marks (tie shaft to 3rd stage rotor.)
- Rotate turbine rotor group approximately 5 degrees, counter-clock-wise, AFT looking forward, to purposely mis-align the curvic couplings.
- Gently continue lowering the turbine module until the FWD curvic coupling of the turbine group comes in contact with the AFT curvic coupling on the compressor group.  
DO NOT COMPLETELY REMOVE ALL THE TENSION FROM THE HOIST AT THIS TIME.
- Gently rotate the turbine module back approximately 5 degrees clock-wise to engage the curvic coupling between the compressor and the turbine. When the curvic coupling engages, you should feel

PS 33

PS 109

ALL HOLES ON THIS FLANGE SHOULD HAVE THE SAME GAP

SAME ENGAGEMENT ALL AROUND

**Parts [2]**

| SEQ | Part         | FindNo | Name                                         | Qty | AltQty |
|-----|--------------|--------|----------------------------------------------|-----|--------|
|     | 3060022-914  | 0033   | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | REF |        |
|     | 651-534-9612 | 0109   | SEAL                                         | REF |        |

Parts [2] Tools [1]

REMOVE ITEMS SHOWN BELOW AND ON NEXT PAGE USING TOOLING SHOWN ON 3RD PAGE (OR EQUIVALENT TOOLING)

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 370         | OP Desc  | THRUST PLATE - INSTALL                       | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

**- TRACEABLE ITEM THIS OPERATION -**

1. Install FN79 thrust plate, FN80 spring, and FN82 ring.
2. Compress items with T272748 & T216769 and install FN81 retainer ring.

Attachments

Parts [4]

| SEQ | Part         | FindNo | Name                     | Qty | AltQty |
|-----|--------------|--------|--------------------------|-----|--------|
|     | 3071296-1    | A0079  | PLATE                    | 1   |        |
|     | 3000005-5    | A0080  | WASHER SPRING            | 1   |        |
|     | M27426-3171D | A0081  | RINGS, RETAINING, SPIRAL | 1   |        |
|     | 3071287-1    | A0082  | RING, RETAINING          | 1   |        |

Parts [4]

Tools [4]

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 360         | OP Desc  | number 6 BEARING - INSTALL                   | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

#### Instructions

Reference Documents [5]

#### - TRACEABLE ITEM THIS OPERATION -

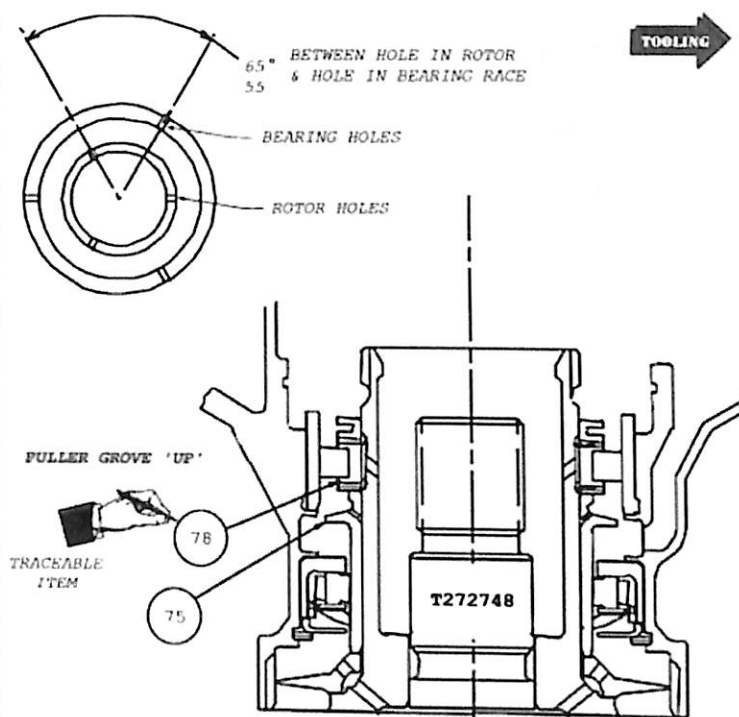
1. Install FN75 seal rotor.

**CAUTION:** Ensure bearing puller grooves are up.

2. Using T272748 & T216732-1  
press inner race of FN78  
bearing onto shaft and  
offset oil passage holes  
as shown.

3. Position bearing rollers  
inward and install outer race  
with chamfered edge down.

#### Attachments



#### Parts [2]

| SEQ | Part      | FindNo | Name            | Qty | AltQty |
|-----|-----------|--------|-----------------|-----|--------|
| 1   | 3070344-2 | A0075  | ROTOR SEAL      | 1   |        |
| 2   | 358750-2  | A0078  | BEARING, ROLLER | 1   |        |

Parts [2] Tools [4]

## TOOLING FOR REMOVAL OF ITEMS ON PREVIOUS PAGE

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 360         | OP Desc  | number 6 BEARING - INSTALL                   | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

Attachments

**- TRACEABLE ITEM THIS OPERATION -**

1. Install FN75 seal rotor.

**CAUTION:** Ensure bearing puller grooves are up.

2. Using T272748 & T216732-1 press inner race of FN78 bearing onto shaft and offset oil passage holes as shown.

3. Position bearing rollers inward and install outer race with chamfered edge down.

Tools [4]

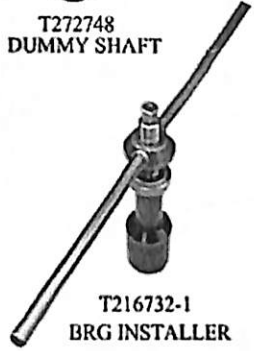
| Rev | Part      | Name                                                                |
|-----|-----------|---------------------------------------------------------------------|
| -   | T216732-1 | BEARING INSTALLER                                                   |
| -   | T216769   | INSTALLATION & REMOVAL TOOL- RETAINING RING (WAVE WASHER DEPRESSOR) |
| -   | T216801   | BEARING PULLER                                                      |
| C   | T272748   | Dummy Tie Rod, Nut and Wrench                                       |

Parts [2]

Tools [4]



**T272748  
DUMMY SHAFT**



**T216732-1  
BRG INSTALLER**



**T216801  
BRG PULLER**



**T216769  
SNAP RING  
INSTALLER**

# REMOVE REAR BEARING SUPPORT HOUSING FROM ASSEMBLY

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 350         | OP Desc  | TAIL PIPE - INSTALL                          | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

**NOTE:** Position off-set hole identified with 'C' at 12 O'CLOCK position.

1. Install FN35 rear bearing support assy.
2. Secure rear bearing support assy to transition duct with hardware shown.

Attachments

BOLT HEAD AFT

301

208

238

35

28

23

18

14

9

6

hole #1

(6)

VIEW X-X

• = LEAVE EMPTY (16) PLACES

Parts [4]

| SEQ | Part      | FindNo | Name                                   | Qty | AltQty |
|-----|-----------|--------|----------------------------------------|-----|--------|
|     | 3073718-3 | A0035  | SUPPORT, BEARING HOUSING, REAR - NO. 6 |     | REF    |
|     | MS9556-07 | A0208  | BOLT                                   | 54  |        |
|     | MS21043-3 | A0238  | NUT, SELF-LOCKING                      | 54  |        |
|     | PCS5720   | A0301  | COMPOUND                               |     | A/R    |

Parts [4]

Tools [0]

# UNSTRETCH THE TURBINE GROUP USING TOOLING SHOWN ON NEXT PAGE (OR EQUIVALENT)

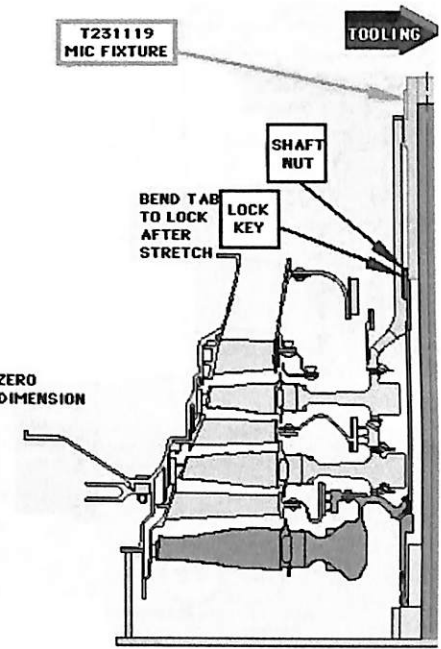
|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 280         | OP Desc  | STRETCH LP TURBINE SHAFT                     | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

Attachments

1. VERIFY TOOLING SET NUMBER.
2. MEASURE ZERO (BEFORE STRETCH) DIMENSION AND RECORD IN BUILD SHEET FORMULA 1.
3. INSTALL T216778 STRETCH FIXTURE. STRETCH TURBINE GROUP TO OBTAIN .0076-.0096 STRETCH.
  - \* BOTTOM OUT TOOLING KNURLED NUT AND WING NUT.
  - \* LOAD TO APPROX 2250 PSIG.
  - \* BOTTOM OUT SHAFT NUT.
  - \* RELEASE PRESSURE AND REMOVE TOOLING.
4. MEASURE STRETCH AND RECORD IN BUILD SHEET FORMULA 1.
5. BEND LOCK KEY THAT BEST LINES UP WITH SLOT IN TURBINE SHAFT.
  - \* ENSURE UNBENT LOCK KEY DOES NOT PROTRUDE OR IS PUSHED BACK AGAINST SHAFT.

Parts [2]

| SEQ | Part      | FindNo | Name     | Qty | AltQty |
|-----|-----------|--------|----------|-----|--------|
| 1   | 3070177-2 | 0018   | NUT      | REF |        |
| 2   | PCS5720   | 0023   | COMPOUND | A/R |        |

Parts [2]

Tools [3]



## TOOLING TO UNSTRETCH TURBINE GROUP

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 280         | OP Desc  | STRETCH LP TURBINE SHAFT                     | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

Attachments

1. VERIFY TOOLING SET NUMBER.
2. MEASURE ZERO (BEFORE STRETCH)  
DIMENSION AND RECORD IN BUILD  
SHEET FORMULA 1.
3. INSTALL T216778 STRETCH FIXTURE.  
STRETCH TURBINE GROUP TO OBTAIN  
.0076-.0096 STRETCH.
  - \* BOTTOM OUT TOOLING KNURLED NUT  
AND WING NUT.
  - \* LOAD TO APPROX 2250 PSIG.
  - \* BOTTOM OUT SHAFT NUT.
  - \* RELEASE PRESSURE AND REMOVE TOOLING.
4. MEASURE STRETCH AND RECORD IN BUILD  
SHEET FORMULA 1.
5. BEND LOCK KEY THAT BEST LINES UP WITH  
SLOT IN TURBINE SHAFT.
  - \* ENSURE UNBENT LOCK KEY DOES NOT PROTRUDE  
OR IS PUSHED BACK AGAINST SHAFT.

Tools [3]

| Rev | Part        | Name                 |
|-----|-------------|----------------------|
| -   | T216773-910 | WRENCH               |
| -   | T216778     | STRETCH FIXTURE BASE |
| -   | T231119     | LP TURB MIC FIXTURE  |

Parts [2]

Tools [3]

T216778-908 STRETCH CART

T216778 STRETCH TOOL

REMOVE 3070177-2 NUT AND 3<sup>RD</sup> STAGE WHEEL ASSEMBLY 3060626-8 USING T216774 LIFTING LUG OR EQUIVALENT.

|      |             |          |                                              |             |                |          |     |   |           |         |             |             |         |      |  |
|------|-------------|----------|----------------------------------------------|-------------|----------------|----------|-----|---|-----------|---------|-------------|-------------|---------|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY |             | Alternate MOT# | 000*     | Rev | C | Rel. Date | Prog_ID | Eng Chg Ltr | P/L Chg Ltr | Outline | Cell |  |
| Step | 220         | OP Desc  | INSTALL SHAFT NUT                            | Work Center | Setup Time     | Run Time |     |   |           |         |             |             |         |      |  |

Instructions

Reference Documents [5]

**\*\*INSTALL OR ASSEMBLE\*\***

1: FN18 SHAFT NUT WITH FN23 COMPOUND TO 3RD STAGE WHEEL ASSEMBLY. SECURE FINGER TIGHT.

\* USE T216773-910 WRENCH TO INSTALL NUT.

Attachments

3rd STAGE WHEEL ASSEMBLY

23 18

LUBRICATE NUT THREADS AND NUT FACE. SECURE FINGER TIGHT

Parts [2]

| SEQ                      | Part      | FindNo | Name     | Qty | AltQty |
|--------------------------|-----------|--------|----------|-----|--------|
| <input type="checkbox"/> | 3070177-2 | 0018   | NUT      |     | REF    |
| <input type="checkbox"/> | PCS5720   | 0023   | COMPOUND |     | A/R    |

Parts [2]

Tools [1]

## REMOVE THIRD STAGE NOZZLE ASSEMBLY

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 180         | OP Desc  | INSTALL 3RD STAGE NOZZLE                     | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

**Instructions**
Reference Documents [5]

**Attachments**

**\*\*RECORD TRACEABILITY\*\***

1: FN4 3RD STAGE NOZZLE.

**\*\*INSTALL OR ASSEMBLE\*\***

2: FN4 THIRD STAGE NOZZLE.  
 \* ALIGN OFFSET HOLE To 12:00.  
 \* SECURE WITH SLAVE HARDWARE.

SECURE TO TRANSITION DUCT WITH SLAVE HARDWARE (4) PLACES

REF 4

**Parts [1]**

| SEQ | Part      | FindNo | Name                 | Qty | AltQty |
|-----|-----------|--------|----------------------|-----|--------|
| 1   | 3060655-6 | 0004   | NOZZLE ASSY LPT 3STG | 1   |        |

Parts [1] Tools [0]

REMOVE SECOND STAGE ROTOR ASSEMBLY 3060608-5.

|      |             |          |                                              |                |      |            |   |           |  |         |  |             |  |             |  |         |  |      |  |
|------|-------------|----------|----------------------------------------------|----------------|------|------------|---|-----------|--|---------|--|-------------|--|-------------|--|---------|--|------|--|
| Part | 3060022-914 | P/N Desc | 3060022 LMAASA -40 TURBINE & REAR MOUNT ASSY | Alternate MOT# | 000* | Rev        | C | Rel. Date |  | Prog_ID |  | Eng Chg Ltr |  | P/L Chg Ltr |  | Outline |  | Cell |  |
| Step | 170         | OP Desc  | INSTALL 2ND STAGE WHEEL ASSY                 | Work Center    |      | Setup Time |   | Run Time  |  |         |  |             |  |             |  |         |  |      |  |

Instructions

Reference Documents [5]

**\*\*INSTALL OR ASSEMBLE\*\***

**\*CAUTION\*** CLEAN CURVICS PRIOR TO INSTALLING WHEEL ASSEMBLIES.

1: FN6 2ND STAGE WHEEL ASSY WITH V-SEAL UP AS SHOWN.

\* ALIGN CURVIC HIGH MARKS HIGH TO HIGH.

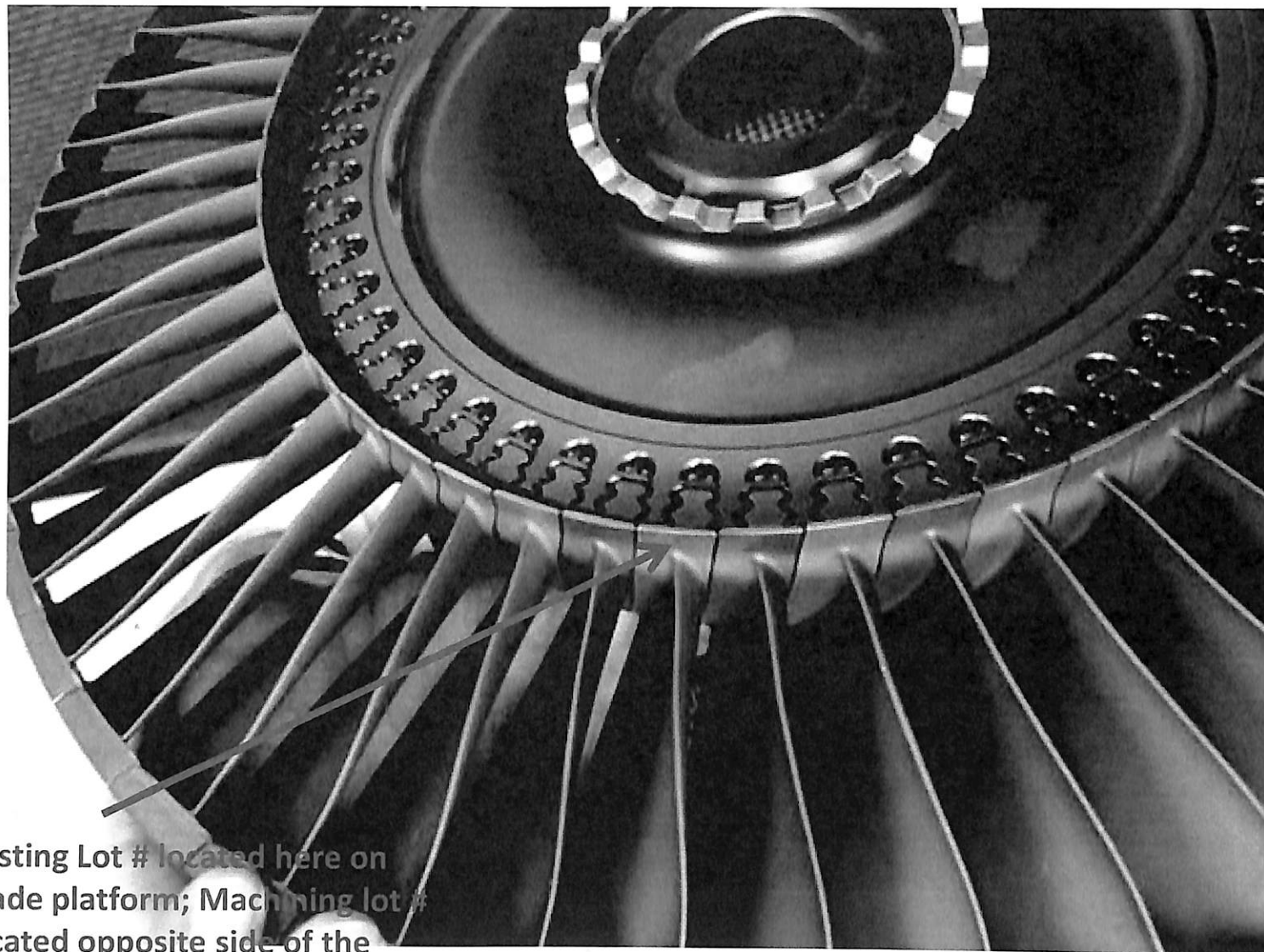
Attachments

Parts [1]

| SEQ | Part       | FindNo | Name                                  | Qty | AltQty |
|-----|------------|--------|---------------------------------------|-----|--------|
| 1   | 70060034-1 | 0006   | TURBINE ROTOR ASSY - LP, SECOND STAGE | 1   |        |

Parts [1]

Tools [0]



Casting Lot # located here on  
blade platform; Machining lot #  
located opposite side of the  
blade platform

# Refer to Supplemental Instructions

- Refer to the following supplemental instructions to eddy current inspect the rotor assembly or replace individual blades.
  - When the inspection or blade replacement is complete, proceed to the next slide for reassembly
- If a replacement rotor assembly was received, proceed to the next slide for reassembly.

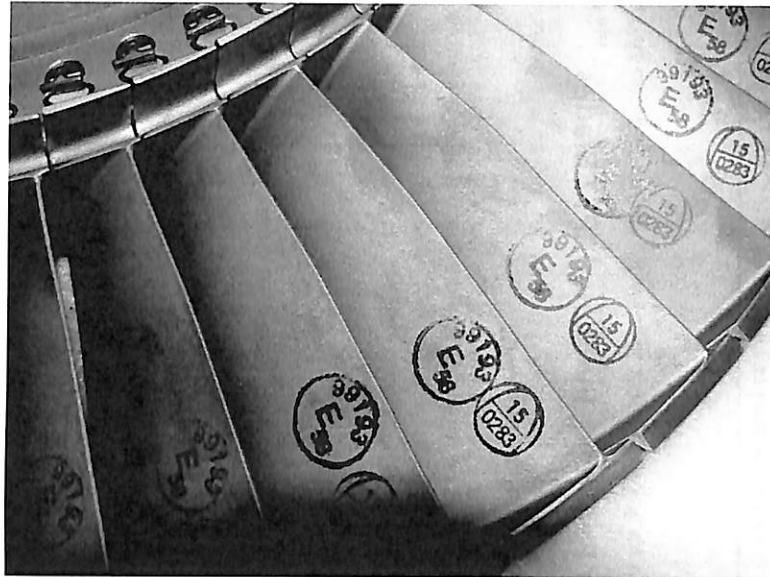
# Reassembly

- Follow disassembly instructions in reverse.
  - Record serial number of LPT2 rotor removed and forward back to Phoenix.
  - Record serial number of LPT2 rotor installed.
    - Record the casting lot numbers (machining lot numbers optional)
    - Validate that each blade on rotor to be installed contains an eddy current (E, ET, EC) inspection stamp – see examples on next slide. If there is no stamp, blade must be eddy current inspected.
  - NOTE: Re-use E-seal (PN 651-534-9612) unless damaged
    - Dents, distortion, cracks or other damage that would cause leaks
  - Perform ground run of engine to verify proper engine operation, leak check and vibration survey
  - When completed, make log book entry per FSE instructions
  - Make entry on LPT2 Life Limit Card: Date of inspection, signature and stamp of certified eddy current inspector (see below)

| LIFE LIMITED PART REPAIR RECORD |                         |                      |
|---------------------------------|-------------------------|----------------------|
| DATE                            | MAINTENANCE PERFORMED   | AUTHORIZED SIGNATURE |
| 6/24/2014                       | Inspection per 21-15477 | Name and UQR stamp   |

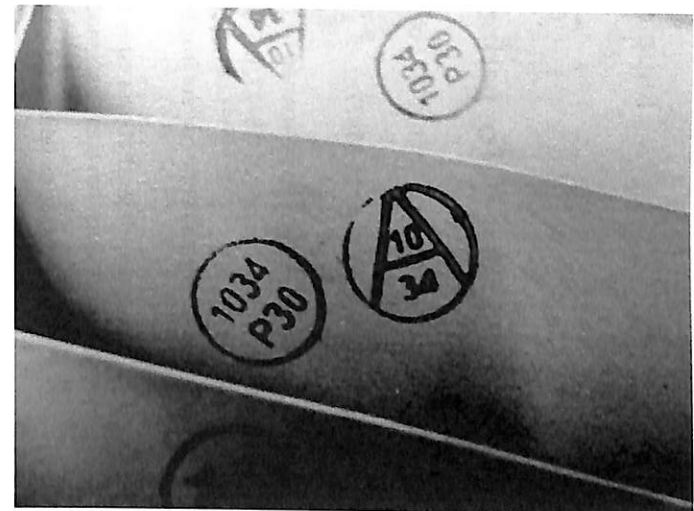


Stamps containing E, ET, EC indicate Eddy Current.



Above – E58 stamp indicates eddy current inspection

Right – These blades have not been  
eddy current inspected



## Supplemental Instructions

### (A) For performing Eddy Current Inspection

Perform eddy current inspection of the LPT2 blades installed in the rotor assembly as follows:

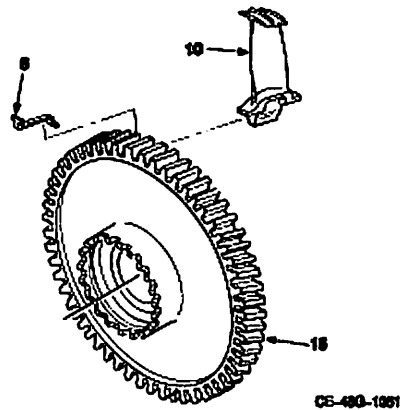
1. Reference Procedure 21-15477.
2. The procedure must be performed by a trained and certified inspector using the equipment and processes specified in procedure 21-15477. Contact Honeywell for a list of certified inspectors.
3. Any blades that fail eddy current inspection shall be replaced by the instructions below.
4. LPT2 Rotor assemblies that pass eddy current inspection (having replaced blades or not) for installation into an engine must have an entry made on the back of the Life Limit Card as follows:
  - a. DATE – date on which the inspection was performed
  - b. Signature and Stamp – shall be that of the eddy current inspector

| LIFE LIMITED PART REPAIR RECORD |                         |                      |
|---------------------------------|-------------------------|----------------------|
| DATE                            | MAINTENANCE PERFORMED   | AUTHORIZED SIGNATURE |
| 6/24/2014                       | Inspection per 21-15477 | Name and UQR stamp   |

### (B) For Replacing Individual LPT2 Blades

1. Remove all blades from the LPT2 rotor assembly

NOTE: Chart the location of all blades on rotor disk to aid balancing after assembly.
2. Chart location of all blades to rotor disk (i.e. using numbers etc.)
3. Remove all turbine blades:
  - a. Bend blade retainers (5, figure 1) away from base of turbine blades (10). Slide turbine blades out, then remove and discard blade retainers.
  - b. Weigh turbine blades that are to be replaced for selecting replacement turbine blades near the same weight.



- 5. BLADE RETAINER
- 10. TURBINE BLADE

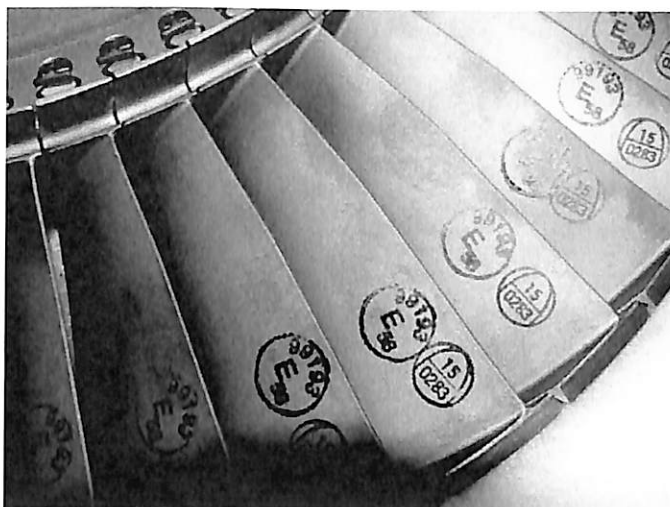
- 15. ROTOR DISC

Figure 1 - LPT2 Rotor Assembly

**NOTE:** Record part number, casting lot number and machining lot number of removed and replacement blades; use pages 3 and 4 of this work instruction.

Record part number and casting lot number of replacement blades on Life Limited Parts Log Card.

**NOTE:** Validate that each replacement blade contains an eddy current (E, ET, EC) inspection stamp; see examples below:



4. Install turbine blades on LPT2 rotor assembly:

- a. With all turbine blades removed, perform single plane balance of LPT2 rotor disk. Mark heavy point and record unbalance in grams for use in selecting blades during assembly.
- b. Select replacement turbine blade as near to the weight of removed turbine blade as possible.
- c. Record on Maintenance Record of Life Limited Part Log Card replacement blade part number followed by casting lot number, date and rotor assembly cycles.

**CAUTION:** DO NOT BEND OR REFORM BLADE RETAINERS PRIOR TO INSTALLATION. ANY REFORMING OR BENDING OF BLADE RETAINERS WILL RESULT IN INCREASED STRESS IN THE INSTALLED RETAINER AND COULD RESULT IN RETAINER CRACKS, SEPARATION, AND POSSIBLE ENGINE DAMAGE.

**NOTE:** Do not reuse any blade retainers.

- d. Install blade retainers into LPT2 rotor assembly rotor disk with all preformed tabs on the same side of rotor disc (figure 2).

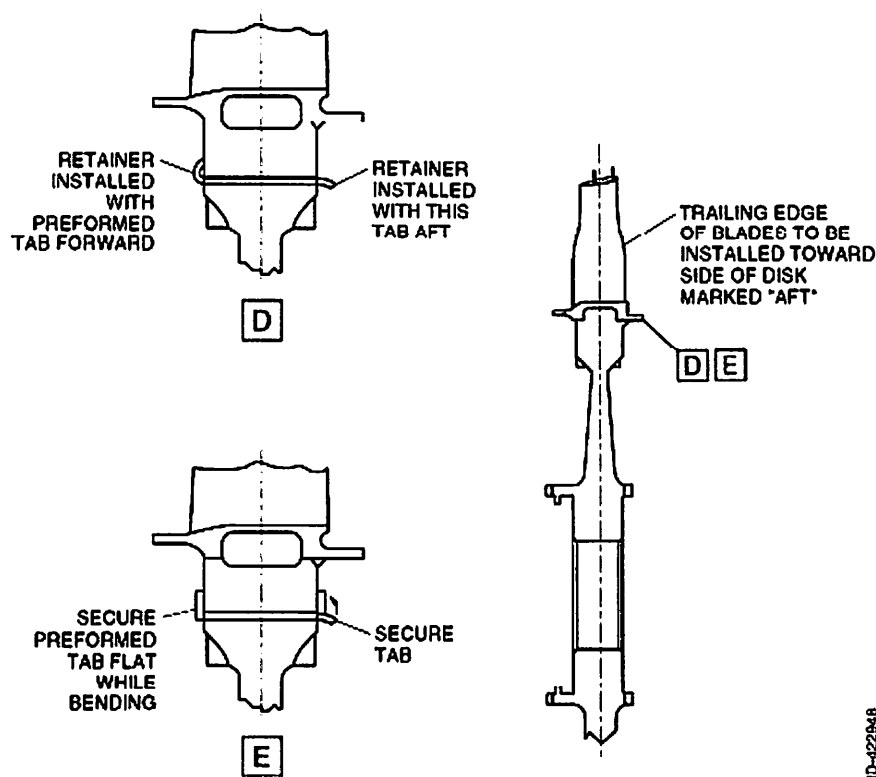


Figure 2 - Blade Retainer Orientation

**CAUTION:** DO NOT FORCE TURBINE BLADES INTO ROTOR DISC. USE ONLY HAND PRESSURE WHEN SLIDING TURBINE BLADES INTO FIR TREE OF ROTOR DISC AND OVER BLADE RETAINERS. IF RESISTANCE IS PRESENT, INVESTIGATE CAUSE (E.G. BENT RETAINER, BURR ON BLADE RETAINER GROOVE, ETC.).

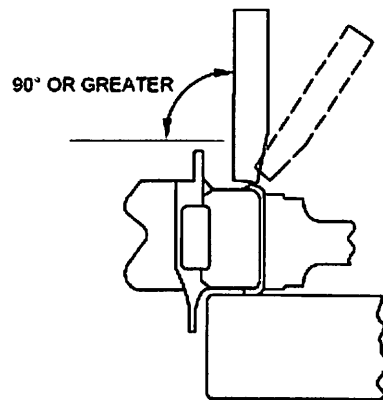
- e. Install blades provided by Honeywell (PN 3075424-2 or -3) as follows:
  - i. Install all turbine blades into rotor disc at location previously charted in Step 2.
  - ii. To install last turbine blade, move approximately 12 to 20 blades (six to ten on either side) opposite direction of preformed end of blade retainer. Start the last turbine blade into rotor disc with Z-gap aligned with adjacent turbine blade, then move all turbine blades in unison forward into rotor disc.
- f. Check for misalignment of knife edges on LPT2 rotor assembly turbine blades. Misalignment of knife edges of adjacent turbine blades must not exceed **0.068 inch (1.73 mm)**.

**CAUTION:** TOOLING USED TO BEND RETAINER MUST BE ALUMINUM OR NICKEL. TOOLING OTHER THAN ALUMINUM OR NICKEL COULD RESULT IN MATERIAL TRANSFER FROM TOOL TO ROTOR DISC OR TURBINE BLADE CAUSING SUBSEQUENT FAILURE OF ROTOR DISC OR TURBINE BLADE.

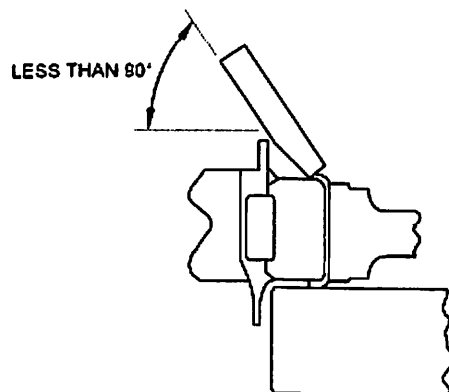
- g. Bend tab of blade retainer on LPT2 rotor assembly as follows:

**NOTE:** Preformed tab of blade retainer (figure 2) must be held flush against face of rotor disc and turbine blade before opposite tab is bent up and into face of disc.

- i. With preformed tab of blade retainer held flush against face of rotor disc, bend tab as shown in figure 2. When forming retainer, do not apply force at less than a 90 degree angle from the rotor disc face; see figure 3.
- ii. Make sure after bending, tab is in firm contact with turbine blade, but not sharply creased at bend line.
- iii. Inspect bent tabs with 10X magnification and if blade retainer tabs are cracked, replace blade retainer. Removal of a cracked blade retainer requires removal and replacement of all blade retainers.



**NOTE:**  
**Acceptable forming technique** - End of tab set with tool held at 90 degrees or greater from the disc axial face.



**NOTE:**  
**Unacceptable forming technique** - End of tab set with tool held less than 90 degrees from the disc axial face. This may create a tighter axial fit of the blade, but results in a reduced retainer bend radius, and a high residual stress condition.

Figure 3 - Blade Retainer Installation

- h. Check turbine blades and blade retainers after installation is completed on LPT2 rotor assembly as follows:
  - i. Check for relative movement between the turbine blade and rotor disc. Relative movement between the turbine blade and rotor disc along the fir tree center axis shall not exceed **0.025 inch (0.64 mm)**.
  - i. Blade retainers may not be reused (reformed) after they have been formed, with the exception of the adjustments that follows. When adjusting blade retainer, do not force at less than a 90 degree angle from the rotor disc face; see figure 3.
    - i. If relative movement is greater than the limit in step h(i), adjust the blade retainer using an approved form tool. Blade retainer can be adjusted only one time to obtain specified value.
    - ii. If blade retainer was adjusted, inspect adjusted tabs with 10X magnification. Cracks are not acceptable.
    - iii. If blade retainer tabs are cracked, replace blade retainer. Removal of a cracked blade retainer requires removal and replacement of all blade retainers.
- 5. Measure and record rotor assembly diameter using LPT blade loading kit 298125-1. Pay particular attention to the locations where the replacement blades are installed.

LPT2 rotor assembly diameter = **15.212 +/- 0.004 inches (386.385 +/- 0.102 mm)**.

If diameter is not met, contact engineering.

- 6. Balance assembly of rotor disc, turbine blades and blade retainers using components of 289905-2 rotor assembly balancing kit as follows:
  - a. Install 290701-1 coupling or 831413-2 adapter on LPT2 rotor assembly forward curvic and 831413-1 adapter on aft curvic. Secure with 290566-4 or 290566-6 tie bolt and 290565-1 nut.
  - b. Install assembly in standard balancing machine.

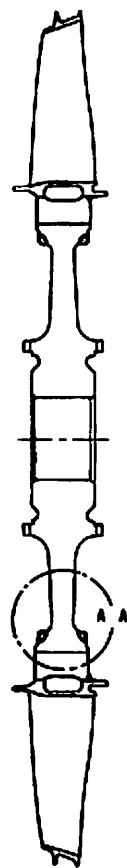
**NOTE:** If unbalance is greater than **0.3 ounce-inch**, reposition blades for optimum balance prior to any grinding of balance material.

When removing material for balancing, two areas at least 1 inch wide and 180 degrees apart must remain without material removal to permit dimensional inspection of rotor assembly.

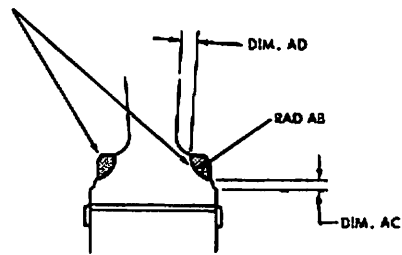
- c. Balance LPT2 rotor assembly on curvic couplings to an accuracy of **0.059 ounce-inch**. Material removal shall leave no corners or edges sharper than **0.125 inch (3.175 mm)** radius and shall have a surface finish of **63 microinches** or better; see figure 4.
- d. Remove LPT2 rotor assembly from standard balancing machine and remove balance kit parts.

**NOTE:** If replacing damaged blade retainers only, balancing of the rotor assembly is not required if the same turbine blades removed are reinstalled in the same location charted at removal.





MATERIAL MAY BE REMOVED FROM THIS AREA FOR BALANCING. DO NOT UNDERCUT BASIC PROFILE. BASIC PROFILE OF DISC IS FORMED BY RADIUS AB, AND STEP DIMENSIONS AC AND AD. REMOVE MATERIAL EQUALLY FROM BOTH SIDES. NO MATERIAL REMOVAL IN TWO AREAS ONE INCH WIDE, 180° APART, TO PERMIT DISC DIMENSIONAL INSPECTION.



DETAIL A A

FIRST AND SECOND STAGE ROTOR ASSEMBLY

1-680-101

#### REPAIRED LIMITS

|        |                              |
|--------|------------------------------|
| RAD AB | 0.200 inch (5.08 mm) minimum |
| DIM AC | 0.030 inch (0.76 mm) minimum |
| DIM AD | 0.040 inch (1.02 mm) minimum |

Figure 4 - Rotor Assembly Balancing Requirements

ENGINE ☒ MODULE ☐

SERIAL NUMBER P137180

## SERVICE RECORD

PAGE 2

DESCRIPTION OF INSPECTIONS, REPAIRS AND OVERHAULS

ENDORSE ALL INSPECTIONS AND REPAIRS WITH NAME, RATING AND CERTIFICATE NUMBER

| DATE                                          | HOURS THIS FLIGHT | CYCLES THIS FLIGHT | ACCUMULATIVE TOTALS                                                                    |           |           |           | DESCRIPTION OF WORK PERFORMED                           |
|-----------------------------------------------|-------------------|--------------------|----------------------------------------------------------------------------------------|-----------|-----------|-----------|---------------------------------------------------------|
|                                               |                   |                    | HOURS                                                                                  |           | CYCLES    |           |                                                         |
|                                               |                   |                    | SINCE NEW                                                                              | SINCE O/H | SINCE NEW | SINCE O/H |                                                         |
|                                               |                   |                    |                                                                                        |           |           |           | TOTALS BROUGHT FORWARD                                  |
|                                               |                   |                    |                                                                                        |           |           |           | Removed From 45-485 To Reallocate to 45-496 16 AUG 2014 |
|                                               |                   |                    |                                                                                        |           |           |           |                                                         |
| 18 Aug 2014                                   |                   |                    |                                                                                        |           |           |           | THIS ENGINE SERVICED WITH Mobil 1 75W-90 OIL            |
|                                               |                   |                    |                                                                                        |           |           |           | ENGINE S/N P137180                                      |
|                                               |                   |                    |                                                                                        |           |           |           | DATE 18 AUG 2014 LEARJET QA                             |
|                                               |                   |                    |                                                                                        |           |           |           | INSTALLED ON ACFT 45-496                                |
|                                               |                   |                    |                                                                                        |           |           |           | ON THE RIGHT SIDE                                       |
|                                               |                   |                    |                                                                                        |           |           |           | DATE 8-21-14 LEARJET QA                                 |
| 13-Nov 2014                                   |                   |                    |                                                                                        |           |           |           | Accomplished FSE WE 731-40-P14-002B IAW HON PC413NM.    |
|                                               |                   |                    |                                                                                        |           |           |           | Replaced LPT2 Assy.                                     |
|                                               |                   |                    |                                                                                        |           |           |           | PN 3060608-5                                            |
|                                               |                   |                    |                                                                                        |           |           |           | SN Off: 12-156103-00730                                 |
|                                               |                   |                    |                                                                                        |           |           |           | SN On: 12-156103-00810                                  |
| 22 Nov 14                                     |                   |                    |                                                                                        |           |           |           | Jim Beardsley HON/FSE                                   |
| TOTAL PRODUCTION FLIGHT TEST TIME             |                   |                    | HOURS: 5.9                                                                             |           | CYCLES: 5 |           |                                                         |
| SIGNATURE:                                    |                   |                    | DATE: November 22, 2014                                                                |           |           |           |                                                         |
| BRENT A. DUNGAN<br>Quality Assurance Engineer |                   |                    | LEARJET INC.<br>One Learjet Way, Wichita, KS 67209<br>FAA Production Certificate 329CE |           |           |           |                                                         |
|                                               |                   |                    | 5.9                                                                                    |           | 5         |           | TOTALS TO DATE                                          |





**SERVICE BULLETIN  
RECORD**

## **Aircraft Customer Engine Production Applicability Report**

According to manufacturer records, the service bulletins listed in this report are applicable to this engine/APU as of the date of this report. After engine/APU delivery, the record of service bulletin compliance depends on compliance reports being forwarded to Honeywell from the service facilities.

NOTE: Check all service bulletins issued after the date of this report for applicability to this engine/APU.

Information regarding this report may be obtained by contacting:

**The Honeywell Aero Customer Care Center**  
**PHONE US/Canada: 800-601-3099    International: 602-365-3099**  
**Fax: 602-365-3345**

**Measurement & Reporting**  
**Reliability, Maintainability & System Safety**  
**Honeywell International, Inc.**

RMSS Measurement & Reporting, Honeywell Aerospace  
Production Applicability, Sorted by Status [#APP105]

Printed on: 02/05/2014

for Engine Serial Number:137180 Model: TFE731-40BR-1B

Run by: E384386

| Status | Number | Latest<br>Rev. | Cmp. | Title | Effectivity |
|--------|--------|----------------|------|-------|-------------|
|--------|--------|----------------|------|-------|-------------|

Status Codes: P = Production incorporation  
S = See service bulletin for effectivity

Effectivity: NA= Not applicable by serial number.  
See service bulletin for effectivity

Compliance Code: R = Recommended Service Bulletin  
O = Optional Service Bulletin  
I = Informative Service Bulletin





**Georgia Crown Aviation Department**

Lear 75  
N877W  
45-496

201 Airport Parkway  
LaGrange, Georgia 31909

Aircraft TTSN: 913.3      Aircraft TCSN: 913  
Engine 2 Part Number: 3062075-1  
Engine 2 Model: TFE731-40BR-1B  
Engine 2 Serial Number: P137180  
Engine TTSN: 913.3  
Engine TCSN: 913  
Date: October 25, 2016

Completed the following tasks:

TFE731-73-5145  
Paragraph E(2)

Fuel Flow Transmitter Line and  
Bracket Service Bulletin  
Inspection

I certify that this engine has been inspected in accordance with current Federal Aviation Regulation Part 91.409 (F) (3) and current aircraft and engine manufacturer's maintenance manuals. All functional and operational checks were satisfactory and this aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: **3481311**

Operator: **GEORGIA CROWN DISTRIBUTING CO.**

Work Card No.: **TFE731-73-5145 PARA E (2)**

Serial No.: **P137180**

Model: **TFE731-40BR-1B**

Part: **3062075-1**

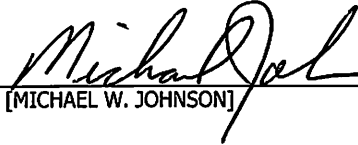
Workorder No.: \_\_\_\_\_

A/C Serial **45-496**

Reg No.: **N877W**

|              | Date(DD/MMM/YYYY) | A/C HRS      | AFL | ENG. HRS     | ENC |  |  |
|--------------|-------------------|--------------|-----|--------------|-----|--|--|
| Due At       |                   | <b>910.9</b> |     | <b>910.9</b> |     |  |  |
| Accomplished | 25-OCT-16         | 913.3        | 913 |              |     |  |  |

TECHNICIAN SIGNATURE:



[MICHAEL W. JOHNSON]

KIND OF CERTIFICATE & NO.: **AP3481311**
**>SB TFE731-73-5145  
PARA E (2)**
**NO. 2 ENGINE - ACCOMPLISH, SERVICE BULLETIN INSPECTION**


COMPLIED WITH



DECLINED



DEFERRED



NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.

Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: Performed recurring 25 Hour Inspection per TFE731-73-5145 part 3B.(1)(2). No defects noted. Next Inspection due 938.3.

TECH



INSP

LABOR-HRS  
HRS.THS



**Georgia Crown Aviation Department**

Lear 75  
N877W  
45-496

201 Airport Parkway  
LaGrange Georgia 30240

Part Number 3062075-1  
Model TFE731-40BR-1B  
Engine TTSN: 915.5

Serial number P137180  
Engine TCSN: 917

Date: November 4, 2016

TFE731-73-5145 PARA E (3) NO. 2 ENGINE - ACCOMPLISH, SERVICE BULLETIN REPLACEMENT  
**Method of Compliance:** COMPLIED WITH  
**Update Note:** COMPLETED SB PER SERVICE BULLETIN INSTRUCTIONS.

I certify that this engine has been inspected or repaired in accordance with current Federal Aviation Regulation Part 91.409 (F) (3). All test operational tests were performed in accordance with the manufacturer's engine maintenance manual or component maintenance manual. This aircraft is approved for return to service.

Michael W. Johnson  
Aircraft Maintenance Technician  
Airframe and Powerplant License Number: **3481311**



**BOMBARDIER**

Operator: **GEORGIA CROWN DISTRIBUTING CO.**

Work Card No.: **TFE731-73-5145 PARA E (3)**

Serial No.: **P137180**

Model: **TFE731-40BR-1B**

Workorder No.: \_\_\_\_\_

Part: **3062075-1**

A/C Serial **45-496**

Reg No.: **N877W**

| Due At       | Date(DD/MMM/YYYY) | A/C HRS | AFL | ENG. HRS | ENC |  |  |
|--------------|-------------------|---------|-----|----------|-----|--|--|
| Accomplished | 04-NOV-16         | 915.50  | 917 |          |     |  |  |

TECHNICIAN SIGNATURE: \_\_\_\_\_

[MICHAEL W. JOHNSON]

KIND OF CERTIFICATE & NO.: **AP3481311**

SB TFE731-73-5145 PARA E  
(3)

NO. 2 ENGINE - ACCOMPLISH, SERVICE BULLETIN REPLACEMENT



COMPLIED WITH



DECLINED



DEFERRED



NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: \_\_\_\_\_

TECH

MJ

INSP \_\_\_\_\_

LABOR-HRS  
HRS.THS \_\_\_\_\_

10-NOV-2016

WID: 21451449-1

Return completed copy via email to fax@campsystems.com

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Total Pages = \_\_\_\_\_, Initial **1** of **1**

|                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  |                                        |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                          |                               |                                                                     |                                |
|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------------------------------|---------------------------------------------------------------------------------------------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-------------------------------|---------------------------------------------------------------------|--------------------------------|
| 1. Approving Civil Aviation Authority/Country:<br><b>FAA/United States</b>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       |                                        | 2.<br><b>AUTHORIZED RELEASE CERTIFICATE</b><br><b>FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</b> |                                                                                                                                                                                                                                                                                                                                                                                                                          |                               | 3. Form Tracking Number:<br>20160002182774Y14<br>8005829392-20      |                                |
| 4. Organization Name and Address: HONEYWELL INTERNATIONAL INC Production Approval<br>1944 E SKY HARBOR CIRCLE PC413NM<br>PHOENIX AZ 85034                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        |                                        |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                          |                               | 5. Work Order/Contract/Invoice Number:<br>5001084364<br>Page 1 of 1 |                                |
| 6. Item:                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         | 7. Description:                        | 8. Part Number:                                                                                   | 9. Quantity:                                                                                                                                                                                                                                                                                                                                                                                                             | 10. Serial Number:            | 11. Status / Work:                                                  |                                |
| 001                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              | TUBE, METAL, FLOW DIVIDER - FLOW METER | 70060172-1                                                                                        | 2                                                                                                                                                                                                                                                                                                                                                                                                                        | N/A                           | NEW                                                                 |                                |
| 12. Remarks:<br>EXPORT AIRWORTHINESS APPROVAL: THIS ARTICLE MEETS THE SPECIAL REQUIREMENTS OF CANADA.<br>SERIALIZATION NOT REFERENCED IN BLOCK 10 IS NOT REQUIRED OR TRACEABLE BY HONEYWELL INTERNATIONAL INC. REFERENCE FAA ORDER 8130.21                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       |                                        |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                          |                               |                                                                     |                                |
| 13a. Certifies the items identified above were manufactured in conformity to:<br><input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation.<br><input type="checkbox"/> Non-approved design data specified in Block 12.                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 |                                        |                                                                                                   | 14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12<br>Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service. |                               |                                                                     |                                |
| 13b. Authorized Signature:<br><i>Florie McCrone</i>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              |                                        | 13c. Approval/Authorization No.:<br><b>ODA-602216-NM</b>                                          |                                                                                                                                                                                                                                                                                                                                                                                                                          | 14b. Authorized Signature:    |                                                                     | 14c. Approval/Certificate No.: |
| 13d. Name (Typed or Printed):<br><b>Florie McCrone</b>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           |                                        | 13e. Date (dd/mm/yyyy):<br><b>24/OCT/2016</b>                                                     |                                                                                                                                                                                                                                                                                                                                                                                                                          | 14d. Name (Typed or Printed): |                                                                     | 14e. Date (dd/mm/yyyy):        |
| <b>User / Installer Responsibilities</b><br>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown. |                                        |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                          |                               |                                                                     |                                |

|                                                                                                                                                                                                                                                                  |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                |                                |                                                                |  |
|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------------------|---------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--------------------------------|----------------------------------------------------------------|--|
| 1. Approving Civil Aviation Authority/Country:<br><b>FAA/United States</b>                                                                                                                                                                                       |                     | 2.<br><b>AUTHORIZED RELEASE CERTIFICATE</b><br><b>FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</b> |                                                                                                                                                                                                                                                                                                                                                                                                                |                                | 3. Form Tracking Number:<br>20160002182773Y14<br>8005829392-10 |  |
| 4. Organization Name and Address: HONEYWELL INTERNATIONAL INC<br>1944 E SKY HARBOR CIRCLE<br>PHOENIX AZ 85034                                                                                                                                                    |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                | Production Approval<br>PC413NM |                                                                |  |
| 5. Work Order/Contract/Invoice Number:<br>5001084364<br>Page 1 of 1                                                                                                                                                                                              |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                |                                |                                                                |  |
| 6. Item:                                                                                                                                                                                                                                                         | 7. Description:     | 8. Part Number:                                                                                   | 9. Quantity:                                                                                                                                                                                                                                                                                                                                                                                                   | 10. Serial Number:             | 11. Status / Work:                                             |  |
| 001                                                                                                                                                                                                                                                              | TUBE ASSEMBLY METAL | 3061288-1                                                                                         | 2                                                                                                                                                                                                                                                                                                                                                                                                              | N/A                            | NEW                                                            |  |
| 12. Remarks:<br>EXPORT AIRWORTHINESS APPROVAL: THIS ARTICLE MEETS THE SPECIAL REQUIREMENTS OF CANADA.<br>SERIALIZATION NOT REFERENCED IN BLOCK 10 IS NOT REQUIRED OR TRACEABLE BY HONEYWELL INTERNATIONAL INC. REFERENCE FAA ORDER 8130.21                       |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                |                                |                                                                |  |
| 13a. Certifies the items identified above were manufactured in conformity to:<br><input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation.<br><input type="checkbox"/> Non-approved design data specified in Block 12. |                     |                                                                                                   | 14a. <input type="checkbox"/> 14. CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12<br>Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service. |                                |                                                                |  |
| 13b. Authorized Signature:<br><i>Florie McCrone</i>                                                                                                                                                                                                              |                     | 13c. Approval/Authorization No.:<br><b>ODA-602216-NM</b>                                          |                                                                                                                                                                                                                                                                                                                                                                                                                | 14b. Authorized Signature:     |                                                                |  |
| 13d. Name (Typed or Printed):<br><b>Florie McCrone</b>                                                                                                                                                                                                           |                     | 13e. Date (dd/mm/yyyy):<br><b>24/OCT/2016</b>                                                     |                                                                                                                                                                                                                                                                                                                                                                                                                | 14c. Approval/Certificate No.: |                                                                |  |
|                                                                                                                                                                                                                                                                  |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                | 14d. Name (Typed or Printed):  |                                                                |  |
|                                                                                                                                                                                                                                                                  |                     |                                                                                                   |                                                                                                                                                                                                                                                                                                                                                                                                                | 14e. Date (dd/mm/yyyy):        |                                                                |  |

### User / Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

**CERTIFIED TRUE COPY** of the original certificate retained on file with Bombardier Q.A.

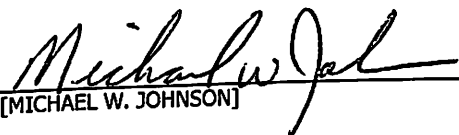
An agency may not conduct or sponsor, and a person is not required to respond to, a collection of information unless it displays a currently valid OMB control number associated with this collection of information is 2120-0018. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW, Washington, DC 20591, Attn: Information Collection Clearance Officer, ABA-20





**BOMBARDIER**

|                                                 |                   |                          |     |                                   |  |  |  |
|-------------------------------------------------|-------------------|--------------------------|-----|-----------------------------------|--|--|--|
| Operator: <b>GEORGIA CROWN DISTRIBUTING CO.</b> |                   |                          |     | Work Card No.: <b>75-73-01 R1</b> |  |  |  |
| Serial No.: <b>45-496</b>                       |                   | Model: <b>LEARJET 75</b> |     | Workorder No.: _____              |  |  |  |
| Reg No.: <b>N877W</b>                           |                   |                          |     |                                   |  |  |  |
| Due At                                          | Date(DD/MMM/YYYY) | A/C HRS                  | AFL | APH                               |  |  |  |
|                                                 |                   |                          |     |                                   |  |  |  |
| Accomplished                                    | 04-NOV-16         | 915.5                    | 917 |                                   |  |  |  |

TECHNICIAN SIGNATURE:  KIND OF CERTIFICATE & NO.: AP3481311  
[MICHAEL W. JOHNSON]

SB 75-73-01 R1

ENGINE FUEL AND CONTROL - ENGINE FUEL FLOW TRANSDUCER BRACKET REPLACEMENT, SERVICE BULLETIN

☒ COMPLIED WITH      ☐ DECLINED      ☐ DEFERRED      ☐ NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

TECH MS INSP \_\_\_\_\_ LABOR-HRS \_\_\_\_\_  
HRS.THS \_\_\_\_\_

CERTIFIED TRUE COPY of the original certificate retained on file with Bombardier Q.A.

An agency may not conduct or sponsor, and a person is not required to respond to, a collection of information unless it displays a currently valid OMB control number associated with this collection of information is 2120-0018. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW, Washington, DC 20591. Attn: Information Collection Clearance Officer. ABA-20



# StandardAero

Page 1 of 1

## SERVICE BULLETIN STATUS THIS VISIT

ENGINE SERIAL: P 137180 ENGINE MODEL NO: TFE731-40BR-1B 3062075-1

TSN: 1193.3 CSN: 1157

| STATUS | DATE     | BULLETIN | REV | CAT | DESCRIPTION                                                  |
|--------|----------|----------|-----|-----|--------------------------------------------------------------|
| C/W    | 7/3/2017 | 72-5247  | R0  | 2   | REPLACE FAN STATOR VANE ASSY 3060163-3/-4/-5 WITH 70060130-1 |
| N/A    | 6/8/2017 | 72-5251  | R0  | 2   | E/C SPECIFIC LPT2 BLADES P/N 3075424                         |
| N/A    | 6/8/2017 | 72-5256  | R0  | 1   | IDENTIFY AND REPLACE SPECIFIC FAN DISCS P/N 3060287-2        |

StandardAero, AUGUSTA, GA  
Certified Repair Station GW4R221M

By:   
7/3/2017

LEGEND: C/W=Complied With; PCW= Previously Complied With; N/A= Not Applicable; DPC= Due At Part Change; NAT= Not Applicable At This Time

GEORGIA CROWN DISTRIBUTING CO.  
5290 ARMOUR ROAD  
COLUMBUS, GA, USA, 31909

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 04-Oct-2017 | 1339.7  | 1279    |           |           | 1290.3    | 1235      |         |            |  |  |

| No. | Task No./Position | Description                                                                                                                                                                                                                                     |
|-----|-------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 1   | TFE731-72-5221    | NO. 2 ENGINE - REPLACE SPECIFIC IDENTIFIED LOW PRESSURE SECOND STAGE TURBINE (LPT2) ROTOR<br>BLADES, PN 3075424-2, -3, SERVICE BULLETIN<br><b>Method Of Compliance:</b> NOT APPLICABLE<br><b>Update Note:</b> NA PER ENGINE MODEL LISTED IN SB. |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F) (3) AND  
APPROVED FOR RETURN TO SERVICE

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:



**BOMBARDIER**

Operator: **GEORGIA CROWN DISTRIBUTING CO.**  
 Serial No.: **P137180** Model: **TFE731-40BR-1B**  
 Part: **3062075-1**  
 A/C Serial **45-496** Reg No.: **N877W**

Work Card No.: **TFE731-72-5221**

Workorder No.: \_\_\_\_\_

| Due At       | Date(DD/MMM/YYYY) | A/C HRS | AFL  | ENG. HRS | ENC |  |  |
|--------------|-------------------|---------|------|----------|-----|--|--|
| Accomplished | 04-OCT-17         | 1339.70 | 1279 |          |     |  |  |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F) (3) AND APPROVED FOR RETURN TO SERVICE

TECHNICIAN SIGNATURE: *Michael W. Johnson* KIND OF CERTIFICATE & NO.: AP3481311  
 [MICHAEL W. JOHNSON]

FAA 2012-16-14

NO. 2 ENGINE - ENGINE - LPT2 ROTOR BLADES - REMOVE/INSPECT

Amendment No: \_\_\_\_\_ Effective Date: 01-OCT-2012 Next Compliance Due Date Hours/Other: \_\_\_\_\_

☐ COMPLIED WITH ☐ DECLINED ☐ DEFERRED ☒ NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
 Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: \_\_\_\_\_  
 \_\_\_\_\_  
 \_\_\_\_\_

TECH *MS* INSP \_\_\_\_\_ LABOR-HRS \_\_\_\_\_  
 HRS.THS \_\_\_\_\_

SB TFE731-72-5221

NO. 2 ENGINE - REPLACE SPECIFIC IDENTIFIED LOW PRESSURE SECOND STAGE TURBINE (LPT2) ROTOR BLADES, PN 3075424-2, -3, SERVICE BULLETIN

☐ COMPLIED WITH ☐ DECLINED ☐ DEFERRED ☒ NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
 Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: \_\_\_\_\_  
 \_\_\_\_\_  
 \_\_\_\_\_

TECH *MS* INSP \_\_\_\_\_ LABOR-HRS \_\_\_\_\_  
 HRS.THS \_\_\_\_\_

|                                                 |                              |                                      |
|-------------------------------------------------|------------------------------|--------------------------------------|
| Operator: <b>GEORGIA CROWN DISTRIBUTING CO.</b> |                              | Work Card No.: <b>TFE731-72-5221</b> |
| Serial No.: <b>P137180</b>                      | Model: <b>TFE731-40BR-1B</b> | Workorder No.: _____                 |
| Part: <b>3062075-1</b>                          | Reg No.: <b>N877W</b>        |                                      |
| A/C Serial <b>45-496</b>                        |                              |                                      |

| Due At       | Date(DD/MMM/YYYY) | A/C HRS | AFL  | ENG. HRS | ENC |  |  |
|--------------|-------------------|---------|------|----------|-----|--|--|
| Accomplished | 04-OCT-17         | 1339.70 | 1279 |          |     |  |  |

|        |                                                          |              |
|--------|----------------------------------------------------------|--------------|
| 725020 | NO. 2 ENGINE - 2ND-STAGE LOW PRESSURE TURBINE (LPT) DISC | LMM 72-50-04 |
|--------|----------------------------------------------------------|--------------|

**LIFE LIMIT**

|                                |                                       |                                  |                                        |                                      |                                                                                                             |
|--------------------------------|---------------------------------------|----------------------------------|----------------------------------------|--------------------------------------|-------------------------------------------------------------------------------------------------------------|
| REASON REMOVED:<br>(CHECK ONE) | <input type="checkbox"/> TIME EXPIRED | <input type="checkbox"/> FAILURE | <input type="checkbox"/> WORN          | <input type="checkbox"/> LOANER      | <input type="checkbox"/> SCHEDULING CONV                                                                    |
|                                | <input type="checkbox"/> MOD/UPGRADE  | <input type="checkbox"/> SERVICE | <input type="checkbox"/> ENGINE CHANGE | <input type="checkbox"/> TIRE CHANGE | <input type="checkbox"/> SWAP/TRBLE SHOOT <input type="checkbox"/> DAMAGED <input type="checkbox"/> UNKNOWN |

*If removed P/N & S/N information is incorrect please provide details below.*

|               |           |  |               |                 |  |                         |      |
|---------------|-----------|--|---------------|-----------------|--|-------------------------|------|
| REMOVED P/N   | 3060611-1 |  | S/N           | 12-156103-00810 |  | LABOR-HRS               |      |
| INSTALLED P/N |           |  | S/N           |                 |  | PART COST\$             |      |
| INSTALLED TSN | MOS       |  | INSTALLED TSO | MOS             |  | TIME SINCE REPAIR       | MOS  |
|               | HRS       |  |               | HRS             |  |                         | HRS  |
|               | LDGS      |  |               | LDGS            |  |                         | LDGS |
|               |           |  |               |                 |  | WARRANTY TIME REMAINING |      |

|       |  |       |  |
|-------|--|-------|--|
| TECH: |  | INSP: |  |
|-------|--|-------|--|

REMARKS : \_\_\_\_\_

\_\_\_\_\_

\_\_\_\_\_

GEORGIA CROWN DISTRIBUTING CO.  
LAGRANGE CALLAWAY AIRPORT  
203 AIRPORT PARKWAY  
LAGRANGE, GA, USA, 30240

TFE731-40BR-1B 3062075-1 S/N: P-137180

On Aircraft Reg: N877W Serial: 45-496

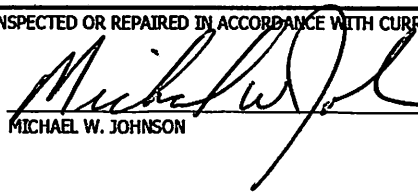
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 21-Sep-2021 | 0       | 0       |           |           | 2949.9    | 2940      |         |            |  |  |

| No. | Task No./Position                      | Description                                                                                                                                                                                                                                     |
|-----|----------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| 1   | TFE731-72-5256 R01 (D201506000058 R01) | NO. 2 ENGINE - BYPASS SECTION - IDENTIFY AND REPLACE SPECIFIC FAN DISKS, PN 3060287-2, WITH SUSPECT SHOT PEENED ROLLED EDGES ON TFE731-20 AND -40 ENGINE MODELS, SERVICE BULLETIN<br>Method Of Compliance: NOT APPLICABLE - NA BY SERIAL NUMBER |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F)(3) AND APPROVED FOR RETURN TO SERVICE.

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

MECHANIC AP3481311

Work Inspected By:

Certificate #:

**AIRWORTHINESS  
DIRECTIVE**



**SERIAL NUMBER** P137180

# AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD

PAGE 1

[illegible]

**This Log is provided for owner/operator use to record A.D. compliance.**

PX3107-52

GEORGIA CROWN DISTRIBUTING CO.  
5290 ARMOUR ROAD  
COLUMBUS, GA, USA, 31909

TFE731-40BR-1B 3062075-1 S/N: P137180

On Aircraft Reg: N877W Serial: 45-496

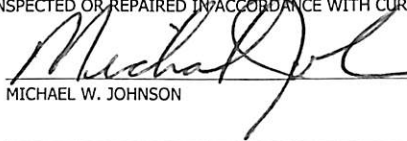
Work Order #:

| Date        | A/C HRS | A/C AFL | ENG 1 HRS | ENG 1 ENC | ENG 2 HRS | ENG 2 ENC | APU HRS | APU Starts |  |  |
|-------------|---------|---------|-----------|-----------|-----------|-----------|---------|------------|--|--|
| 04-Oct-2017 | 1339.7  | 1279    |           |           | 1290.3    | 1235      |         |            |  |  |

| No. | Task No./Position | Description                                                |
|-----|-------------------|------------------------------------------------------------|
| 1   | FAA 2012-16-14    | NO. 2 ENGINE - ENGINE - LPT2 ROTOR BLADES - REMOVE/INSPECT |
|     |                   | Method Of Compliance: NOT APPLICABLE                       |
|     |                   | Update Note: NA PER ENGINE MODEL LISTED IN AD              |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F) (3) AND APPROVED FOR RETURN TO SERVICE

Work Performed By:

  
MICHAEL W. JOHNSON

Certificate #:

AP3481311

Work Inspected By:

Certificate #:

**BOMBARDIER****CHECK CURRENT DUE LIST FOR DUE TIME CHANGES**Operator: **GEORGIA CROWN DISTRIBUTING CO.**Work Card No.: **FAA 2012-16-14**Serial No.: **P137180**Model: **TFE731-40BR-1B**Part: **3062075-1**

Workorder No.: \_\_\_\_\_

A/C Serial **45-496**Reg No.: **N877W**

| Due At       | Date(DD/MMM/YYYY)  | A/C HRS       | AFL  | ENG. HRS    | ENC |  |  |
|--------------|--------------------|---------------|------|-------------|-----|--|--|
|              | <b>01-OCT-2017</b> | <b>3049.4</b> |      | <b>3000</b> |     |  |  |
| Accomplished | 04-OCT-17          | 1339.70       | 1279 |             |     |  |  |

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED OR REPAIRED IN ACCORDANCE WITH CURRENT FEDERAL AVIATION REGULATION 91.409 (F) (3) AND APPROVED FOR RETURN TO SERVICE

TECHNICIAN SIGNATURE: \_\_\_\_\_

[MICHAEL W. JOHNSON]

KIND OF CERTIFICATE & NO.: **AP3481311**> **FAA 2012-16-14****NO. 2 ENGINE - ENGINE - LPT2 ROTOR BLADES - REMOVE/INSPECT**

Amendment No: \_\_\_\_\_

Effective Date: **01-OCT-2012**

Next Compliance Due Date Hours/Other: \_\_\_\_\_

☐ COMPLIED WITH☐ DECLINED☐ DEFERRED☒ NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: AD2012-16-14 is not applicable due to engine model was not listed in AD.AD2012-16-14 is not applicable due to engine model was not listed in AD.AD2012-16-14 is not applicable due to engine model was not listed in AD.

TECH

MS

INSP

LABOR-HRS

HRS.THS

SB TFE731-72-5221

**NO. 2 ENGINE - REPLACE SPECIFIC IDENTIFIED LOW PRESSURE SECOND STAGE TURBINE (LPT2) ROTOR BLADES, PN 3075424-2, -3, SERVICE BULLETIN**☐ COMPLIED WITH☐ DECLINED☐ DEFERRED☐ NOT APPLICABLE

All text added to the "Note" field will be presented as part of the MOC selection through the application.  
Ex: MOC of "Complied With" and a Note of "At Manufacture" will display as "Complied With - At Manufacture"

Compliance Note: AD2012-16-14 is not applicable due to engine model was not listed in AD.AD2012-16-14 is not applicable due to engine model was not listed in AD.AD2012-16-14 is not applicable due to engine model was not listed in AD.

TECH

MS

INSP

LABOR-HRS

HRS.THS

**04-OCT-2017**  
**WID: 26608033-1**Return completed copy via email to [fax@campsystems.com](mailto:fax@campsystems.com)

&gt; INDICATES DUE ITEMS

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Total Pages = \_\_\_\_\_, Initial **1** of **2**



ation from the ground up

**BOMBARDIER**

**CHECK CURRENT DUE LIST FOR DUE TIME CHANGES**

Operator: **GEORGIA CROWN DISTRIBUTING CO.**  
 Serial No.: **P137180** Model: **TFE731-40BR-1B**  
 Part: **3062075-1**  
 A/C Serial **45-496** Reg No.: **N877W**

Work Card No.: **FAA 2012-16-14**

Workorder No.:

| Due At       | Date(DD/MMM/YYYY)  | A/C HRS       | AFL  | ENG. HRS    | ENC |  |  |
|--------------|--------------------|---------------|------|-------------|-----|--|--|
|              | <b>01-OCT-2017</b> | <b>3049.4</b> |      | <b>3000</b> |     |  |  |
| Accomplished | 04-OCT-17          | 1339.70       | 1279 |             |     |  |  |

725020

NO. 2 ENGINE - 2ND-STAGE LOW PRESSURE TURBINE (LPT) DISC

LMM 72-50-04

**LIFE LIMIT**

REASON REMOVED: (CHECK ONE) ☐ TIME EXPIRED ☐ FAILURE ☐ WORN ☐ LOANER ☐ SCHEDULING CONV  
☐ MOD/UPGRADE ☐ SERVICE ☐ ENGINE CHANGE ☐ TIRE CHANGE ☐ SWAP/TRBLE SHOOT ☐ DAMAGED ☐ UNKNOWN

If removed P/N & S/N information is incorrect please provide details below.

|               |  |           |        |               |  |      |                 |            |  |      |             |               |      |        |
|---------------|--|-----------|--------|---------------|--|------|-----------------|------------|--|------|-------------|---------------|------|--------|
| REMOVED P/N   |  | 3060611-1 |        |               |  | S/N  | 12-156103-00810 |            |  |      | LABOR-HRS   | 14000         |      |        |
| INSTALLED P/N |  |           |        |               |  | S/N  |                 |            |  |      | PART COST\$ | 14000         |      |        |
| INSTALLED TSN |  | MOS       | 040000 | INSTALLED TSO |  | MOS  | 040000          | TIME SINCE |  | MOS  | 040000      | WARRANTY TIME | MOS  | 040000 |
|               |  | HRS       | 040000 |               |  | HRS  | 040000          | REPAIR     |  | HRS  | 040000      |               | HRS  | 040000 |
|               |  | LDGS      | 040000 |               |  | LDGS | 040000          |            |  | LDGS | 040000      |               | LDGS | 040000 |

TECH:

INSP:

REMARKS : AD2012-16-14 is not applicable due to engine model was not listed in AD.

AD2012-16-14 is not applicable due to engine model was not listed in AD.

AD2012-16-14 is not applicable due to engine model was not listed in AD.

04-OCT-2017

WID: 26608033-2

Return completed copy via email to fax@campsystems.com

> INDICATES DUE ITEMS

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Total Pages = 2 of 2  
 Initial

**LIFE LIMITED PARTS**



**LIFE LIMITED PART MAINTENANCE RECORD**

| DATE | MAINTENANCE PERFORMED | AUTHORIZED SIGNATURE |
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# Honeywell

## LIFE LIMITED PART LOG

ASSEMBLY NAME: FAN ROTOR ASSEMBLY

PART NUMBER: 3060155-6

SERIAL NUMBER: 13-162246-00720

LIFE LIMITED PART NAME: DISK, FAN

PART NUMBER: 3060287-2

SERIAL NUMBER: 13-162246-00720

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

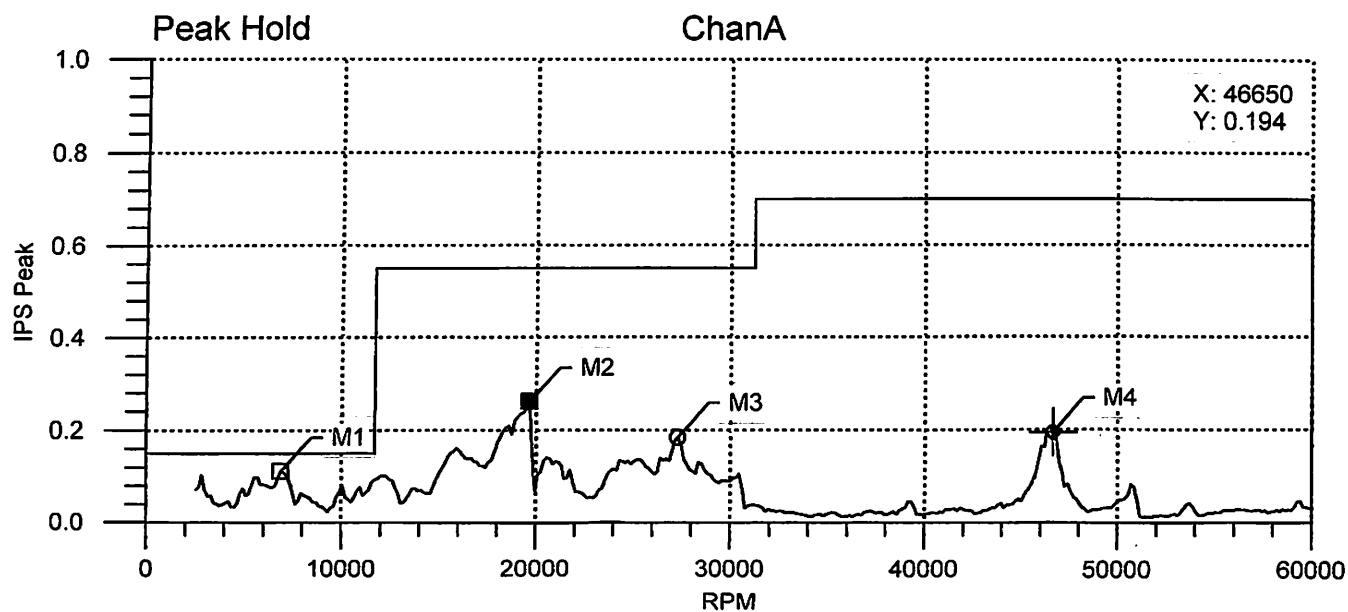
PX-3107-76C  
YQML FRMQM00004



[illegible]

Customer: 137180  
Started: 07/15/2016 17:49.47  
A/C Reg:  
Eng S/N: 137180  
Eng Pos: 2  
Condition: ACCEL  
Channel A: ChanA

Status: D  
Finished: No  
A/C Total TM: 0.0  
TSO: 0.0  
TSN: 0.0



| Peak List  |                 |                    |                         |       |
|------------|-----------------|--------------------|-------------------------|-------|
| X<br>(RPM) | Y<br>(IPS Peak) | Frequency<br>(RPM) | Amplitude<br>(IPS Peak) | Label |
| 6900       | 0.112           | 6900.00            | 0.112                   | M1    |
| 19650      | 0.261           | 19650.00           | 0.261                   | M2    |
| 27300      | 0.184           | 27300.00           | 0.184                   | M3    |
| 46650      | 0.194           | 46650.00           | 0.194                   | M4    |

**Honeywell Diskbal Compressor/Turbine**

DiskBal Version 2.60 07-15-2016 15:07:17  
Part File Version 2.6 Operator 63767

Disk Part Number 3060287 Bypass Civil  
Balance Radius 1  
Disk Serial Number 13-162246-00720  
Assembly Part Number 3060155\*\*

Engine Serial Number 137180  
Work Order 327791

**Sample Statistics (gm)**

Minimum 3406.100  
Maximum 3515.100  
Average 3465.970  
Range 109.000  
Disk Unbalance(oz-inch) 0.000  
Disk Unbalance Angle 0  
Assy Unbalance Amt.(oz-inch) 0.000  
Assy Unbalance Angle 104

| Disk Slot | Blade No. | Weight(gm) |
|-----------|-----------|------------|
| 1         | Blade 15  | 3512.100   |
| 2         | Blade 17  | 3494.500   |
| 3         | Blade 6   | 3437.500   |
| 4         | Blade 25  | 3515.100   |
| 5         | Blade 14  | 3459.700   |
| 6         | Blade 26  | 3428.100   |
| 7         | Blade 22  | 3440.800   |
| 8         | Blade 2   | 3447.100   |
| 9         | Blade 4   | 3450.400   |
| 10        | Blade 30  | 3450.300   |
| 11        | Blade 18  | 3458.200   |
| 12        | Blade 19  | 3458.500   |
| 13        | Blade 11  | 3507.600   |
| 14        | Blade 7   | 3475.600   |
| 15        | Blade 5   | 3472.400   |
| 16        | Blade 20  | 3440.800   |
| 17        | Blade 27  | 3514.300   |
| 18        | Blade 29  | 3498.300   |
| 19        | Blade 28  | 3424.300   |
| 20        | Blade 13  | 3506.500   |
| 21        | Blade 21  | 3497.800   |
| 22        | Blade 8   | 3406.100   |
| 23        | Blade 16  | 3423.800   |
| 24        | Blade 1   | 3493.900   |
| 25        | Blade 12  | 3426.300   |
| 26        | Blade 10  | 3439.400   |
| 27        | Blade 24  | 3432.600   |
| 28        | Blade 3   | 3487.200   |
| 29        | Blade 9   | 3505.800   |
| 30        | Blade 23  | 3474.100   |

63767



[illegible]



[illegible]





[illegible]



# Honeywell

## LIFE LIMITED PART LOG

ASSEMBLY NAME: COMPRESSOR ROTOR ASSY, AXIAL - STAGE 4

PART NUMBER: 3075351-12

SERIAL NUMBER: 13-162053-54870

LIFE LIMITED PART NAME: DISK, AXIAL COMPRESSOR, - STAGE 4

PART NUMBER: 3075326-5

SERIAL NUMBER: 13-162053-54870

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

PX-3107-76C  
YQML\_FRMQM00004

[illegible]



[illegible]

## LIFE LIMITED PART LOG

| ASSEMBLY NAME:                                                  |                 | PART NUMBER:                       |                                | SERIAL NUMBER:                 |                    |       |                        |
|-----------------------------------------------------------------|-----------------|------------------------------------|--------------------------------|--------------------------------|--------------------|-------|------------------------|
| LIFE LIMITED PART NAME: SEAL PLATE, HPT                         |                 | PART NUMBER: 3060514-1             |                                | SERIAL NUMBER: 13-182449-08997 |                    |       |                        |
| MANUFACTURED PER FAR PART 21 UNDER PRODUCTION CERTIFICATE PC413 |                 | SIGNATURE OR ACCEPTANCE STAMP: 295 |                                |                                |                    |       |                        |
| DATE<br>INSTALLED                                               | DATE<br>REMOVED | ENGINE<br>SERIAL NUMBER            | TIME ON PART THIS INSTALLATION |                                | TOTAL TIME ON PART |       | SIGNATURE / FAA NUMBER |
|                                                                 |                 |                                    | CYCLES*                        | HOURS                          | CYCLES*            | HOURS |                        |
| 02/12/2014                                                      |                 | P137180                            |                                |                                |                    |       |                        |
|                                                                 |                 |                                    |                                |                                |                    |       |                        |
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\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

[illegible]





[illegible]

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## LIFE LIMITED PART LOG

|                       |                           |                     |                  |                       |                        |
|-----------------------|---------------------------|---------------------|------------------|-----------------------|------------------------|
| <b>ASSEMBLY NAME:</b> | <b>HPT ROTOR ASSEMBLY</b> | <b>PART NUMBER:</b> | <b>3060555-5</b> | <b>SERIAL NUMBER:</b> | <b>12-156111-00365</b> |
|-----------------------|---------------------------|---------------------|------------------|-----------------------|------------------------|

|                |                    |              |           |                |                 |
|----------------|--------------------|--------------|-----------|----------------|-----------------|
| ASSEMBLY NAME: | HPT ROTOR ASSEMBLY | PART NUMBER: | 3060555-5 | SERIAL NUMBER: | 12-156111-00365 |
|----------------|--------------------|--------------|-----------|----------------|-----------------|

|                       |                           |                     |                  |                       |                        |
|-----------------------|---------------------------|---------------------|------------------|-----------------------|------------------------|
| <b>ASSEMBLY NAME:</b> | <b>HPT ROTOR ASSEMBLY</b> | <b>PART NUMBER:</b> | <b>3060555-5</b> | <b>SERIAL NUMBER:</b> | <b>12-156111-00365</b> |
|-----------------------|---------------------------|---------------------|------------------|-----------------------|------------------------|

|                                                         |                               |                                       |
|---------------------------------------------------------|-------------------------------|---------------------------------------|
| <b>LIFE LIMITED PART NAME:</b> DISK, TURBINE ROTOR - HP | <b>PART NUMBER:</b> 3060841-I | <b>SERIAL NUMBER:</b> 12-156111-00365 |
|---------------------------------------------------------|-------------------------------|---------------------------------------|

|                                                         |                               |                                       |
|---------------------------------------------------------|-------------------------------|---------------------------------------|
| <b>LIFE LIMITED PART NAME:</b> DISK, TURBINE ROTOR - HP | <b>PART NUMBER:</b> 3060841-I | <b>SERIAL NUMBER:</b> 12-156111-00365 |
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|                                                         |                               |                                       |
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| <b>LIFE LIMITED PART NAME:</b> DISK, TURBINE ROTOR - HP | <b>PART NUMBER:</b> 3060841-I | <b>SERIAL NUMBER:</b> 12-156111-00365 |
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| MANUFACTURED PER FAR PART 21 UNDER<br>PRODUCTION CERTIFICATE PC413 | SIGNATURE OR ACCEPTANCE STAMP: 295 |
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| MANUFACTURED PER FAR PART 21 UNDER<br>PRODUCTION CERTIFICATE PC413 | SIGNATURE OR ACCEPTANCE STAMP: 295 |
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[illegible]

SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

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[illegible]

PX-3107-76 BACK

## LIFE LIMITED PART LOG

ASSEMBLY NAME: TURBINE ROTOR ASSY - LP, FIRST STAGE

PART NUMBER: 3060610-5

SERIAL NUMBER: 12-156103-01094

LIFE LIMITED PART NAME: DISK, ROTOR - LP TURBINE, 1ST STA

PART NUMBER: 3060625-1

SERIAL NUMBER: 12-156103-01094

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

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[illegible]

SERIAL NUMBER: 12-156103-00810

SERIAL NUMBER: 12-156103-00810

**SIGNATURE OR ACCEPTANCE STAMP: A-75**

PX-3107-76C  
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[illegible]

## LIFE LIMITED PART LOG

ASSEMBLY NAME: TURBINE ROTOR ASSY - LP, THIRD STAGE

PART NUMBER: 3060626-8

SERIAL NUMBER: 12-156103-00929

LIFE LIMITED PART NAME: DISK, ROTOR - LP TURBINE, STAGE 3

PART NUMBER: 3060671-2

SERIAL NUMBER: 12-156103-00929

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

PX-3107-76C  
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[illegible]



# Honeywell

## LIFE LIMITED PART LOG

ASSEMBLY NAME: NOZZLE ASSY, LPT, STAGE 1

PART NUMBER: 3060666-11

SERIAL NUMBER: 13-3300-02512

LIFE LIMITED PART NAME: SUPPORT, STATIONARY SEAL, AIR

PART NUMBER: 3060693-1

SERIAL NUMBER: 13-3300-03364

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

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[illegible]

## LIFE LIMITED PART LOG

|                |              |                |
|----------------|--------------|----------------|
| ASSEMBLY NAME: | PART NUMBER: | SERIAL NUMBER: |
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|                                                       |                               |                                |
|-------------------------------------------------------|-------------------------------|--------------------------------|
| <b>LIFE LIMITED PART NAME:</b> TIE ROD, THREADED ENDS | <b>PART NUMBER:</b> 3074744-I | <b>SERIAL NUMBER:</b> 13P62792 |
|-------------------------------------------------------|-------------------------------|--------------------------------|

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

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[illegible]

## LIFE LIMITED PART LOG

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

PX-3107-76C  
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## LIFE LIMITED PART MAINTENANCE RECORD

| DATE | MAINTENANCE PERFORMED | AUTHORIZED SIGNATURE |
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SIGNATURE OR ACCEPTANCE STAMP: 295

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**LIFE LIMITED PART MAINTENANCE RECORD**

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## LIFE LIMITED PART LOG

**ASSEMBLY NAME:**

**PART NUMBER:**

**SERIAL NUMBER:**

LIFE LIMITED PART NAME: SHAFT SHOULDERED

PART NUMBER: 3060366-3

SERIAL NUMBER: 14P64042

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

PX-3107-76C  
YQML\_FRMQM00004

[illegible]

## LIFE LIMITED PART LOG

ASSEMBLY NAME:

PART NUMBER:

**SERIAL NUMBER:**

LIFE LIMITED PART NAME: COUPLING, SHAFT HP TURBINE

PART NUMBER: 3060502-2

SERIAL NUMBER: 13-1034-40068

MANUFACTURED PER FAR PART 21 UNDER  
PRODUCTION CERTIFICATE PC413

SIGNATURE OR ACCEPTANCE STAMP: 295

[illegible]

\* SEE SERVICE LIFE LIMITS OF CRITICAL LIFE LIMITED COMPONENTS, ENTRIES SHALL COMPLY TO FAR 43.

PX-3107-76C  
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## LIFE LIMITED PART MAINTENANCE RECORD

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**MPI/CZI Inspection  
Record**

Note: The MPI and CZI inspection intervals identified herein are valid as of the time of MPI or CZI maintenance ONLY. These, and other engine inspection intervals, WILL change over time and, as a result, owner/operator is expressly advised NOT to rely on this information for engine maintenance planning and performance. Owner/operator is solely responsible for reviewing all subsequent approved data changes, and for managing any impact such changes may have on engine inspection intervals.

## Honeywell TFE731 Turbofan Engine

### MPI/CZI Inspection Record

Engine Model: TFE731-40BR-1B

Engine Serial No.:

P137180

Enter initials or stamp in the C/W block for each inspection accomplished.

Enter calculated hours in the next due block

|             | MPI |          | CZI |          | Facility | Signature/Stamp |
|-------------|-----|----------|-----|----------|----------|-----------------|
|             | C/W | Next Due | C/W | Next Due |          |                 |
| Date: _____ |     |          |     |          |          |                 |
| TSN: _____  |     |          |     |          |          |                 |
| Date: _____ |     |          |     |          |          |                 |
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| Date: _____ |     |          |     |          |          |                 |
| TSN: _____  |     |          |     |          |          |                 |

## NOZZLE DATA SHEET

## PRODUCTION

Honeywell International Inc.  
Engines & SystemsENGINE S/N: P137180Outline : 3062075-1Assy : 3061704-2Rev : A

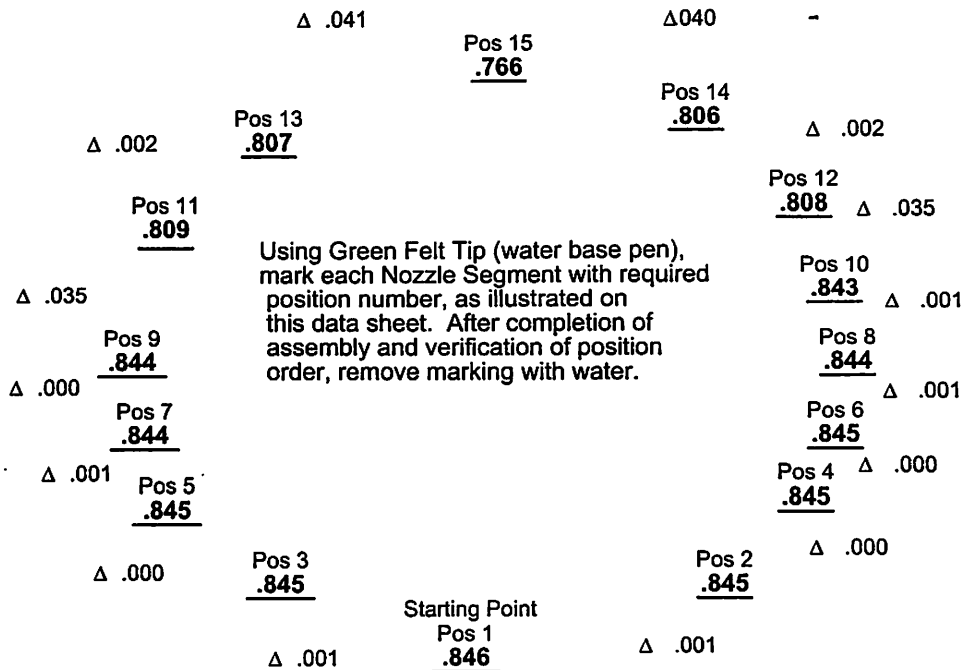
Date: 02/04/2014

Time: 08:24:59 AM

Prod Eng: EVELYN MALDONADOApproved: JOE THORNER

| POS | PART      | AE   | ID.NO   | LOT         |
|-----|-----------|------|---------|-------------|
| 1   | 3060840-6 | .846 | 4426077 | 14-1082-027 |
| 2   | 3060840-6 | .845 | 4423070 | 14-1082-024 |
| 3   | 3060840-6 | .845 | 4426072 | 14-1082-027 |
| 4   | 3060840-6 | .845 | 4426082 | 14-1082-027 |
| 5   | 3060840-6 | .845 | 4415191 | 14-1082-016 |
| 6   | 3060840-6 | .845 | 4423068 | 14-1082-024 |
| 7   | 3060840-6 | .844 | 4415185 | 14-1082-016 |
| 8   | 3060840-6 | .844 | 4426076 | 14-1082-027 |
| 9   | 3060840-6 | .844 | 4415188 | 14-1082-016 |
| 10  | 3060840-6 | .843 | 4415189 | 14-1082-016 |
| 11  | 3060840-4 | .809 | 3724084 | 13-1082-325 |
| 12  | 3060840-4 | .808 | 3724091 | 13-1082-325 |
| 13  | 3060840-4 | .807 | 3724088 | 13-1082-325 |
| 14  | 3060840-4 | .806 | 3724087 | 13-1082-325 |
| 15  | 3060840-5 | .766 | 3613366 | 13-1082-213 |

Total AE:12.442

A5 S/N: 13-3300-02512 With an AE of: 24.189Engineer Approved: Joe Thorne

A total of (15) two-vane segments with any combination of 3060840-4, 3060840-5 & 3060840-6 that is required to achieve a flow area of 12.00 to 12.720. The effective flow area of adjacent segments must not vary more than .040 sq in. A maximum variation in adjacent segments of .085 sq in is allowed at two locations only.

## A4/A5 Engine Data Sheet

A5 Area: \_\_\_\_\_ sq. in. Date: \_\_\_\_\_

### TFE731-20/40/50/60

Position Nozzle Segment  
Number Part and Dash

### Engine Models

Effective Flow  
Area (AE), sq. in

Ascending AE

Decending AE

|     |       |       |
|-----|-------|-------|
| 1.  | _____ | _____ |
| 2.  | _____ | _____ |
| 3.  | _____ | _____ |
| 4.  | _____ | _____ |
| 5.  | _____ | _____ |
| 6.  | _____ | _____ |
| 7.  | _____ | _____ |
| 8.  | _____ | _____ |
| 9.  | _____ | _____ |
| 10. | _____ | _____ |
| 11. | _____ | _____ |
| 12. | _____ | _____ |
| 13. | _____ | _____ |
| 14. | _____ | _____ |
| 15. | _____ | _____ |

Total AE: \_\_\_\_\_

Pos 15  
Pos 13

Pos 14  
Pos 12

Pos 11

Pos 10

Pos 9

Pos 8

Pos 7

Pos 6

Pos 5

Pos 4

Pos 3 Starting Point Pos 2  
Pos 1

Aft Looking Forward

Note: - See applicable maintenance manual for  
associated procedures.

- Retain completed data sheet with engine  
log book.

**ACCESSORIES**





## COMPONENT MAINTENANCE / MODIFICATION RECORD CARD

(RECORD CARD MUST TRAVEL WITH COMPONENT, REPAIR STATION OR ENGINE LOG BOOK)

## PART NOMENCLATURE

## FUEL CONTROL

PART NUMBER

3070800-47

SERIAL NUMBER

20398

**SERIES NUMBER**

6

CHANGE /MOD RECORD

[illegible]







## COMPONENT MAINTENANCE / MODIFICATION RECORD CARD

(RECORD CARD MUST TRAVEL WITH COMPONENT, REPAIR STATION OR ENGINE LOG BOOK)

PART NOMENCLATURE

## ACCESSORY GEARBOX

PART NUMBER

3060827-4

SERIAL NUMBER

S-4238

SERIES NUMBER

1

CHANGE /MOD RECORD

[illegible]

# COMPONENT MAINTENANCE / MODIFICATION RECORD CARD

(RECORD CARD MUST TRAVEL WITH COMPONENT, REPAIR STATION OR ENGINE LOG BOOK)

## PART NOMENCLATURE

L. P. FIRST STAGE TURB. NOZZLE ASSY.

PART NUMBER

3060666-11

SERIAL NUMBER

13-3300-02512

SERIES NUMBER

N/R

CHANGE /MOD RECORD

[illegible]

# COMPONENT MAINTENANCE / MODIFICATION RECORD CARD

(RECORD CARD MUST TRAVEL WITH COMPONENT, REPAIR STATION OR ENGINE LOG BOOK)

## PART NOMENCLATURE

## TRANSFER GEARBOX

PART NUMBER

3070093-16

SERIAL NUMBER

14P64285

SERIES NUMBER

N/R

CHANGE /MOD RECORD

[illegible]

(RECORD CARD MUST TRAVEL WITH COMPONENT, REPAIR STATION OR ENGINE LOG BOOK)

## Motive Flow Pump

7006000.3-

12-3000 3-00010

**0 SERIES NUMBER**

### CHANGE/MOD RECORD

**ENGINE  
SERIAL NO.**

**AIRCRAFT  
SERIAL NO.**

**ENGINE HOURS  
AT INSTALLATION**

### ENGINE HOURS AT REMOVAL

TOTAL HOURS  
ON PART[illegible]

P/37180

45-496

1186.9

## COMPONENT MAINTENANCE / MODIFICATION RECORD CARD

**(SERVICE BULLETIN COMPLIANCE, REPAIRS, INSPECTIONS, ETC.)**

[illegible]