The subject aircraft was involved in an incident / accident on 6/28/2023 rendering it a Constructive Total Loss as not repairable within its insured value. For further details refer to insurance claim: 3005671925Us.

Signed: Underwriters Salvage Company

Date: 11/30/2023
AIG Aerospace Adjustment Services, Inc. 3500 Lenox Road, Suite 1100
Atlanta, GA 30326

TOTAL broughten said from the vio FAR 91.411 ALTIMETERS ALTITUDE REPORTING EQUIPMENT STATIC PRESSURE SYSTEM HAVE BEEN CHECKED AND MEET SPECIFICATIONS OF APPENDIX E OF PART 43 ALTIMETER CERTIFIED TO FEET RIGHT FEET LEFT 70,000 9.4994 S/N S/N INSPECTOR W/O DATE: 2404 3-25-PF Flyers, Inc. FAA APPROVED REPAIR STATION # Y3FR325Y FAR 91.413 THE ATC TRANSPONDERS AND INSPECTIONS AS REQUIRED BY FAR 91.413 WERE PERFORMED THIS DATE AND FOUND TO COMPLY WITH PART 43 APPENDIX F, AND APPENDIX E PARAGRAPH C. S/N MODEL: TRANSPONDER MAKE: 465,00085 GNX 375 GARMIN #1 MODEL: S/N MAKE: TRANSPONDER #2 INSRECTOR: W/O DATE 2504 03-15-2011 .PF Flyers, Inc. FAA APPROVED REPAIR STATION # Y3FR325Y REMOVE & REPLACE ASD. POWER FACK AF TESTING. FLUSHED SYSTEM AND SERVE 5-12-22 W 9606, LEAR CK AND OPS CK GOOT . SUB-TOTAL this page

The money rue amount with COMES THAT AND I GO CONTAIN AND AND ADDRESS OF THE PARTY OF N 28 of become set that godingent 1466.6 INSP. AIRCRAFT YOW AN ANNUAL INSPECTION LUBED AIRFRAME & GEAR TERFORMED GEAR RETRACTION AND ENERGENG EXT. REPCACED TAIL MULAMP. C/W ADS: 12-02-10, 15-16-1/ AND 93-05-06. SEE AD LIST FOR DETAILS. REPLACED ELY BATT (MAY 2024) AND TESTED ELT IAN 91. 2076 AND FOUND TO BE IN SAT. COND. 1 Certify That This Thas Been Ispected In Accordance With ALANDAL Inspection And Was Determed To Be In Airworthy Condition. Time In Service: 24.68.4 Day 5-6-22 Signed 1 Ca. A 3595343 JA

WAS REPAIRED AND FUNCTIONAL TESTED IAW 73500 REV. G DRAWING AND COMMANDER EPAS 795101-501 10/12/93. NO DEFECTS WERE NOTED. ------

13a. Certifies the items identified above were	manufactured in conformity to:	14a. ☑14 CFR 43.9 Return to Service	☐ Other regulation specified in Block 12
Approved design data and are in a Non-approved design data specifie	condition for safe operation, d in Block 12.	described in Block 12 was accomp	ed in Block 12, the work identified in Block 11 olished in accordance with Title 14, Code of espect to that work, the items are approved for
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.: 7CXR376B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): BRYAN A COX	14e. Date (dd/mmm/yyyy): 04/MAR/2022
	User/Installer	Responsibilities	

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

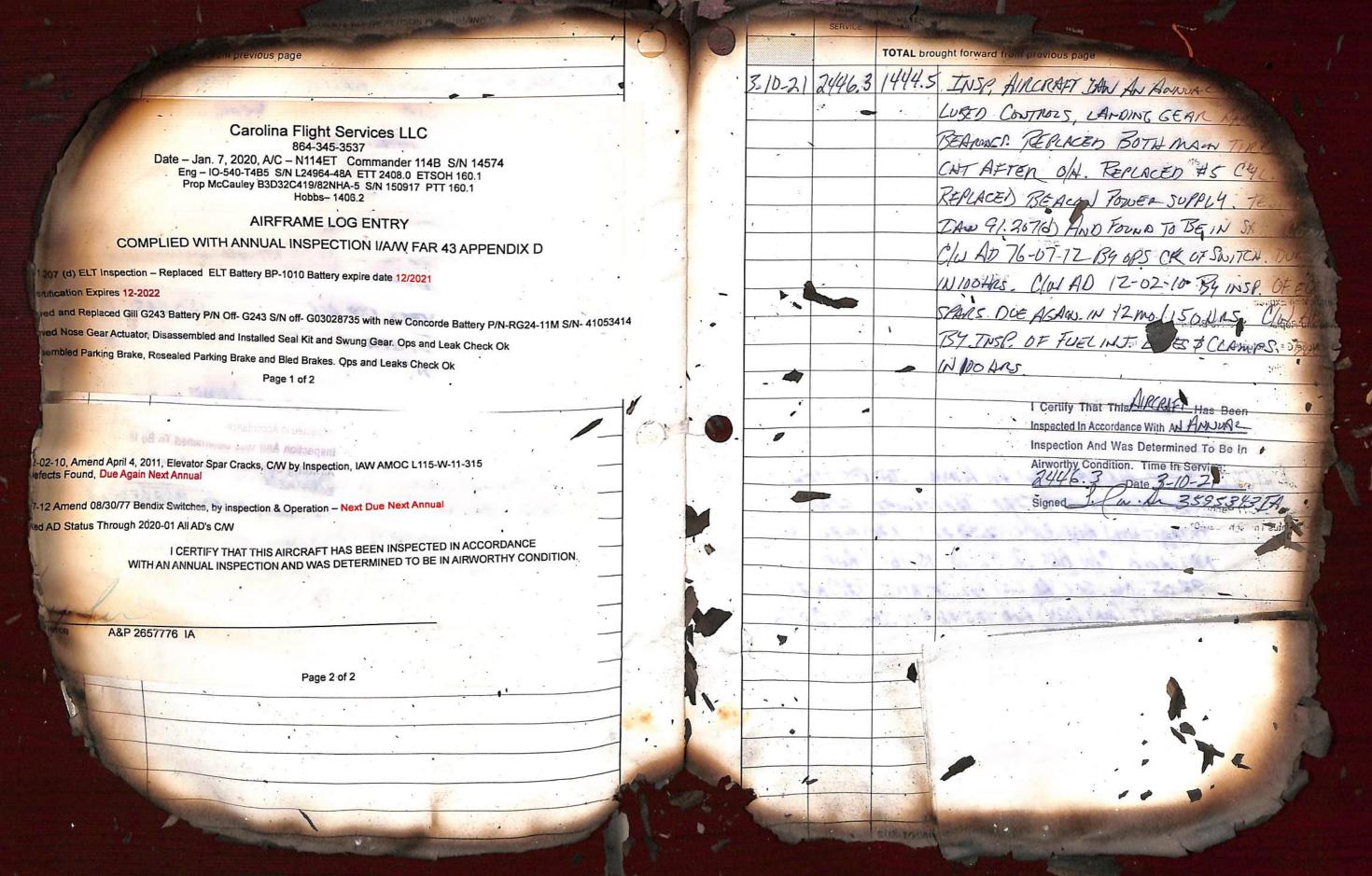
FAA Form 8130-3 (02-14)

NSN: 0052-00-012-9005

	Airparts, LLC	
	chita, Ks 67217	FAA Cert. No. 7CXR376B
501 Description	POWER PACK	Serial Number AH5410
GHT SERVICES		PO
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	11 K	
X	] <u> </u>	
Repair	a 5 - 10 .	Overhaul Inspect & Test
Customer Purch	ase Order Instruc	ctions / Comments
a We		*
4 4	Work Performed	di e ,
AND INSPECTED I ADDED	AND POLISHED AL	LL REQUIRED SURFACES. REPLACED ALL
TE INSTALLED NEW SEAL	KIT REASSEMBLE	D AND TESTED IAW 73500 REV. G DRAWING
95101-501 10/12/93. NO	DEFECTS NOTED.	
95101-301 10/12/33.110	.7.	
	*	No.
	-	
		The state of the s
	-Parts Replaced	
1.11		P/N & Description
& Description	Qty	
A Property of the Park of the		H.
		Repairman's Signature
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den Damage	Be 1/2	6/11
nspection	Be Yu	S.Aller
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30 Issued:	Bc 3/4.	Number 2-0033
	1	
	THE RESERVE TO SERVE THE PARTY OF THE PARTY	Date MAR 0.4 2022
CWAX-		
		Form Number

Form Number'
Revision Number:
Revision Date:

QAF001 Initial Release 07/27/2009



JAMES GUIDA 213 RIVEREDGE DR MOORE, SOUTH CAROLINA 29363 USA

· Termin Account: JAGU

Your PO# :

Our TEL: 478-788-30 Your TEL: 864-216-33

Tail#: N114ET

A/C Make: COMMANDER

Model: 114-B Serial: 14574

**DETAILS** -

WO; 4192 TASK: 1 AF T/S KFC200 START: 10/07/2020 FINISH: 10/21/2020 SER#: 11808

PART#: 065-0034-00

DISCREPANCY: T/S KFC200

ACTION: TROUBLESHOT AUTOPILOT NOT ENGAGING. GAINED ACCESS TO KC295 AUTOPILOT COMPUTER S/N:11808 AND REMOVE BENCH CHECK/REPAIR. COMPLETED VISUAL INSPECTION ON SERVO WIRING AND SERVO INSTALLATIONS. INSPECTED DISCONN AND TRIM SWITCH. REPAIRED WEAK ENGAGE SWITCH ON AUTOPILOT CONTROL HEAD TO PREVENT AUTOPILOT FROM DISCONNECTION IN FLIGHT. REINSTALLED REPAIRED KC295 S/N:11808. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW KFC200 INSTALLATION MANUAL 006-00728-0000 REV 1 DATED DEC 1, 1992. HOURS: Flat Rate LABOR: \$995.00 PARTS/EQUIP/OTHER: \$47.44 TASK TOTAL: \$1,042.44

WO: 4192 TASK: 2 AF FIX STEC YAW START: 10/12/2020 FINISH: 10/21/2020 PART#: 0106-Y10 SER#: 6757A

DISCREPANCY: FIX STEC YAW

ACTION: INSPECTED STEC YAW SERVO AND FOUND THAT BRIDLE CABLE HAD COME OF CAPSTAN AND WRAPPED AROUND THE THE SERVO. REMOVED SERVO AND INSPECTED. FOUND CLUTCH TO BE TO HIGH, RESET CLUTCH TO MANUFACTURERS SPECS REWRAPPED BRIDLE CABLE. REINSTALLED YAW SERVO AND TESTED. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW BULLETIN NO. 582 REV 2 AND DRAWING NO. 05166.

HOURS: Flat Rate LABOR: \$995.00

PARTS/EQUIP/OTHER: \$0.00

TASK TOTAL: \$995.00

WO: 4192 TASK: 3 AF PANEL LIGHTS START: 10/13/2020 FINISH: 10/21/2020

DISCREPANCY: PANEL LIGHTS

ACTION: TROUBLESHOT PANEL LIGHTS NOT DIMMING. GAINED TO DIMMER POTENTIOMETERS AND TESTED. REMOVED EXISTING AND INSTALL 2 NEWS POTS P/N:RESSORE. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW COMMANDER 1148 MAINTEN MANUAL P/N:MM-114B/TC.

LABOR: \$630.00

HOURS: 6.00

PARTS/EQUIP/OTHER: \$89.84

TASK TOTAL: \$719.84

### MAINTENANCE RELEASE / RETURN TO SERVICE CRS# ETER699D

The Aircraft, Component, Propeller or Appliance Identified above was repaired in accordance with cureent Federal Aviation Regulations and is hereby approved for return to service.

Shipping & Handling: Sub Total:

Parts & Equip:

Labor:

Invoice TOTAL:

Inspector / Authorized Signature :

Certificate #:

Date:

ETER699D

THANK YOU FOR YOUR BUSINESS. YOUR CONTINUED PATRONAGE IS APPRECIATED AND WE LOOK FORWARD TO DOING BUSINESS THANK YOU FUR YOUR BUSINESS TOOK CONTINUES ARE DUE UPON RECIEPT. 1.5% INTEREST WILL BE CHARGED FOR ANY INVOICES OVER 

#### Carolina Flight Services LLC TOTAL brought forward from previous page 864-345-3537 December 17, 2018, A/C - N114ET Commander 114B S/N 14574 Eng - IO-540-T4B5 S/N L24964-48A ETT 2408.0 ETSOH 129.9 Prop McCauley B3D32C419/82NHA-5 S/N 150917 PTT 129.9 1. L'almetto Avionics Aircraft Make & Model: COMMANDER 114-B Hobbs- 1876.0 FAA CRS# 3PXP870B AIRFRAME LOG ENTRY Aircraft Reg No.: ! .. 14ET COMPLIED WITH ANNUAL INSPECTION I/A/W FAR 43 APPENDIX D 1-REMOVED THE FOLLOWING EQUIPMENT: KING KI-203 CDI, ACK A-30 ENCODER, MIDCONTINENT MD-ANNUNCIATOR, KING KX-155 NAV/COM, KING KI-76C TRANSPONDER, KING KLN-90B GPS LT Inspection - Replaced, ELT Battery BP-1020 Battery expire date 12/2019 2-INSTALLED THE FOLLOWING EQUIPMENT: A)WAAS NAVIGATOR WITH ADS-B COMPLIANT TRANSPONDER MODEL GNX-375 MEG'D BY GARMIN S/N: 5GJ001 Expires 10-24-2019 GARMIN WAAS ANTENNA MODEL GA-35 S/N 171646 IN ACCORDANCE WITH STC# 5A0? 36.7E. B)COM TRANSCEIVER MODEL GTR-225 MFG'D BY GARMIN S/N: 2A5007035. Replaced Right Hand Position Light P/N- W1290-28 C)COURSE DEVIATION INDICATOR MODEL KI-200 MFG'D BY KING S/N 1449. Replaced Gill G243 Battery with new P/N- G243 S/N Off G02943730 S/N On- G03028735 THE INSTALLED ADS-B OUT SYSTEM WAS SHOWN TO MEET THE EQUIPMENT PERFORMANCE REQUIREMENTS OF CFR PART 91.227. GARMIN GNX-375 AFMS POCUMENT# 190-02207-A3, REV. 2 WAS INSERTED INTO THE AIRCRA Replaced Nose Tire P/N- 505C66-5 S/N Off-.71470480 S/N- 83292162 POH. GARMIN GNX-375 ICA L OCUMENT# 190-02207-A2, RE VAS INSERTED INTO THE AIRCRAFT'S MAINTENA 10, Amend April 4, 2011, Elevator Spar Cracks, C/W by Inspection, IAW AMOC L115-W-11-315 ound, Due Again Next Annual The aircraft and/or component identified ab. . . . as repaired and/or inspected in accordance with the current regulations. of the Federal Aviation Administration and is approved for return of vice for the work performed. Letails are on · 1 - 1 - 10 Outer 1 Page 1 of 2 Auu. iized Signature FAR 91.411 ALTIMETERS, ALTITUDE REPORTING EQUIPMENT AND STATIC PRESSURE SYSTEM HAVE BEEN CHECKED AND MEET ALL 12 Amend 08/30/77 Bendix Switches, by inspection & Operation - Next Due Next Annual SPECIFICATIONS OF APPENDIX E OF PART 43 D Status Through 2018-25 All AD's C/W I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE ALTIMETER CERTIFIED TO WITH AN ANNUAL INSPECTION AND WAS DETERMINED TO BE IN AIRWORTHY CONDITION. LEFT RIGHT 20000 FEET. S/N 914994 S/N DATE: W/O INSPECTOR! A&P 2657776 IA 1/12/2020 2171 PF Flyers, Inc. FAA APPROVED REPAIR STATION # Y3FR325Y Page 2 of 2 FAR 91.413 THE ATC TRANSPONDERS AND INSPECTIONS AS REQUIRED BY FAR 91.413 WERE PERFORMED THIS DATE AND FOUND TO COMPLY WITH PART 43 APPENDIX F, AND APPENDIX E PARAGRAPH C. GARMIN. TRANSPONDER S/N GNX 375 MAKE 595000853 MODEL: DATE SIN W/O 1/17/2020 INSPECTOR 2171

### Work Order/Invoice Work Order No. . Date Aircraft Manufacturer Aircraft Model 2167 06/02/2020 COMMANDER 114 Description of Work Performed Tech Insp 1-REMOVED AND ROUTED FLIGHT COMPUTER MODEL KC-295 MFGD' BY KING S/N 11808-TO-AUTOPILOT RTS RTS CENTRAL FOR REP IR. UNIT WAS REPAIRED UNDER AUTOPILOT CENTRAL WORK ORDER# SH20-33880. UNIT WAS REINSTALLED INTO AIRCRAFT AND A GROUND CHECK WAS PERFORMED. SYSTEM INDICATED FUNCTIONING PROPERLY. Parts Description, **Unit Price** Qty. Extend AUTOPILOTCENTRAL REPAIR 01 1152.45 'Credit ~ Freight Parts Subtotal Deposit Tax-Labor Subtotal Invoice Total **Balance Due** 1152.45 0.00 1452.45 300.00 1452.45 The aircraft and/or component identified above was repaired and/or inspected in accordance with the current regulations of the Federal Aviation Administration and is approved for return to service for the work performed. Signature of Authorized Individual

**AVIONICS** AIRFRAME TIME IN TIME IN SERVICE SERVICE

#### DESCRIPTION OF WORK PERFORMED SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK

Palmetto Avionics FAA CRS# 3PXR870B

Aircraft Make & Model: COMMANDER 114B

Aircraft Hobbs: 1417.9

Tach:

Aircraft Reg No.: N114ET

Date: 06/02/2020

1-REMOVED AND ROUTED FLIGHT COMPUTER MODEL KC-295 MFGD' BY KING S/N 11808 TO AUTOPILOT CENTRAL FOR REPAIR. UNIT WAS REPAIRED UNDER AUTOPILOT CENTRAL WORK ORDER# SH20-33880. UNIT WAS REINSTALLED INTO AIRCRAFT AND A GROUND CHECK WAS PERFORMED. SYSTEM INDICATED

The aircraft and/or component identified above was repaired and/or inspected in accordance with the current regulations of the Federal Aviation Administration and is approved for return to service for the work performed. Details are on file at this

Authorized Signature

LOWE AVIATION SERVICES,LLC 2178 FLIGHTLINE DR MACON,GA.31216 ETER699D WO# 4192 HOBBS 1435.2 S/N 14574 COMMANDER 114-B TROUBLESHOT AUTOPILOT NOT ENGAGING. GAINED ACCESS TO KC295 AUTOPILOT COMPUTER S/N:11808 AND REMOVED FOR BENCH CHECK/REPAIR. COMPLETED VISUAL INSPECTION ON SERVO WIRING AND SERVO INSTALLATIONS. INSPECTED DISCONNECT AND TRIM SWITCH. REPAIRED WEAK ENGAGE SWITCH ON AUTOPILOT S/N:11808. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW KFC200 INSTALLATION MANUAL OOG-00728-0000 REV 1 DATED DEC 1, 1992. Pg 1/1

INSPECTED STEC YAW SERVO AND FOUND THAT BRIDLE CABLE HAD COME OFF CAPSTAN AND WRAPPED AROUND THE BASE OF THE SERVO. REMOVED SERVO AND INSPECTED. FOUND CLUTCH TO BE TO HIGH, RESET CLUTCH TO MANUFACTURERS SPECS AND REWRAPPED BRIDLE CABLE. REINSTALLED YAW SERVO AND TESTED. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW STEC BULLETIN NO. 582 REV 2 AND DRAWING NO.05166.

TROUBLESHOT PANEL LIGHTS NOT DIMMING. GAINED TO DIMMER POTENTIOMETERS AND TESTED. REMOVED EXISTING POTS AND INSTALL 2 NEWS POTS P/N:RES5ORE. SYSTEM RAMP CHECKED OPERATIONAL. WORK DONE IAW COMMANDER 114B MAINTENANCE MANUAL P/N:MM-114B/TC.

MAINTENANCE RELEASE / RETURN TO SERVICE CRS# ETER699D

GINE COMPONENT PROPELLER OR APPLIANCE IDENTIFIED ABOVE WAS REPAIRED IN CORRENT FEDERAL ANTATION AGENCY REGULATIONS AND IS HEREBY APPROVED FOR RETURN THE AIRCRAFT, ENGINE COMPONENT ACCORDANCE WITH TO SERVICE.

INSPECTOR LAUTHORIZED SIGNATUR

ETER699D

Work Order:

SH

Acct Number:

(803)257-6242

Sold To: Palmetto Avionics P.O. Box 2846 Aiken, SC 29802 Ship to: Palmetto Avionics

322 Terminal Road

Hgr A-1

Greenwood, SC 29649

#### **General Comments:**

Received Via: UPS Ground

Shipped Via: Fed-X

Tracking Number: 1198-1204-7770

#### Discrepancy: 1

#### **Unit Description:**

KC295 Flight Computer S/N 11808 (065-0034-00)

#### Action Taken:

PRELIMINARY INSPECTION: Bench tested unit and found engage transistor shorted; traces on power supply board burne looks to have happened once before.

WORK PERFORMED: Troubleshot-flight computer; found Q505 (engage transistor) on power supply board damaged; replation transistor. Repaired damaged circuit trace on power supply board. Repaired, tested and certified reference King manual 006-5558-03, Rev 3, 1983.

\*\*NOTE: THIS DAMAGE WAS LIKELY DUE TO A SHORTED SOLENOID WIRE (ROLL, PITCH OR TRIM) IN AIRCRAFT.

Charges	This Item:	:01 		3.00 Hours @	115:00 \$
Part Number	Description	Credit	Quantity Units	List Price Disc	Unit Price
12039-0010	Transistor (MJE2955)		1.00 Each	115.870	115.870 \$
14				Total For This	Discrepancy: \$
Miscellaneous Ch	arges:				
		1911		M	isc Supplies: \$
				Ger	neral Freight: \$
Totals:					40000000000000000000000000000000000000
i Otais.	Laboration and the control of the co		SubTotal:	1	\$
B. Committee			Charges:		\$
			Paid-AHT-\	/ISA	\$

Terms: C.O.D.

The articles and/or work hereon described is/are certified airworthy (unless otherwise specified).

Autopilots Central, Inc. FAA CRS CM2R747K

Authorized Signature\_

PARTS INSTALLED MAY BE FAA/PMA'D OR STC'D PARTS AND NOT OF "ORIGINAL EQUIPMENT MANUFACTURE"
AS SUCH ARE NOT COVERED BY THE OEM'S WARRANTY.

PAST DUE INVOICES ARE SUBJECT TO A 1.5% FINANCE CHARGE.

Palmetto Avionics Printed: 5/06/2020

1:38 PM

Work Order: SH20-33880

Entered By: RACHAEL

Page

Authority/C	Civil Aviation ountry: ED STATES	2. A	UTHORIZED RE FAA Form 8130-3, AIRW			3H20- 33 880
	on Name and A #CM2R747K		AUTOPILO HANGAR 23, TULSA INT'L A	TS CENTR	AL, INC. SA, OKLAHOMA 74115	5. Work Order/Contract/Invoct Number: Same #3
6. Item: 7. De			8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
	HOUTER		065-0034-00	1	11808	REPARKED
			report for work performed.	to: 14a.⊠14 C	FR 43.9 Return to Service Ot	her regulation specified in Block 12
	ved design data a		condition for safe operation. ied in Block 12.	Title 14,		in Block 12, the work identified in accomplished in accordance with rt 43 and in respect to that work, ice.
	ed Signature:		13c. Approval/Authorization N	No.: 14b. Author	Signature:	. 14c. Approval/Certificate No.: CM2R747K
13d. Name (1	Typed or Printed	i):	13e. Date (dd/mmm/yyyy):	STEVE	ENL Chapkan	14e. Date (dd/mmm/yyyy): 78-A-R2-2270
G.				r Responsibi		Francisco Company
			tence of this document alone does			

country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Work Order: SH20-33717 Acct Number: (803)257-6242

Sold To: Palmetto Avionics

P.O. Box 2846 Aiken, SC 29802 Ship to: Pa

Palmetto Avionics

322 Terminal Road Hgr A-1

Greenwood, SC 29646

#### **General Comments:**

Received Via: UPS Ground

Shipped Via: Tracking Number:

#### Discrepancy: 1

#### **Unit Description:**

KC295 Flight Computer S/N 11808 (065-0034-00)

#### Action Taken: .

PRELIMINARY INSPECTION: Bench tested unit and found trim fail problems; gets stuck in test mode (-5vdc at trim drive output continuous trim fail lamp; has all newer filter capacitors.

(918) EL 6418 - FHX (9.14)

WORK PERFORMED: Troubleshot flight computer; found C403 and C404 on adapter board faulty; replaced. Repaired 14/28v power switch. Bench tested and certified reference King manual 006-5558-03, Rev 3, 1983.

Charges This	s Item:	40011	
Part Number	Description	Credit ' Quantity Units Units	
	Cap Tn 2.2uf10%15v	Credit ' Quantity Units Unit Price	Ex
096-01030-0003	00p 2.2di 10 % 13V	2,00 Each 32.100s	11000000
		. Total For This Discrepancy: \$	
Miscellaneous Charg	es:		N. T. C.
•		Misc Supplies: \$	
Summary:			
Tatal Darte	- 4.00 Hours: \$	64.20 Misc Supplies: \$ 460.00	
Totals:			
-		SubTotal:	
		Total Charges: \$	
Torms: COF	,	Ψ	

Terms: C.O.D.

Administration	on .	ALCOHOL:			414.1	LUNE TEA HPI	THE PROPERTY OF			
INSTRUC	TIONS: Print	or type an	entric	es. See Title 14 CF. This report is requi	FRS	43.9, Part 43	Appendix B.	1		
Such viola	ation. (49 U.S	.C. §46301(	torm. (a))	This report is requi	red c	by law (49 U.S	S.C. §44701). F	allure to rer	CT THE TAN	A CONTRACTOR OF THE PARTY OF TH
	Nationalit	y and Regis	stratio	n Mark	7		Serial No.		ort can N	Marie Control of the
1. Aircraft	N114E	Γ.			24	自由	14574			
1. All Crait	Iviake		15				Model	100		THE PARTY OF THE P
		ANDER					114-B			Series
	Name (A	s shown on	regist	tration certificate)			Address (As .	shown on re	gistration	
2. Owner				(90)			70diess 213 F	RIVEREDG	E DR	cennicate)
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4. Ty	уре		22		5.	Unit Identifica	ation			
Repair	Alteration	Unit	1	М	lake			Model		Serial No.
	<b>V</b>	AIRFRAM	IE	-			(As described in Item 1 above)			
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	ENWOOD	LISA		State SC	1					
Zip 2964		untry USA	_				Maintenance Organ		3PXR8	
have	been made in	accordance	e with	ion made to the unit( the requirements of to the best of my kno	f Part	t 43 of the U.S.	5 above and de . Federal Aviatio	escribed on t on Regulation	the revers	e or attachments hereto at the information
Extended ra	ange fuel		Signa	ature/Date of Authori	ized	Individual				
per 14 CFR	Part 43			1	-			2110		- 2 2246
дрр. В				ととと		)		AUC	SUST	9, 2019
				7. Ar	prov	val for Return	to Service			
Pursuant	to the author ator of the Fer	ority given deral Aviatio	perso on Adr	ons specified below ministration and is	v, the	e unit identifie	ied in item 5 v	was inspecte	ed in the	e manner prescribed
TF	AA Fit. Stand	lards		nufacturer	M	laintenance Org	ganization	Depar	rtment of Tra	d by Canadian ansport
	AA Designee	· /	Repa	air Station	In	spection Autho	orization	Other (Spec	cify)	7
1 1	1000		Signa	ature/Date of Authori	ized I	Individual .				
Certificate or Designation		*	1	20	1		1	AUG	GUS <sup>-</sup>	Г 9, 2019
-70	2		_		1		25		300	1 3, 2013
FAA Form	337 (10-06)									

# Work Order/Invoice

Work Order No.	Date	Aircraft Manufacturer	Aircraft Model	Regis	trati		
1992							
	Descrip	otion of Work Performed		Mary Control	Tech	Insp	
1-REMOVED THE FOLL 328 GPS ANNUNCIATO	OWING EQUIPMEN OR, KING KX-155 N	T: KING KI-203 CDI, ACK A-3 <mark>0 E</mark> AV/COM, KING KT-76C TRANSPO	NCODER, MIDCONTINENT ONDER, KING KLN-90B GP:	MD41	RTS		
5GJ000853 AND GARM SA02636SE.	VITH ADS-B COMPL IN WAAS ANTENNA	LIANT TRANSPONDER MODEL GI A MODEL GA-35 S/N 171646 IN A	ACCORDANCE WITH STC#	I S/N:	RTS	RTS	
C)COURSE DEVIATION	INDICATOR MODE	FG'D BY GARMIN S/N: 2A500703 EL KI-209 MFG'D BY KING S/N 14 ATION SEE FAA FORM 337 DATI	49.				
a DEDECRMED 24 MON	THE CENTIFICA	TION ON THE FOLLOW FOLITOM	ENT/CVCTENC				

3-PERFORMED 24 MONTH IFR CERTIFICATION ON THE FOLLOW EQUIPMENT/SYSTEMS: A)TRANSPONDER MODEL GNX-375 MFG'D BY GARMIN S/N: 5GJ000853.

B)ALTIMETER MODEL 5934D-1 MFG'D BY UNITED S/N 9H994.

C)ALTITUDE REPORTER.

D)STATIC SYSTEM.

ALL UNITS/SYSTEMS LISTED ABOVE WERE TESTED AND ARE IN COMPLIANCE WITH FAR 91.411 AND 91.413 PART 43 APPENDIX E AND F.

Parts Description	Qty.	Unit Price	
GARMIN GNX-375 WAAS GPS/TRANSPONDER GARMIN GTR-225 COM GARMIN GAE-12 ALTITUDE REPORTER KING KI-209 CDI	01 01 01 01	7000.00 1800.00 275.00 900.00	7000 1800 27' 90
			. 1

Labor Subtotal	Parts Subtotal	Deposit	Credit	Freight	Tax	Invoice Total	Con 1
2850.00	9975.00	-7000.00			0.00	12825.00	Balance Due
The aircraft a	nd/or component ide	ntified above w	as repaired an	d/or inspected i	n _	12023.00	5825.00

The aircraft and/or component identified above was repaired and/or inspected in accordance with the current regulations of the Federal Aviation Administration and is approved for return to service for the work performed.

Signature Chuchorized Individual

RTS RTS

		7.1	Total	- Contract	The same	
		DES SIGNATURE &	CRIPTION OF CERTIFICATE N	WORK PERFOR	RMED— REFORMING WORK	
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april 1	4.					
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attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N114ET

AUGUST 9, 2019

Nationality and Registration Mark

Date

MOVED THE FOLLOWING EQUIPMENT: KING KI-203 CDI, ACK A-30 ENCODER, MIDCONTINENT MD41-328 GPS INCIATOR, KING KX-155 NAV/COM, KING KT-76C TRANSPONDER, KING KLN-90B GPS

TALLED THE FOLLOWING EQUIPMENT:

WAAS NAVIGATOR WITH ADS-B COMPLIANT TRANSPONDER MODEL GNX-375 MFG'D BY GARMIN S/N: 5GJ000853 AND RMIN WAAS ANTENNA MODEL GA-35 S/N 171646 IN ACCORDANCE WITH STC# SA02636SE.

OM TRANSCEIVER MODEL GTR-225 MFG'D BY GARMIN S/N: 2A5007035.

OURSE DEVIATION INDICATOR MODEL KI-209 MFG'D BY KING S/N 1449.

TALLATION ACCOMPLISHED IN ACCORDANCE WITH GARMIN GNX-375 MASTER DRAWING LIST DOCUMENT# 005-01206-REV. 2, DATED MARCH 27, 2019; GARMIN GTR-225 INSTALLATION DOCUMENT# 190-01182-02, REV. H DATED SEPT. 17; KING KI-209 INSTALLATION MANUAL DOCUMENT# 006-00140-0004, REV. 4, DATED AUGUST 2002; AC43.13-1B, APTER 7, SECTIONS 1-5, 11, CHAPTER 10, SECTIONS 1 AND 2, CHAPTER 11, SECTIONS 1,3-20, CHAPTER 12, SECTIONS 1 AC43.13-2B, CHAPTERS 1,3, AND 11.

THE INSTALLED ADS-B OUT SYSTEM WAS SHOWN TO MEET THE EQUIPMENT PERFORMANCE REQUIREMENTS OF 14 CFR

TRUCTURAL INSTALLATION IS IN COMPLIANCE WITH FAR 23.561. NSTALLATION IS IN COMPLIANCE WITH FAR 23.1301 AND 23.1431.

ARMIN GNX-375 AFMS DOCUMENT# 190-02207-A3, REV. 2 WAS INSERTED INTO THE AIRCRAFT'S POH.

ARMIN GNX-375 ICA DOCUMENT# 190-02207-A2, REV. 2 WAS INSERTED INTO THE AIRCRAFT'S MAINTENANCE RECORDS.

VEIGHT AND BALANCE DATA, EQUIPMENT LIST, AND AIRFRAME LOGS UPDATED TO REFLECT THIS CHANGE THIS DATE.
FREORMED AN ELECTRICAL LOAD ANALYSIS AND FOUND TOTAL LOAD DOES NOT EXCEED 80% OF RATED CAPACITY.

N OPERATIONAL CHECK WAS PERFORMED ON THE GROUND AND THE INSTALLED SYSTEMS INDICATED FUNCTIONING ROPERLY. INSTALLED SYSTEMS DID NOT ADVERSELY AFFECT THE EXISTING SYSTEMS OR COMPONENTS IN THE INCRAFT.



### Department of Transportation Federal Aviation Administration

# Supplemental Type Certificate

Number: SA02636SE

This certificate issued to: Garmin International, Inc. 1200 East 151st Street

Olathe, KS 66062

certifies that the change in the type design for the following product with the limitations and conditions therefore a specified hereon meets the airworthiness requirements of Part 23 of the Federal Aviation Regulations.

Original Product - Type Certificate Number:

Model:

Multiple - AML STC

See Approved Model List (AML) SA02636SE for approved aircraft models and applicable airworthiness standards.

Description of Type Design Change:

Installation of Garmin GPS 175 Navigation System, GNX 375 Navigation System with ADS-B In/Out Transponder, and GNC 355 Navigation System with VHF COM Radio. Data required includes Garmin Master Drawing List (MDL) 005-01206-A1 Revision 1, dated March 22, 2019 or later FAA-approved revision; Garmin Airplane Flight Manual Supplement (AFMS) 190-02207-A3 Revision 1, dated March 22, 2019 or later FAA-approved revision; Garmin Maintenance Manual (Includes Instructions for Continued Airworthiness (ICA)) 190-02207-A2 Revision 1, dated March 22

#### Limitations and Conditions:

(1) The installer must determine whether this design change is compatible with previously installed modifications.

(2) If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other

(3) A copy of this certificate, the approved AFMS, and the approved ICA must be maintained as part of the permanent

(See Continuation Sheet on Page 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: June 7, 2018

Date Reissued:

Date of Issuance: March 22, 2019

Date Amended: July 25, 2019

By Direction of the Administrator

JR Brownell Title

ODA STC Unit Administrator

ODA-240087-CE

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. This certificate may be Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. This certificate m transferred or made available to third persons by licensing agreements in accordance with 14 CFR 21.47. Possession of this Supplemental Type (STC) document by persons other than the STC holder does not constitute rights to the design data nor to alter an aircraft aircraft. Any altered or made available to third persons by licensing agreements in accordance with 17 or 18 2000 ossession or this supplemental Type certificate (STC) document by persons other than the STC holder does not constitute rights to the design data nor to alter an aircraft, aircraft engine, or the STC's supporting documentation (drawings, instructions, specifications, flight manual supplements, etc.) is the property of the STC. Certificate (STC) document by persons other trial tile of C holder does not constitute rights to the design data nor to alter an aircraft, aircraft engine. The STC's supporting documentation (drawings, instructions, specifications, flight manual supplements, etc.) is the property of the STC holder who allows a person to use the STC to alter an aircraft, aircraft engine, or propeller must provide that person with propeller. The STC's supporting documentation (grawings, instructions, specifications, ingrit manual supplements, etc.) is the property of the STC propeller. An STC holder who allows a person to use the STC to alter an aircraft, aircraft engine, or propeller must provide that person with written

#### 4. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

#### 4.1 Airworthiness Limitations

There are no new (or additional) airworthiness limitations associated with this equipment and/or installation.

The Airworthiness Limitations section is FAA approved and specifies maintenance required under §§ 43.16 and 91.403 of the Federal Aviation Regulations unless an alternative program has been FAA approved.

**FAA APPROVED** 

JR Brownell

7/25/2019

JR Brownell

Date

ODA STC Unit Administrator

ODA-240087-CE

GPS 175/GNX 375 in accordance with Section 5.

#### 21 Periodic Maintenance

the GPS 175/GNX 375/GNC 355 are designed to detect internal failures. A thorough self-test is executed automatically upon application of power to the units. The built-in tests (BIT) are continuously executed. Detected errors are indicated as failure annunciations, system messages or a combination of the two.

Antenna installations are not covered under this STC. Inspect and maintain all antennas in accordance with the data provided for that specific antenna installation.

### 4.2.2 Special Tools

A milliohm meter with an accuracy of  $\pm 0.1~\text{m}\Omega$  (or better) is required to measure the electrical bonding between the GPS 175/GNX 375/GNC 355 system components and aircraft ground.

Table 4-1below shows the systems and components, installed by this STC, which the checks at specific intervals. The inspections based on calendar elapsed time have specifically intervals.

### Table 4-1 Periodic Maintenance

Item	Description/Procedure	Interval
Equipment Removal and Replacement	Removal and reinstallation of the GPS 175/GNX 375/GNC 355 LRUs,	On Condition
Cleaning	The GPS 175,GNX 375 and GNC 355 display and bezel may be cleaned periodically.  The front bezel and display should be cleaned with a soft cotton cloth dampened with clean water.  DO NOT use any chemical cleaning agents. Avoid scratching the surface of the display.	On Condition
Electrical Bonding Test	An electrical bonding test must be performed on equipment installed by this STC.	10 years or 2000 flight hours
Testing	The GPS 175/GNX 375/GNC 355 must be tested and shown to comply with Title 14 CFR Part 91.277.	Replacement of GPS Position source(s).
Visual Inspection	A visual inspection of the equipment installed by this STC must be performed.	12 Calendar Months

masses to ensure that they continue to comply with the STC SA02636SE.

inspect the GPS 175/GNX 375/GNC 355 unit for security of attachment, including visual inspection of mounting racks and other supporting structure attaching the racks to aircraft instrument panel.

- GPS 175/GNX 375/GNC 355 Verify the countersunk fastener heads are in full contact with the unit mounting rack holes. Re-torque the mounting screws 8.0 ± 1.0 in-lb, if required.
- Flight Stream 510 Ensure the data card is properly oriented (label facing left), fully inserted, and locked into position in the card slot on the front right side of the unit.
- Inspect for corrosion.
- 3. Inspect switches, knobs, and buttons for damage.
- 4. Inspect condition of wiring, shield terminations, routing, and attachment/clamping.
- Check the fan intake/outlet slots on the GPS/GNX/GNC unit's bezel for dust, dirt, or obstructions. Clean as needed.
- 6. Conduct a visual check of the GPS/SBAS antenna cable overbraid (if installed).
- 7. Conduct a visual check of and bonding strap or conductive tape used for electrical bonding.
- 8. Replace any damaged or torn straps. Refer to Section 6.8 for details.
- Replace any torn bonding tape using a heavy duty aluminum foil tape, such as 3M P/N 436 or 438 or another foil with aluminum that is 7.2 mils thick or greater. If strap termination hardware is loose, tighten and re-test bonding. Refer to Section 6.8 for details.

#### 4.5 Electrical Bonding Test

### 4.5.1 GPS 175/GNX 375/GNC 355 Bonding Check (Metallic or Tube-and-Fabric Alreas



#### NOTE

A bonding test failure may occur if a fastener is not torqued to the specified torque value. For installations that use screws in lieu of rivets to secure the rack to surrounding structure, verify that the screws are torqued to the appropriate value before proceeding to remove the rack. Refer to Section 4.4 for torque values.

Perform an electrical bonding check as follows:

- 1. Remove the GPS 175, GNX 375 or GNC 355 from the mounting rack.
- 2. Remove the backplate assembly from the rack.
- 3. Measure the resistance between the mounting rack and nearby exposed portion of aircraft metallic structure and verify it is less than or equal to  $5~\text{m}\Omega$ .

In the event of bonding test failure, remove the GPS 175/GNX 375/GNC 355 rack, clean the attachment points with a bonding brush at both the GPS 175/GNX 375/GNC 355 rack and the aircraft, and re-attach the rack to the rails in the panel. Measure the resistance between the mounting rack and nearby exposed portion of aircraft metallic structure and ensure that the resistance is less than or equal to 2.5 m $\Omega$ .

4. Re-install the backplate assembly and re-install the GPS 175/GNX 375/GNC 355 in the mounting rack.

### 4.5.2 GPS 175/GNX 375/GNC 355 Bonding Check (Composite Aircraft)



#### NOTE

A bonding test failure may occur if a fastener is not torqued to the specified torque value. For installations that use screws in lieu of rivets to secure the rack to surrounding structure, verify that the screws are torqued to the appropriate value before proceeding to remove the rack. Refer to Section 4.4 for torque values.

Perform an electrical bonding check as follows:

- 1. Remove the GPS 175, GNX 375 or GNC 355 from the mounting rack.
- 2. Remove the backplate assembly from the rack.
- 3. Measure the resistance between the mounting rack and the instrument panel, verify it is less than or equal to 10 m $\Omega$ .

In the event of bonding test failure, remove the GPS 175/GNX 375/GNC 355 rack, clean the attachment points with a bonding brush at both the GPS 175/GNX 375/ GNC 355 rack and the aircraft, and re-attach the rack to the rails in the panel. Measure the resistance between the mounting rack and the instrument panel and ensure that the resistance is less than or equal to  $5~\text{m}\Omega$ .

 Re-install the backplate assembly and re-install the GPS 175/GNX 375/GNC 355 in the mounting rack.

#### 4.6 Overhaul Period

The system does not require overhaul at a specific time period. Power on self-test and BIT will monitor the health of the GPS 175/GNX 375/GNC 355 system. If any LRU indicates an internal failure, the unit may be removed and replaced. See Section 5 of this document for fault corrective actions.

### 4.7 Additional Instructions

None.

#### A59100 Certified Comple

20732 Lassen Street, Chatsworth

Phone: (818) 993-3842 Fax: (818) 993-4524

#### CERTIFICATION OF CONFORMANCE

AAT Part Number: AT575-93GW-TNCF-000-RG-27-NM

Customer PO Number: 287153

Customer Part Number: 013-00235-00 GA35

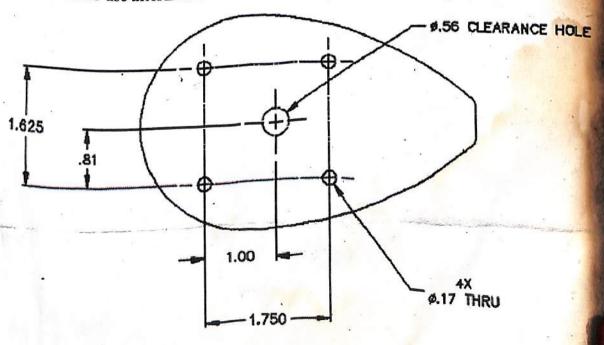
Serial Number: 171646

AeroAntenna Technology, Inc. certifies that the above mentioned antenna was manufactured in accordance with specifications and standard quality assurance practices and procedures, and meets all requirements as stated in the above purchase order as well as shipment and delivery instructions.

Quality Assurance: Benjamin Rubeni Date: 7/16/2019

### CERTIFICATION OF CONFORM

NOT TO SCALE. Use an accurate measuring instrument for installation. A calibrated torque wrench must be used during installation. Torque value for mounting screws is 13-15 in·lbs. The 4 supplied mounting screws must be used and no substitutes are allowed. Do not deviate from these installation methods or use alternate screws as damage to the antenna may result.



### worthiness Directive Compliance Record

AIRCRAFT RECORDS - DO NOT DESTROY

airframe

Company

Carolina Flight Services LLC

Manufacturer

GULFSTREAM (SEE ALSO GRUMMAN, SCHWEIZER, FRAKES, FLYING BOAT)

Model

114B

Tail# Serial # N114ET

**Current Time Total Time** A/C Cert. Date 14574 1330.9

\*12-02-10

03/15/12

LEFT AND RIGHT ELEVATOR SPARS/

Method of Compliance / C/W By Inspection, no cracks found SB#

Date 11-01-217

Next Due Next Annual Cert. # A&P 2657776 IA

Signature

prepared by Robert T. Herce

Date 11-01-2017

## Airworthiness Directive Compliance Record

AIRCRAFT RECORDS - DO NOT DESTROY

appliances

Company

Carolina Flight Services LLC

Tail#

N114ET

\*76-07-12

08/30/77

BENDIX **SWITCHES** 

[R1] IGNITION SWITCHES/

Method of Compliance C/W by inspection & Operation

SB# Signature Date 11-1-2017

Next Due Next Annual Cert. # A&P 2657776 IA

DESCRIPTION OF WORK PERFORMED-SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK

Avionics West, Inc. dba Avwest Atlanta

Mid F. FAR 91.411, FAR 91.413, Altimeter tosted per FAR 43 (b). This A/C and/or component was repaired and inspected

FARs and is Approved for return to service

#### Avwest Atlanta

P.O. Box 2032

Kennesaw, GA 30156 FAX: 678-354-6248

Phone: 678-354-6255

CHE LA

Date: 7-31-14 W.O.# 1438 N# /14FT

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114-B, N114ET S/N 14574

HOBB: 1246.1

AGE BY REPLACING A LOWER FRAME - RH, P/N 43481-6. SEE 337

16. REMOVED & REINSTALLED INTERIOR TO FACILITATE REPAIR OF RH
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AGE STEP AREAS. REPAIRED AFT FUSELAGE DAMAGE AROUND CABIN
RED NOSE GEAR DOOR BOX. INSTALLED NOSE GEAR DOORS - LH & RH.
NER ENGINE COWL. REMOVED & REINSTALLED LYCOMING ENGINE 10-54024964-48A AFTER OVERHAUL (TRIAD AVIATION, INC.) INSTALLED NEW
OPELLER B3D32C419/82NHA-5 S/N 150917. REMOVED & REINSTALLED
FLER EXHAUST 9910758-101 AFTER REPAIRED BY ACORN WELDING LTD.
TAG. REMOVED & REINSTALLED PROP GOVERNOR C290D3-J/T41 S/N
ER OVERHAUL BY H & H PROPELLER SERVICE (SEE TAG). REPLACED
C1-105-16 (2 EACH). GROUND RUN-UP SATISFACTORY.

THOMA R. STOIA, A&P 267830477 IA

## Carolina Flight Services LLC

Date - October 3, 2016; A/C - N114ET Commander 114B S/N 14574 ACTT - 27-11.4 Eng - IO-549-T4B5 S/N L24964-48A ETT 2278.1 ETSN 2278.1 Prop McCauley B3D32C419-C S/N 922747 PTT 2278.1

### AIRTRAME LOG ENTRY

## COMPLIED WITH ANNUAL INSPECTION I/AW FAR 43 APPENDIX D

7 (d) ELT Inspection - Replaced ELT Battery BP-1045 Battery expire date 11/2017,

02-10, Amend April 4, 2011, Elevator Spar Cracks, C/W by Inspection, IAW AMOC 1.115-W-11-315

Found, Due Again 2851.4 ' OTT or October 2017 whichever occurs first.

Amend 08/30/77 Bendix Switches, by inspection & Operation'- Next us notes 2841

19ry PIN G243 SIN GO2441064

TY PIN G243 SIN GO2933464

tus Through 2016-19 All AD's CM

VITH AN ANNUAL INSPECTION AND WAS DETERMINED TO BE IN AIRWORTHY CONDITION.

Carolina Flight Services LLC 864-345-3537 Date - November 1, 2017, A/C - N114ET Commander 114B S/N 14574 Hobbs- 1330.9 Eng - IO-540-T4B5 S/N L24964-48A ETT 2362.9 ETSOH 84.8 Prop McCauley B3D32C419/82NHA-5 S/N 150917 PTT 84.8 AIRFRAME LOG ENTRY COMPLIED WITH ANNUAL INSPECTION I/A/W FAR 43 APPENDIX C/W 91.207 (d) ELT Inspection - Replaced ELT Batter BP-1020 Battery expire date 12/2019 IFR Certification Expires 10-24-2019 AD 2012-02-10, Amend April 4, 2011, Elevator Spar Cracks, C/W by Inspection, IAW AMOC L115-W-11-315 No Defects Found, Due Again Next Annual AD 76-07-12 Amend 08/30/77 Bendix Switches, by inspection & Operation - Next Due Next Annual Page 1 of 1 Removed & Replaced BA-4005 Air Filter Element Checked AD Status Through 2017-20 All AD's CAV I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED I WITH AN ANNUAL INSPECTION AND WAS DETERMINED TO BE IN

A&P 2657776 IA

Robert J. Pierce

pener, ops & Land Thecks OK, All W. 1275.4 HOBBS AND AD LAGE. DRILLED OUT FOUR RIVETS - CLEANED AREA. INSTALLED RIVETS AND TWO RIVETS ON OIL DOOR HINGE. SERVICED TIRES, LUBRICATED AIRCRAFT. SEVICED LUBRICATED AIRCRAFT. SEVICED LUBRICATED AIRCRAFT. SEVICED LUBRICATED AIRCRAFT. SEVICED TIRES - NO LEAKS NOTED. DOOR WINDOWS. LUBRICATED RIGHT DOOR OPEN ARM. SECURED LEFT AND RIGHT TWO RIVETS ON OIL DOOR HINGE. SERVICED TIRES, LUBRICATED AIRCRAFT. SINCH SEVICED TIRES, LUBRICATED AIRCRAFT. SECURED LEFT AND RIGHT SINCH SEVICED TIRES, LUBRICATED AIRCRAFT. SECURED LEFT AND REPACKED AIRCRAFT. SHECKED NORMAL AND EMERGENCY GEAR - OKAY. BLED BRAKES. VISUALLY INSPECTION AND DETERMINED TO BE IN AIRWORTHY COND. TO THE SECURED LEFT AND SECURED LEFT AN Checked AD Status Through 2015-16 All AD'S CIW - I CERTIEY THAT THIS AIRCRAFT HAS BEEN INSPECT WITH AN ANNUAL INSPECTION AND WAS DETERMINED TO BE IN AIRCRAFT ENGINE, OR APPLIANCE IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED NORK PERFORMED UNDER WO#14291 AT REPAIR STATION DK7R795J. PERTINENT DETAILS Robert J. Pierce A&P 2657776 IA without the market to the BARS 1912 - 16 . 11525 211 LE AT THIS REPAIR STATION. DLK AVIATION, INC. sna Lane NW Kennesaw, GA 30144 770-427-4954 16178 09-September-2014 WO# 14360 S/N 14574 COMMANDER 114-B Pg 1/1 Carolina Flight Services LLC 864-641-3907 Date - December 10, 2015, A/C - N114ET Commander 114B S/N Hobbs - 1243.6 PRESSURE GAGE WITH SERVICABLE UNIT PN 855102-1 SN 028. RAN ENGINE MANIFOLD INGS CORRECT - INSTALLED INLINE ANOTHER MALIFOLD PRESSURE GAGE AND BOTH GAGE'S AIRFRAME LOG ENTRY Removed & Replaced Alternator with Overhauled Hartzell P/N DOFF10300B, S/N On H-P060457, S/N On FOR WORK PERFORMED UNDER WO#14360 AT REPAIR STATION DK7R795J. PERTINENT DETAILS Ran Engine, Ops Checks Ok. All work I/A/W manufacturers Service Manua ON FILE AT THIS REPAIR STATION. Aircraft Returned to service at this time. OR DLK AVIATION, INC. a Flight 864-64 Date - September 13, 2015, A/C - N114ET Commander 114B S/N 14574 ACTT = 2278.1 Hobbs - 1231.0 Eng - IO-540-T4B5 S/N L24964-48A ETT 2278.1 ETSN 2278.1 Prop McCauley B3D32C419-C S/N 922747 PTT 2278.1 AIRFRAME LOG ENTRY COMPLIED WITH ANNUAL INSPECTION I/A/W FAR 43 APPENDIX D ection - Replaced ELT Battery BP-1045 Battery expire date 11/2017, April 4, 2011, Elevator Spar Cracks, C/W by Inspection, IAW AMOC L115-W-11-315 in 2428.1 ACTT or September 2016 whichever occurs first. 25/12 Fuel injector lines - by inspection, Next Due ACTT 2378.1 or 12 Months 77 Bendix Switches, by inspection & Operation - Next Due Hobbs 1331.0 SUB-TOTAL thi

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Commander: 114B

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Hobbs: 861.8

SN#: 17258914

ormed an Annual Inspection IAW Commander Service Manual. ELT tested IAW 91-207(d). Replaced ELT battery Exp. Date Feb 2010. Replaced Vac relief filter VB3-5-1. Nose tire replaced with Goodyear Flt. Custom III 5.00-5 tire Sycled ding gear through an extension, retraction and emergency extension. searched using Avantex software through 2008-04. I certify that this aircraft as inspected IAW an Annual Inspection and was determined to be in an airworthy

Paul S Ridgeway A&P/IA 249947097

May 12, 2008 Airframe: Commander 1148

N114ET

Installed O/H Weldon Fuel Boost PUMP PN# A8120-D from AIRCRAFT

ACCESSORIES, INC on this date. Pump OPS checked satisfactory

S APPROVAL TAG

3. Form Tracking Number:

, AA 050708

5. .Work order/Contract/Invoice Number: 32556

12. Status / Work: Serial/Batch Number: Quantity 11. **OVERHAULED** 0035

AA APPROVED DATA. THERE ARE NO APPLICABLE AIR RFORMED IS ON FILE AT THIS FACILITY.

March 17, 2009

Airframe: Commander 114B

VIII4ET

Performed an ANNUAL INSPECTION IAW FAR 43, Appendix D. Serviced nose's with nitrogen. Replaced dirty Vac relief filter Pn. B3-5-1. Applied sealant to pilots real. Cleaned leaking Rt wing sump drain. Cycled Landing Gear through a retraction extension and an emergency extension no defects noted. Lecertify that this aircraft was nspected IAW an ANNUAL INSPECTION and was determined to be in an airworthy

Hobbs: 914.5

REGIONAL ALR

Paul S Ridgeway A&P249947097IA

REGION

April 10, 2009

Airframe: Commander 1148

N114ET

Owner/customer installed new

Hobbs: 924

Hobbs: 995.0

Ser#: 14574

Ser#: 14574

Battery on this date

AFTT: 2043.0

April 1, 2010

- Airframe: Commander 114B

N114ET

Performed an ANNUAL INSPECTION IAW FAR 43, Append d. ELT ops checked Pullies and cables checked and IAW 91.207(d).

Jubricated as required. Brake pads and fluid checked and serviced as required With bearings checked and greased. Replaced leaking RH wing fuel drain. Replaced both static wicks. Replaced anticollision light bulb and filere. Induction airfilter thecks and cleaned. Cycled geer thru 4 cycles. All AD's checked for compliance using AVANTEX Ad software Ver 2010-06. No new AD's applicable at this time. I certify that I have inspected this aircraft IAW an Annual Inspection and were found it to be in an airworthy condition on this date.

Mustet A. Mala Christopher A. Maxwell 2846774 A&P/I

SUB-TOTAL

March 21, 2005 Hobbs/ Fach: 614. / Airframe: Rockwell Commander 114B S/N 14574 N114ET

Complied with an Annual Inspection on this date. Serviced all struts with 5606 hydraulic fluid and nitrogen as required. Cleaned fuel strainer assembly as needed. Lubed landing gear and all wheel bearings. Cleaned and checked all brake linings. Cycled gear through 5 fault free cycles. Inspected ELT, battery expires March 2006. Checked AD Compliance through 2005-05 using Avantex AD Research CD's.

I certify I have inspected this aircraft in accordance with an Annual Inspection and found it in an airworthy condition on this date.

James D. Nielsen 2840818 A&P, I.A.



May 27, 2005 Hobbs: 623.8

N114ET

Replaced left main tire. Repacked wheel bearings with grease. Installed flight Custom III, 6.00 x 6 6 ply tire.

James D. "Doug" Nielsen 2840818 A&P, I.A.



March 8, 2006, Hobbs: 666.8

Airframe: Commander 114B S/N 14574

Replaced ELT battery pack. T battery expires April 2008. Complied w inspection as per FAR 91.207(d). Tested satisfactory. Replaced nose gea tube due to crack in scissor mount attach point. Replaced all seals and ser Mil\_H-5606 Hydraulic fluid and nitrogen as per maintenance manual. gear through 12 fault free cycles. Taxi-tested aircraft turning radius. Ops satisfactory. Checked AD compliance through 2006-04 using Avantex AD

I certify that I have inspected this aircraft in accordance with an Annual Inspection and determined it to be in an airworthy condition on this date.

JAMES D. WIELSEN

James D. "Doug" Nielsen 2840818 AGP, I.A.

Hobbs: 733.1

Ser. 14574

March 15, 2007.

Airframe: Commander 114B

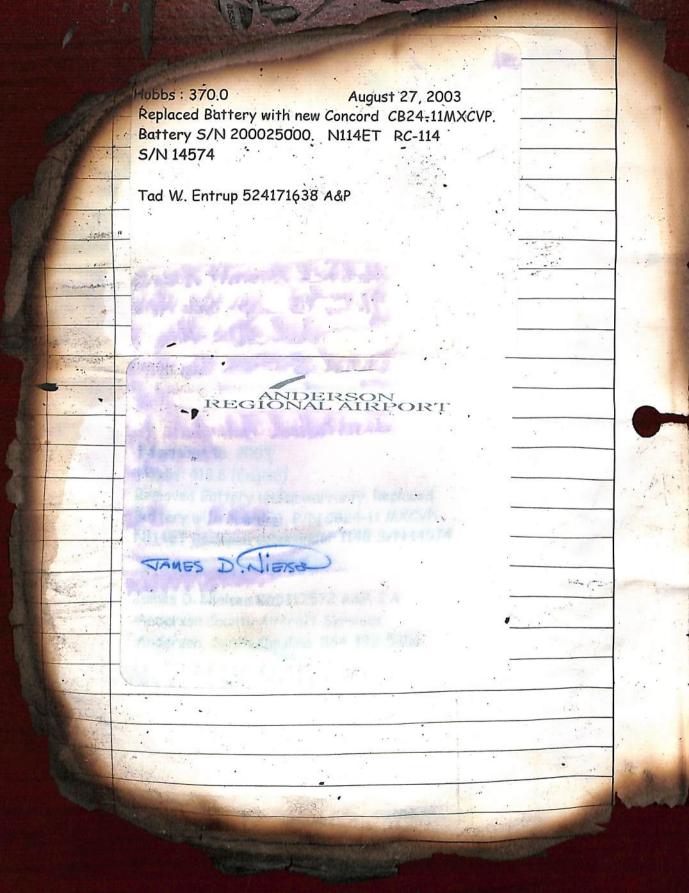
N114ET

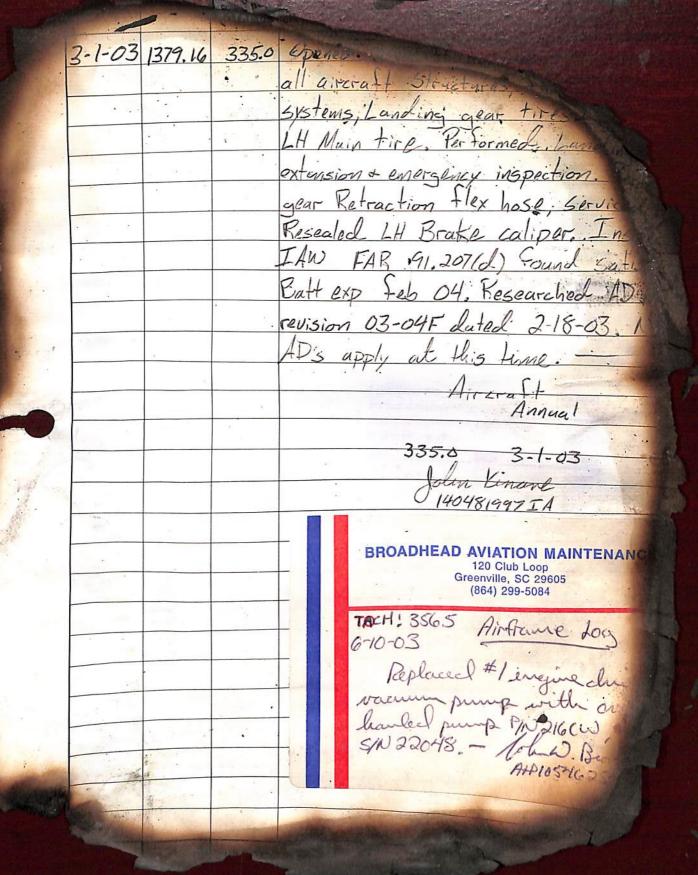
Performed ANNUAL INSPECTION on this date IAW Commander Service Manu All pullies, cables and turnbuckles checked and lubed as needed. ELT tested IA FAR 91-207(d). ELT Batt. Exp April 2008. Hydraulic reservoir serviced with 560 Wheel bearings cleaned and packed. All AD's checked for compliance using Avail AD listings (2007-04) Refer to attached listing. I certify that I have inspected aircraft IAW an annual inspection and have found it to be in an airworthy condition

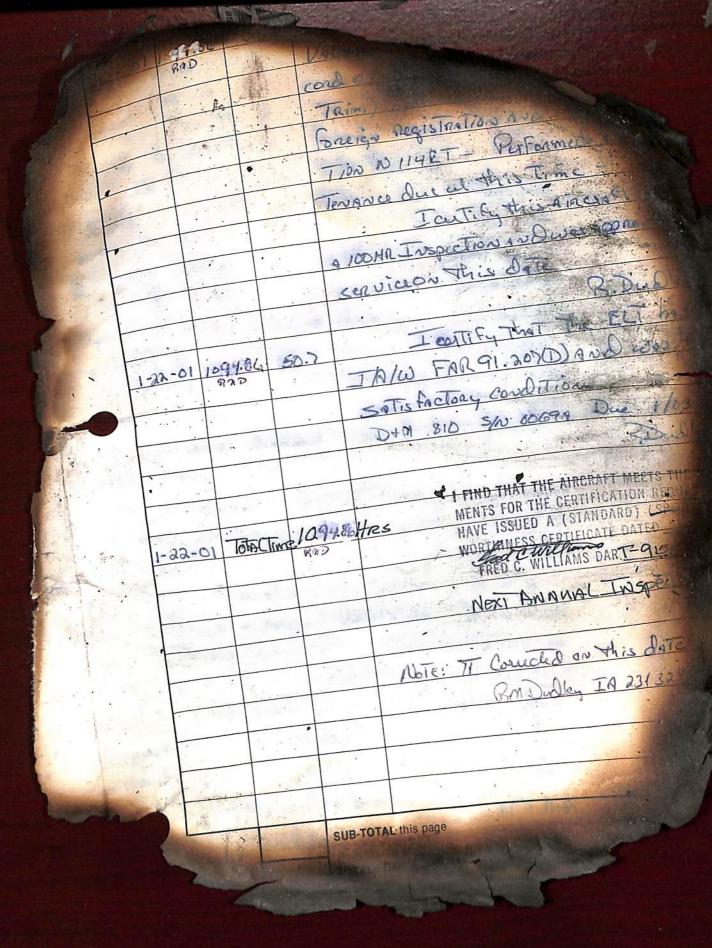
Christofe A. Majurell

Christopher A. Maxwell 2846774 A&P, IA

-	IN SERVICE	TIME IN SERVICE	DESCRIPTION OF WORK PERFORMED—
10-12-04	Garage Contraction		SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
			10-12-04 - FACH- HOBBS
			REMOVE/REPLACE KING KI256 FLIGHT COMMAND ATT. GYRO W/ OVERHAULED UNIT (SNX22984)
			overhauled Unit (SNX22984)
			OPS TESTS Ingmal Co
1	A .		RELATED documents in Aircraft RECORDS.
			John Coop
			TOCCOA,GA 30577
	V.		36577
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	ATT	The B	The Altimeter system has been to be	
	A	9.0 10	in accordance with FAR part 91.411	
			The ATC Transponder has been tested	
			in accordance with FAR part 91.413	
			The ELT battery due date is 5/94	
		A Priva	All outstanding Airworthiness Directiv	ves
Charlest at the			complied with as of 02/22/93 93-041	
			fue 5	1
	+13	A JAMES	Ste	ve s
			Vice President Qua	lity
	1.	1 1 2 A 1	Commander Ai	100
	-47	1 1		٠.
A - California	50 50 E	n Alexander	I find that this Aircraft meets the r	egu
MAR 2 8 19	93 7.3	7.3		
24VF 3	HARLS.	17574.	for an EXPORT Airworthiness Certifica	n o
	wo 52	1 1000	accordance with FAR 21.329, dated MA	D 9
ALBH CONTRACTOR		10.11	The next annual inspection is due MA	m 4
		1	The "E" card # is E253356	hu
	A North		Wa	yne
			DA	1R-4
	1912			
		2 17	1.15	1 1
			2. 1 3 1 3 1 3 1 3 1 3 1 3 1 3 1 3 1 3 1	

rection - Checked of on of movement, security. nd proper taguel - tubed all controls -Cheched All lights, Times wheels brakes Master colinders, bottery - Jacked sinconft - Cleared Fregetto + Reprehed All wheel bearings - Installed New 500.5 GPhy Goodyean line on Nose wheel assy and Removed RT Main Genn Actuator Acing AND Replaced All "O" Rives Beinstalled Actuator and checked Rigging-Per Formed Gern Retarction Test and emergency extension - All Andio + Usual warrings check ok - Lubed landing gean. Bemoved Roda - Replaced upper Rudder bearing and Reinstalled - Removed RH Elevator Assy- Replaced out & bearing And bushing - Reinstalled elevator -Replaced bearings in elevator control Rod STAFT bulkhend (upper + lower) And Reinstalled ROD - Checked elevator and Ruddel For proper Track - Ketorqued Full Attach Solt on Rt. Main Genr - Replaced outbo God end on It sileron - Removed Foreign external Data plate From side a Air craft And installed aga Rivets Tolled wine obcords in cock

## AIRFRAME

Company Manufacturer GULFSTREAM (SEE ALSO GRUMMAN, SCHWEIZER, FRAKES, FLYING BOAT)
114B
14574
Tach Time 1466.6

Model Serial #

A/C Cert. Date

**Total Time** 

Tail Number N114ET

AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance		R	Next Comp @ Hrs/Date	Authorized Signature & Number	
A STATE OF THE STA	SUPERSEDED BY AD 2012-02-10/		NA SUPERSEDED .	×		N/A	2/4	- 3595.243 Ar
04/04/11 *12-02-10 03/15/12	LEFT AND RIGHT ELEVATOR SPARS/	3/23/22 1466.6	C/W BY INSPECTION I/C/W AMOC L115W -11-315		X	100 HRS/ 12MO	Sen. 1	J58534340

## APPLIANCES

Company Tail #

N114ET

AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance	0	R	Next Comp @ Hrs/Date	Authorized Signature & Number
* <b>76-07-12</b> 08/30/77	[R1] IGNITION SWITCHES/		N/A BY P/N	-	X	N/A	
*93-05-06 04/29/93	IGNITION SWITCHES/	3/23/22 1466.6	C/W BY INSTALLING KIT A-3659-2 ICW SB92-01		X	2K HRS	1 L. L 3595343H

## FAA Approved Model List (AML) STC SA02636SE

	Airplane Make (TCDS holder) [common name or previous make]	Aircraft Model (Alias)	Type Certificate Number	TC Certification Basis [1]	MDL 005-01206-A1 Approved Rev. [2]	AML Amended Date	AFMS Part Number	AFMS Revision [2]
	Cessna (Cessna Aircraft Company)	T-50 (Army AT-17, UC-78 series, and Navy JRC-1)	A-722	CAR 4a	1	Original	190-02207-A3	. 1
	Cirrus Design Corporation (Cirrus Design Corporation)	SR20, SR22	A00009CH	FAR 23	1	Original	190-02207-A3	1
	Commander (Commander Aircraft Corporation) [CPAC, Inc.]	112, 112TC, 112B, 112TCA, 114, 114A, 114B, 114TC	A12SO	FAR 23	1	Original	190-02207-A3	. 1
	Costruzioni Aeronautiche Tecnam S.P.A (Costruzioni Aeronautiche Tecnam S.P.A) [Costruzioni Aeronautiche Tecnam srl]	P2006T	A62CE	FAR 23	1	Original	190-02207-A3	1
	Cub Crafters (Cub Crafters, Inc.)	CC19-180	A00053SE	FAR 23	1	Original	190-02207-A3	1
	Diamond (Diamond Aircraft Industries Inc.)	DA20-A1, DA20-C1	TA4CH	FAR 23	1	Original	190-02207-A3	1
1	Diamond (Diamond Aircraft Industries Inc.) [Diamond Aircraft Industries GmbH]	DA 40	A47CE	FAR 23	1 ,	Original	190-02207-A3	
	Discovery (Discovery Aviation, Inc.) [Liberty Holdings, LLC]	XL-2	A00008DE	FAR 23	1	Original	190-02207-A3	
"	EADS-PZL "Warszawa- Okęcie" (EADS-PZL Warszawa-Okęcie" S.A.) Panstwowe Zaklady Lotnicze]	PZL-104 WILGA 80, PZL- 104M WILGA 2000, PZL- 104MA WILGA 2000	A55EU	FAR 23	1	Original	190-02207-A	