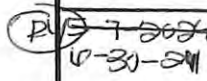


HISTORICAL RECORD - TECHNICAL INSTRUCTION COMPLIANCE RECORD

Lebenslaufakte - Nachweis der Durchführung technischer Anweisungen

PAGE OF PAGES
Seite von Seiten

1. NOMENCLATURE OF EQUIPMENT Bezeichnung des Geräts Hauptrotorblatt / Main Rotor Blade			2. MODEL AND MANUFACTURER L621 M 1010 055 Baumuster und Hersteller Eurocopter Deutschland GmbH		3. SERIAL NUMBER Werk-Nr. 1389		4. ACCEPTANCE DATE Übernahme-Datum 06. Aug. 02
DATE Datum A	NUMBER Nr. B	TITLE Titel C	DATE COMPLETED Durchgeführt am D	INSPECTOR Prüfer E	MAINTENANCE OFFICER Werfleiter F	ORGANIZATION Einheit G	
30. Jun. 04	SB EC135-62-016	Abdichtung der Stirnwand am Ende der Steuertüte Sealing of end face wall inside pitch control cuff	13. Sep.05			Eurocopter Deutschland GmbH Donauwörth	
09. Jul. 04	SB EC135-62-015	Einführung eines verbesserten Erosionsschutzes und Umkennzeichnung des Hauptrotorblattes Introduction of improved erosion protection and reidentification of main rotor blade	13. Sep.05			Eurocopter Deutschland GmbH Donauwörth	
12. Okt. 04	SB EC135-62-017	Einführung eines Sand-Erosionsschutzes an der Blattspitze Introduction of a sand erosion protection at the blade tip					
07-17-98	ASB- EC 135-62A-003	Inspect Pitch Control Cuff -----	3/27/06		-----	American Eurocopter LLC CRS#VY1R575K	
03-20-03	SB-EC 135-62-014	Visual inspection IAW Section 3A -----	3/27/06		-----	American Eurocopter LLC CRS#VY1R575K	
09-11-2006	ASB EC135- 62A-022	The main rotor blade is not affected.-----	09/29/06		-----	American Eurocopter LLC CRS#VY1R575K	
07-17-98	ASB EC135- 62A-003	Inspect Pitch Control Cuff.-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
09-11-06	ASB EC135- 62A-005	Complied with-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
10-18-01	SB EC135- 62-008	Visual Plate Washer Check.-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
11-08-01	SB EC135- 62-009	Inspect Static Discharger.-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
03-20-03	SB EC135- 62-014	Visual inspection per section 3A preflight check.-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
06-30-04	SB EC135- 62-016	Inspection of sealed cuff.-----	09/05/07		-----	American Eurocopter LLC CRS#VY1R575K	
2007-05-02	SB EC135- 62-024	MODIFICATION OF LOWER FACE OF PITCH CONTROL CUFF	3-7-08		-----	American Eurocopter LLC CRS#VY1R575K	
 7-5084 10-30-04	SB-EC135- 62-040	Improved MRB bearing support Assembly	5-7-24	RL		PHE	

HISTORICAL RECORD - TECHNICAL INSTRUCTION COMPLIANCE RECORD Lebenslaufakte - Nachweis der Durchführung technischer Anweisungen

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1. NOMENCLATURE OF EQUIPMENT Bezeichnung des Geräts		2. MODEL AND MANUFACTURER Baumuster und Hersteller	3. SERIAL NUMBER Werk-Nr.		4. ACCEPTANCE DATE Übernahme-Datum	
DATE Datum A	NUMBER Nr. B	TITLE Titel C	DATE COMPLETED Durchgeführt am D	INSPECTOR Prüfer E	MAINTENANCE OFFICER Werfleiter F	ORGANIZATION Einheit G
08. Feb. 10 08. Nov. 13	SB EC135-62-023 Rev. 02	Hauptrotorsystem - Hauptrotorblatt - Nachrüstung zusätzlicher PU-Schutzfolien Main rotor system - Main rotor blade - Retrofitting additional PU-strips	By customer			Checked by AHD, 23. Nov. 20 
31. May 17	SB EC135-62-039	Hauptrotorsystem – Hauptrotorblatt - Umrüstung zu verbesserten Trimmblechen Main rotor system – Main rotor blade – Retrofit of the improved trim tabs	23. Nov. 20			Airbus Helicopters Deutschland GmbH, Donauwörth

HISTORICAL RECORD - SIGNIFICANT HISTORICAL DATA

Lebenslaufakte - Wichtige Lebenslaufdaten

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PAGES 1
Seiten 1

1. NOMENCLATURE OF EQUIPMENT Bezeichnung des Geräts Hauptrotorblatt / Main Rotor Blade		2. MODEL AND MANUFACTURER Baumuster und Hersteller L621 M 1010 055 056 Eurocopter Deutschland GmbH	3. SERIAL NUMBER Werk-Nr. 1389	4. ACCEPTANCE DATE Übernahme-Datum 06. Aug. 02
DATE A	REMARKS B	ORGANIZATION Einheit C		
06. Aug. 02	Stückprüfung durch ECD - DON am 06.08.02 durchgeführt / Production inspection accomplished by ECD. Gewichts- und Schwerpunktsdaten G (W) = 38.869 kg, GTL (WTR) = 18.412 kg, ML = 85800 kgmm. Weight and balance data. Vulkanisations-Datum-Dämpfer / Vulkanisations-Date-Damper: LK2934 1st QR 02 LK2937 1st QR 02	Eurocopter Deutschland GmbH Donauwörth ECD D-111		
23 March 2005	Inspected as per AMM 62.11.00.6-1. Blade has crack at station 1975 bottom surface. Return blade to ECD for repair. Blade is found repairable this date on AE LLC WO#W096489.	American Eurocopter LLC CRS#VY1R575K		
13. Sep. 05	Instandsetzung gem. RV L621M1018 D01 bei TT: 1569:24 von ECD durchgeführt. Repair in acc. with RV L621M1018 D01 at TT: 1569:24 accomplished by ECD. PU-Erosionsschutz erneuert, Trimmbleche erneuert, Gelenklager für Lagerhalter erneuert. Hautreparatur Fahne (1-Meter-Rep. / Abschnitt 05-01-04) durchgeführt. Replaced PU-erosion strips, replaced trim tabs, replaced spherical bearing of support reassy. repaired T/E delamination (1-meter-rep. / chapter 05-01-04). CT-Prüfung gem. PV L621 M1106 X01 durchgeführt / CT-inspection performed in accordance with PV L621 M1106 X01 Nachprüfung durch ECD durchgeführt / Reinspection accomplished by ECD Gewichts- und Schwerpunktsdaten G (W) = 38,981 kg, GTL (WTR) = 18.423 kg, ML = 85851 kgmm Weight- and balance data. Überprüfung auf helle Hautlagen gem. RV L621M1018 D01 Abschn. 04-01-05 "Index D008" durchgeführt. Inspection of dry fabric ply's accomplished in acc. with RV L621M1018 D01 chapter 04 01-05 "Issue D008". Alle periodischen Inspektionen wurden im Rahmen dieser Instandsetzung gem. dem gültigen Aircraft-Maintenance-Manual (AMM) durchgeführt. All periodic inspections were executed during this maintenance in accordance with the valid Aircraft-Maintenance-Manual (AMM). Wegen Durchführen SB-EC135-62-015 neues TKZ: L 621 M 1010 056 Due to performing of SB-EC 135-62-015 new part-no. L 621 M 1010 056	Eurocopter Deutschland GmbH Donauwörth Qualified Technician No. 313		
3/27/06	Inspected as per AMM 62-11-00,6-1, Paint touch-up per AMM 62-11-00, 6-1. Blade found to be serviceable this date on American Eurocopter work order W112778.	American Eurocopter LLC CRS#VY1R575K		
09/29/06	Inspected per ASB EC135-62A-022. The main rotor blade is not affected.	American Eurocopter LLC CRS#VY1R575K		
09/05/07	Inspected per 62-11-00, 6-1, and paint touch up. Details on file on AELLC WO# W134214.	American Eurocopter LLC CRS#VY1R575K		
Sep 16 2019	Return Blade to AHD for repair Due to cracked and de-bonded out of limits. Detail on file at AHI S0#11000481086.	Airbus Helicopters CRS#VY1R575K.		

HISTORICAL RECORD - SIGNIFICANT HISTORICAL DATA

Lebenslaufakte - Wichtige Lebenslaufdaten

PAGE 2 OF 2
Seite 2 von 2

1. NOMENCLATURE OF EQUIPMENT Bezeichnung des Geräts		2. MODEL AND MANUFACTURER Baumuster und Hersteller		3. SERIAL NUMBER Werk-Nr. 1389		4. ACCEPTANCE DATE Übernahme-Datum	
DATE A		REMARKS Bemerkungen B				ORGANIZATION Einheit C	
23. Nov. 20		Instandsetzung gem. RI L621M1018 E01 "F" bei TT: 6292:13 von AHD durchgeführt. Repair in acc. with RI L621M1018 E01 "F" at TT: 6292:13 accomplished by AHD. - siehe Work Report Nr. 101188186 - see work report- no. 101188186 Torsionsprüfung gem. RI L621M1018E01 Index „F“ durchgeführt / Torsional inspection accomplished in accordance with RI L621M1018E01 Issue „F“. CT-Prüfung gem. PV L621 M1106 X01 durchgeführt und zulässig gemäß RDAS Nr. MRBL 110/2020 Index A. CT-inspection performed in accordance with PV L621 M1106 X01 and permitted in acc. with RDAS No. MRBL 110/2020 Issue A. E-Schutz erneuert, PU-Erosionsschutz erneuert (innen/ausen), Trimmbleche erneuert, Gfk-Tape erneuert, Entlader erneuert (vollst.), Teflontape erneuert (Saugseite/ Druckseite), Lochleibungslaminat erneuert (Saugseite), Bolzen erneuert, Flanschbolzen erneuert, Hautreparatur durchgeführt, Hautreparatur Formstoss durchgeführt, Fahnenreparatur durchgeführt (Trimmblechbereich), Fahnenreparatur durchgeführt (Ableiterbereich), Hautreparatur Tiefenwuchtkammerbereich(Abschnitt 05-01-02) durchgeführt, Fahnenreparatur (bei R 1540) durchgeführt (Abschnitt 05-04-00), Hautreparatur Fahne Abschnitt 05-01-04 durchgeführt, Hauptbolzenbuchsen erneuert, statisch und dynamisch gewuchtet. Durchmesser der Bohrlöcher im Anschlussbereich zulässig gem. RDAS Nr. MRBL 110/2020 Index A. Bohrung für Flanschbolzen gem. RDAS Nr. MRBL 110/2020 Index A instandgesetzt. renewed L/E strip, replaced PU-erosion strips (inside/outside), replaced trim-tabs, replaced auxiliary tab, replaced discharger (assy.), renewed teflon liner (upper side / lower side), replaced bearing stress lay-up (upper side), renewed bolts, renewed flange bolt, repaired skin delamination, repaired skin delamination at mould interface, repaired T/E delamination (trim tab area), repaired T/E delamination (discharger area), repaired skin delamination weight chamber area (Chapter 05-01-02), repaired (at R 1540) T/E delamination (Chapter 05-04-00), repaired skin delamination at T/E chapter 05-01-04, replaced main bolt bushings, statically and dynamically balanced. Diameter of the boreholes in attachment area permitted in acc. with RDAS No. MRBL 110/2020 Issue A. Repaired boreholes for flange bolt in acc. with RDAS No. MRBL 110/2020 Issue A. Kontrolle der Verklebung der E-Schutz-Schalen nach 5-10 Flugstunden gemäß AMM 62-11-00, 6-1. Check the bonding of the erosion protection shell after 5-10 flight hours according to AMM 62-11-00, 6-1. Gewichts- und Schwerpunktsdaten G (W) = 39.091 kg, GTL (WTR) = 18.422 kg, ML = 85848 kgmm Weight- and balance data. Pretracking of Pitchlink - 0,2 mm Lead Lag at 0 degree + 2,0 mm Lead Lag at 10,4 degree + 1,1 mm Alle Inspektionen wurden im Rahmen dieser Instandsetzung gemäß dem gültigen MSM EC 135 Rev. 001 Kapitel 05-22-00, 62-11; Kapitel 05-24-00, 62-11 und Kapitel 05-25-00, 62-11 und Airworthiness Limitation Section (ALS) EC 135 Rev. 001 Kapitel 04-20-00, 62-11 durchgeführt. All inspections were executed during this maintenance in accordance with the valid Master Servicing Manual (MSM) EC 135 Rev. 001 chapter 05-22-00, 62-11; chapter 05-24-00, 62-11 and chapter 05-25-00, 62-11 and Airworthiness Limitations Section (ALS) EC 135 Rev. 001 chapter 04-20-00, 62-11.				Airbus Helicopters Deutschland GmbH, Donauwörth 	

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PHI HEALTH, LLC
Certificate No. VE7A070N
2800 N 44th Street, Suite 800
Phoenix, AZ 85008 USA

COMPONENT STATUS REPORT

Comp. PN: L621M1003101
Comp. SN: 24-LK41750
Comp. ID: MRS3-006212-2025

Part Description: DAMPER
Part Category ~ Family: EC135P2 ~ EC135P2
Manufacture Date:

Total Hours: 0.00
Total Take Offs: 0
Report Type: Regulatory

Lvl	Pos	Work Unit	Part No ~ Part Description ~ Part Serial	Service Life		T.T	TS-IN/OH	Rem	Due at	
				Usage	Date	T.C	CS-IN/OH	Units	Comp Hr	Date
		MR-DAMPER EC135 RT	L621M1003101 ~ DAMPER ~ 24 LK41750	8000 FH RTR	120 ME	0 00		8000.00	8,000.00	3/31/2035

MS



6										Modification and Service Bulletin checking Contrôle d'exécution des modifications et des Services Bulletins										LOG CARD / FICHE MATRICULE*										Log card N° 1 Fiche N°																													
No. Numéro					Type of modification Nature de la modification					Performing unit or contractor Unité ou société d'exécution					Date of embodiment and inspection stamp Date d'exécution et tampon de contrôle					Follow up Sheet for new equipment / Fiche suivie Matériel neuf																																							
																				1 Material identification / Identification du matériel																																							
																				Name Dénomination										MAIN ROTOR BONDED DAMPER																													
																				NATO Nomenclature Nomenclature OTAN																																							
																				AIRBUS HELICOPTERS Part number Référence										L621M1003101																													
																				Manufacturer's Part number Référence fabricant										J-18862-10										NATO Manuf code Code OTAN fabricant 76005																			
																				Delivery configuration Configuration de livraison																																							
																														Version																													
																				Serial number Numéro de série										24-LK41433										Amendments Amendements																			
																				Supplier Fabricant										LORD Corporation										Type Typ																			
																				2 Contract or order / Marché ou commande																																							
																				Reference Référence																																							
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																				Issuing agency Organisme émetteur										Batch N° N° de lot																													
																				Contrator Fournisseur										Address Adresse																													
																				3 Guarantee / Garantie																																							
																				Equipment Matériel					Date of delivery Date de livraison					Guaranteed storage period Durée garantie de stockage					Service date Date de mise en service					Guarantee operation period Durée Garantie de fonctionnement																			
5					Contractor Unité ou société					Date Date					Aircraft Version and S/N Version et No Appareil					Operation Fonctionnement					Certification/ Attestation					4 Special information / Renseignements particuliers																													
																														Appendix to table 4 : Annexe au tableau 4										YES ☐ NO ☒ OUI NON										Nb of sheets : Nb de fiches									
																														Inventory of lifed components: Inventaire des pièces à Durée de Vie										YES ☐ NO ☒ OUI NON										Nb of sheets : Nb de fiches									
																														Transfer sheet: Fiche Matricule de Transfert										YES ☐ NO ☒ OUI NON										Nb of sheets : Nb de fiches									
LORD Corp.					1Q-25										0 Hr					Manuf/Receipt* Fab/Reception																																							
																				Depreservation/* Destockage																																							
																				Period Overhaul/* V.P																																							
																				Fitting/* Montage																																							
																														Operating limit Limite de fonctionnement										Life limit Limite de vie										ATA									



PHI HEALTH, LLC
Certificate No. VE7A070N
2800 N 44th Street, Suite 800
Phoenix, AZ 85008 USA

COMPONENT STATUS REPORT

Comp. PN: L621M1003101
Comp. SN: 24-LK41433
Comp. ID: MRS3-006213-2025

Part Description: DAMPER
Part Category ~ Family: EC135P2 ~ EC135P2
Manufacture Date:

Total Hours: 0.00
Total Take Offs: 0
Report Type: Regulatory

Lvl	Pos	Work Unit	Part No ~ Part Description ~ Part Serial	Service Life		T.T	TS-IN/OH	Rem	Due at	
				Usage	Date	T.C	CS-IN/OH	Units	Comp Hr	Date
		MR-DAMPER-EC135-RT	L621M1003101 ~ DAMPER ~ 24 LK41433	8000 FH RTR	120 ME	0.00		8000 00	8,000.00	3/31/2035

Handwritten signature/initials



LOG CARD

MRJ-15364

Log Card No:

1. MATERIAL IDENTIFICATION

Name	Main Rotor Blade	
NATO Nomenclature		
ECD Part number	L621 M1010 056	
Manufacturer's P/N		NATO Code
Delivery config.		Version "E006"
Serial number	2605	Amendments
Supplier	EUROCOPTER DEUTSCHLAND GmbH	Typ

2. CONTRACT OR ORDER

Reference	Date
Issuing agency	Kit No.
Contractor	Address

3. GUARANTEE

Material	Date of delivery	Guaranteed storage period	Service date	Guarantee operation period

4. SPECIAL INFORMATIONS

Monitoring of life limited components	YES ☒	NO ☐	No. of sheet	: 1
Appendix to table 4	YES ☐	NO ☒	No. of sheet	:
Transfer sheet	YES ☐	NO ☒	No. of sheet	:

Remarks:

Production inspection accomplished by ECD, Date Apr.24.2006

Weight and balance data : $G(W) = 39,064 \text{ kg}$

Weight and balance data : GTL(WTR) = 18,422 kg

Weight and balance data : ML = 85848 kgmm.

Operating limit (TBO)	Life limit (TCI):	ATA 100 Ch 62
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[illegible]

5. OPERATIONS

Contractor	Date	Aircraft Version and S/N	Support	Partial	Total	Certification
						ECD D-121
ECD	Apr.2006	-----	new	-----	new	Manufacturing
						Depreservation
						Period.Overhaul
ECD	May.2006	EC 135 P2 0489	new	-----	new	Fitting

7. SUCCESSIVE STATUS, MAINTENANCE AND OVERHAUL OPERATIONS

Contractor	Date	Aircraft Version and S/N	Operation Unit (hours)			Reason for transfer (code and symptoms) - Accomplished work - Replaced parts Successive status before ACFT delivery - see front page of table 5
			Support	Partial	Total	
PHI	10-06	EC135P2 0489	New		New	Received installed new on N329PH
PHI	9-17-19	EC135P2 0489	5800.56		5800.56	Removed.
AHI	28 Oct 19	CRS#VY1R575K	-	TSN:	5800.56	Blade was sent in due to Deflection incident. Return blade to AHD for CT scann. Detail on file at AHI under SO#11000487590.-----
AHD	21. Sep. 20				5800:34	Repair in acc. with RI L621M1018E01 Issue F accomplished by AHD. - see work report- no. 101203317 CT-inspection performed in accordance with PV L621 M1106 X01 .
						Replaced RFP shell, replaced PU-erosion strips (inside/outside), replaced trim-tabs, replaced auxiliary tab, replaced discharger (assy.), replaced bearing stress lay-up (upper side), renewed teflon liner (upper side/ lower side), renewed bolts, replaced spherical bearing of support, repaired skin delamination at mould interface,
						repaired T/E delamination (trim tab area), repaired skin delamination Profil Nose-Seal assy (chapter 05-03-01), repaired skin delamination, repaired skin delamination at T/E chapter 05-01-04, replaced main bolt bushings, statically and dynamically balanced.
						Inspection after deflection incident in acc. with RDAS No. MRBL 145/2020 Issue A. Weight and balance data G (W) = 39,241 kg; GTL (WTR) = 18,422 kg; ML = 85848 kgmm. Pretracking of Pitchlink ± 0,0 mm Lead Lag at 0 degree + 2,9 mm Lead Lag at 10,4 degree + 3,5 mm
						All inspections were executed during this maintenance in accordance with the valid Master Servicing Manual (MSM) EC 135 Rev. 001 chapter 05-22-00, 62-11; chapter 05-24-00, 62-11 and chapter 05-25-00, 62-11 and Airworthiness Limitations Section (ALS) EC 135 Rev. 001 chapter 04-20-00, 62-11 .
PHI	2-1 2021	EC135P2 0389	9489.37		5800.56	Repaired, Airbus, Ro-057449-2019, 850649817, 12-4-20, By B JPH Installed on N307PH.

INVENTORY OF COMPONENTS

PAGE 1

- 1 - Date of installation
- 2 - Serial number
- 3 - Operating time of component
- 4 - Operating time of assembly
- 5 - Component number of cycles
- 6 - Assembly number of cycles
- 7 - Version

INST./REPLACE (*)

1	3	5
2	4	6

ASSEMBLY DESIGNATION	ASSEMBLY REFERENCE	SERIAL NUMBER	VERSION
Main Rotor Blade	L621 M1010 056	2605	"E006"

COMPONENTS

[illegible]



Comp Category: EC135P2

Comp ID: MRS-15364

Part No.: L621M1010056

PHI, Inc.

Component Status Report

Part Desc: Blade, M/R (See Remarks)

Comp Serial#: 2605

Comp TT: 5,800.56

Comp. ST:

Pos	Work Unit	Part No ~ Part Desc. ~ Part Serial	Service Life Usage	Date	T.T TC	TS-IN/OH CS-IN/OH	Rem Units	Due at Comp Hr. Date
	10H-M/R BLADE-COND-INSP	L621M1010056-Blade, M/R (See Remarks)-2605	5810.56 FH-INS		5,800.56	5,800.56	10.00	5,810.56
	200H-M/R BLADE-COND-INSP	L621M1010056-Blade, M/R (See Remarks)-2605	6000.56 FH-INS		5,800.56	5,800.56	200.00	6,000.56
	M/R BLADE INSP	L621M1010056-Blade, M/R (See Remarks)-2605	1100 FH-INS		5,800.56	0.00	1,100.00	6,900.56
	✓ MR-BLADE-EC135-1000H-36M-IN	L621M1010056-Blade, M/R (See Remarks)-2605	1000 FH-INS	36 ME	5,800.56	0.00	1,000.00	6,800.56
621102010-UP	✓ MR DAMPER EC135 RT	L621M1003101-Damper-24-LK13220	8000 FH-RTR	10 YME	2,797.84		5,202.16	11,002.72 06/30/22
621102010-LWR	✓ MR DAMPER EC135 RT	L621M1003101-Damper-24-LK13601	8000 FH-RTR	10 YME	2,797.84		5,202.16	11,002.72 06/30/22

2-29-21
ID

2-5-21

6 Modification and Service Bulletin checking <i>Contrôle d'exécution des modifications et des Services Bulletins</i>						LOG CARD / FICHE MATRICULE*				Log card N° 1 Fiche N°									
No. Numéro		Type of modification Nature de la modification		Performing unit or contractor Unité ou société d'exécution		Date of embodiment and inspection stamp Date d'exécution et tampon de contrôle		Follow up Sheet for new equipment / Fiche de suivi de Matériel neuf											
								1 Material identification / Identification du matériel											
								Name / Dénomination MAIN ROTOR BONDED DAMPER											
								NATO Nomenclature / Nomenclature OTAN											
								AIRBUS HELICOPTERS Part number / Référence L621M1003101											
								Manufacturer's Part number / Référence fabricant J-18862-10		NATO Manuf code / Code OTAN fabricant 76005									
								Delivery configuration / Configuration de livraison											
								Serial number / Numéro de série 24-LK30078		Amendments / Amendements									
								Supplier / Fabricant LORD Corporation		Type / Typ									
								2 Contract or order / Marché ou commande											
								Reference / Référence				Date / Date							
Issuing agency / Organisme émetteur				Batch N° / N° de lot															
Contrator / Fournisseur				Address / Adresse															
								3 Guarantee / Garantie											
								Equipment / Matériel		Date of delivery / Date de livraison		Guaranteed storage period / Durée garantie de stockage		Service date / Date de mise en service		Guarantee operation period / Durée Garantie de fonctionnement			
								4 Special information / Renseignements particuliers											
								Appendix to table 4 : / Annexe au tableau 4				YES ☐ NO ☒		Nb of sheets : / Nb de fiches :					
								Inventory of lifed components: / Inventaire des pièces à Durée de Vie				YES ☐ NO ☒		Nb of sheets : / Nb de fiches :					
								Transfer sheet: / Fiche Matricule de Transfert				YES ☐ NO ☒		Nb of sheets : / Nb de fiches :					
								Operating limit / Limite de fonctionnement		Life limit / Limite de vie		ATA							
								5 Contractor Unité ou société		Date Date		Aircraft Version and S/N Version et No Appareil		Operation Fonctionnement Support Partial Total		Certification/ Attestation 			
LORD Corp.		4Q-21						0 Hr		Manuf/Receipt* / Fab/Reception									
										Depreservation/* / Destockage									
										Period Overhaul/* / V.P									
										Fitting/* / Montage									

*Delete as necessary / Supprimer la mention inutile

AIRBUS HELICOPTERS F 16-04B3



PHI HEALTH, LLC
Certificate No. VE7A070N
2800 N 44th Street, Suite 800
Phoenix, AZ 85008 USA

COMPONENT STATUS REPORT

Comp. PN: L621M1003101
Comp. SN: 24-LK30078
Comp. ID: MRS3-004233-2022

Part Description: DAMPER
Part Category ~ Family: EC135P2 ~ EC135P2
Manufacture Date:

Total Hours: 0.00
Total Take Offs: 0
Report Type: Regulatory

Lvl	Pos	Work Unit	Part No ~ Part Description ~ Part Serial	Service Life		T.T	TS-IN/OH	Rem	Due at	
				Usage	Date	T.C	CS-IN/OH	Units	Comp Hr	Date
		MR-DAMPER-EC135-RT	L621M1003101 ~ DAMPER ~ 24-LK30078	8000 FH RTR	120 ME	0.00		8000.00	8,000.00	12/31/2031

6/15/22
AB

6

Modification and Service Bulletin checking

Contrôle d'exécution des modifications et des Services Bulletins

LOG CARD / FICHE MATRICULE*

Log card N°

1

Fiche N°

Follow-up Sheet for new equipment / Fiche suivie Matériel neuf

1

Material identification / Identification du matériel

Name

Dénomination

MAIN ROTOR BONDED DAMPER

NATO Nomenclature

Nomenclature OTAN

AIRBUS HELICOPTERS Part number

Référence

L621M1003101

Manufacturer's Part number

Référence fabricant

J-18862-10

NATO Manuf code

Code OTAN fabricant

76005

Delivery configuration

Configuration de livraison

Version

Version

Serial number

Numéro de série

24-LK30014

Amendments

Amendements

Supplier

Fabricant

LORD

Corporation

Type

Typ

2

Contract or order / Marché ou commande

Reference

Référence

Date

Date

Issuing agency

Organisme émetteur

Batch N°

N° de lot

Contractor

Fournisseur

Address

Adresse

3

Guarantee / Garantie

Equipment

Matériel

Date of

delivery

Date de

livraison

Guaranteed

storage period

Durée garantie de

stockage

Service date

Date de mise

en

service

Guarantee operation

period

Durée Garantie de

fonctionnement

4

Special information / Renseignements particuliers

Appendix to table 4 :

Annexe au tableau 4

YES

OUI

NO

NON

☒

Nb of sheets

Nb de fiches

:

Inventory of lifed components:

Inventaire des pièces à Durée de Vie

YES

OUI

NO

NON

☒

Nb of sheets

Nb de fiches

:

Transfer sheet:

Fiche Matricule de Transfert

YES

OUI

NO

NON

☒

Nb of sheets

Nb de fiches

:

Operating limit

Limite de fonctionnement

Life limit

Limite de vie

ATA

5

Contractor
Unité ou sociétéDate
DateAircraft
Version
and S/N
Version et
No AppareilOperation
FonctionnementSupport
SupportPartial
PartielTotal
TotalCertification/
AttestationManuf/Receipt*
Fab/ReceptionDepreservation/*
DestockagePeriod Overhaul/*
V.PFitting/*
Montage

LORD Corp.

1Q-22

0 Hr

*Delete as necessary / àayer la mention inutile

AIRBUS HELICOPTERS F 16-04B3

**Delete as necessary / Rayer la mention inutile*



PHI HEALTH, LLC
Certificate No. VE7A070N
2800 N 44th Street, Suite 800
Phoenix, AZ 85008 USA

COMPONENT STATUS REPORT

Comp. PN: L621M1003101
Comp. SN: LK30014
Comp. ID: MRS3-004341-2022

Part Description: DAMPER
Part Category ~ Family: EC135P2 ~ EC135P2
Manufacture Date: 1/15/2022

Total Hours: 0.00
Total Take Offs: 0
Report Type: Regulatory

Lvl	Pos	Work Unit	Part No ~ Part Description ~ Part Serial	Service Life		T.T	TS-IN/OH	Rem	Due at	
				Usage	Date	T.C	CS-IN/OH	Units	Comp Hr	Date
		MR-DAMPER-EC135-RT	L621M1003101 ~ DAMPER ~ LK30014	8000 FH RTR	120 ME	0.00		8000.00	8,000.00	1/31/2032

1Q-22

gm

3-31-32

R

U-15-22

6 Modification and Service Bulletin checking Contrôle d'exécution des modifications et des Services Bulletins						
No. Numéro	Type of modification Nature de la modification	Performing unit or contractor Unité ou société d'exécution	Date of implementation and inspection stamp Date de mise en œuvre et attestation de contrôle			
5 Contractor Unité ou société		Date Date	Aircraft Version and S/N Version et No Appareil	Operation Fonctionnement		Certification Attestation
				Support Support	Partial Partiel	Total Total
						Manuf/Receipt / * Fab/Reception
						Depreservation / * Destockage
						Period.Overhaul / * V.P
						Fitting / * Montage

LOG CARD / FICHE MATRICULE * Log card N° Fiche N°				2
Follow-up Sheet for new equipment / Fiche suiveuse Matériel neuf *				
1 Material identification/ Identification du matériel				
Name Dénomination		MAIN ROTOR BLADE		
NATO Nomenclature Nomenclature OTAN				
AIRBUS HELICOPTERS Part num Référence		L621M1010057		
Manufacturer's Part number Référence fabricant		NATO Manuf code Code OTAN fabricant		
Delivery configuration Configuration de livraison		Version "B1" Version		
Serial number Numéro de série		6128		
Supplier Fabricant		Eurocopter Deutschland GmbH		
		Amendments Amendements		
		Type Type		
2 Contract or order/ Marché ou commande				
Reference Référence		Date Date		
Issuing agency Organisme émetteur		Batch N° N° de lot		
Contrator Fournisseur		Address Adresse		
3 Guarantee / Garantie				
Equipment Matériel	Date of delivery Date de livraison	Guaranteed storage period Durée garantie de stockage	Service date Date de mise en service	Guarantee operation period Durée Garantie de fonctionnement
4 Special information / Renseignements particuliers				
Appendix to table 4 : Annexe au tableau 4		YES ☐ NO ☒	Nb of sheets : Nb de fiches :	
Inventory of lifed components: Inventaire des pièces à Durée de Vie		YES ☒ NO ☐	Nb of sheets : Nb de fiches :	
Transfer sheet: Fiche Matricule de Transfert		YES ☐ NO ☒	Nb of sheets : Nb de fiches :	
		OUI NON		
		OUI NON		
		OUI NON		
Operating limit Limite de fonctionnement		Life limit Limite de vie	ATA 62	

7 Successive status and minor/major maintenance and overhaul operations									
Positions successives et opérations d'entretien et de remise en état mineures et majeures									
Contractor Unité ou société	Date Date	Aircraft Version and S/N Version et N° Appareil	Operations / Fonctionnement						Reason for transfer (code and symptoms) - Accomplished work - Replaced parts Motif du mouvement (code symptômes) - Travaux effectués - Pièces changées Successive status before A/C delivery - See front page of table 5 Positions successives avant livraison avion - Voir recto tableau 5
			Unit / Unité: Support	Partial Partiel	Total Total	Unit / Unité: Support	Partial Partiel	Total Total	
AHI	10Aug2022	CRS# VY1R575K	———	TSN:	4759.84	———			Inspected per R.I. L621M1018E01 Issue F1. Not repairable at AHI due to L/E impact and cracks on the trailing edge. Return to AHD for further inspection and repair. Details on file at AHI under S/O# 11000657258
AHD	12. Jul. 23				4759:49				Repair in acc. with RI L621M1018E01 Issue F1 accomplished by AHD. - see work report- no. 101462992
									Torsional inspection accomplished in accordance with RI L621M1018E01 Issue F1. CT-inspection performed in accordance with PV L621 M1106 X01
									Repaired skin delamination at mould interface, replaced erosion protection, replaced trim-tabs, replaced PU-erosion strips (inside/outside), replaced auxiliary tab, replaced discharger (assy.), replaced bearing stress lay-up (upper side), renewed teflon liner (upper side/ lower side), replaced spherical bearing of support,
									repaired T/E delamination (trim tab area), repaired T/E delamination (discharger area), repaired skin delamination, repaired skin delamination at T/E in acc. with RDAS No. MRBL 302/2020 Issue A, replaced main bolt bushings,
									statically and dynamically balanced.
									Check the bonding of the erosion protection shell after 5-10 flight hours according to AMM 05-52-00, 6-2.
									Weight and balance data G(W)= 39,075 kg ; GTL(WTR)= 18,422 kg;ML= 85848 kgmm.
									Pretracking of Pitchlink - 1,9 mm Lead Lag at 0 degree - 1,2 mm Lead Lag at 10,4 degree - 0,8 mm
									All inspections were executed during this maintenance in accordance with the valid Master Servicing Manual (MSM) EC 135 Rev. 004 chapter 05-22-00, 62-11;
									chapter 05-24-00, 62-11; chapter 05-25-00, 62-11 and Airworthiness Limitations Section (ALS) EC 135 Rev. 003 chapter 04-20-00, 62-11.
PHI	9-13-23	EC135P2 0398	11338.41		4759.84				Installed on N307PH

7 Successive status and minor/major maintenance and overhaul operations Positions successives et opérations d'entretien et de remise en état mineures et majeures									
Contractor Unité ou société	Date	Aircraft Version and S/N Version et No Appareil	Operations / Fonctionnement			Reason for transfer (code and symptoms) - Accomplished work - Replaced parts Motif du mouvement (code symptômes) - Travaux effectués - Pièces changées			Successive status before A/C delivery - See front page of table 5 Positions successives avant livraison avion - Voir recto tableau 5
			Unit / Unité: Support	Partial Partiel	Total Total	Unit / Unité: Support	Partial Partiel	Total Total	
									RCVD-REPAIRED/FRM-AEC/PS-843522609 4-29-2013 / (w) Ro-031331-2013
									ASB EC135-62-022 13 Not applicable b, the Serial number of this main rotor blade is 4-21-2013
									Complied with Eurocopter EC135 Service Bulletin Main Rotor System - Main Rotor Blade - Retrofit of additional PU
									Protective Foils SB-EC135-62-023 dated 02-08-2010. TT: 0:00 Jessie Paul Landry R.C. #3195382 04-30-2013
MAINTENANCE RELEASE PHI, INC. Regional Airport, Lafayette, Louisiana Certificate No. HEER617E The aircraft, airframe, aircraft engine, propeller, or appliance identified above was repaired and inspected in accordance with current regulations of the Federal Aviation Agency and is approved for return to service. Pertinent details of the repair are on file at this agency under Work Order No. <u>WS-063750-2013</u> Signed <u>Walter J. Remondet</u> 0700 for Date <u>04-30-2013</u> (Authorized Signature For Station)									
PHI	12/29/13	EC135P2 S/n 0434	4157:56	—	—	Ø	—	—	Installed on N323PH
PHI	11/23/15	EC135P2 0434	5569:24	—	—	1411:29	—	—	Removed from N323PH
AHI	11-Jan-16	CRS#VY1R575K		TSN:	1411:29				Inspected blade per RI L621M1018E01 Chapter 4 rev F. Replaced I/B & O/B PU erosion strip (05-11-00), Replaced I/B & O/B metal trim tab (05-08-00), replaced spherical bearing (05-22-02), reinstalled support S/N 7801 P/N L621M1009206 (05-22-00), reinstalled dampers P/N L621M1003101, S/N 24LK17728 & S/N 24K117062 (05-20-00), replaced root tapes (06-01-00), & closed up weight chamber (06-06-02), applied additional PU (SB62-023 rev.2), paint refinished (FV105-15103.00.06 rev. H1), Static balanced (FV105-10000.00.08 rev K). All inspections were validated and/or executed during this maintenance IAW MSM EC 135 rev.00200 chapter 05-22-00 62-1 & chapter 05-24-00 62-1 including SB's & AD's. GTL = 18,423g, total weight = 38,982g. Details on AHI SO#11000286112
									REC-RP, FR- AIRBUS, RO-041896-2015, P/S 846406615, 1-14-2016, MTM Don H
PHI	1/15/16	EC135P2 0456	5396:47	—	—	1411:29	—	—	Installed on N343PH
PHI	5-16-22	EC135P2 0456	8745:14			4759:84			Removed - Damaged

INVENTORY OF COMPONENTS
INVENTAIRE DES COMPOSANTS

ASSEMBLY DESIGNATION
DESIGNATION ENSEMBLE

MAIN ROTOR BLADE

ASSEMBLY REFERENCE
REFERENCE ENSEMBLE

L621M1010057

SERIAL NUMBER
No DE SERIE

6128

VERSION
VERSION

"B1"

- 1 - Date of instalation / Date de pose
- 2 - Serial number / Numéro de série
- 3 - Operating time of component /
Heures de fonctionnement du composant
- 4 - Operating time of assembly /
Heures de fonctionnement de l'ensemble
- 5 - Component number of cycles /
NB cycles du composant
- 6 - Assembly number of cycles /
NB cycles ensemble
- 7 - Version / Version

PAGE 01

INSTALLATION

POSE (*)

1	3	5	7
2	4	6	

COMPONENTS
COMPOSANTS

DESIGNATION REFERENCE	INSTALLATION POSE (*)				REPLACEMENTS ECHANGES (*)				SPECIAL INFORMATION RENSEIGNEMENTS PARTICULIERS
L621M1003101 DAMPER <i>Bottom</i> CURE DATE: 1st Qt- 2014	Apr.2015	NEW	/////	AHI 42					Removed 6-3-17 See attached log Cards for install
	24-LK17062	1038:51 5196:47	/////						
L621M1003101 DAMPER <i>top</i> CURE DATE: 2nd Qt- 2014	Apr.2015	NEW	/////	AHI 42					Removed 4-3-17 See attached log Cards for install
	24-LK17728	1038:51 5196:47	/////						
L621M1009206 SUPPORT ✓	Oct.2012	NEW	/////	AHI 42					
	7801	NEW	/////						
END OF REPORT									
L621M1003101 DAMPER Cure Date : 1 st QR 2016	Mar. 2017	NEW	/////						Performed by customer
	24-LK21533	2409:25	/////						
L621M1003101 DAMPER Cure Date : 3 rd QR 2016	Mar. 2017	NEW	/////		Jul. 2023	2350:24	/////		Performed by customer
	24-LK22294	2409:25	/////		24-LK22294	4759:49	/////		
L621M1003101 DAMPER Cure Date : 3 rd QR 2022	Jul. 2023	NEW	/////						
	24-LK31264	4759:49	/////						

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: BOOTH / BUME
O'HAUL: _____
INSP: _____
RINS: _____

COMPONENT NAME: Damper
PART NUMBER: L621M1003101
SERIAL NUMBER: 24-LK21533
CMP#: MR5-42661
DATE OF MANUFACTURE: 1Q-2014

INSTALLATION DATA					REMOVAL DATA				
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES		REASON FOR REMOVAL
4/3/17	S.N. 0456	A/C T.T. 0394.41	T.T. New	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N 343RH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	

TECHNICAL DIRECTIVES AND HISTORY OF OVERHAUL

[illegible]



PHI HEALTH, LLC
Certificate No. VE7A070N
2800 N 44th Street, Suite 800
Phoenix, AZ 85008 USA

COMPONENT STATUS REPORT

Comp. PN: L621M1010057
Comp. SN: 6128
Comp. ID: MRS3-000416-2021

Part Description: Blade, M/R
Part Category ~ Family: EC135P2 ~ EC135P2
Manufacture Date: 4/29/2013

Total Hours: 4,760.08
Total Take Offs: 16,570
Report Type: Regulatory

Lvl	Pos	Work Unit	Part No ~ Part Description ~ Part Serial	Service Life		T.T	TS-IN/OH	Rem	Due at	
				Usage	Date	T.C	CS-IN/OH	Units	Comp Hr	Date
		AM-MR-BLADE-EC135-1000H-36M-IN	L621M1010057 ~ Blade, M/R ~ 6128	1000 FH INS	36 ME	4,760.08	348.61	651.39	5,411.47	3/31/2024
		SB-EC135-62-040	L621M1010057 ~ Blade, M/R ~ 6128	OTH	1645 D					12/31/2025
1	621102010-UP	MR-DAMPER-EC135-RT	L621M1003101 ~ DAMPER ~ 24-LK31264	8000 FH RTR	120 ME	0.00		8000.00	12,760.08	9/30/2033
1	621102010-LWR	MR-DAMPER-EC135-RT	L621M1003101 ~ DAMPER ~ 24-LK21533	8000 FH RTR	120 ME	2,350.76		5649.24	10,409.32	3/31/2026



HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: 16,000 FH

O'HAUL: _____

INSP: _____

RINS: _____

COMPONENT NAME: BOLT - BLADE

PART NUMBER: L621M1001205

SERIAL NUMBER: 3022

CMP#: MRS- 15085

DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA				
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES		REASON FOR REMOVAL
8/2/05	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	

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HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: 16,000 FH
O'HAUL: _____
INSP: _____
RINS: _____

COMPONENT NAME: BOLT - BLADE
PART NUMBER: L621M1001205
SERIAL NUMBER: 2993
CMP#: MRS-15089
DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA				
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES		REASON FOR REMOVAL
8/2/05	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	

September 12, 2006

PAGE 1 OF 2

TECHNICAL DIRECTIVES AND HISTORY OF OVERHAUL

[illegible]

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: 16,000 FH

O'HAUL: _____

INSP: _____

RINS: _____

COMPONENT NAME: BOLT - BLADE

PART NUMBER: L621M1001205

SERIAL NUMBER: 2966

CMP#: MRS- 15090

DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA			
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES	REASON FOR REMOVAL
8/2/105	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T. T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC. T.S.I.	

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Y OF OVER

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIREES: 16,000 FH

O'HAUL: _____

INSP: _____

RINS: _____

COMPONENT NAME: BOLT - BLADE

PART NUMBER: L621M1001205

SERIAL NUMBER: 3003

CMP#: MRS- 15091

DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA			
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES	REASON FOR REMOVAL
8/2/05	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N 30704	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.

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Handwritten signature/initials.

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRE: 16,000 FH
O'HAUL:
INSP:
RINS:

COMPONENT NAME: BOLT - BLADE
PART NUMBER: L621M1001205
SERIAL NUMBER: 3025
CMP#: MRS- 15092
DATE OF MANUFACTURE:

INSTALLATION DATA					REMOVAL DATA				
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES		REASON FOR REMOVAL
8/2/05	S.N. 0398	A/C NEW	T.T. NEW	T.S.O.		A/C			
	N 307PH	A/C CYC.	CYC.	T.S.I.		T.T.	T.T.	T.S.O.	
						A/C			
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	
	S.N.	A/C T.T.	T.T.	T.S.O.		CYC.	CYC.	T.S.I.	
	N	A/C CYC.	CYC.	T.S.I.		A/C			
						T.T.	T.T.	T.S.O.	

W. S. [Signature]

W. S. [Signature]

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: 16,000 FH
 O'HAUL: _____
 INSP: _____
 RINS: _____

COMPONENT NAME: BOLT - BLADE
 PART NUMBER: L621M1001205
 SERIAL NUMBER: 2985
 CMP#: MRS- 15093
 DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA			
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES	REASON FOR REMOVAL
8/21/05	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.

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P.S.

HISTORICAL SERVICE RECORD



PHI, Inc.
P.O. BOX 90808
LAFAYETTE, LA 70509-0808

RETIRES: 16,000 FH

O'HAUL: _____

INSP: _____

RINS: _____

COMPONENT NAME: BOLT - BLADE

PART NUMBER: L621M1001205

SERIAL NUMBER: 3047

CMP#: MRS- 15094

DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA			
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES	REASON FOR REMOVAL
8/2/05	S.N. D398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.

[Signature]

[Signature]

RINS: _____

DATE OF MANUFACTURE: _____

INSTALLATION DATA					REMOVAL DATA				
DATE	INSTALLED ON	INSTALLED AT	COMPONENT TIME		DATE	REMOVED AT	COMPONENT TIMES		REASON FOR REMOVAL
8/2/05	S.N. 0398	A/C T.T. NEW	T.T. NEW	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N 307PH	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	
	S.N.	A/C T.T.	T.T.	T.S.O.		A/C T.T.	T.T.	T.S.O.	
	N	A/C CYC.	CYC.	T.S.I.		A/C CYC.	CYC.	T.S.I.	

PH *[Signature]*
P.S.

PH *[Signature]*
P.S.

DUPLICATE

6 Modification and Service Bulletin checking Contrôle d'exécution des modifications et des Services Bulletins						
No. Numéro	Type of modification Nature de la modification	Performing unit or contractor Unité ou société d'exécution	Date of embodiment and inspection stamp Date d'exécution et tampon de contrôle			
COMPUTER LO3285	STATE OF SOFTWARE P/N EG-26 LOADED.	SAGEM TEXAS	24 APR 15 RB 245			
5						
Contractor Unité ou société	Date Date	Aircraft Version and S/N Version et No Appareil	Operation Fonctionnement			Certification/ Attestation
			Support Support	Partial Partiel	Total Total	
						Manuf/Receipt* Fab/Reception
						Depreservation/* Destockage
						Period Overhaul/* V.P
						Fitting/* Montage

LOG CARD* / FICHE MATRICULE				Log card no Fiche no
2				
Follow-up Sheet for new equipment / Fiche suivieuse Matériel neuf				
1 Material identification / Identification du matériel				
Name Dénomination				
NATO Nomenclature Nomenclature OTAN				
APM 2000 AUTOPILOT MODULE				
AIRBUS HELICOPTERS Part number Référence				
Manufacturer's Part number Référence fabricant				
416-00297-007				
NATO Manuf code Code OTAN fabricant				
Delivery configuration Configuration de livraison				
Serial number Numéro de série				
1119				
Supplier Fabricant				
Amendments Amendements				
A				
Typ Type				
2 Contract or order / Marché ou commande				
Reference Référence				
Issuing agency Organisme émetteur				
Contrator Fournisseur				
Date Date				
Kit no No ou lot				
Address Adresse				
3 Guarantee / Garantie				
Equipment Matériel	Date of delivery Date de livraison	Guaranteed storage period Durée garantie de stockage	Service date Date de mise en service	Guarantee operation period Durée Garantie de fonctionnement
4 Special information / Renseignements particuliers				
Appendix to table 4 : Annexe au tableau 4				
YES ☐ NO ☒				
Inventory of lifed components: Inventaire des pièces à Durée de Vie				
YES ☐ NO ☒				
Transfer sheet: Fiche Matricule de Transfert				
YES ☐ NO ☒				
No of sheet : 0 Nb de page				
No of sheet : 0 Nb de page				
No of sheet : 0 Nb de page				
Operating limit Limite de fonctionnement			Life limit Limite de vie	ATA

*Delete as necessary / Rayer la mention inutile

AIRBUS HELICOPTERS F 16-04 B2

7 Successive status and minor/major maintenance and overhaul operations									P/N: 416-00297-007
Positions successives et opérations d'entretien et de remise en état mineures et majeures									S/N: 1119
Contractor Unité ou société	Date Date	Aircraft Version and S/N Version et No Appareil	Operations / Fonctionnement						Reason for transfer (code and symptoms) - Accomplished work - Replaced parts Motif du mouvement (code symptômes) - Travaux effectués - Pièces changées Successive status before A/C delivery - See front page of table 5 Positions successives avant livraison avion - Voir recto tableau 5
			Unit / Unité: Support	Partial Partiel	Total Total	Unit / Unité: Support	Partial Partiel	Total Total	
AHI	19 Mar 2015	CRS#VY1R575K		TSN:	UNK				LOG CARD DRAWN-UP THIS DATE DUE TO LOSS OF ORIGINAL INFORMATION IS THAT KNOW AS OF THIS DATE. AHI 46
SAGEM	24 APRIL 2015		TEST	---	---	---	---	RB 245	UNIT CLEANED, TESTED & INSPECTED AT SAL, TEXAS
BALP	18-NOV 2015	SN 1154 EC135T2+	645.1						INSTALLED IN C-GBAI - R. FRITH 133-95-05 (RENTAL RETURN)
BALP	25-NOV 2015	SN 1154 EC135T2+	663.9	18.8	UNK	---	---	---	REMOVED FOR OPS CONVENIENCE T. FRANCIS 133-95-33
SAGEM	18 JANUARY 2016		RECERTIFICATION	---	---	---	---	---	TESTED & INSPECTED AT SAL, TEXAS - RB 245. PAGE 27
REACH	9 AUGUST 2016	EC135P2+ 1148	1049.8						INSTALLED ON N316RX ACTT: 1052.2 DATE: 8/9/16 ER PAGE ON #91304 ER PAGE OFF 97173
REACH	11/30/16	EC135 P2+ SN 1148 N316RX	1251.1						REMOVED AFTER 201.3 HOURS ON INSTALLATION Nov. 30 2016 <i>Adrian</i> ADP #1781336
SED AVIONICS	21 JULY 2017		TEST	---	---	---	---	NFF	UNIT TESTED AND INSPECTED AT SEDA, TEXAS - RB 245. Repairman 3390001
Uniflight	9-18-16	EC135 P2+ N113MT S/N: 1103	1251.1			1841.6			Installed on N113MT S/N: 1103 ACTT: 1841.1 9-18-16 <i>MS</i> CRS#UNFR783K 27
MTA	10/6/17	EC135 P2+ 113MT Concord	1281.2			1871.2			removed from N113MT @ ACTT 1871.2 10/6/17 <i>Colleen</i> AP 3557280
SED AVIONICS	16 DECEMBER 2017		REPAIR	---	---	---	---	---	UNIT REPAIRED, TESTED, INSPECTED AT SEDA, TEXAS - RB 245. 3398#3596922
PHI	11-20-18	EC135P2 0398	8039.83			1871.2			Installed on N307PH

Modification and Service Bulletin checking <i>Contrôle d'exécution des modifications et des Services Bulletins</i>						
No. Numéro	Type of modification Nature de la modification	Performing unit or contractor Unité ou société d'exécution	Date of implementation and inspection stamp Date de mise en œuvre et attestation de contrôle			

Contractor Unité ou société	Date Date	Aircraft Version and S/N Version et No Appareil	Operation Fonctionnement			Certification Attestation
			Support Support	Partial Partiel	Total Total	
						Manuf/Receipt / * Fab/Reception
						Depreservation / * Destockage
						Period.Overhaul / * V.P
						Fitting / * Montage

LOG CARD / FICHE MATRICULE*				Log card N° Fiche N°	1
Follow-up Sheet for new equipment / Fiche suiveuse Matériel neuf*					
1 Material identification/ Identification du matériel					
Name <i>Dénomination</i>		APMS2010 AN/NVG			
NATO Nomenclature <i>Nomenclature OTAN</i>					
AIRBUS HELICOPTERS Part num <i>Référence</i>		416-00298-010			
Manufacturer's Part number <i>Référence fabricant</i>		416-00298-010			
Delivery configuration <i>Configuration de livraison</i>					
Serial number <i>Numéro de série</i>		4014			
Supplier <i>Fabricant</i>					
		NATO Manuf code <i>Code OTAN fabricant</i>			
		Version <i>Version</i>			
		Amendments <i>Amendements</i>			
		Type <i>Type</i>			
2 Contract or order/ Marché ou commande					
Reference <i>Référence</i>				Date <i>Date</i>	
Issuing agency <i>Organisme émetteur</i>				Batch N° <i>N° de lot</i>	
Contrator <i>Fournisseur</i>				Address <i>Adresse</i>	
3 Guarantee / Garantie					
Equipment <i>Matériel</i>	Date of delivery <i>Date de livraison</i>	Guaranteed storage period <i>Durée garantie de stockage</i>	Service date <i>Date de mise en service</i>	Guarantee operation period <i>Durée Garantie de fonctionnement</i>	
4 Special information / Renseignements particuliers					
Appendix to table 4 : <i>Annexe au tableau 4</i>		YES ☐ NO ☒ OUI NON	Nb of sheets : <i>Nb de fiches :</i>		
Inventory of lifed components: <i>Inventaire des pièces à Durée de Vie</i>		YES ☐ NO ☒ OUI NON	Nb of sheets : <i>Nb de fiches :</i>		
Transfer sheet: <i>Fiche Matricule de Transfert</i>		YES ☐ NO ☒ OUI NON	Nb of sheets : <i>Nb de fiches :</i>		
Operating limit <i>Limite de fonctionnement</i>			Life limit <i>Limite de vie</i>		ATA

Successive status and minor/major maintenance and overhaul operations

Positions successives et opérations d'entretien et de remise en état mineures et majeures

*Delete as necessary/ Rayer la mention inutile

FICHE MATRICULE LOG GARD*																							
					Fiche n° _____ Log card n° <u> 4 </u>																		
Fiche Suiveuse Matériel neuf Equipement et card New equipment																							
1 Identification du matériel Material identification																							
Dénomination <i>Name</i>		S.E.M.A. +I- 2,5 mm																					
Nomenclature <i>Nomenclature</i>		<u>841654</u>																					
Référence article <i>Part number</i>		Code entreprise <i>Manuf code</i>		F0386																			
Numéro de série <i>Serial number</i>		Amendements <i>Amendments</i>		AB																			
Marque <i>Trade mark</i>		Type <i>Type</i>		EC135																			
Identification du support ou du sous-support de rattachement <i>Related assembly or sub assembly identification</i>																							
Code <i>Code</i>		Dénomination <i>Name</i>																					
2 Marché ou commande / Contract or order																							
Référence <i>Reference</i>		Date <i>Date</i>		23/10/2003																			
Organisme émetteur <i>Issuing agency</i>		N° ou lot <i>Kit n°</i>		ECD																			
Fournisseur <i>Contractor</i>		Adresse <i>Address</i>		SAGEM S.A. MNT																			
3 Garantie Guarantee																							
Matériel <i>Material</i>	Date de livraison <i>Date of delivery</i>	Durée garantie de stockage <i>Guarantee storage period</i>	Date de mise en service <i>Service date</i>	Durée garantie de fonctionnement <i>Guarantee operation period</i>																			
Neuf/new																							
4 Renseignements particuliers Special information																							
(Document réception usine, dérogations, précaution à la mise en service, etc...) (Acceptance document, concessions, precaution before putting into service...)																							
SAGEM SA Emplacement de Manteaux CONTROLE FINAL Date P. PIOGER		Positions <i>Location</i> 	<table border="1" style="width: 100%; border-collapse: collapse;"> <thead> <tr> <th colspan="3" style="text-align: center;">Fonctionnement <i>Operation</i></th> </tr> <tr> <th style="width: 33%;">Support <i>Support</i></th> <th style="width: 33%;">Partiel <i>Partial</i></th> <th style="width: 33%;">Total <i>Total</i></th> </tr> </thead> <tbody> <tr> <td> </td> <td> </td> <td>Ch.</td> </tr> <tr> <td> </td> <td> </td> <td> </td> </tr> <tr> <td> </td> <td> </td> <td> </td> </tr> <tr> <td> </td> <td> </td> <td> </td> </tr> </tbody> </table>			Fonctionnement <i>Operation</i>			Support <i>Support</i>	Partiel <i>Partial</i>	Total <i>Total</i>			Ch.									
Fonctionnement <i>Operation</i>																							
Support <i>Support</i>	Partiel <i>Partial</i>	Total <i>Total</i>																					
		Ch.																					
SAGEM S.A. MNT		1104		<table border="1" style="width: 100%; border-collapse: collapse;"> <tbody> <tr> <td style="width: 50%;">Fab/Manuf. - Récept/Receipt. *</td> </tr> <tr> <td>Déstockage / Depreservation *</td> </tr> <tr> <td>V.P. / Periodic Overhaul *</td> </tr> <tr> <td>Montage. / Fitting *</td> </tr> </tbody> </table>		Fab/Manuf. - Récept/Receipt. *	Déstockage / Depreservation *	V.P. / Periodic Overhaul *	Montage. / Fitting *														
Fab/Manuf. - Récept/Receipt. *																							
Déstockage / Depreservation *																							
V.P. / Periodic Overhaul *																							
Montage. / Fitting *																							
Limite de fonctionnement <i>Operating limit</i>			Limite de vie <i>Life limit</i>																				

5

Positions successives et opérations d'entretien et de remise en état mineures et majeures

Successive location minor and major maintenance and overhaul operations

[illegible]

[illegible]

Fiche suiveuse Matériel neuf Equipment record card New equipment*

1	Identification du matériel	<i>Material identification</i>
---	----------------------------	--------------------------------

Dénomination
Name OPTICAL FIBER RATE GYRO UNIT FOG-P1-E1

Nomenclature	1020914
Nomenclature	

Référence fabricant Part number	438-00100-030	Code fabricant Manuf code	F6158
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Numéro de série Serial number	04/01545	Amendements Amendments	//
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Marque
Trade mark

SFIM INDUSTRIE

Type
Type

Identification du support ou du sous-support de rattachement
Related assembly or sub assembly identification

Code	Dénomination
Code	Name

2	Marché ou commande / <i>Contract or order</i>
---	---

Référence	Date
<i>Reference</i>	<i>Date</i>
Organisme émetteur	N° du lot
<i>Issuing agency</i>	<i>Kit n°</i>
Fournisseur	Adresse
<i>Contractor</i>	<i>Address</i>

3	Garantie	<i>Guarantee</i>
---	----------	------------------

Matériel <i>Material</i>	Date de livraison <i>Date of delivery</i>	Durée garantie de stockage <i>Guarantee storage period</i>	Date de mise en service <i>Service date</i>	Durée garantie de fonctionnement <i>Guarantee operation period</i>
Neuf/New				
	-			

4	Renseignements particuliers	<i>Special information</i>
---	-----------------------------	----------------------------

(Document réception usine, dérogations, précaution à la mise en service, etc...)
(Acceptance document, concessions, precaution before putting into service...)
6679 C04 ST IND.01

SM
317

5	Unité ou société Contractor	Date Date	Positions Location	Fonctionnement Operation		
				Support Support	Partiel Partial	Total Total
	SAGEM	01/05	NEUF			OH
						Fab/Manuf.-Récept/Receipt. *
						Déstockage/Depreservation *
						V.P. / Periodic Overhaul *
						Montage/Fitting *

Limite de fonctionnement
Operating limit

Limite de vie
Life limit

* Rayer la mention inutile * Delete as necessary

Positions successives et opérations d'entretien et de remise en état mineures et majeures
Successive location minor and major maintenance and overhaul operations

Unité ou société Contractor	Date Date	Positions Location	Fonctionnement Operation			Motif du mouvement (code, symptômes - Travaux effectués - Pièces changées Reason of transfert (code and symptoms) - Accomplished work - Replaced parts
			Support Support	Partiel Partial	Total Total	
						Positions successives avant livraison avion, voir au recto rubrique n° 5 Successive location before A / C delivery - See front page box number 5
SAGEN	04/05				04	Matériel neuf
ECD	02/2005	ECB5 P2/0398	new	—————	new	Installed

Positions successives et opérations d'entretien et de remise en état mineures et majeures
Successive location minor and major maintenance and overhaul operations

[illegible]